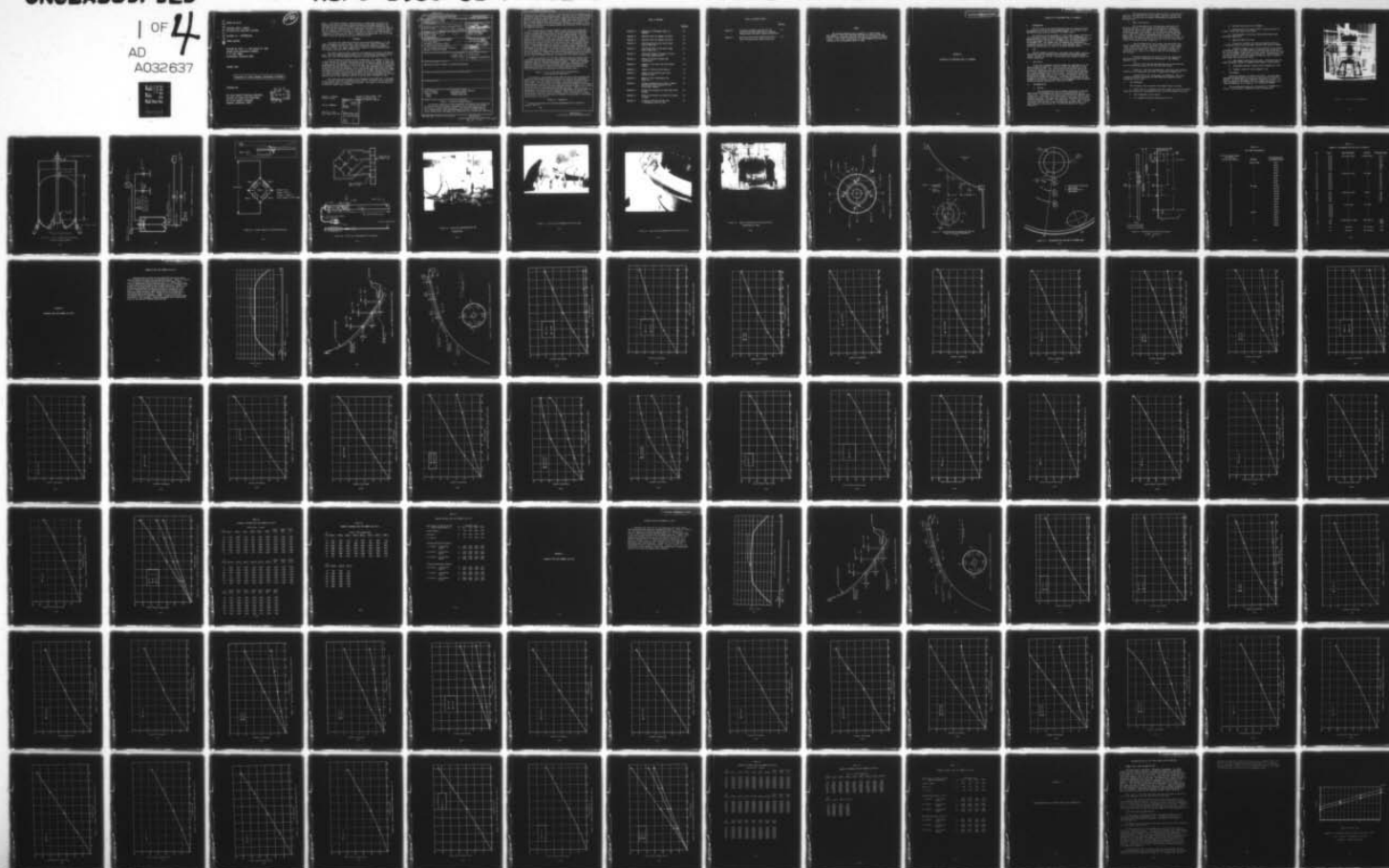


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FLEXIBLE CASE - GRAIN  
IN BALLISTIC WEAPON SYSTEMS

VOLUME III - APPENDIXES

FINAL REPORT

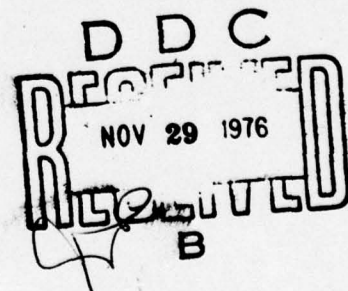
Kenneth W. Bills, Jr. and Samuel W. Jang  
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P. O. Box 13400  
Sacramento, California, 95813

October 1976

Approved for Public Release; Distribution Unlimited

Prepared for:

Air Force Rocket Propulsion Laboratory  
Director of Science and Technology  
Air Force Systems Command  
Edwards, California, 93523





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#### FOREWORD

This report was submitted by Aerojet Solid Propulsion Company, P. O. Box 13400, Sacramento, California, 95813, under Contract No. F04611-72-C-0055, Job Order No. 305910JU with the Air Force Rocket Propulsion Laboratory, Edwards, California, 93523. The report summarizes the technical efforts conducted under this contract from April 1972 to March 1976.

The efforts reported herein represent the combined activities of the Aerojet Solid Propulsion Company, Harold Leeming, Ph.D. and Associates, Konigsberg Instruments, Inc., the Texas A&M Research Foundation, and the University of Texas.

The key technical personnel on this program were: Mr. Kenneth W. Bills, Jr. of ASPC, who was the Principal Investigator on the Program; Mr. Samuel W. Jang, also of ASPC, who was the program's Principal Engineer; Dr. Harold Leeming, of HL&A, who coordinated the instrumentation of the motors and, later the acquisition of gage data during motor testing; Mr. Herman P. Briar, of ASPC, who conducted much of the gage diagnostic work; Mr. Eph Konigsberg, of KI, who supplied the stress and strain gages and supporting consultation; Dr. Scott W. Beckwith, of TAMRF, who provided an extensive study of flexible case materials and their constitutive relations; and Drs. Eric Becker and Robert Dunham, of the University of Texas, who developed an advanced computer code for the structural analysis of grains held in fiberglass cases.

This report has been reviewed by the Information Office/DOZ and is releasable to the National Technical Information Service (NTIS). At NTIS it will be available to the general public, including foreign nations. This report is unclassified and suitable for general public release.

Durwood I. Thrasher  
Project Engineer

FOR THE COMMANDER

Charles R. Cooke  
Solid Rocket Division

Charles E. Payne, Major, USAF  
Chief, Surveillance and  
Mechanical Behavior Section

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(thermal cycling, handling, vibration, pressurization) and then comparing the experimental results with predicted values. This was to be a "closed envelope" approach in which the reduced data from the gages were to be kept secret until the final assessment phase of the program. Midway through the motor testing phase serious anomalies were detected in the gage data. This led to a suspension of the original plan and an eventual redirection and change in scope to concentrate efforts on the identification and correction of the sources of the anomalies observed.

The revised program included diagnostic evaluations to isolate error sources, a system rework to correct and/or modify questionable components, a parallel laboratory investigation of specific gage characteristics and, finally, verification of the stability of the reworked system. From the results of the diagnostic tests it was determined that the major anomalies observed could be traced to the gages-lead-wire-solder junction combinations. Use of an acid flux in soldering the stainless steel leadwires provided a potential for corrosion to occur as the junction aged. The reworked system, which included crimped spade-lug junctions in place of the leadwire solder joints, a new DAS and revised operational procedures showed a substantial improvement in system stability, exhibiting an average drift rate of 0.5% of full scale output per month. This value is consistent with that for the gages alone as quoted by the manufacturer, and converts to about 0.75 psi per month for the 150 psi gage. However, this drift is considered excessive for measurement of long term thermal stresses (as required by the original program).

Laboratory evaluations of the gages addressed potential problems associated with exposed semi-conductor strain gages on the normal stress gage diaphragm, gage self-heating and hysteresis effects. These tests indicated potential transducer response differences between the calibration and the high rate pressurization situations which would require experimentally determined corrections to achieve the accuracy required to accomplish the original program goals for the high rate pressurization tests.

#### Volume II - Solid Propellant Grain Instrumentation System Design and Application

The experience and knowledge gained from this and similar programs were compiled in this volume, which was designed as a guide to the experimental stress analysis of solid propellant grains. The effort was divided into six major phases. The first phase is directed to the program manager and the project engineer, who must make the initial decision to conduct such an effort. The second and third phases are more elaborate versions of Phase I, but involve realistic plans for instrumenting and testing the units. Phase III is an evaluation of these plans to assure that the measurements can be obtained with the available facilities and test equipment. Phases IV and V define the extensive work required to carry out the test plans, while Phase VI includes the reduction of the test data and an assessment of the quality of the testing and the value of the results.

#### Volume III - Appendices

Seventeen appendices give detailed supplemental data in support of Volumes I and II.

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## FORMAT

The attached appendixes were prepared in a simple format. In each appendix a brief text explains the nature of the work, the test results and any relevant conclusions that may not be covered in Volumes I and II. All of the illustrations of a given appendix are given next, followed by the corresponding tables of data.

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## APPENDIX A

### HYDROTESTS OF MINUTEMAN STAGE III CHAMBERS



## HYDROTESTS OF MINUTEMAN STAGE III CHAMBERS

## A. INTRODUCTION

The first step in the motor fabrication and test program involved the hydrotest of the full scale Minuteman chambers to characterize the deformation versus pressure response of the motor cases.

The first full scale Minuteman III, Third Stage chamber (S/N 30113-1), a 52-inch diameter glass filament wound structure, was hydrotested to a maximum pressure of 200 psi on 25 September 1972. The second Minuteman III, Third Stage chamber (S/N 30114-1) was hydrotested the following week on 29 September 1972. The chambers were instrumented and hydrotested in accordance with AGC Test Plan 1826-26-TP, dated September 1972, to measure strains and deflections as a result of internal pressure effects.

## B. TEST

The two chambers, identified as S/N 30013 and S/N 30014, purchased from Thiokol, were of ASPC design and conformed to AGC Drawing Numbers 1147251, 1147250, and 1147278. They had been proof tested to 720 psi by Thiokol Corporation.

## C. TEST SET-UP

In the test arrangement used the chamber was supported by its aft skirt as shown in Figure A-1. Figure A-2 shows the hydrotest tooling (P/N 1017623) installed. This tooling reduces the pressure load on the aft dome by transferring that part acting on the piston to the forward skirt, thus simulating the firing test load. A high pressure water pump was connected through a port in the piston of the hydrotest tooling to pressurize the chamber (Figure A-3). A complete listing of the tooling and materials required for hydrotest, plus a detailed procedure for the assembly of the tooling, are contained in the documentary files (Minuteman Production Standard Procedure No. 113A) sent to AFRPL.

## D. INSTRUMENTATION

## 1. General

The instrumentation used to measure deflections of the motor cases during the hydrotests were cantilever beam-deflection devices (Figure A-4). These deflection devices were fabricated by mounting four conventional strain gages, B.L.H type FAB-25-12, onto the .030 inch thick tempered steel beam. The gages were located at the highest stress point to provide maximum sensitivity. Two gages (Nos. 1 and 2, Figure A-4) were installed on the top side of the steel beam and two gages (Nos. 3 and 4) were installed on the bottom side. Fabrication details are shown in Figure A-5.



The instrumentation used to measure strain in the aft dome consisted of conventional foil strain gages. They were the BLH type FAB-25-12. The gages used to measure circumferential growth were the BLH type PA-52.

## 2. Motor Installations

The instrumentation for the hydrotests consisted of 25 deflection devices, 11 strain gages, and two pressure transducers. The deflection devices were mounted on one side to a rigid external framework and then to the chamber. Six of these units were located in the forward dome area and three were installed in the forward boss area, Figure A-6. Similar instrumentation was installed in the aft chamber area to measure aft dome and aft closure deflections, Figure A-7. Additionally, six deflection devices were used to measure longitudinal growth in the barrel section, Figure A-8.

Three circumferential strain gages, BLH type PA-52, were used to measure hoop strain in the barrel, Figure A-9. Eight conventional strain gages, BLH type FAB-25-12, were used to measure strain in the aft head, Figure A-10. They were installed at an angle of approximately  $23^\circ$ , which is the angle of wrap for the fiberglass in the aft head.

Pressure transducers were placed at both the forward and aft closures and were independent of the pressure in the incoming hydraulic lines.

Table A-1 lists the code designations used to correlate the test plan numbering system with the Test Area code system.

Table A-2 lists the instrumentation used during the hydrotests. Locations of instrumentation are shown in Figures A-10 through A-13.

Installation of the strain gages and deflection reeds were standard operational procedures for ASPC (they are contained in the documentary files, also).

## E. TEST PROCEDURES

The procedures used to hydrotest the chambers were as follows:

1. Verify that all transducers and strain gages are installed properly. Range and calibrate all instrumentation prior to filling chamber with water.
2. Take photographs of test setup.
3. Fill chamber with water and bleed out all air.

4. Perform visual leak test of chamber.
5. Pressurize with city water to  $55 \pm 5$  psig and return to 0 psig. Verify that there are no leaks.
6. Verify chamber is at 0 psig, then perform balance and calibration operations.
7. Make sure pumps are ready.
8. Turn on all recorders and with oscillograph running at .1 inches per second using city water, perform final bleed of chamber.
9. Pressurize chamber case to 50 psig and hold for 60 seconds to record all data. Increase pressure in 50 psig steps to 200 psig, hold for 60 seconds at each step to record all data and return to 0 psig in a similar manner, recording at each hold point. Pressurization rise and decay rates will be in the order of minutes per step, to achieve good steady state calibration levels.
10. Vent chamber system and drain water. (Precaution must be taken to prevent collapse of chamber during removal of test fluid).
11. Disassemble hydrotest tooling as per Procedure No. 113A.
12. Prepare a report of tooling damage if any.

F. TEST RESULTS

During the hydrotest of Chamber S/N 30114-1, two minor leaks developed at approximately 150 psig, near the middle of the barrel section. The leaks were water "weeping" through the case wall. All leaks discovered in the hydrotest of this chamber were marked on the case and documented in an historical I.R. (Inspection Report).

The case deflection data for the hydrotests of Chambers No. 1 and No. 2 are contained in Appendices B and C, respectively.

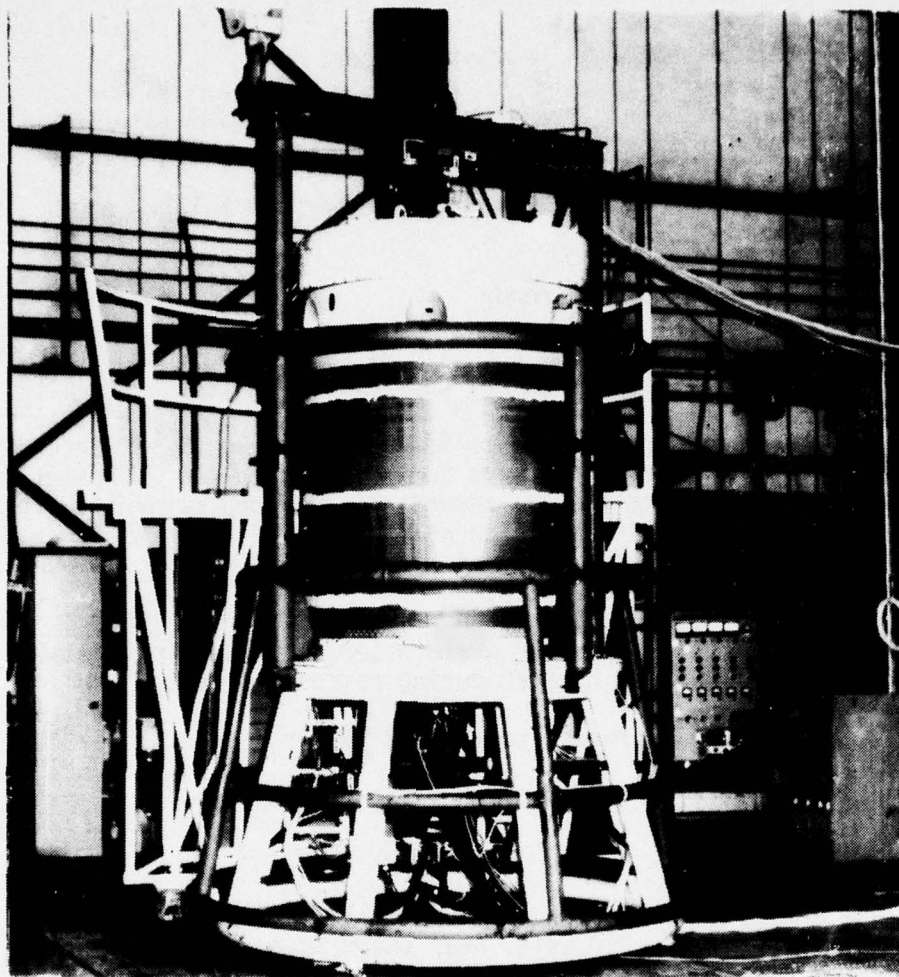


FIGURE A-1. TEST SET-UP FOR HYDROTESTS



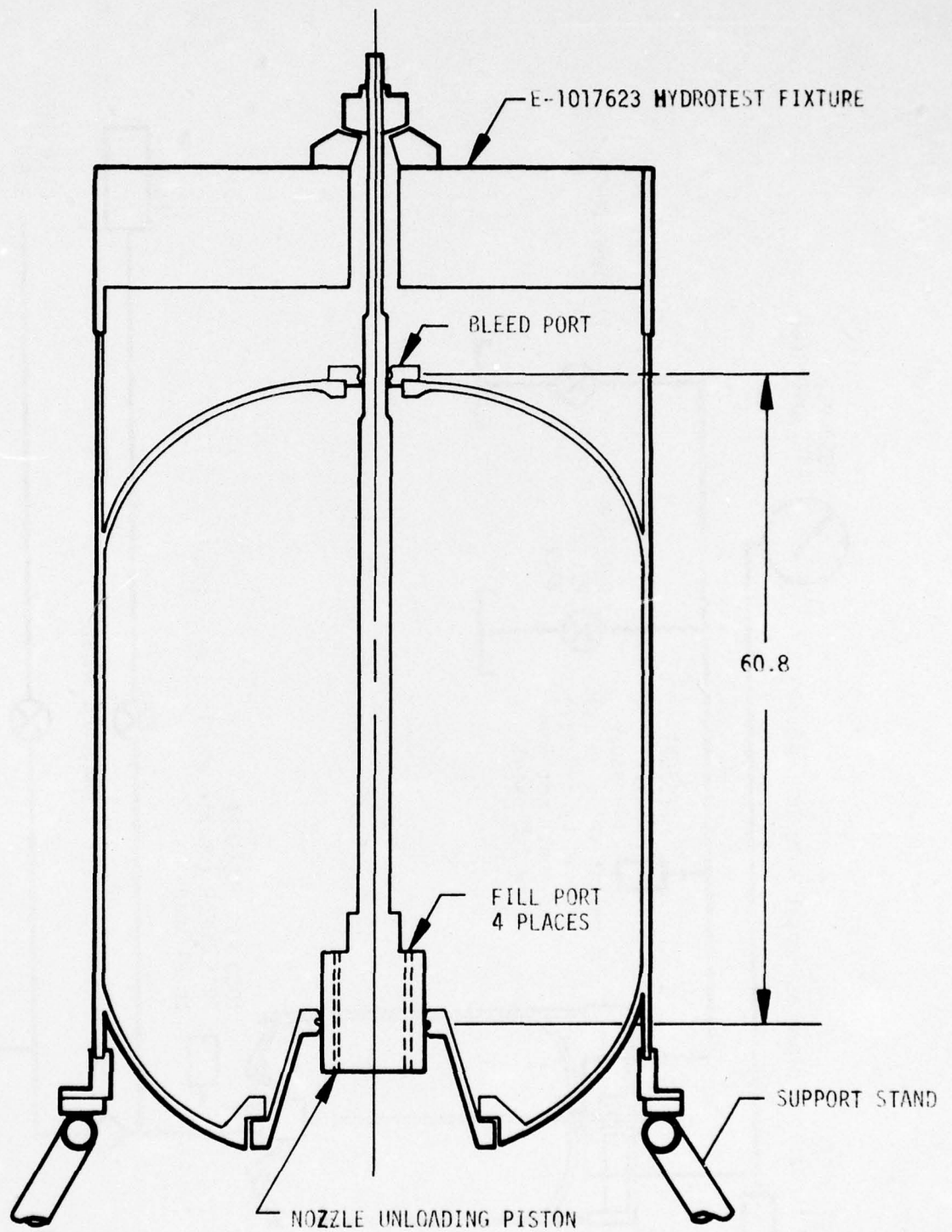


FIGURE A-2. SKETCH OF MOTOR WITH SUPPORTING  
EQUIPMENT DURING HYDROTEST



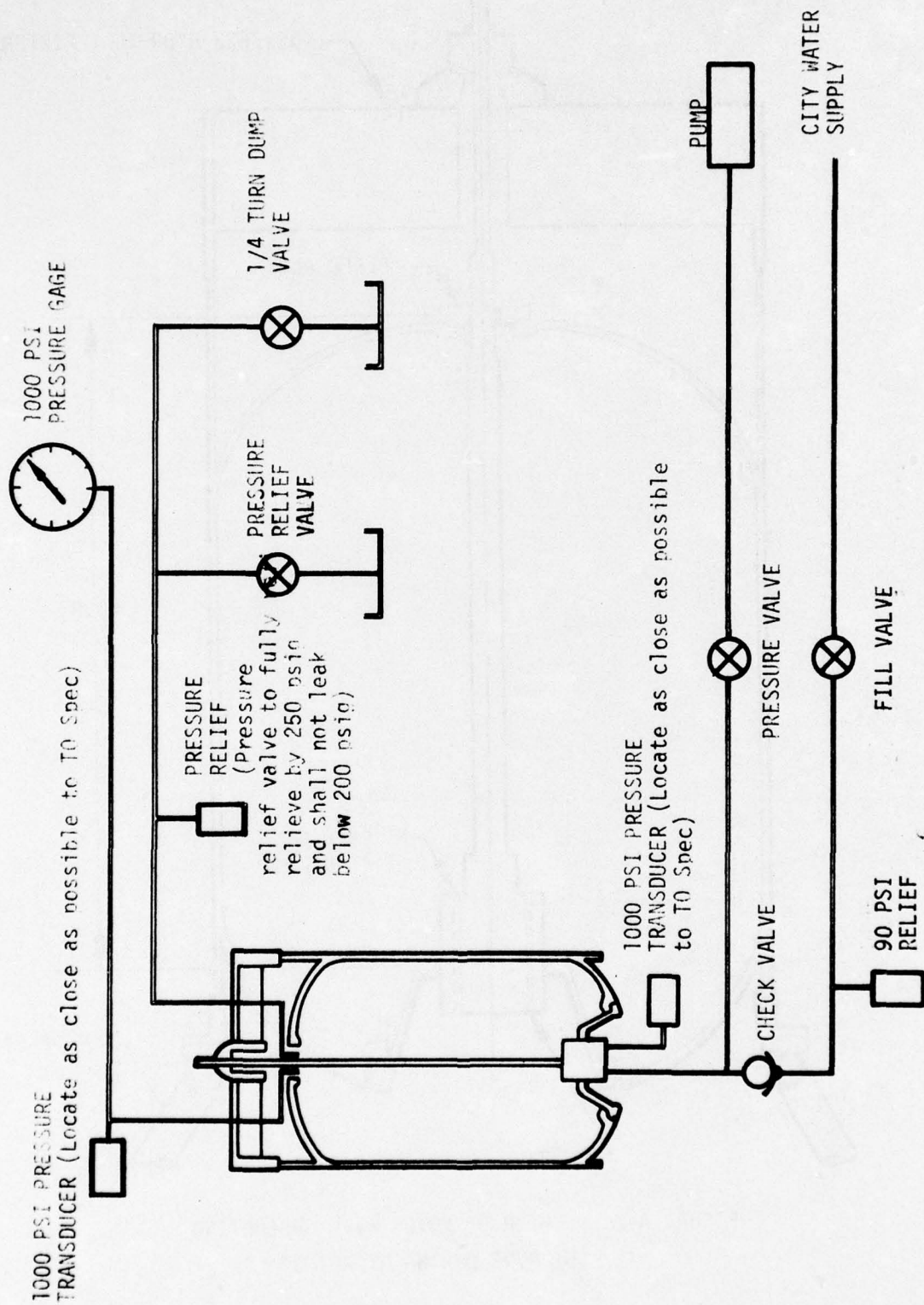
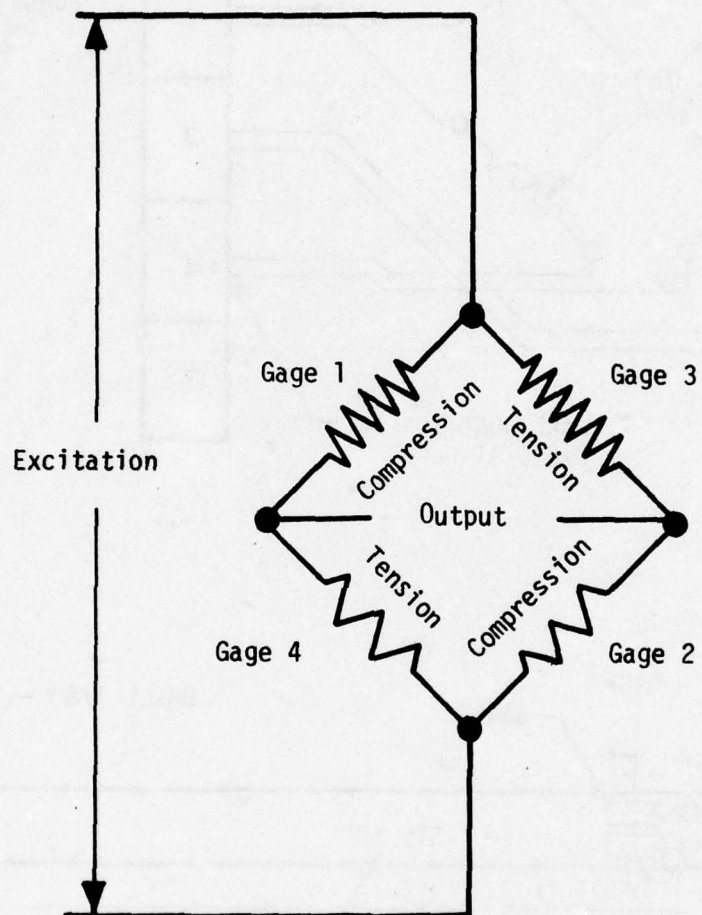
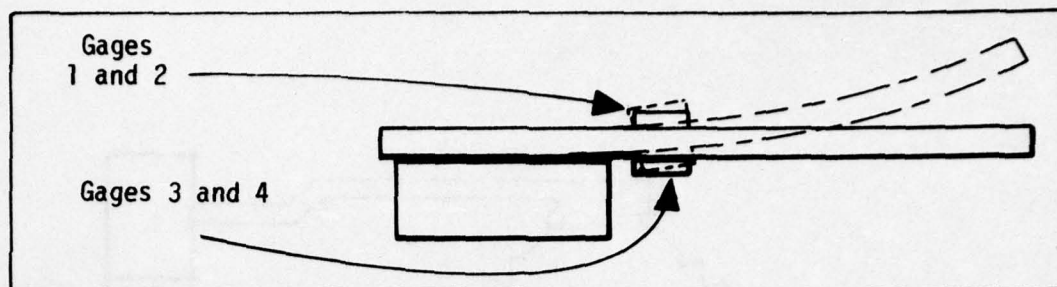


FIGURE A-3. SCHEMATIC OF FULL-SCALE MOTOR HYDROTEST SETUP



Output in mv =  
 (Gages 3 and 4) -  
 (-Gage 1 - Gage 2) =  
 Sum of Output of Four Gages

FIGURE A-4. CIRCUIT ANALYSIS OF DEFLECTION DEVICE

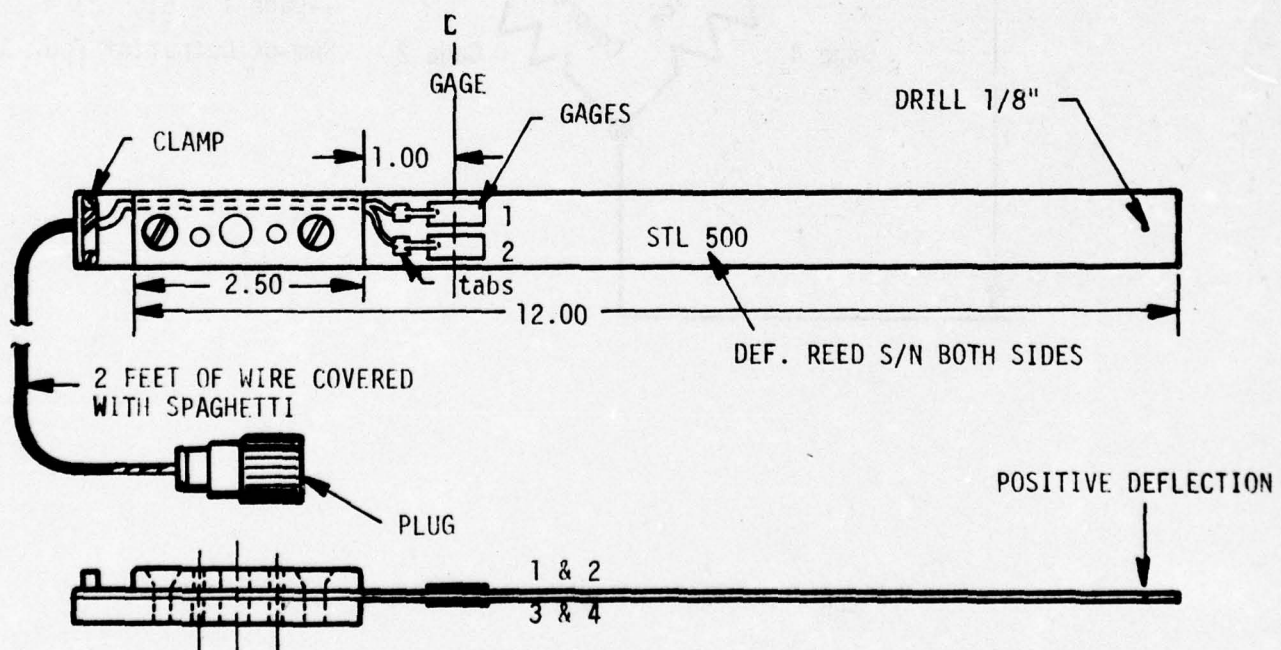
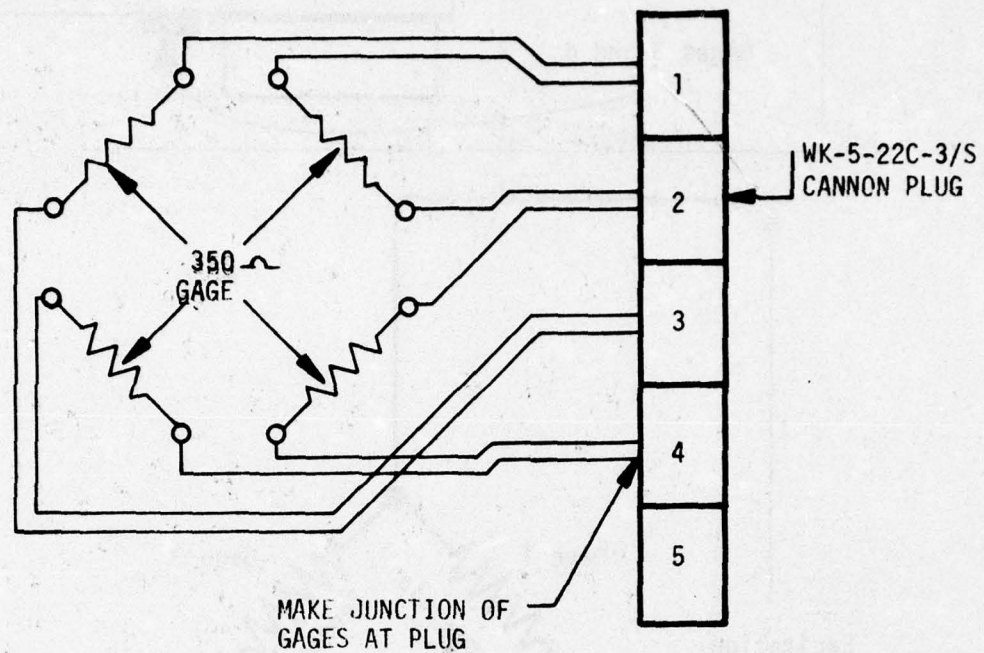


FIGURE A-5. DEFLECTION REED (FABRICATION PROCEDURE)



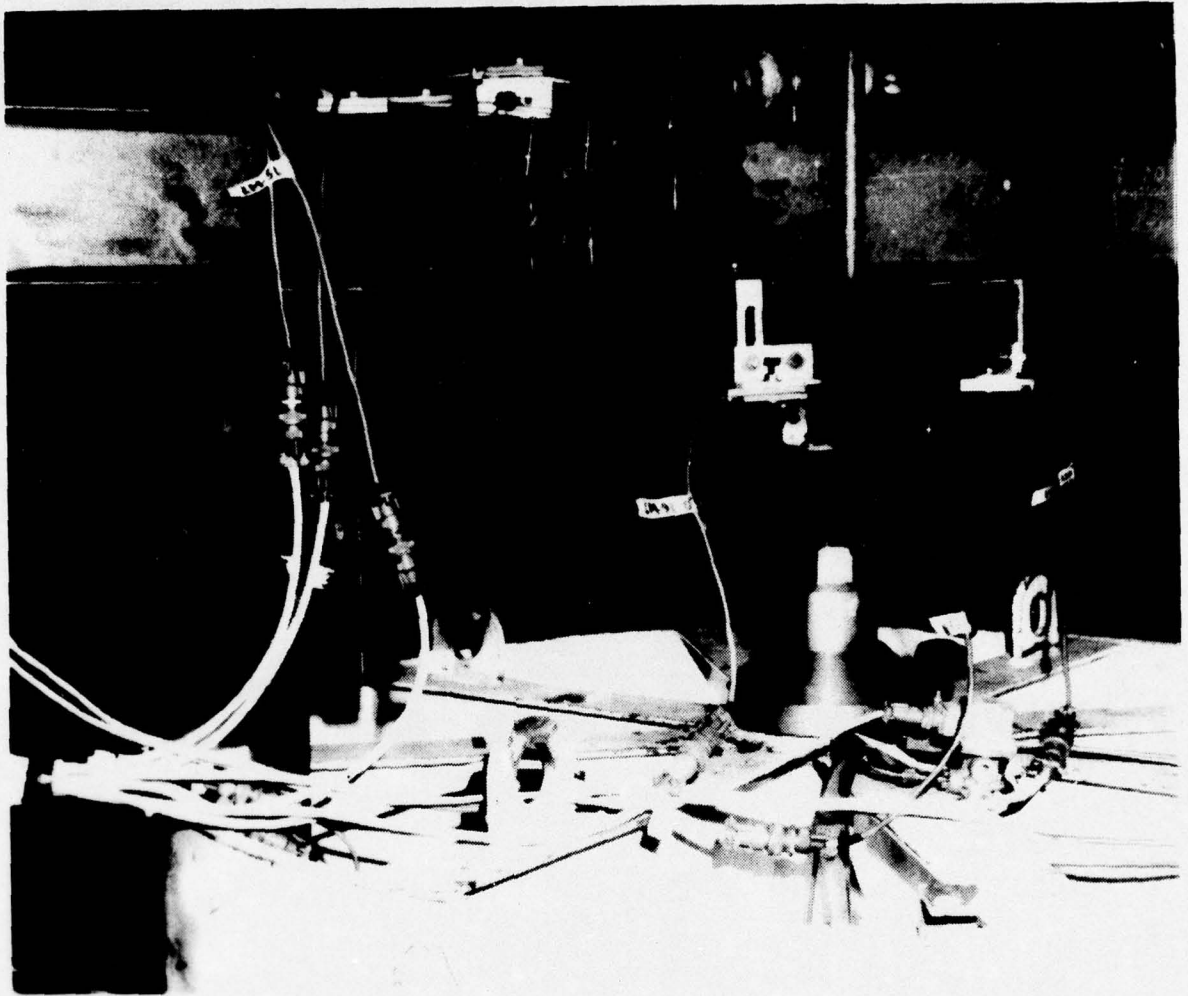


FIGURE A-6. DEFLECTION INSTRUMENTATION FOR  
FORWARD DOME



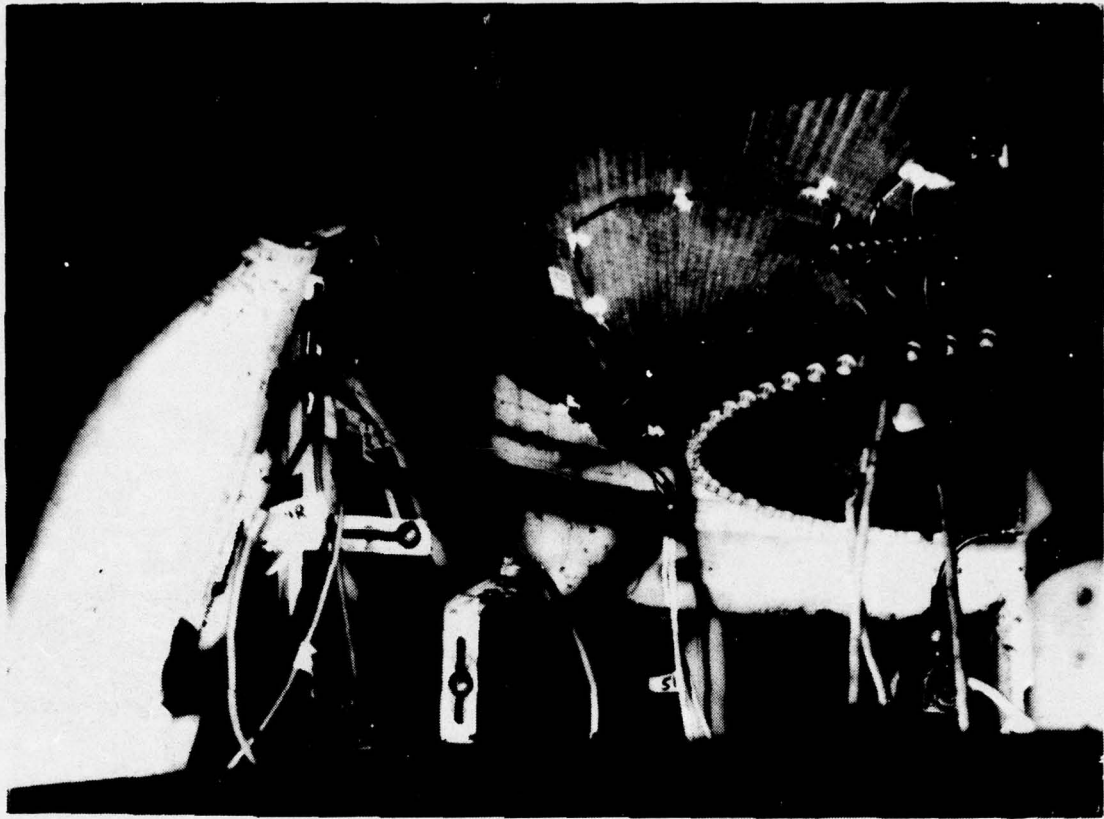


FIGURE A-7. DEFLECTION INSTRUMENTATION ON AFT DOME

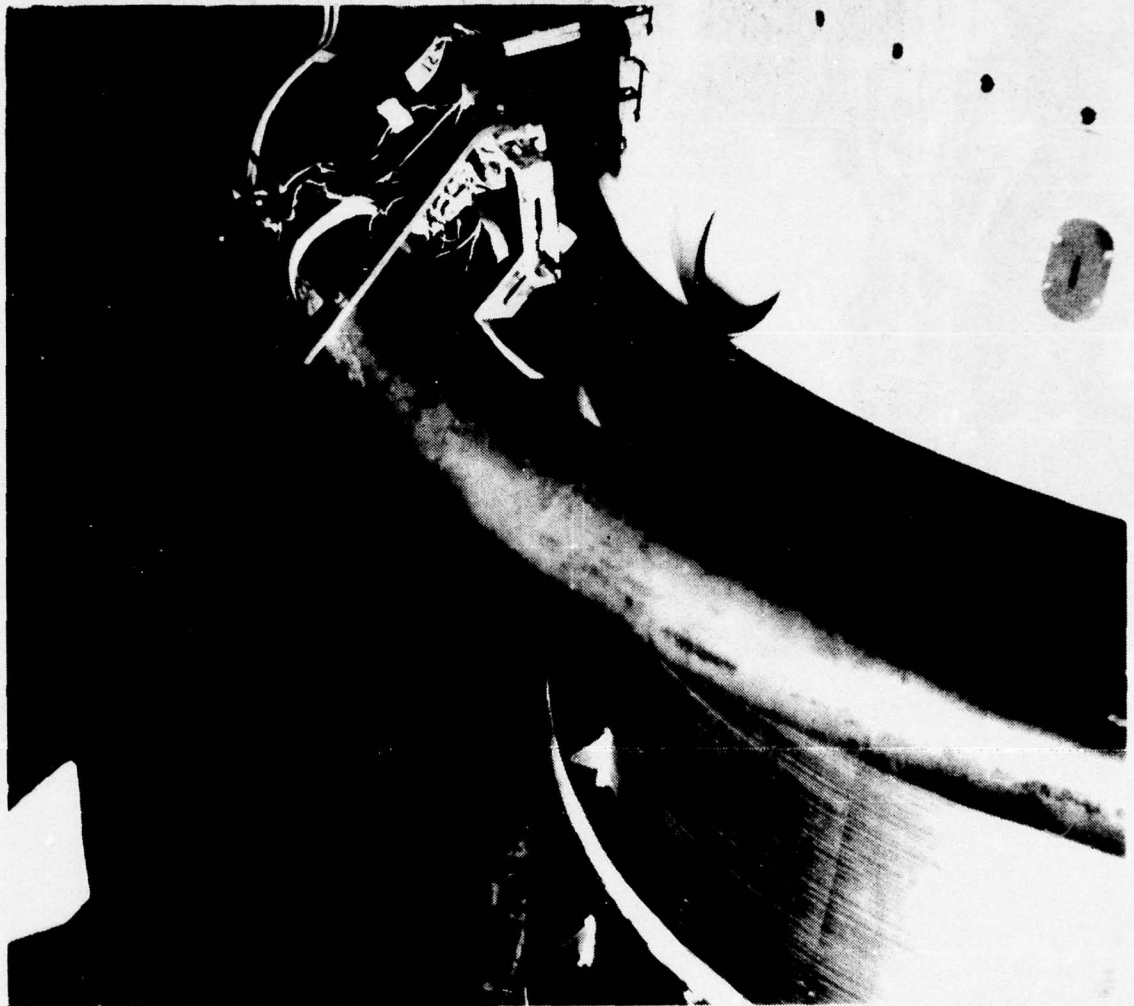


FIGURE A-8. DEFLECTION INSTRUMENTATION FOR BARREL SECTION

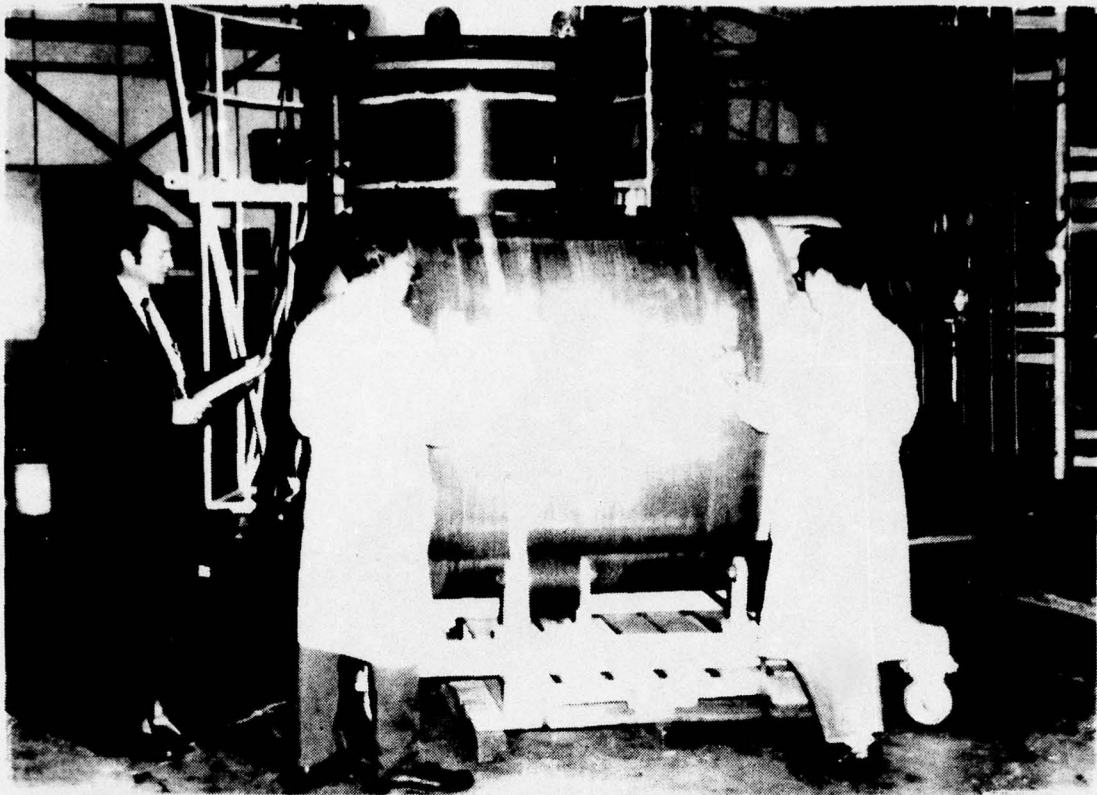


FIGURE A-9. SURFACE PREPARATION FOR INSTALLATION OF  
CIRCUMFERENTIAL GAGES



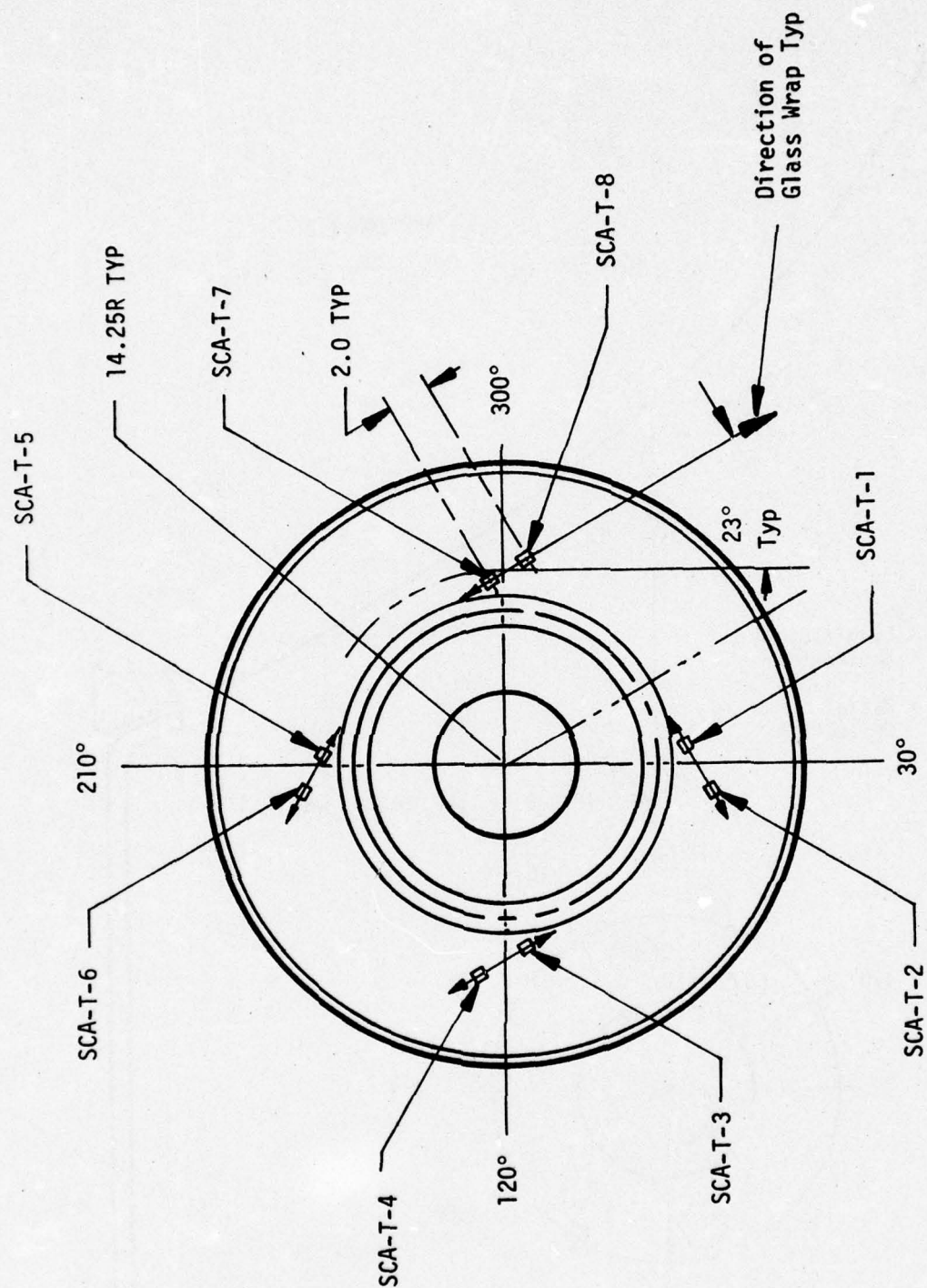


FIGURE A-10. INSTRUMENTATION LOCATIONS ON AFT DOME

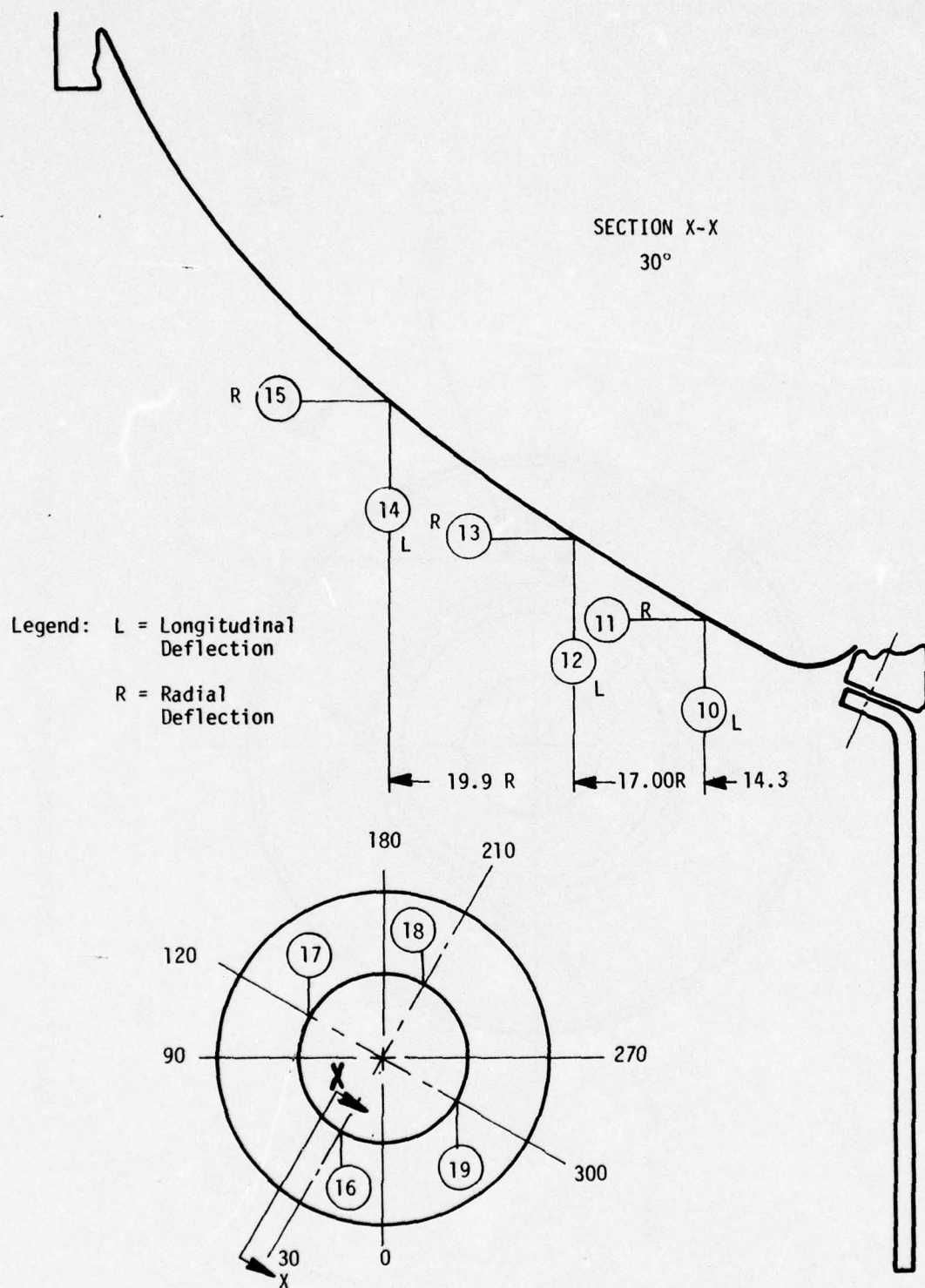


FIGURE A-11. INSTRUMENTATION LOCATIONS ON AFT DOME AND BOSS FOR DEFLECTION MEASUREMENTS  
A-16

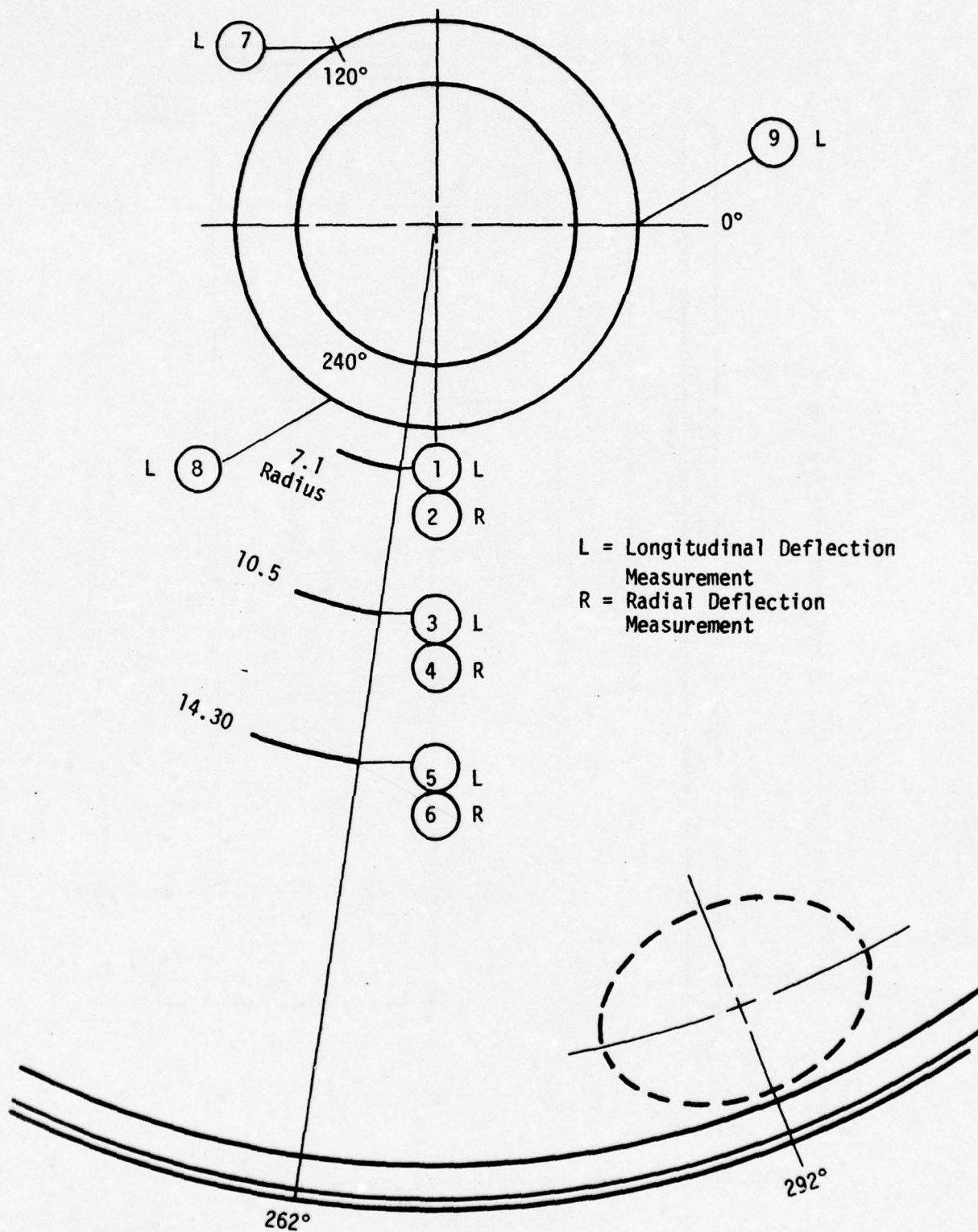
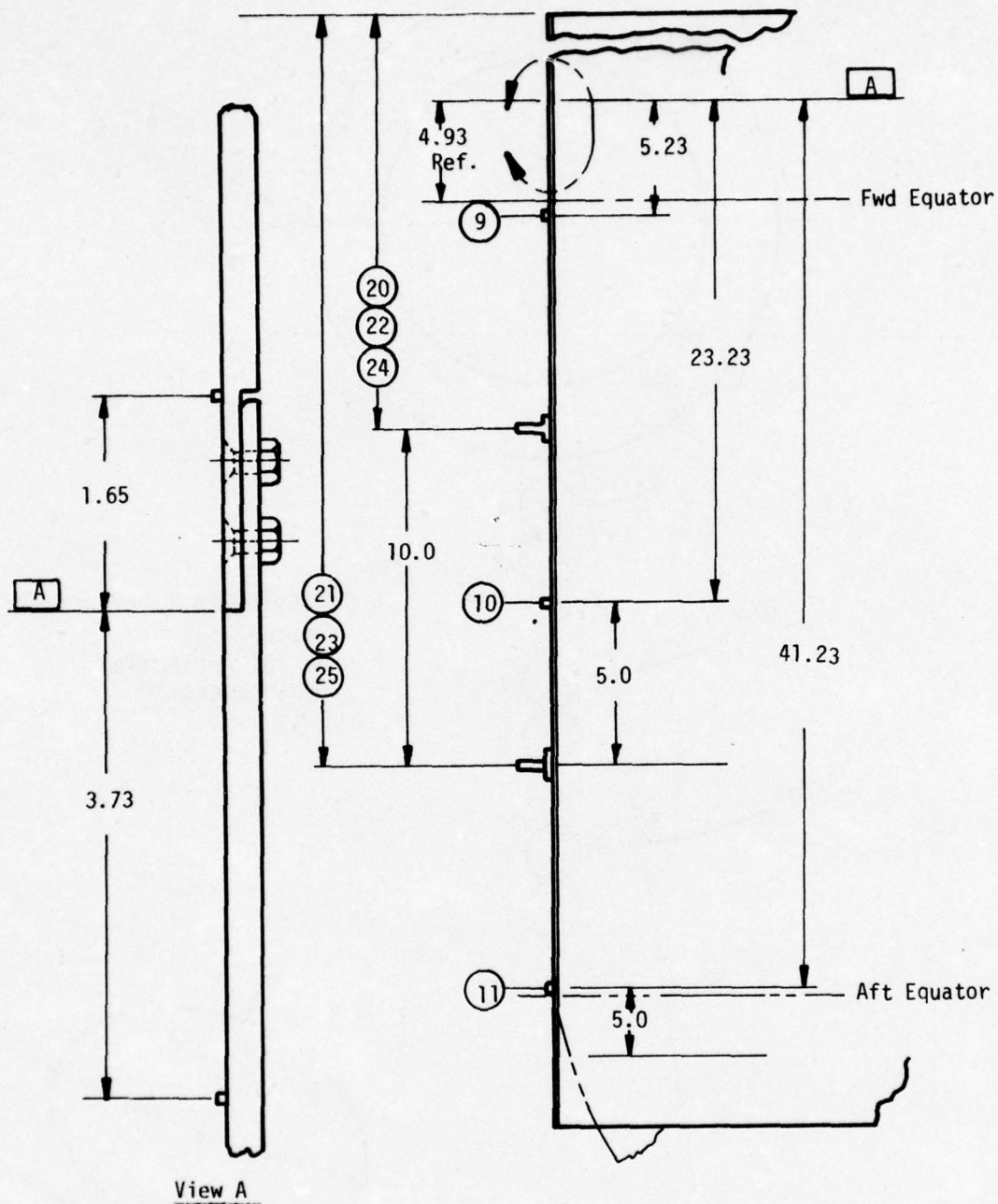


FIGURE A-12. INSTRUMENTATION LOCATIONS ON FORWARD DOME





View A  
20-25 are deflections,  
9, 10, 11 are circum-  
ferential strain gages.

FIGURE A-13. INSTRUMENTATION LOCATIONS ON CYLINDER  
SECTION - SECTION 45°  
A-18

TABLE A-1  
TEST CODE RELATIONSHIPS

<u>External Growth Channel Designation (As per Test Plan)</u>	<u>Location</u>	<u>Code Designations of Plotted Curves</u>
1	Forward Dome	LM-1L
2	"	LM-2R
3	"	LM-3L
4	"	LM-4R
5	"	LM-5L
6	"	LM-6R
7	"	LM-7L
8	"	LM-8L
9	"	LM-9L
10	Aft Dome	LM-10L
11	"	LM-11R
12	"	LM-12L
13	"	LM-13R
14	"	LM-14L
15	"	LM-15R
16	"	LM-16
17	"	LM-17
18	"	LM-18
19	"	LM-19
20	Barrel	LM-20
21	"	LM-21
22	"	LM-22
23	"	LM-23
24	"	LM-24
25	"	LM-25

TABLE A-2

## SUMMARY OF INSTRUMENTATION FOR 200 PSI HYDROTEST

<u>No.</u>	<u>Figure</u>	<u>Type Transducer</u>	<u>Location</u>	<u>Estimated Range</u>
1	A-12	Deflection Reed	Forward Dome	.4 in.
2	A-12	"	"	.03
3	A-12	"	"	.35
4	A-12	"	"	.035
5	A-12	"	"	.25
6	A-12	"	"	.036
7	A-12	"	"	.0466
8	A-12	"	"	.46
9	A-12	"	"	.48
10	A-11	Deflection Reed	Aft Dome	.23
11	A-11	"	"	.08
12	A-11	"	"	.2
13	A-11	"	"	.08
14	A-11	"	"	.15
15	A-11	"	"	.08
16	A-11	"	"	.32
17	A-11	"	"	.32
18	A-11	"	"	.32
19	A-11	"	"	.32
20	A-9	Deflection Reed	Barrel	-.04
21	A-9	"	"	-.08
22	A-9	"	"	-.04
23	A-9	"	"	-.08
24	A-9	"	"	-.04
25	A-9	"	"	-.08
1	A-10	Strain Gage	Aft Head	
2	A-10	"	"	
3	A-10	"	"	
4	A-10	"	"	
5	A-10	"	"	
6	A-10	"	"	
7	A-10	"	"	
8	A-10	"	"	
9	A-9	Circumferential Gage	Gage Barrel	3000
10	A-9	"	"	5400
11	A-9	"	"	3000
1	A-3	Pressure	Aft Closure	1000
2	A-3	Pressure	Fwd Closure	1000



APPENDIX B

HYDROTEST DATA FOR CHAMBER S/N 30113

## HYDROTEST DATA FOR CHAMBER S/N 30113

Tabulated data and plots of case deflection and strain versus internal pressure are given for each measuring device. Tables B-1 and B-2 list the data of the deflection devices and strain gages, while Table B-3 summarizes the more pertinent data; i.e., of the hoop strains in the barrel, and the forward and aft dome deflections. Graphic illustrations of the hoop strains in the cylinder section and the aft and forward dome deflections at 200 psig are shown in Figures B-1, B-2, and B-3. Figures B-4 to B-9 show the curves for the forward head deflection versus pressure, while Figures B-10 to B-16 show the aft head deflections versus pressure. Figures B-17 to B-19 show the longitudinal strain in the barrel section. Figures B-20 to B-27 show the strains in the aft head. Figure B-28 presents the circumferential growth of the case with chamber pressurization.

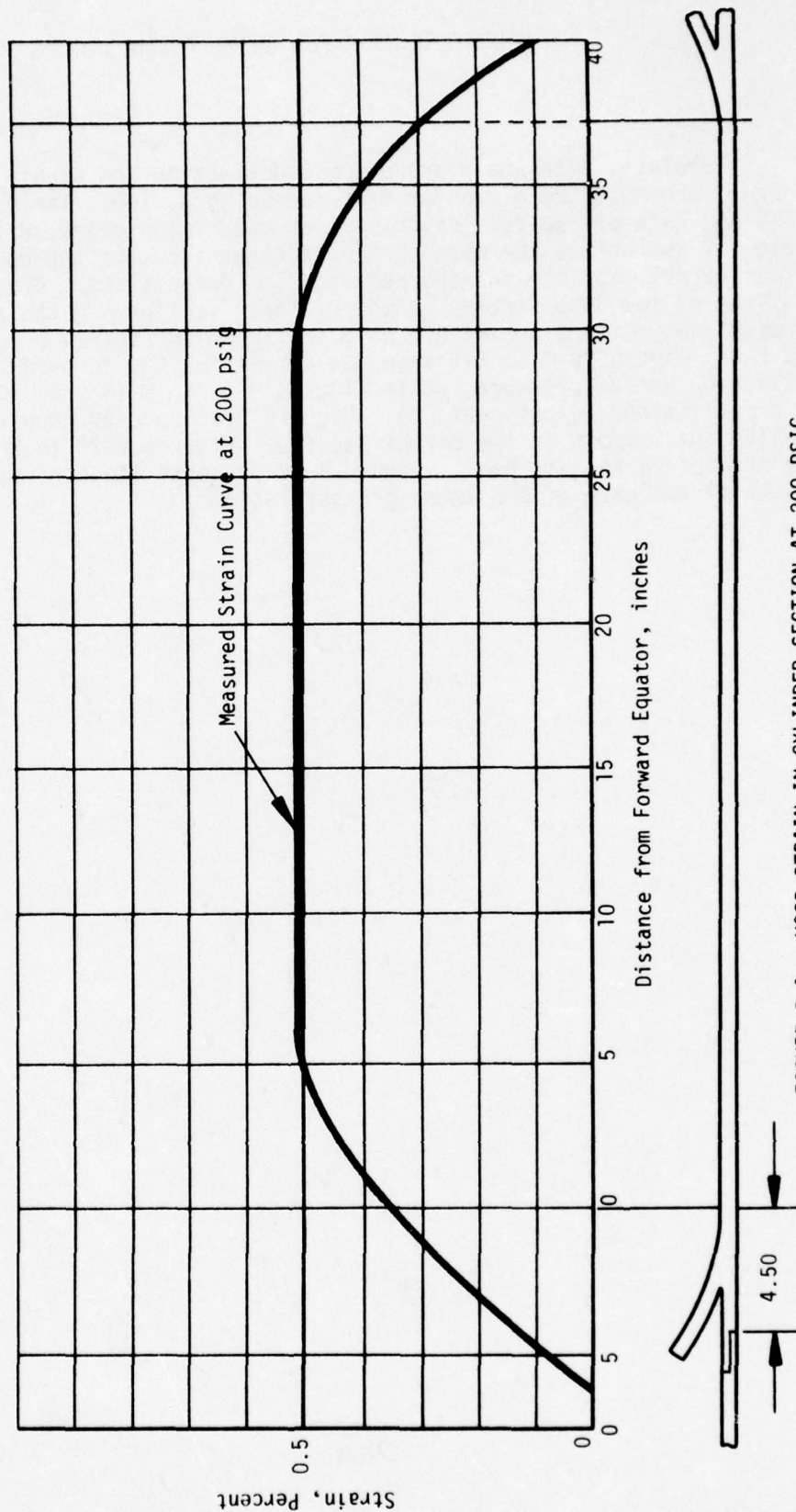


FIGURE B-1. HOOP STRAIN IN CYLINDER SECTION AT 200 PSIG



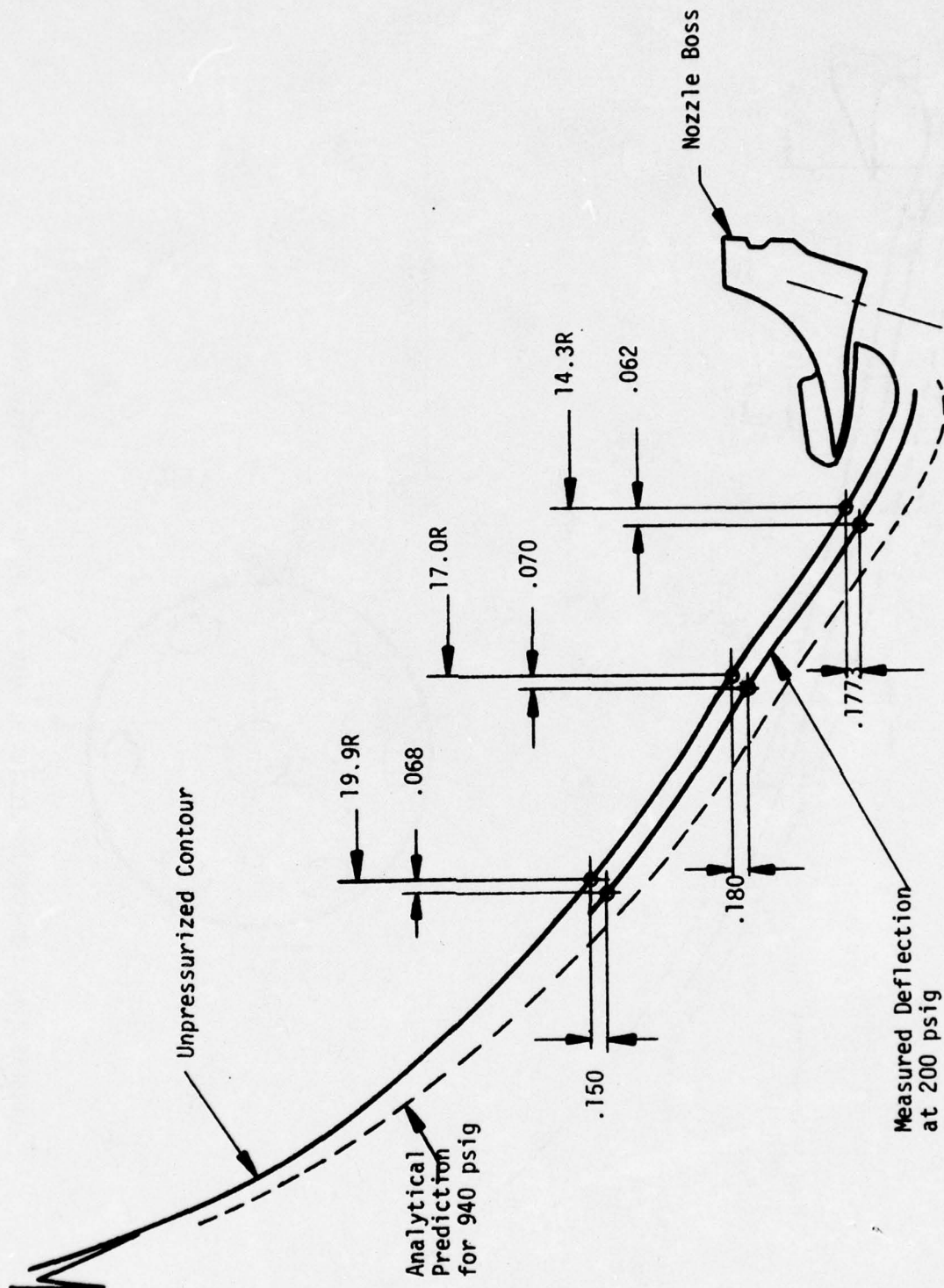


FIGURE B-2. AFT DOME DEFLECTION AT 200 PSIG (INCHES)

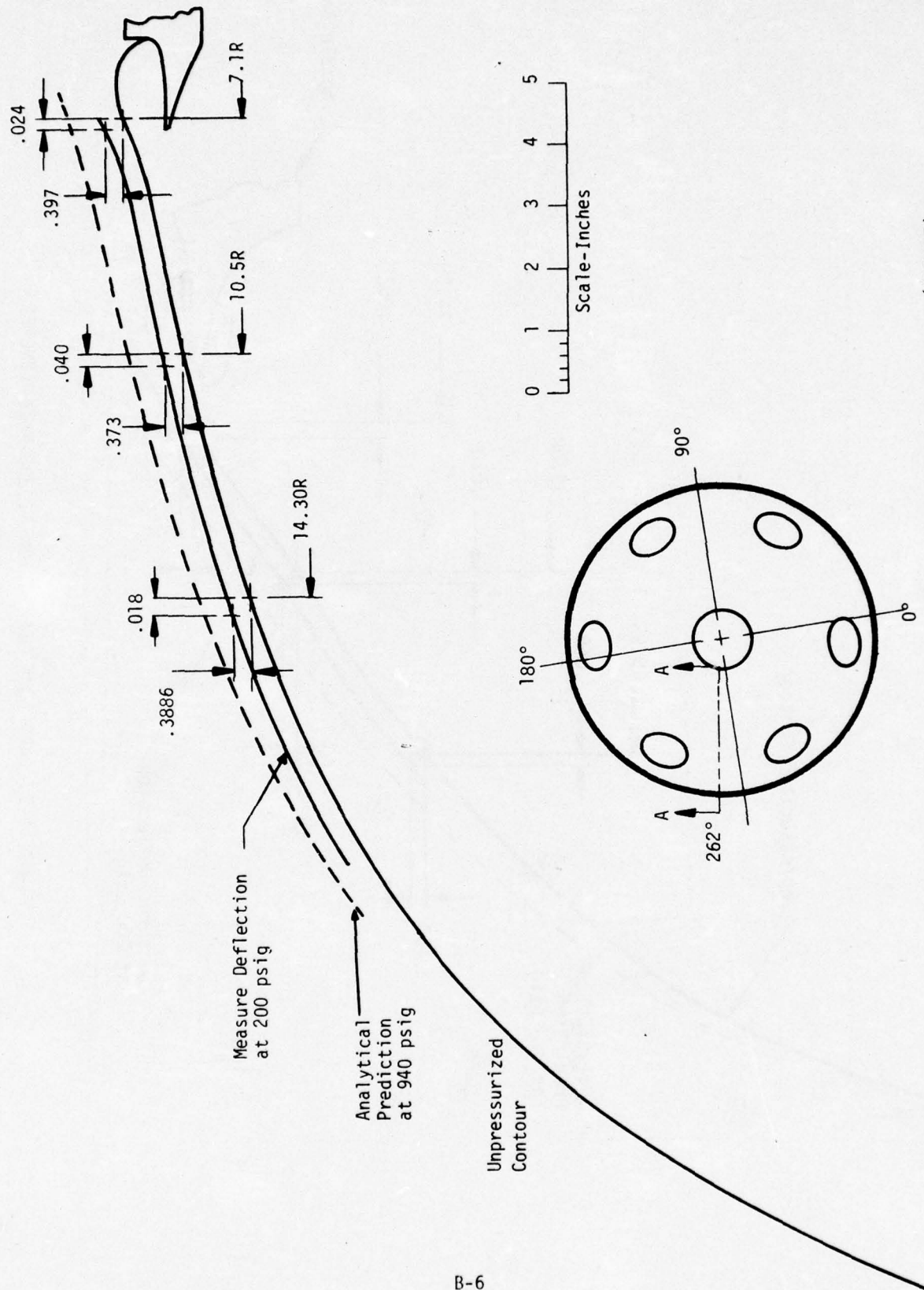


FIGURE B-3. FWD DOME DEFLECTION BETWEEN T.T. PORTS AT 200 PSIG (INCHES)

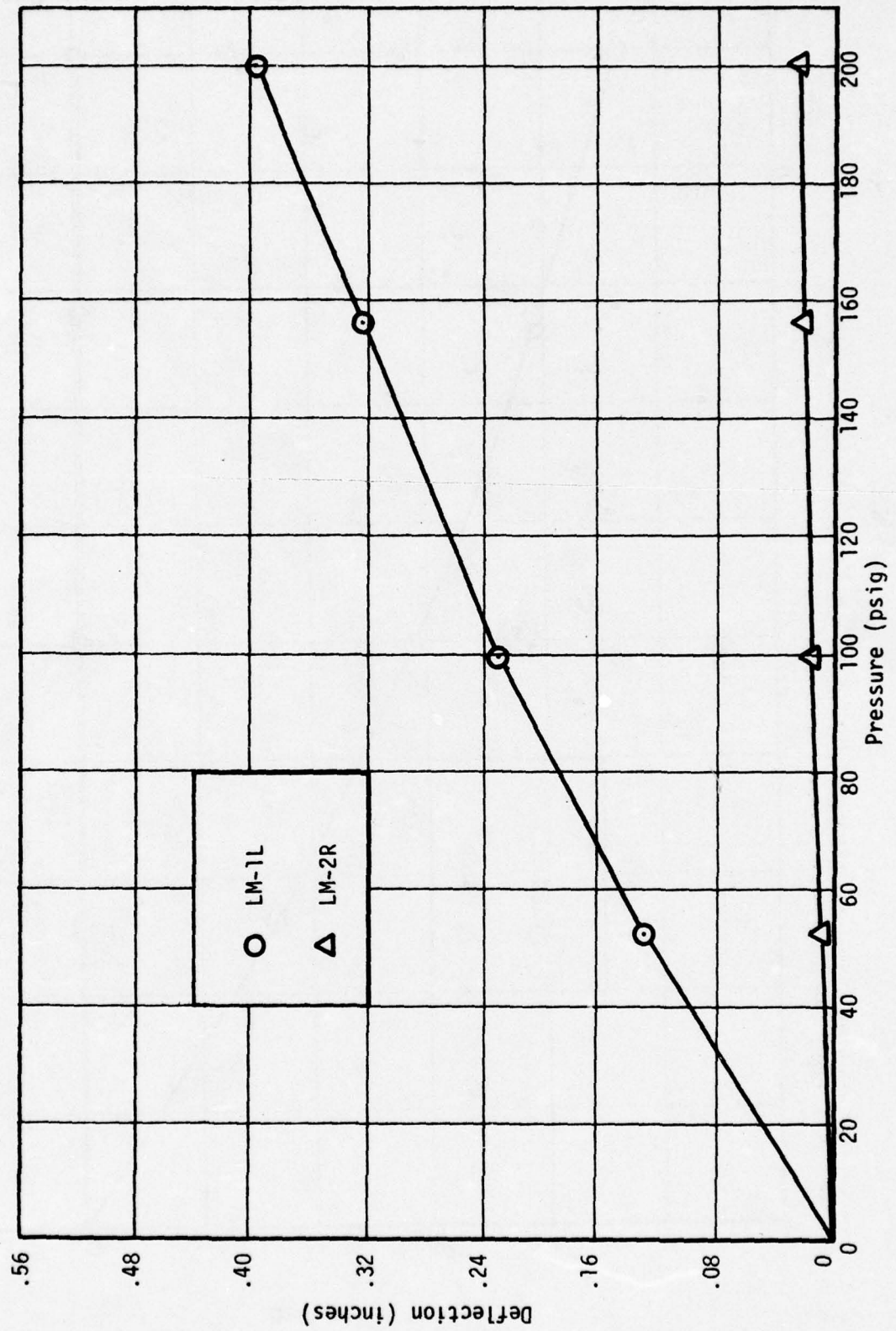


FIGURE B-4. DEFLECTION VS PRESSURE FOR DEVICES LM-1L & LM-2R



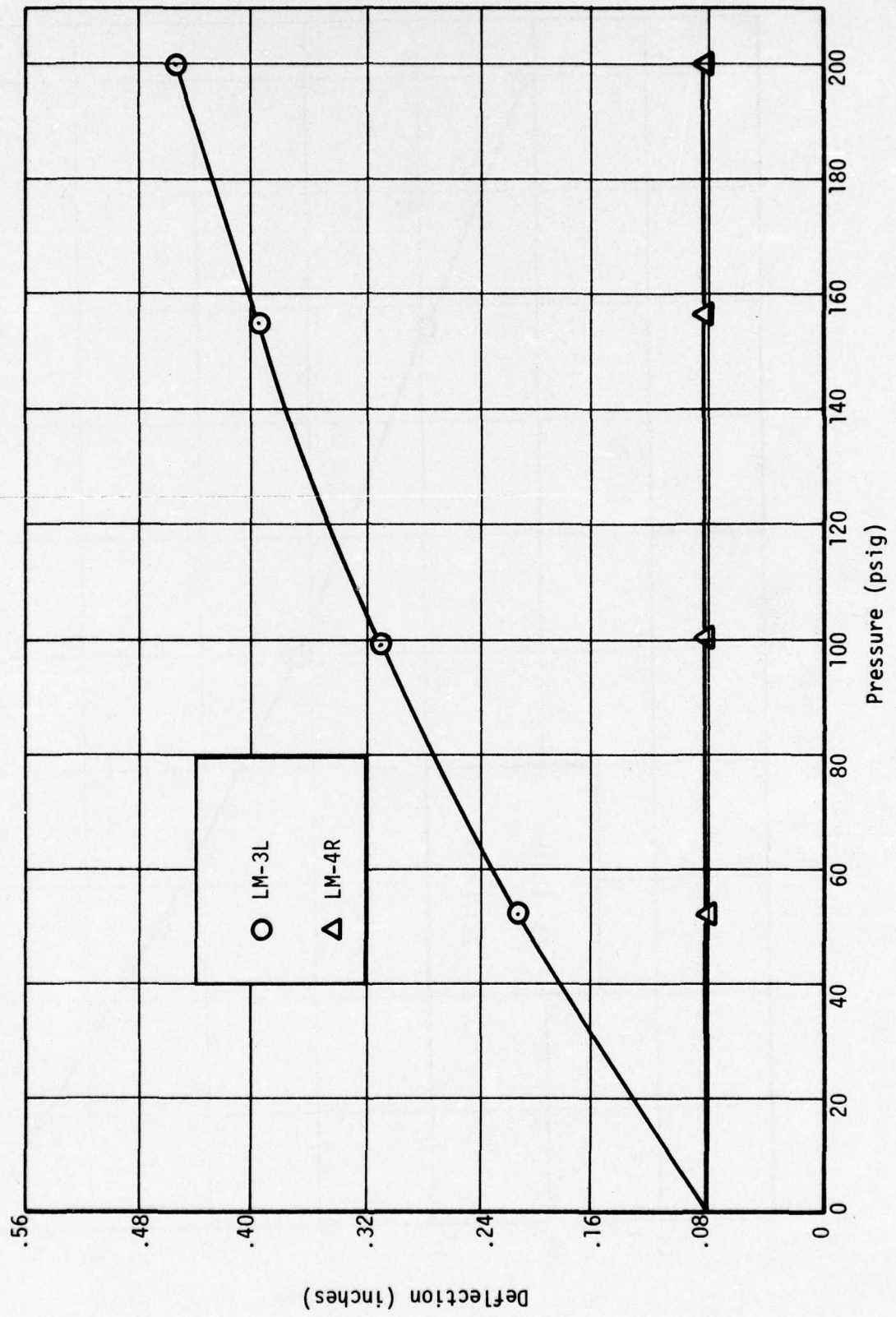


FIGURE B-5. DEFLECTION VS PRESSURE FOR DEVICES LM-3L & LM-4R

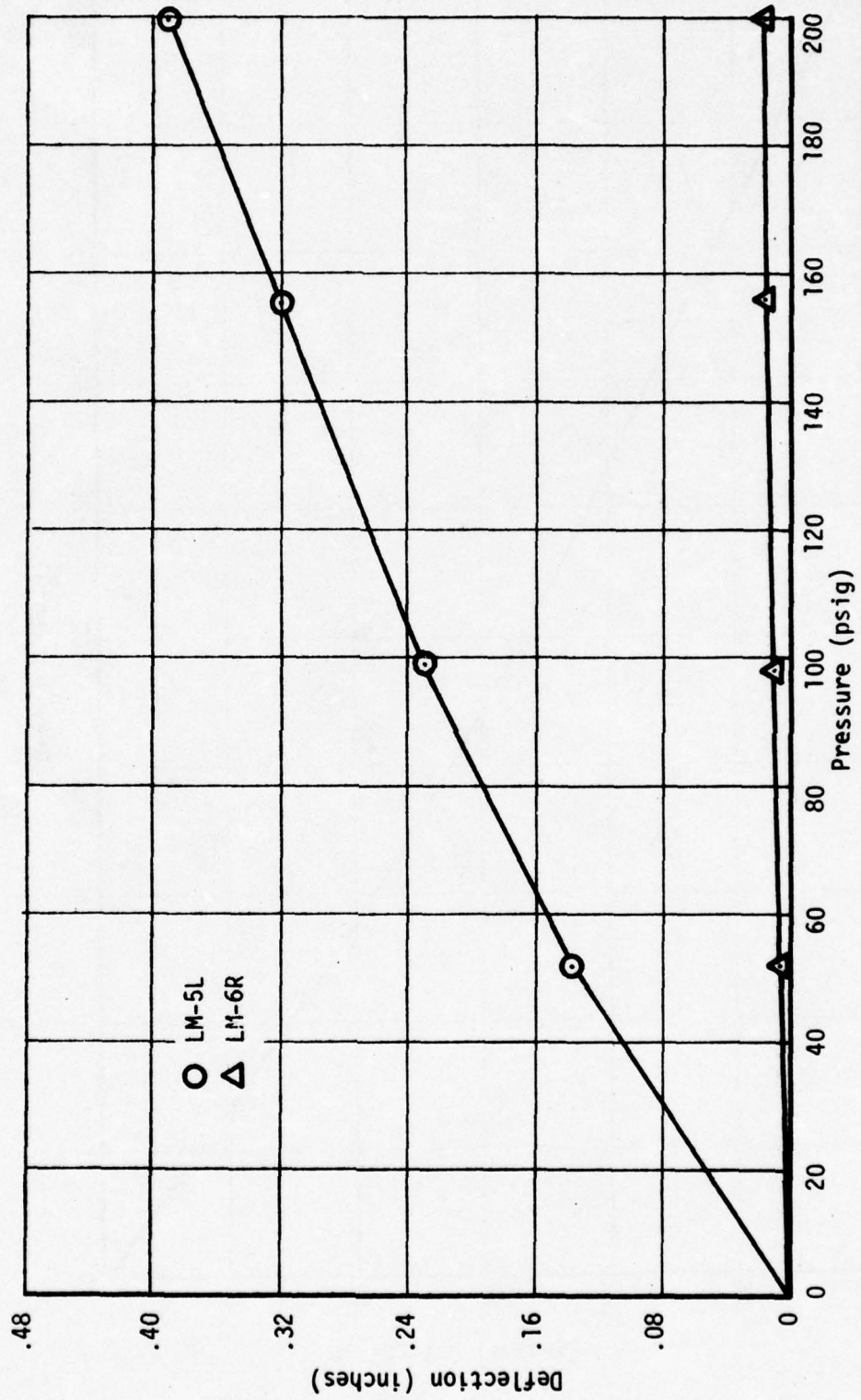


FIGURE B-6. DEFLECTION VS PRESSURE FOR DEVICES LM-5L & LM-6R

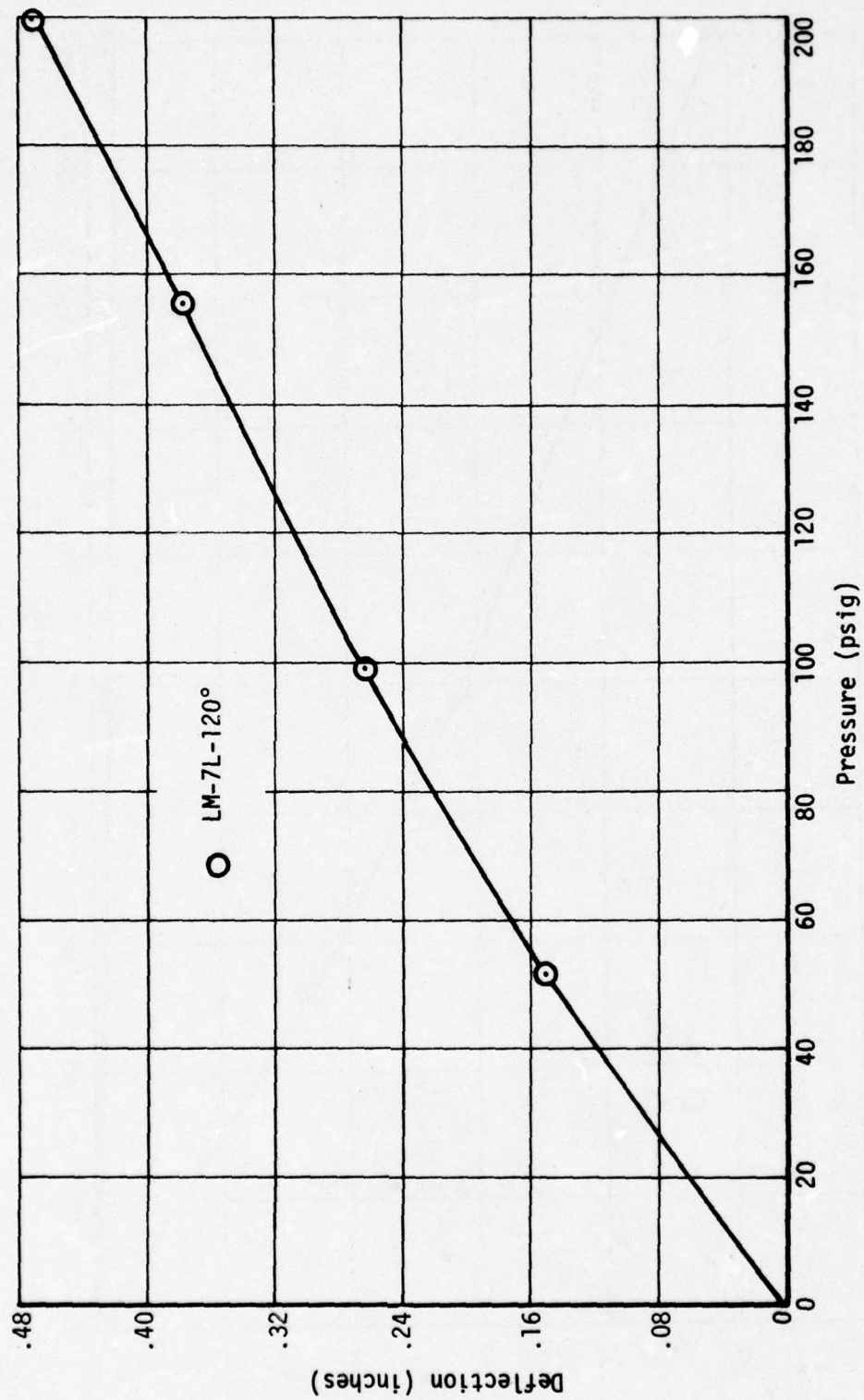


FIGURE B-7. DEFLECTION VS PRESSURE FOR DEVICE LM-7L



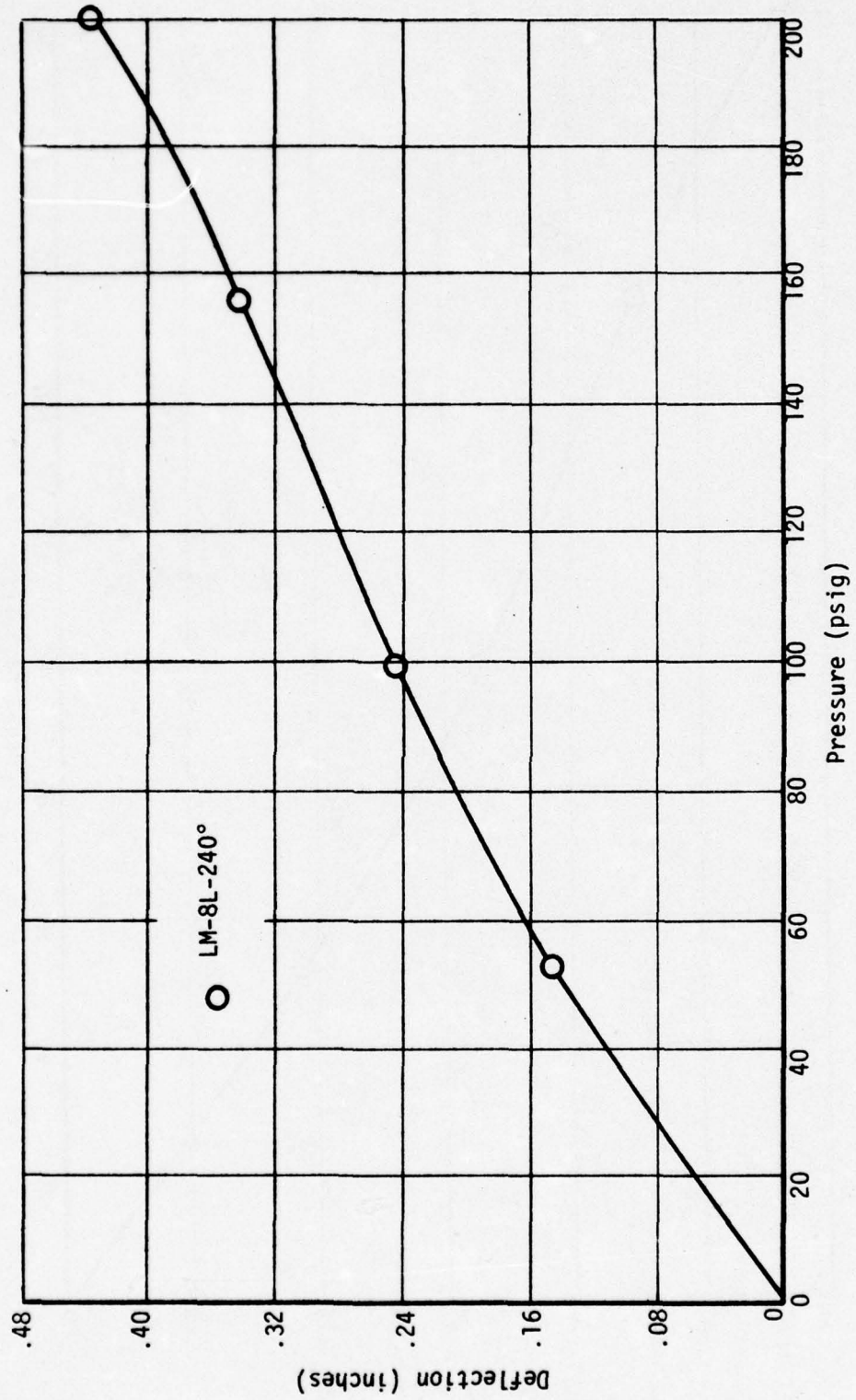


FIGURE B-8. DEFLECTION VS PRESSURE FOR DEVICE LM-8L

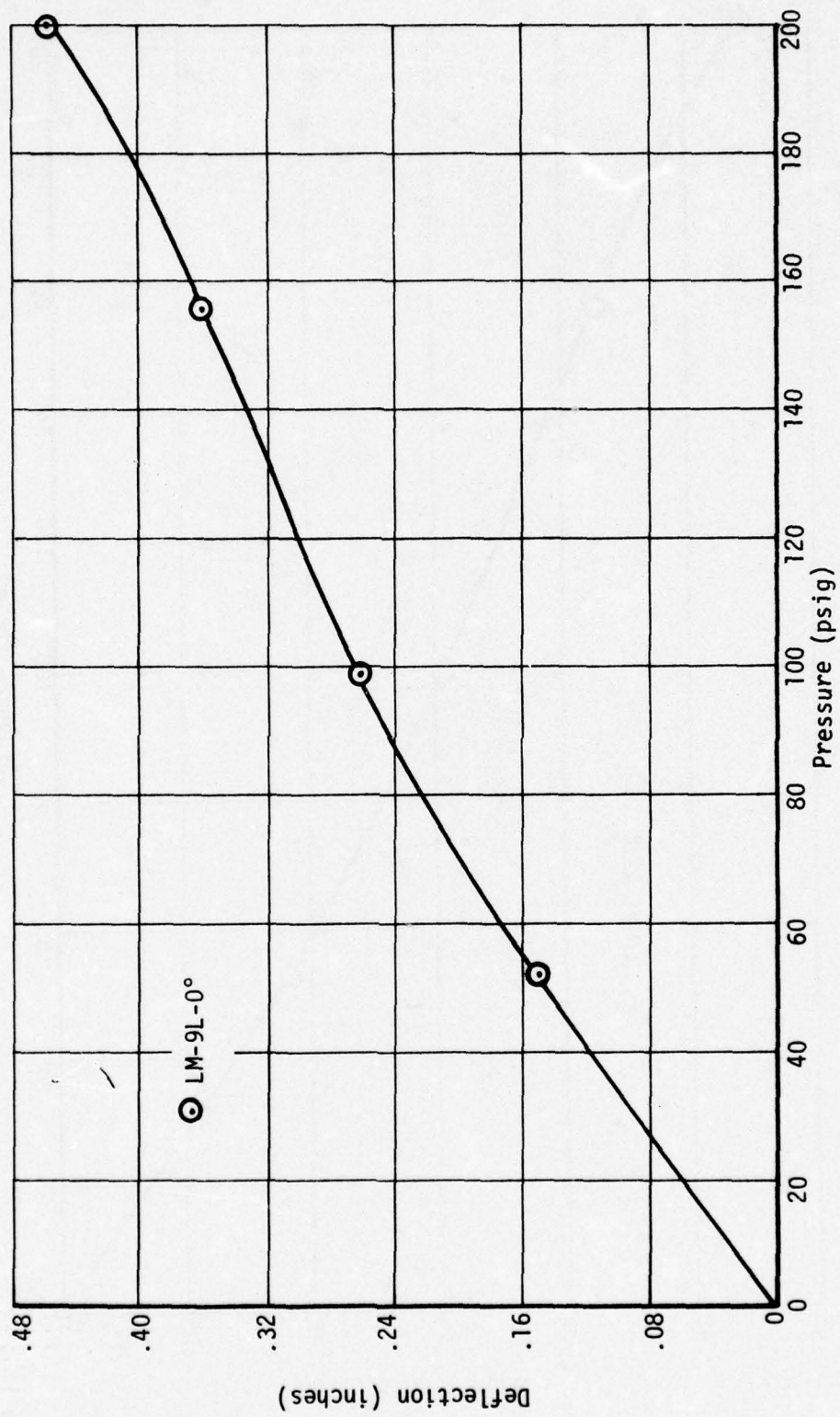


FIGURE B-9. DEFLECTION VS PRESSURE FOR DEVICE LM-9L

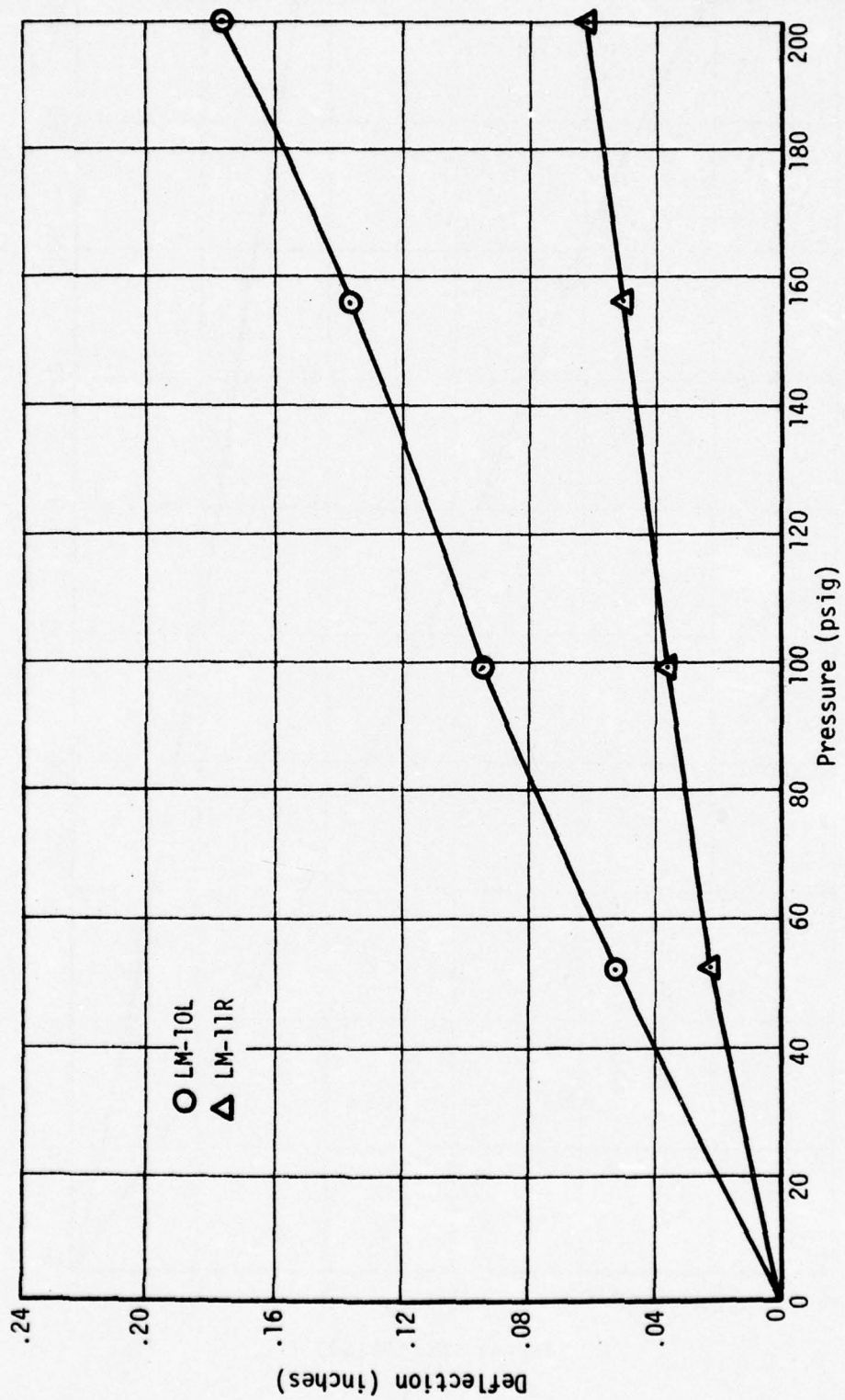


FIGURE B-10. DEFLECTION VS PRESSURE FOR DEVICES LM-10L & LM-11R



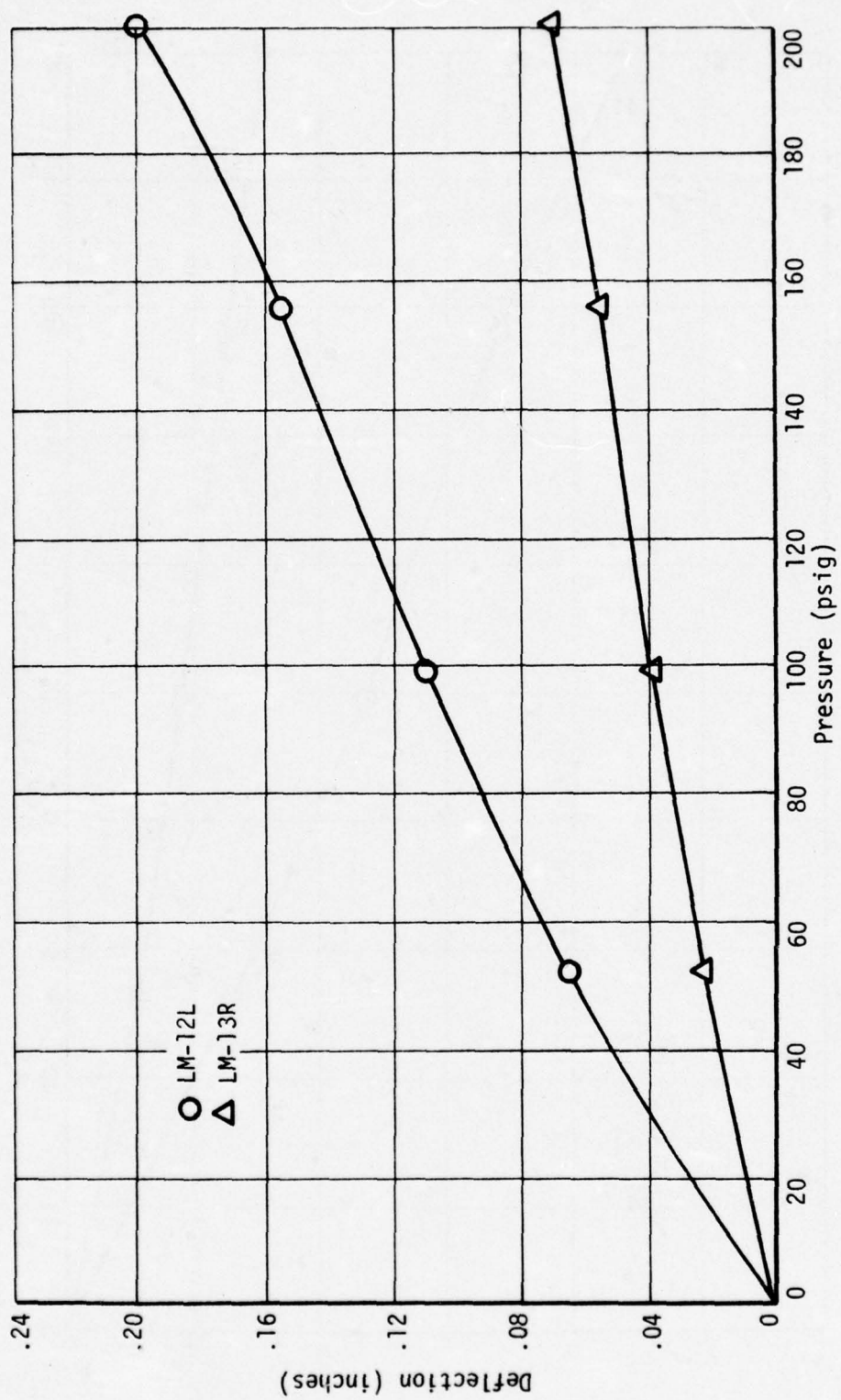


FIGURE B-11. DEFLECTION VS PRESSURE FOR DEVICES LM-12L & LM-13R

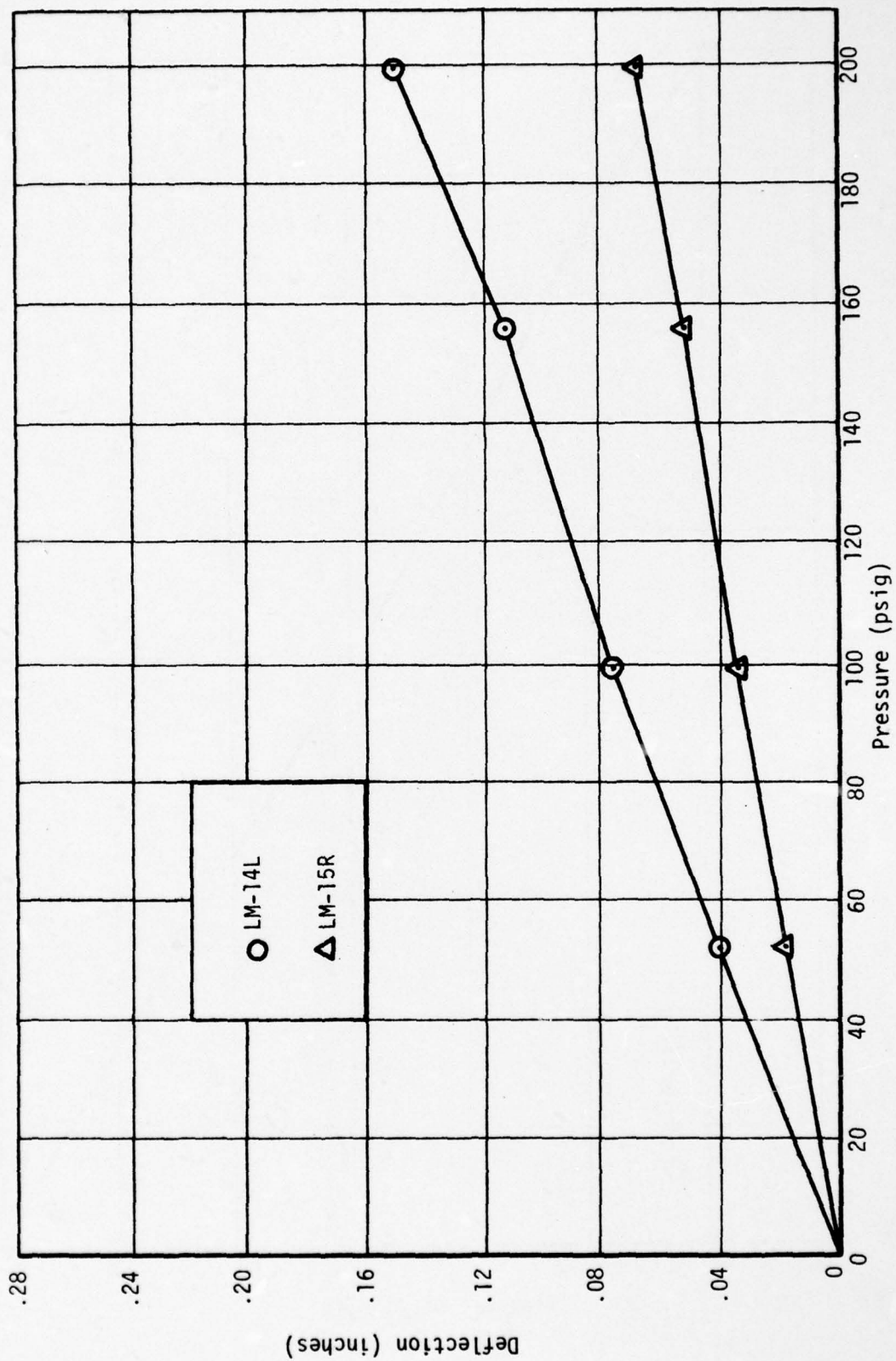


FIGURE B-12. DEFLECTION VS PRESSURE FOR DEVICES LM-14L & LM-15R

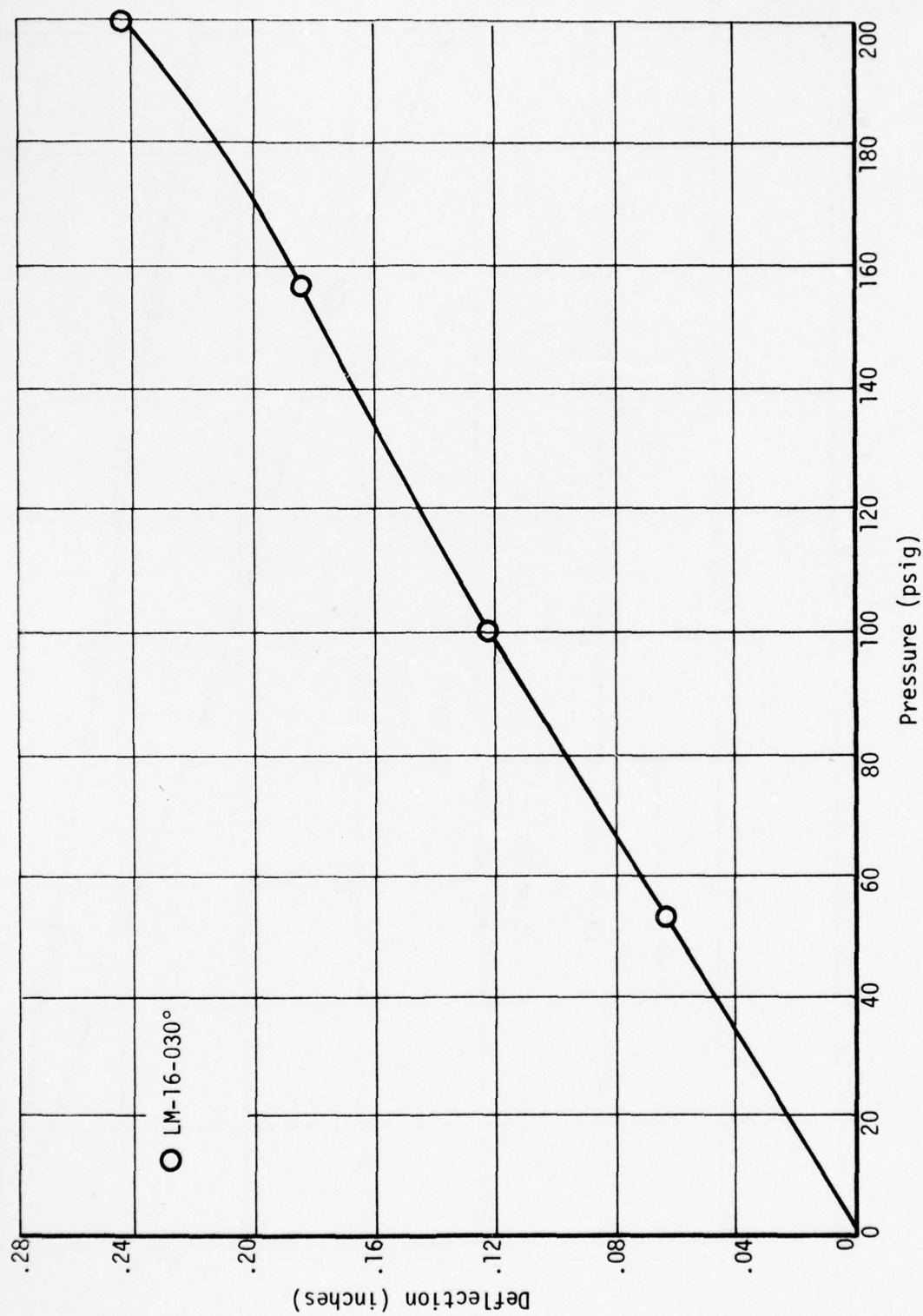


FIGURE B-13. DEFLECTION VS PRESSURE FOR DEVICE LM-16



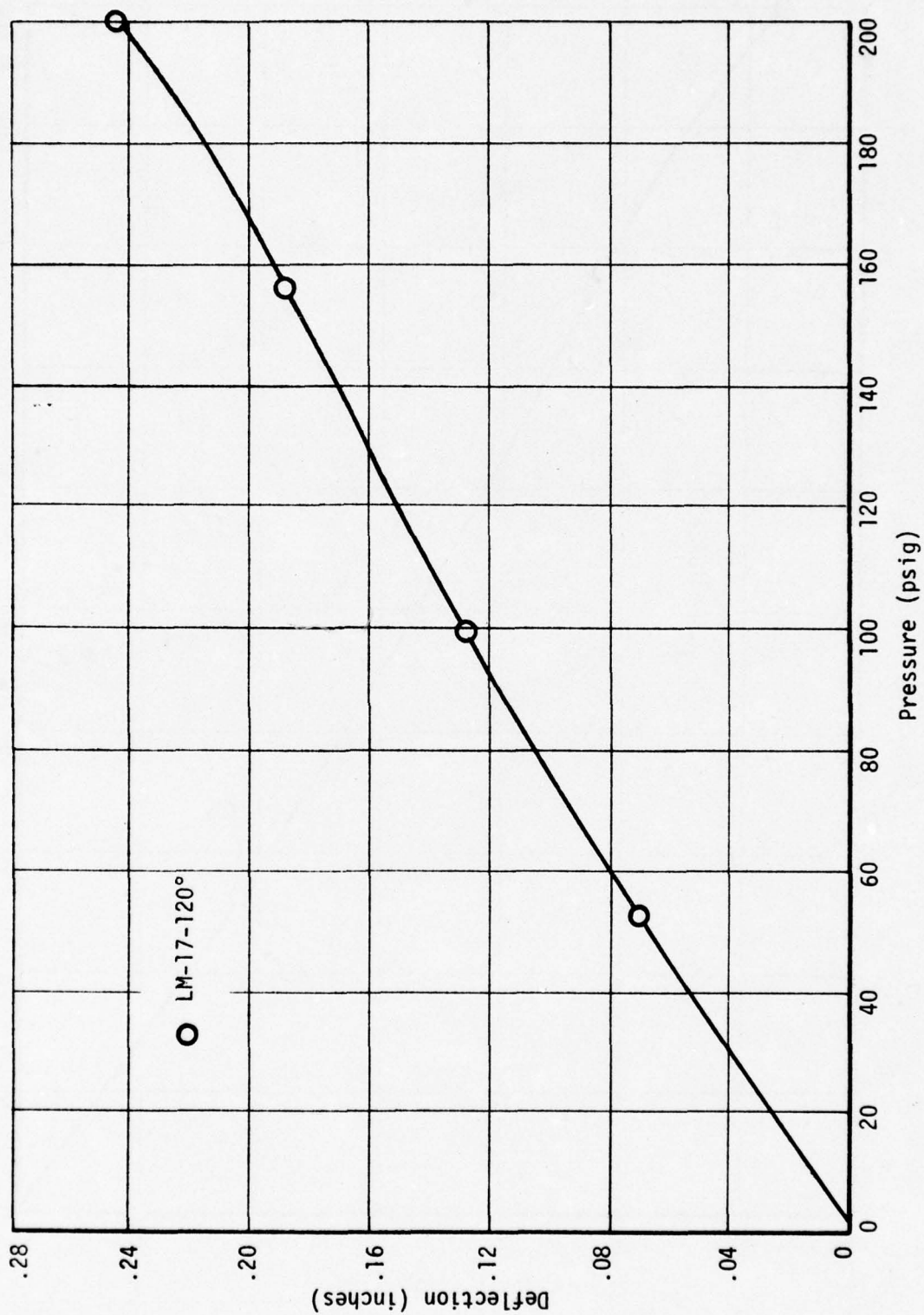


FIGURE B-14. DEFLECTION VS PRESSURE FOR DEVICE LM-17

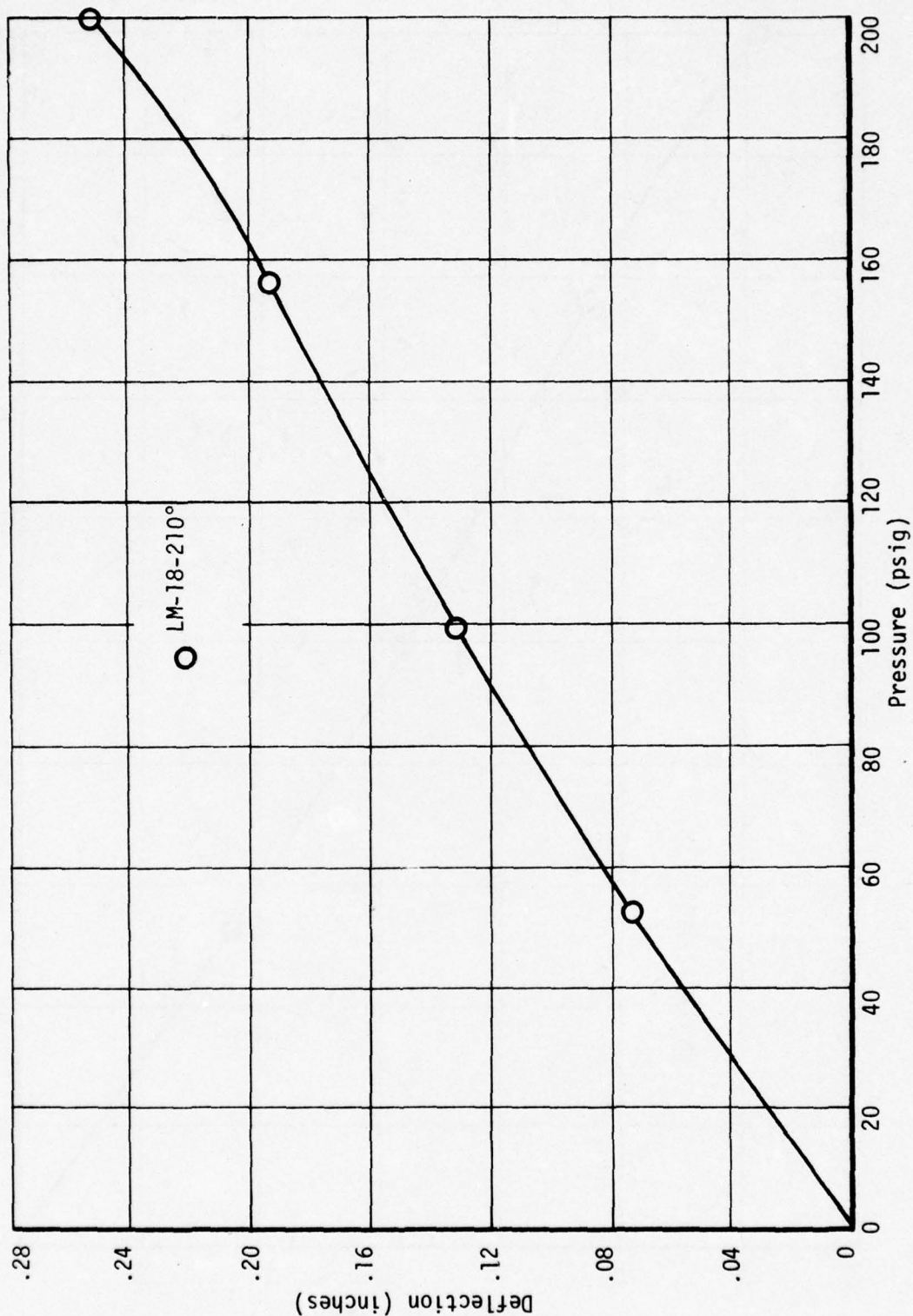


FIGURE B-15. DEFLECTION VS PRESSURE FOR DEVICE LM-18

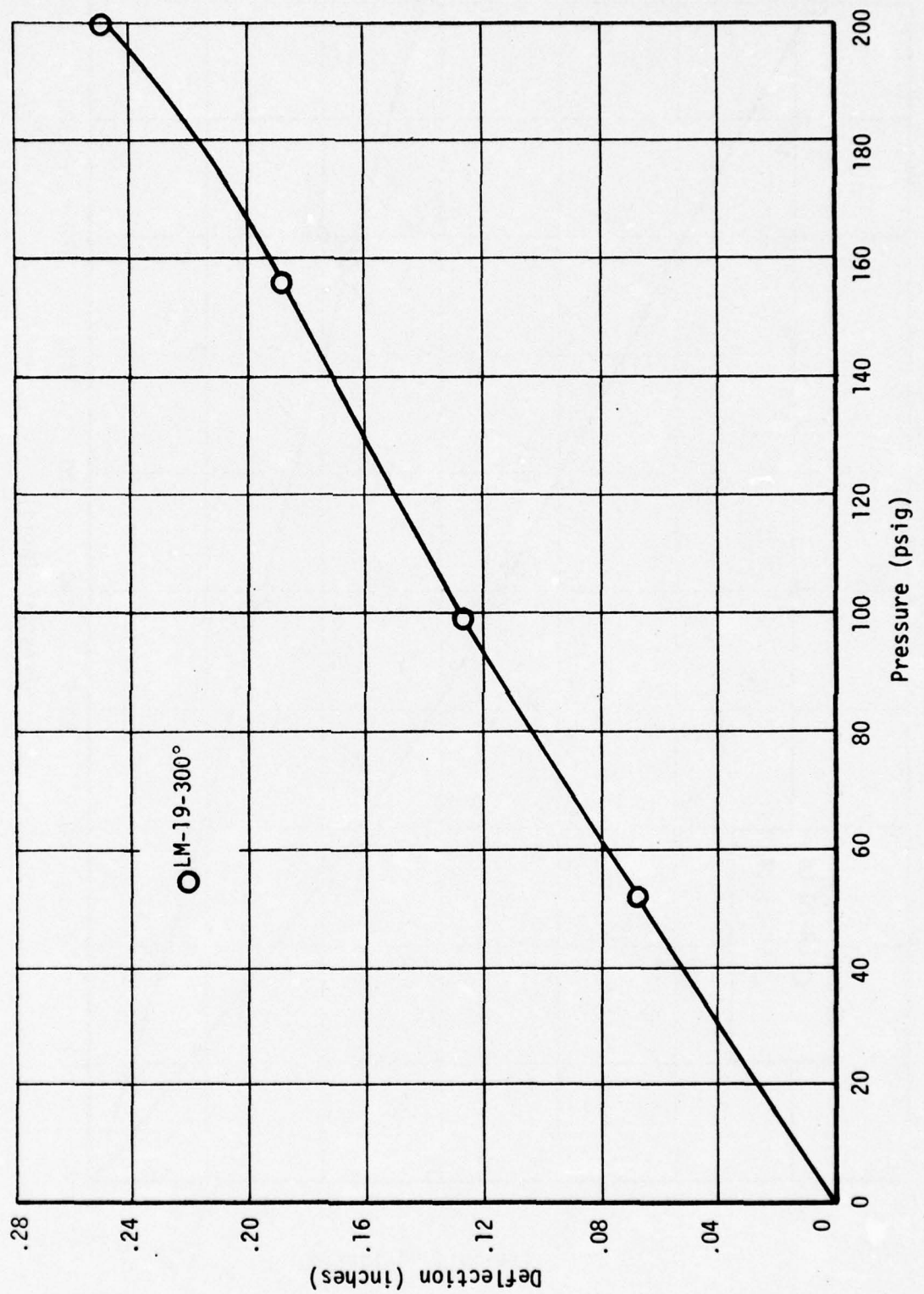


FIGURE B-16. DEFLECTION VS PRESSURE FOR DEVICE LM-19



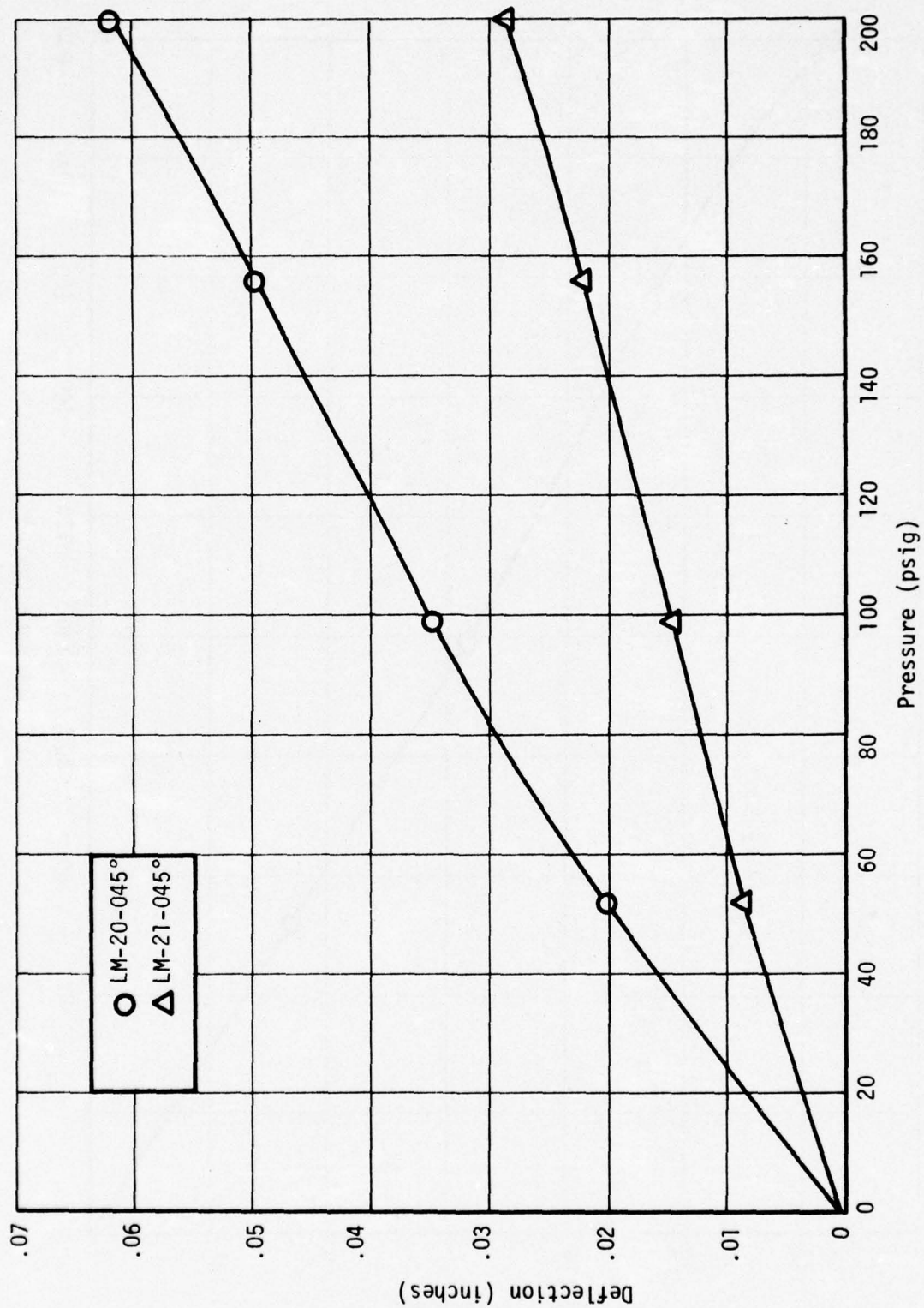


FIGURE B-17. DEFLECTION VS PRESSURE FOR DEVICES LM-20 & LM-21

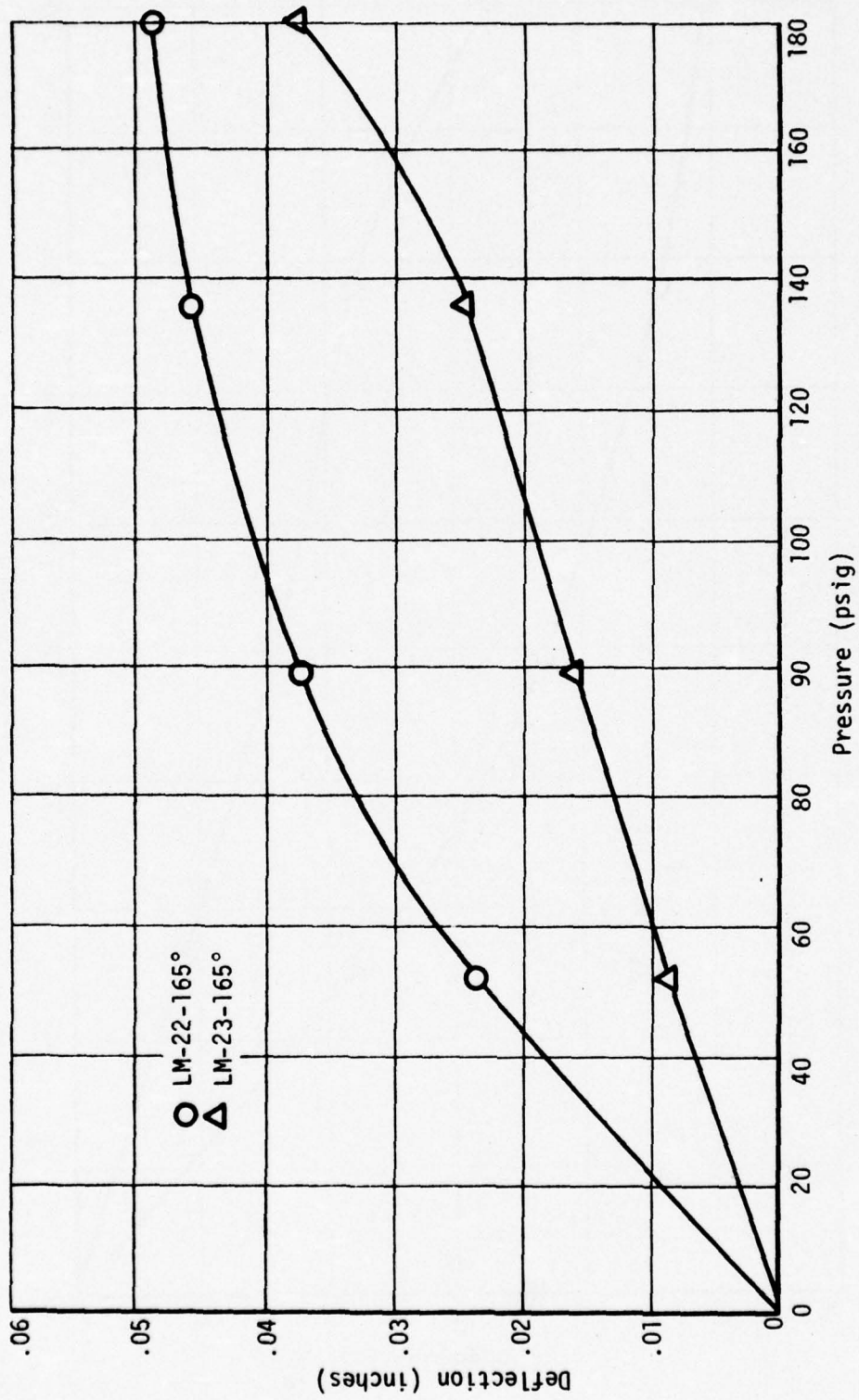


FIGURE B-18. DEFLECTION VS PRESSURE FOR DEVICES LM-22 & LM-23

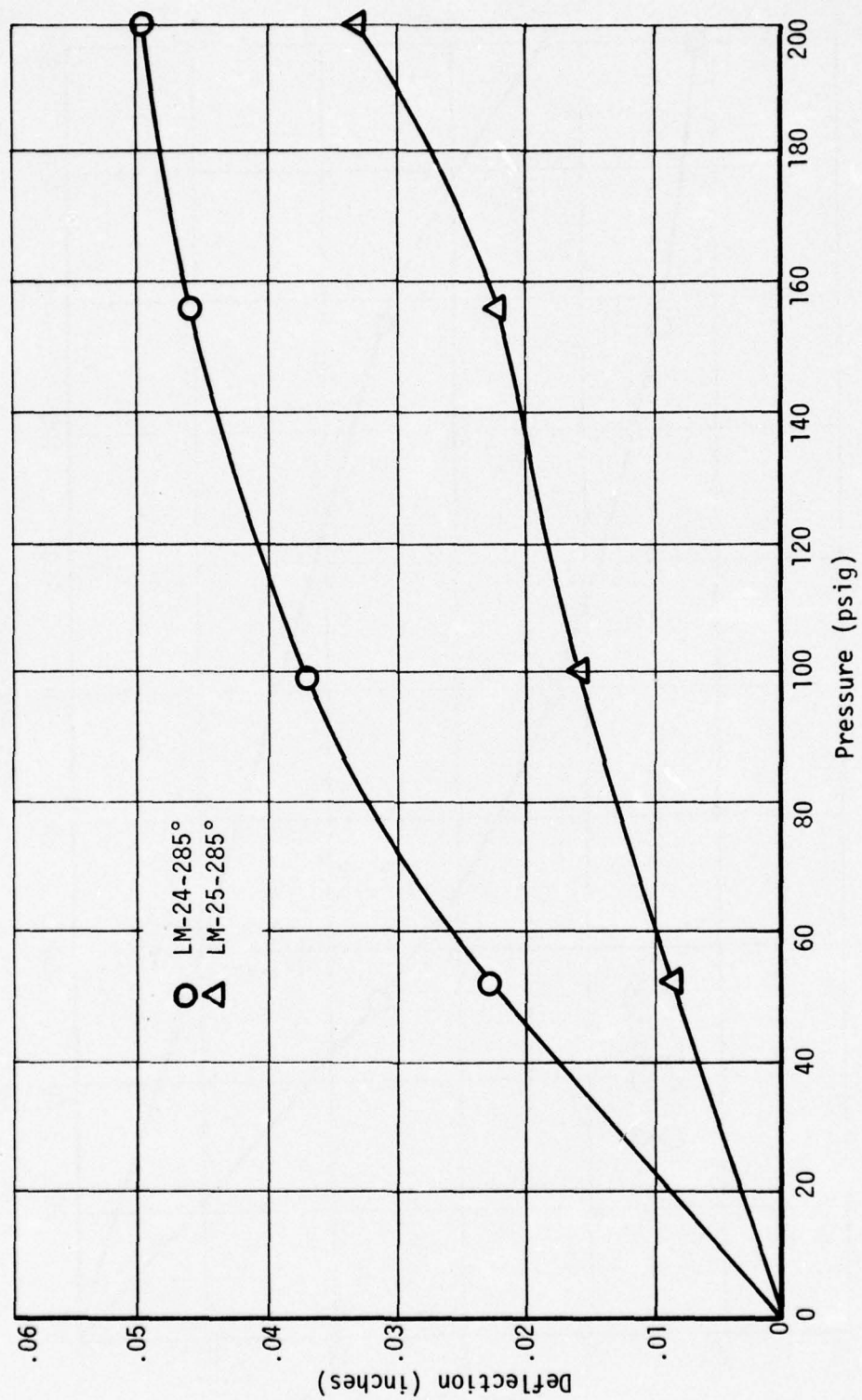


FIGURE B-19. DEFLECTION VS PRESSURE FOR DEVICES LM-29 & LM-25



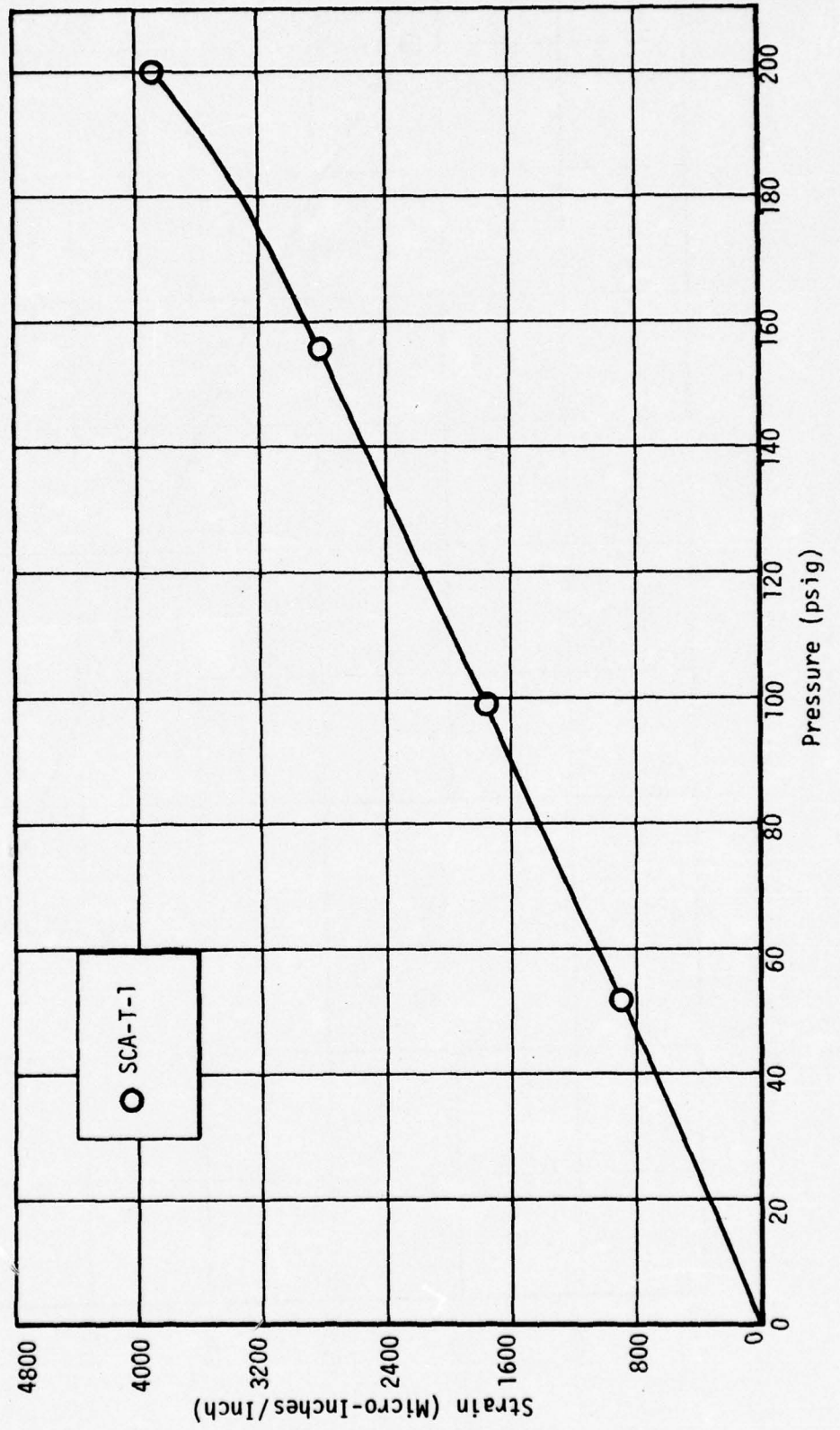


FIGURE B-20. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-1

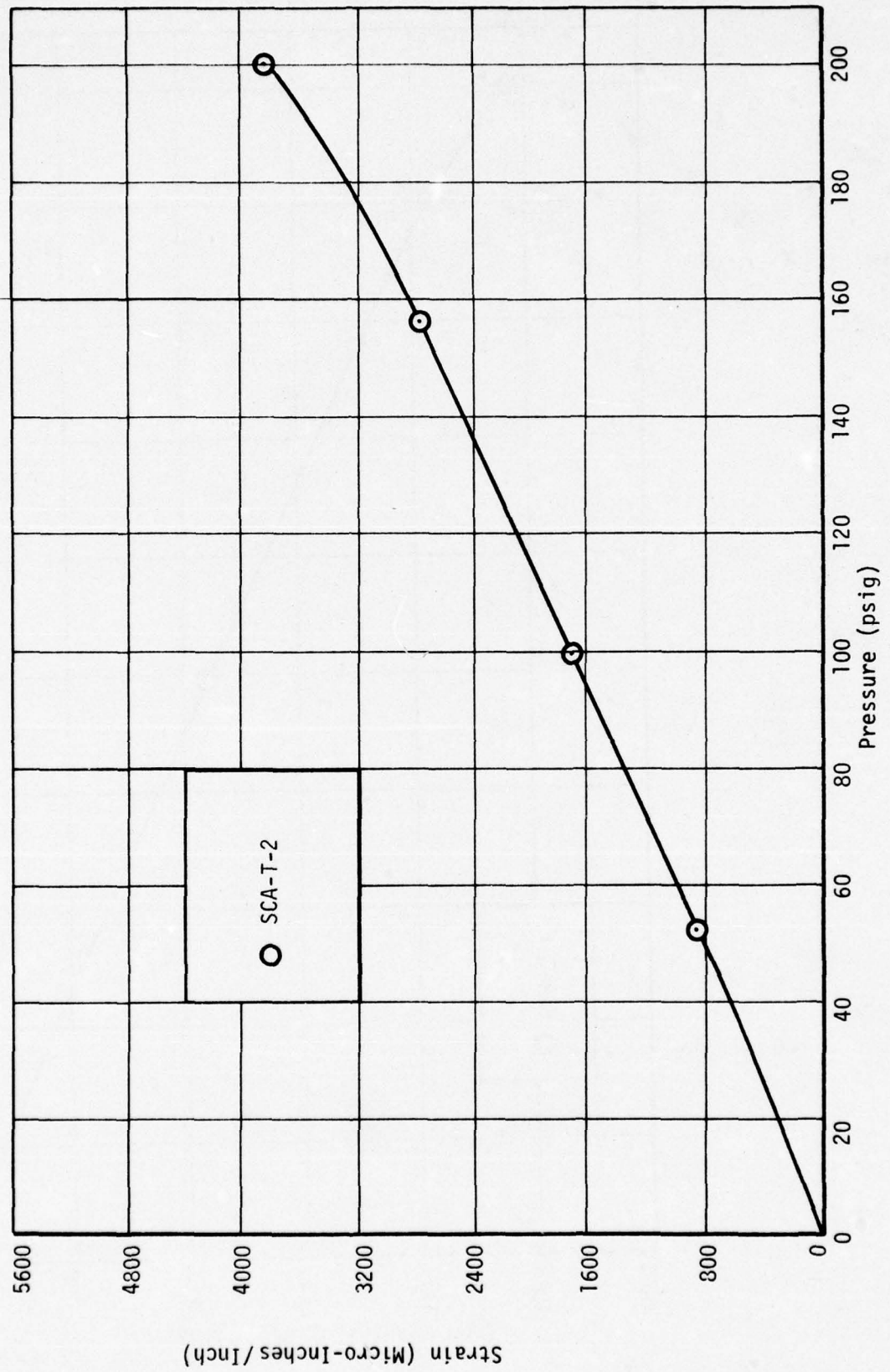


FIGURE B-21. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-2

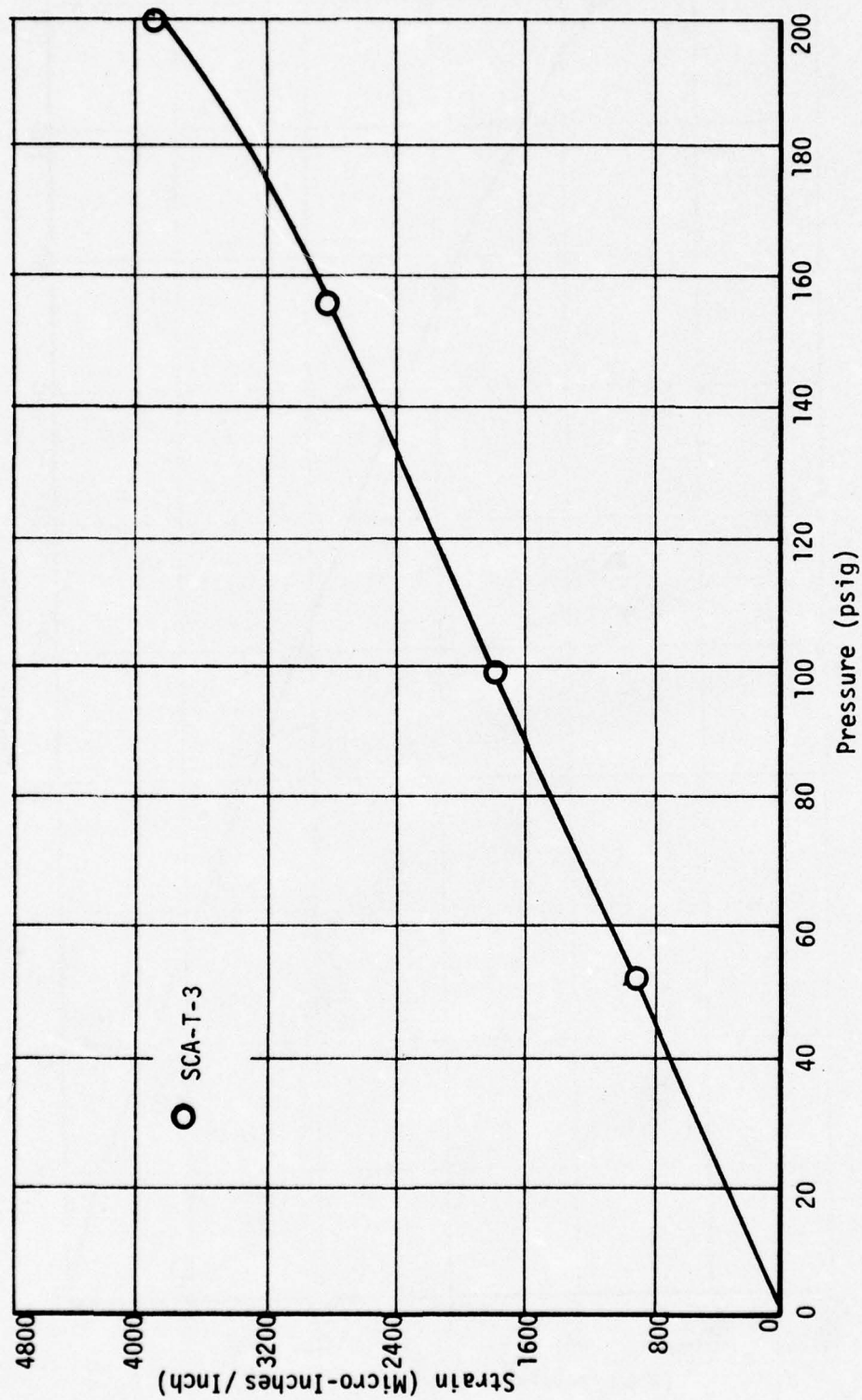


FIGURE B-22. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-3



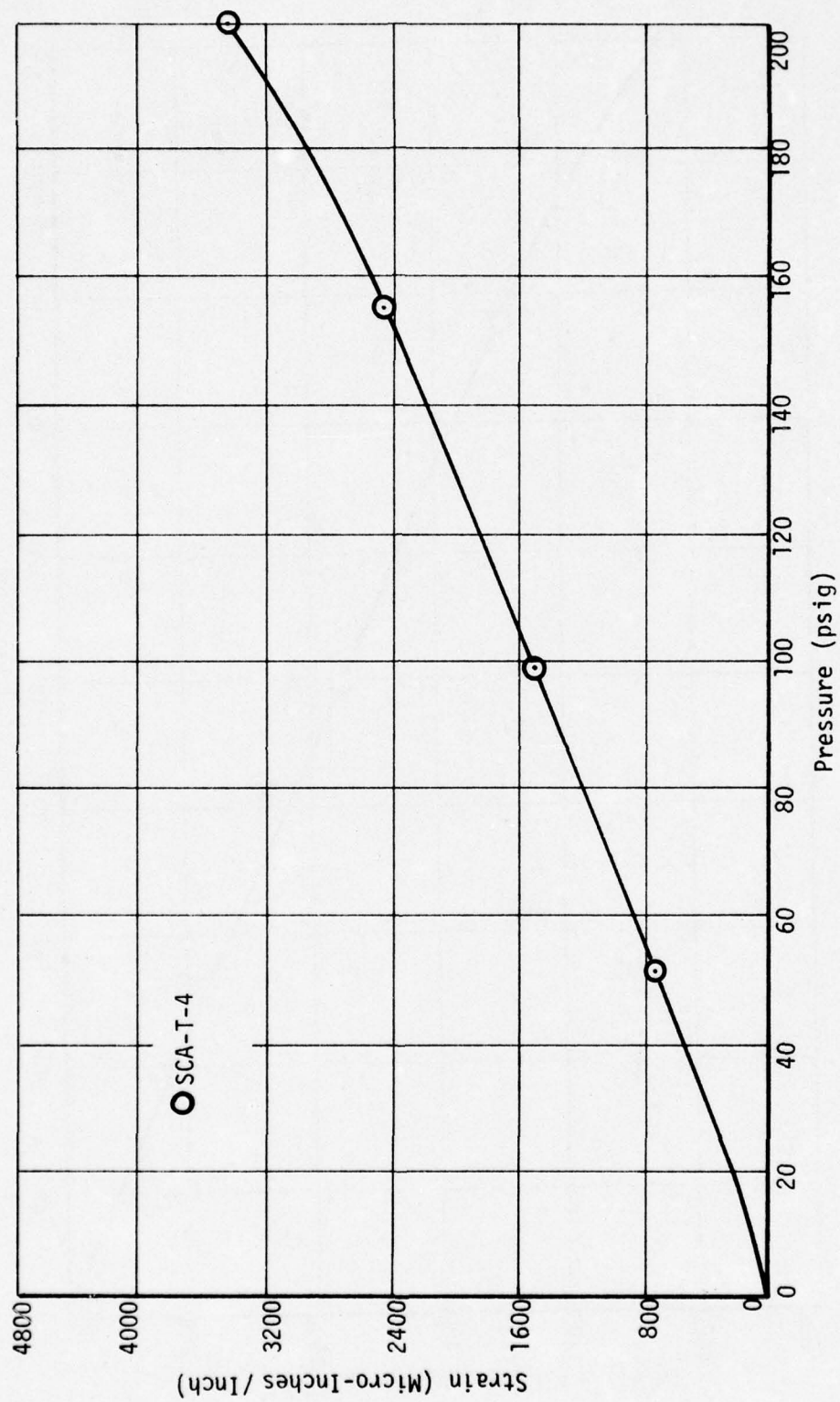


FIGURE B-23. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-4

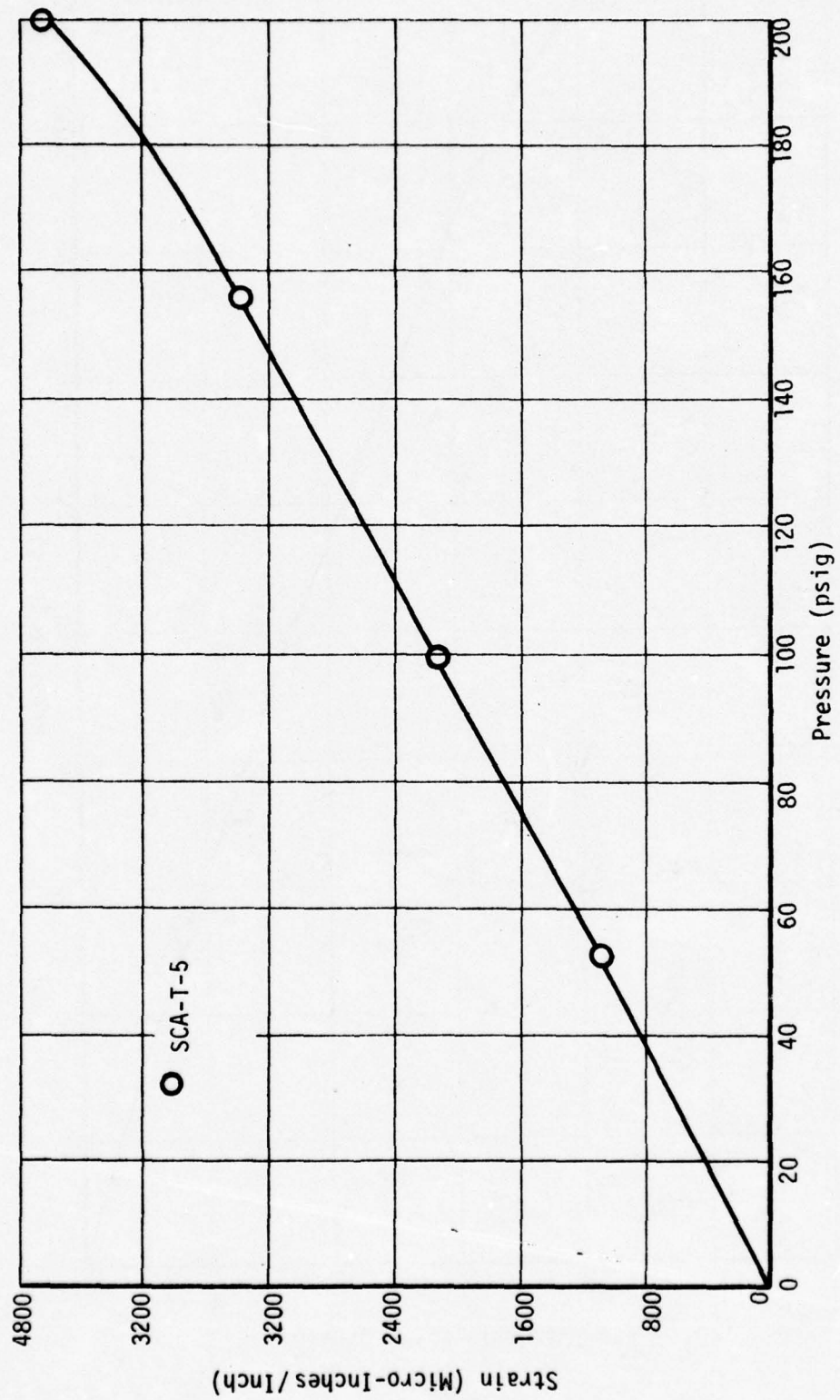


FIGURE B-24. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-5

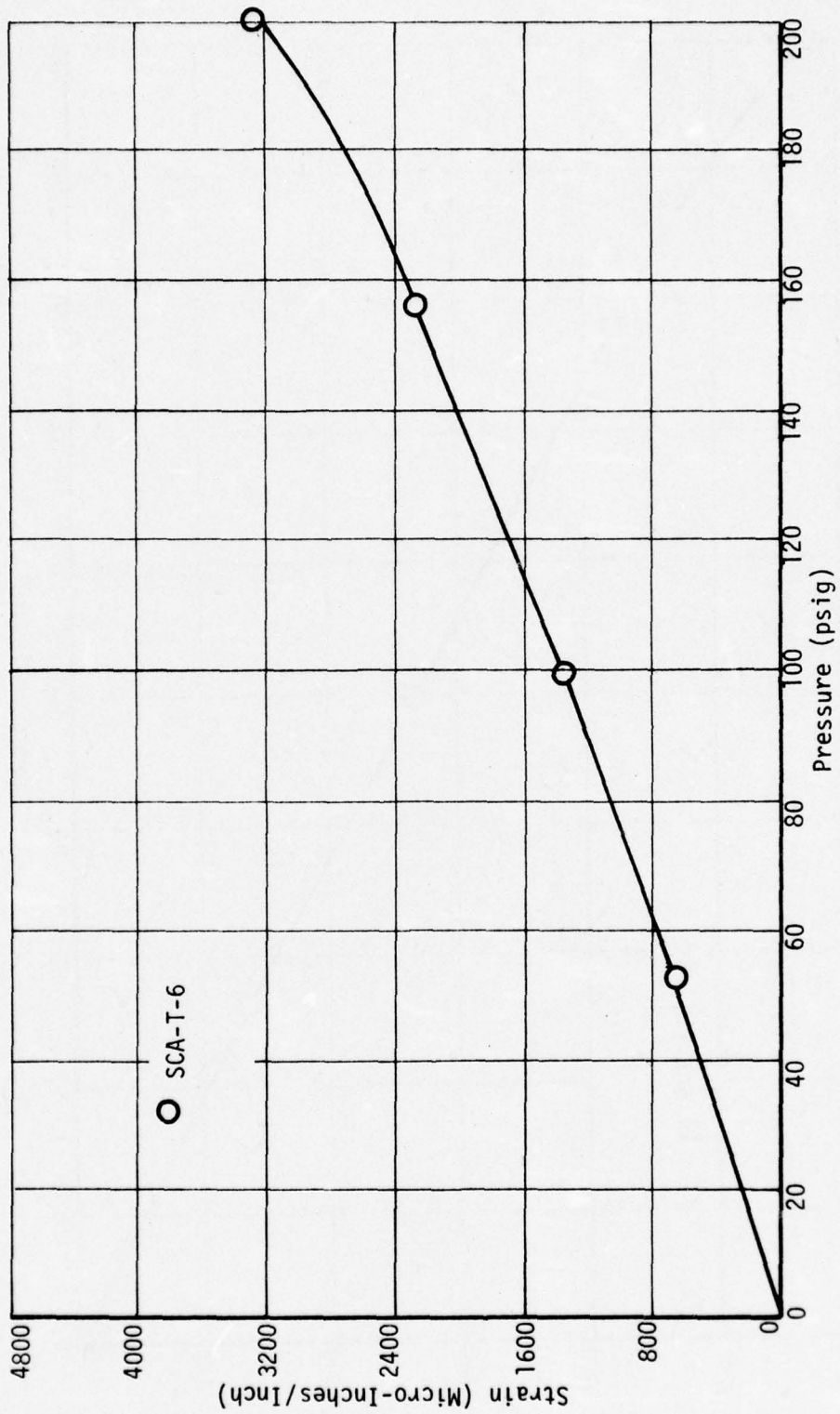


FIGURE B-25. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-6



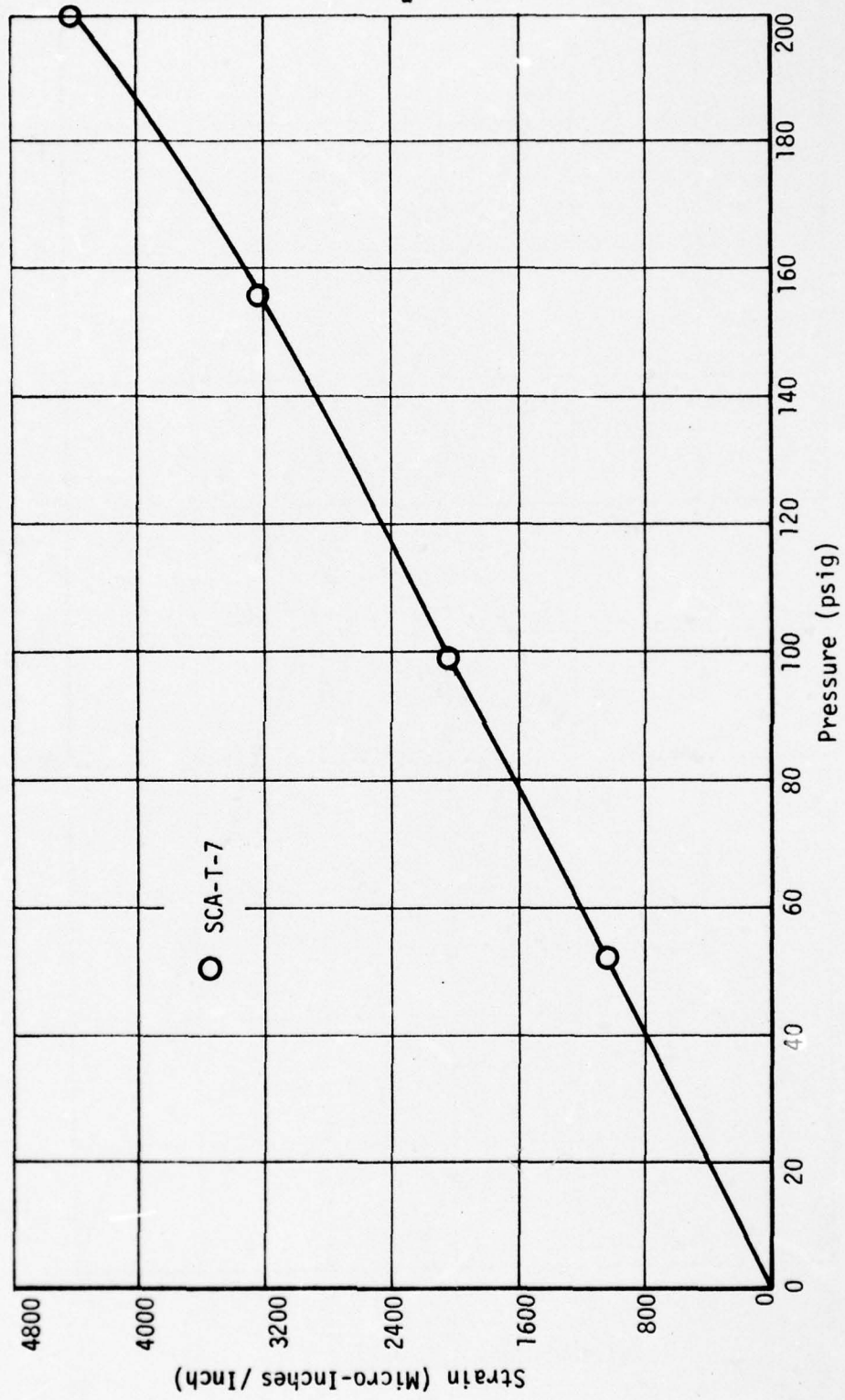


FIGURE B-26. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-7

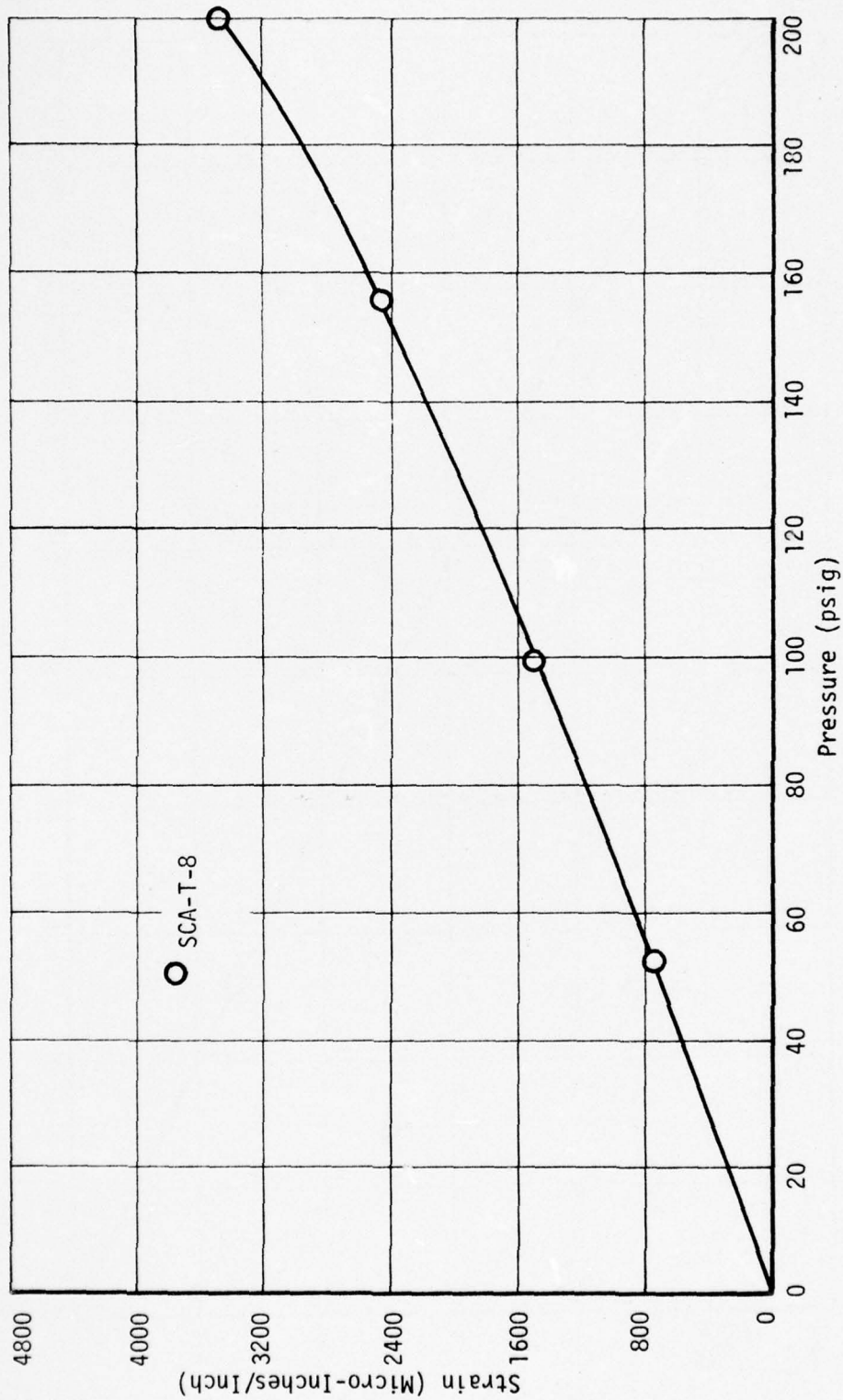


FIGURE B-27. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-8

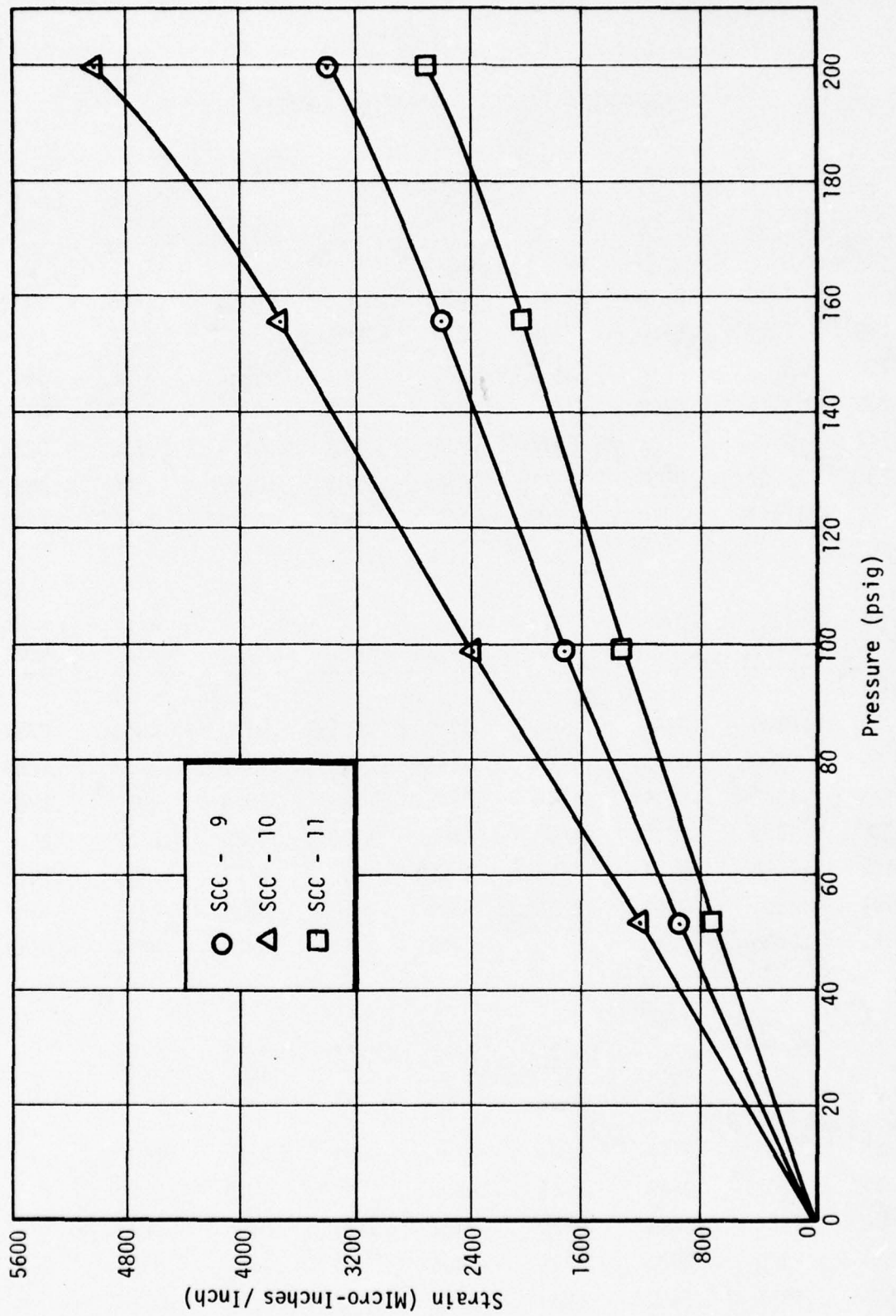


FIGURE B-28. STRAIN VS PRESSURE FOR CIRCUMFERENTIAL GAGE NOS. SCC-9, SCC-10, & SCC-11



TABLE B-1  
HYDROTEST RESPONSE DATA FOR CHAMBER S/N 30113

PC (psig)	Deflections - Inches						LM-7L- 120°	LM-8L- 240°	LM-9L- 0°
	LM-1L	LM-2R	LM-3L	LM-4R	LM-5L	LM-6R			
52	.1310	.0098	.1341	.002	.1358	.0058	.1500	.1450	.1500
99	.2310	.0168	.2295	.020	.2286	.0105	.2625	.2452	.2614
156	.3238	.0218	.3143	.032	.3182	.0152	.3775	.3428	.3605
200	.3976	.0240	.3727	.040	.3886	.0181	.4714	.4350	.4579
142	.3905	.0182	.2952	.036	.3068	.0138	.3600	.3214	.3405
101	.2357	.0138	.2273	.034	.2405	.0107	.2725	.2381	.2568
52	.1310	.0070	.1318	.028	.1500	.0058	.1591	.1300	.1350

PC (psig)	LM-10L						LM-16- 30°	LM-17- 120°	LM-18- 210°
	LM-10L	LM-11R	LM-12L	LM-13R	LM-14L	LM-15R			
52	.0525	.0235	.0443	.0220	.0405	.0190	.0643	.0714	.0738
99	.0950	.0373	.0895	.0381	.0762	.0345	.1238	.1286	.1310
156	.1364	.0504	.1348	.0545	.1125	.0520	.1857	.1886	.1928
200	.1773	.0622	.1800	.0700	.1500	.0685	.2452	.2452	.2524
142	.1318	.0477	.1276	.0515	.1100	.0475	.1810	.1818	.1881
101	.0950	.0359	.0895	.0381	.0810	.0330	.1310	.1357	.1405
52	.0500	.0222	.0419	.0205	.0428	.0170	.0714	.0786	.0833

PC (psig)	LM-19- 300°						LM-25- 285°
	LM-19- 300°	LM-20- 45°	LM-21- 45°	LM-22- 165°	LM-23- 165°	LM-24- 285°	
52	.0667	.0198	.0084	.0236	.0089	.0228	.0087
99	.1262	.0348	.0145	.0370	.0162	.0370	.0160
156	.1881	.0495	.0220	.0457	.0245	.0460	.0223
200	.2500	.0620	.0285	.0486	.0376	.0497	.0332
142	.1857	.0463	.0205	.0443	.0273	.0406	.0227
101	.1357	.0340	.0150	.0368	.0200	.0320	.0160
52	.0762	.0182	.0095	.0225	.0114	.0205	.0083

TABLE B-2  
HYDROTEST RESPONSE DATA FOR CHAMBER S/N 30113

PC (psig)	Strain - Microinches/Inch							
	SCAT-1	SCAT-2	SCAT-3	SCAT-4	SCAT-5	SCAT-6	SCAT-7	SCAT-8
52	887	835	910	719	1092	647	1039	736
99	1749	1693	1778	1489	2135	1351	2023	1497
156	2810	2751	2812	2448	3383	2265	3211	2462
200	3888	3809	3879	3482	4631	3276	4399	3476
142	2736	2679	2746	2316	3276	2192	3085	2397
101	1923	1860	1943	1555	2299	1472	2149	1652
52	987	938	1001	719	1174	696	1078	834

PC (psig)	SCC-9	SCC-10	SCC-11
52	953	1223	711
99	1748	2397	1332
156	2595	3742	2030
200	3389	5039	2715
142	2489	3522	1952
101	1814	2495	1409
52	1006	1296	763

TABLE B-3

## SUMMARY RESPONSE DATA FOR CHAMBER S/N 30113

<u>Hoop Strains at Barrel Section</u> (Micro inches/inch)		<u>Pressure, psig</u>				
		<u>0</u>	<u>52</u>	<u>99</u>	<u>156</u>	<u>200</u>
Forward Equator		0	953	1748	2595	3389
Mid Barrel		0	1223	2397	3742	5039
Aft Equator		0	711	1332	2030	2715
<u>Fwd Dome Deflections (inches)</u>						
7.1 Radius	Longitudinal	0	.1310	.2310	.3238	.3976
	Radial	0	.0098	.0168	.0218	.0240
10.5 Radius	Longitudinal	0	.1341	.2295	.3143	.3727
	Radial	0	.0020	.0200	.0320	.0400
14.3 Radius	Longitudinal	0	.1358	.2286	.3182	.3886
	Radial	0	.0058	.0105	.0152	.0181
<u>Aft Dome Deflections (inches)</u>						
14.3 Radius	Longitudinal	0	.0525	.0950	.1364	.1773
	Radial	0	.0235	.0373	.0504	.0622
17.0 Radius	Longitudinal	0	.0443	.0895	.1348	.1800
	Radial	0	.0220	.0381	.0545	.0700
19.9 Radius	Longitudinal	0	.0405	.0762	.1125	.1500
	Radial	0	.0190	.0345	.0520	.0685



APPENDIX C

HYDROTEST DATA FOR CHAMBER S/N 30114

## HYDROTEST DATA FOR CHAMBER S/N 30113

Tabulated data and plots of case deflection and strain versus internal pressure are given for each measuring device. Tables C-1 and C-2 list the data of the deflection devices and strain gages, while Table C-3 summarizes the more pertinent data; i.e., of the hoop strains in the barrel, and the forward and aft dome deflections. Graphic illustrations of the hoop strains in the cylinder section and the aft and forward dome deflections of 200 psig are shown in Figures C-1, C-2, and C-3. Figures C-4 to C-9 show the curves for the forward head deflection versus pressure, while Figures C-10 to C-16 show the aft head deflections versus pressure. Figures C-17 to C-19 show the longitudinal strain in the barrel section. Figures C-20 to C-27 show the strains in the aft head. Figure C-28 presents the circumferential growth of the case with chamber pressurization.

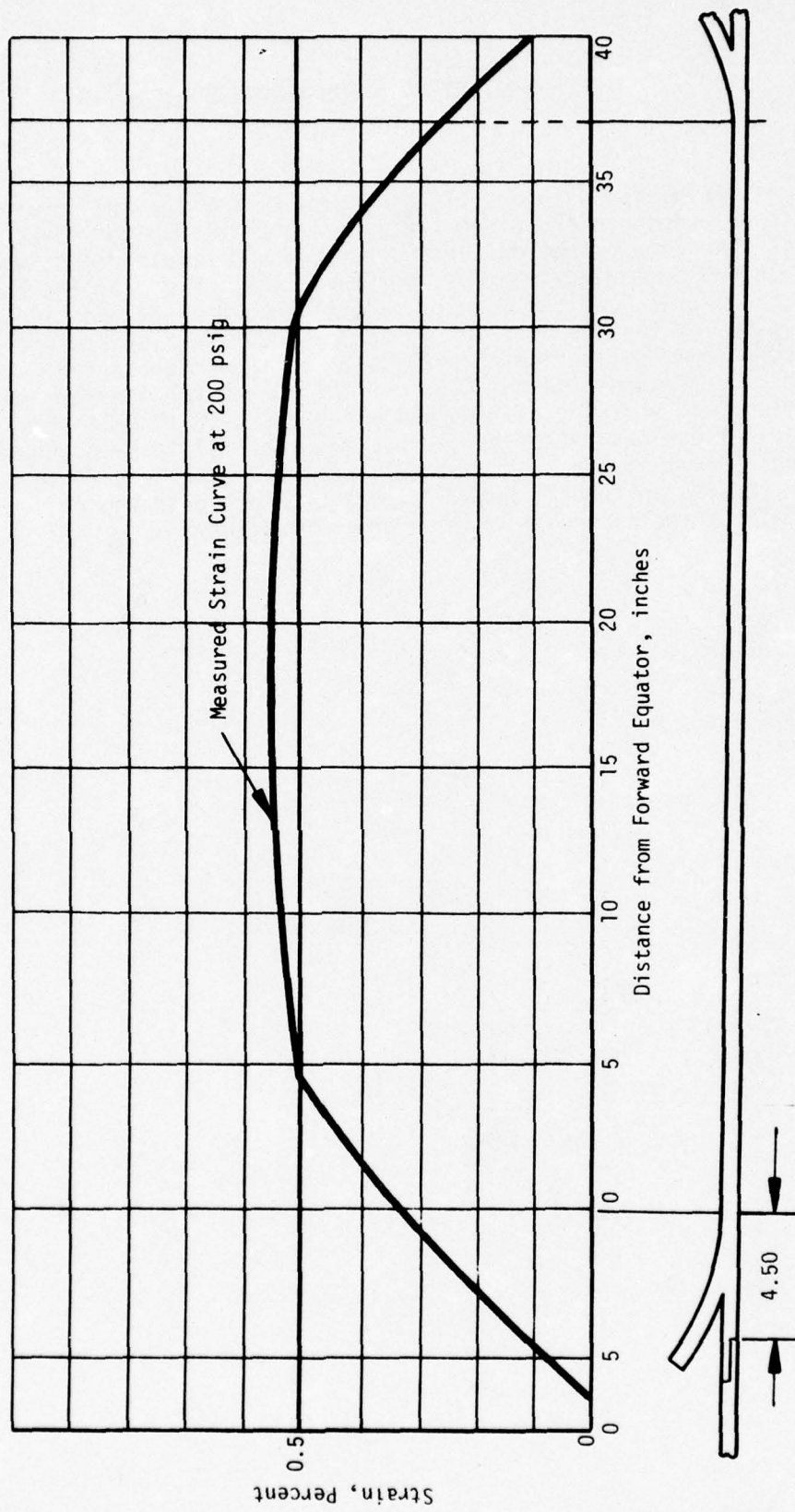


FIGURE C-1. HOOP STRAIN IN CYLINDER SECTION AT 200 PSIG



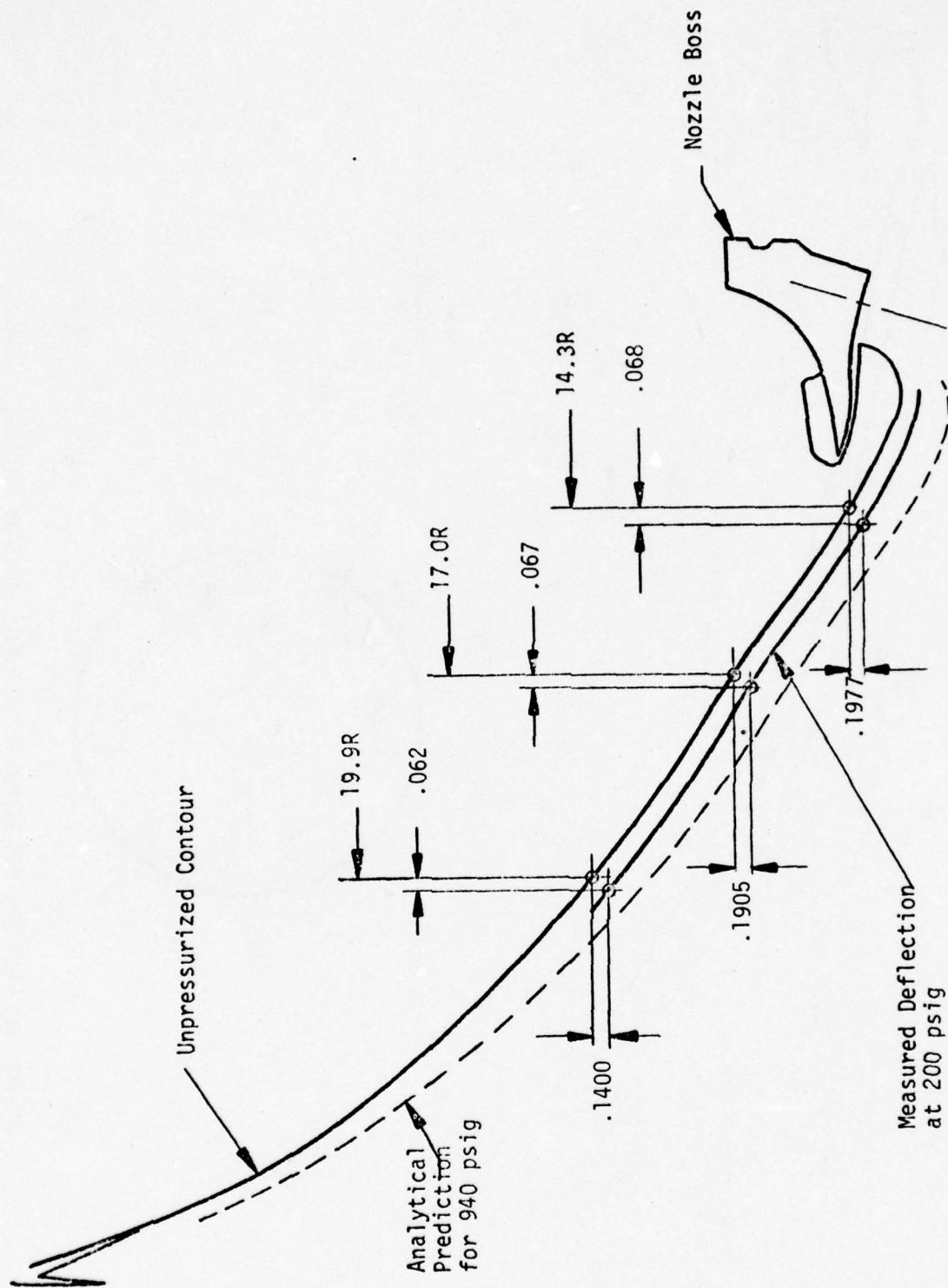


FIGURE C-2. AFT DOME DEFLECTION AT 200 PSIG (INCHES)

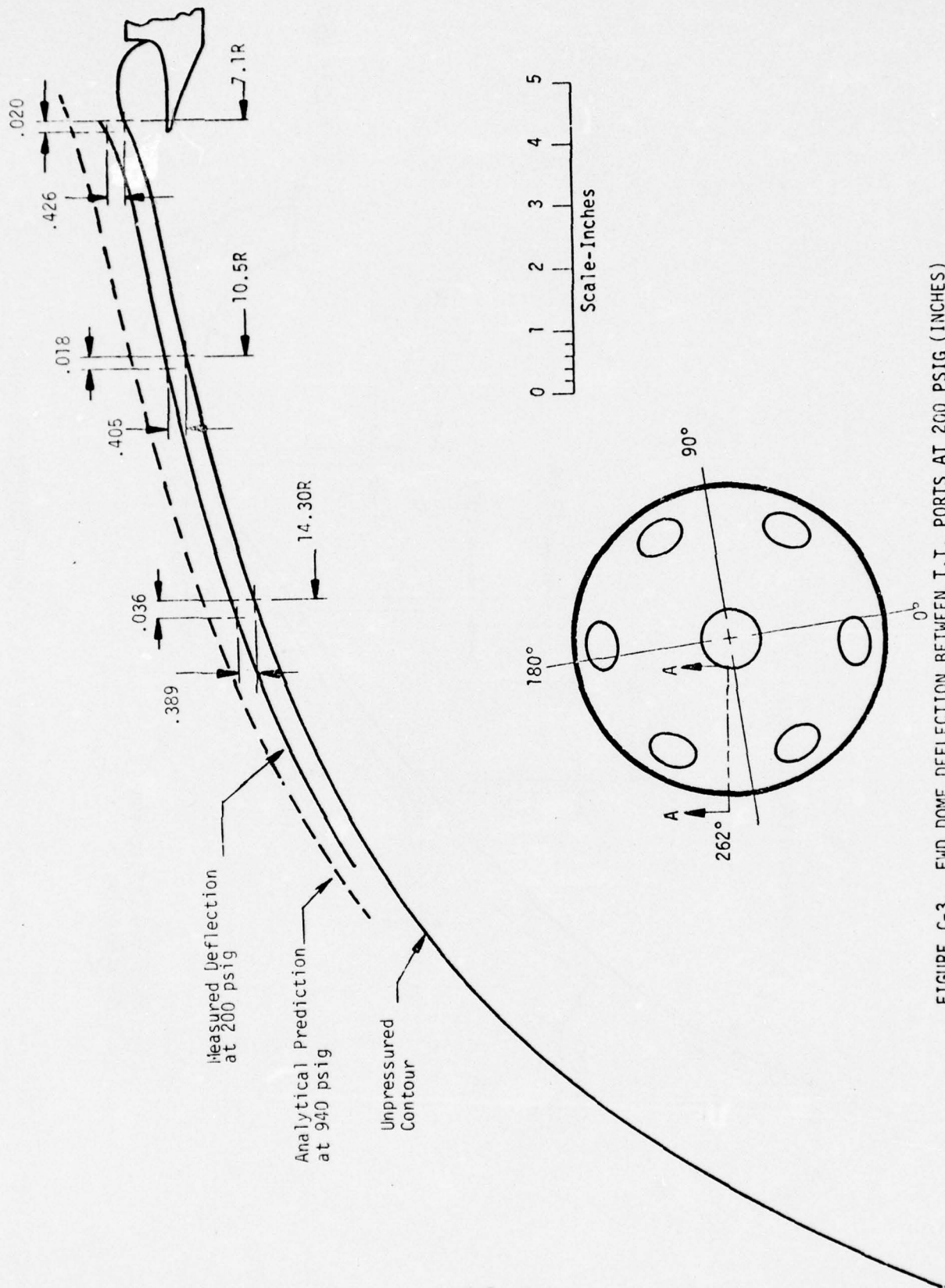


FIGURE C-3. FWD DOME DEFLECTION BETWEEN T.T. PORTS AT 200 PSIG (INCHES)

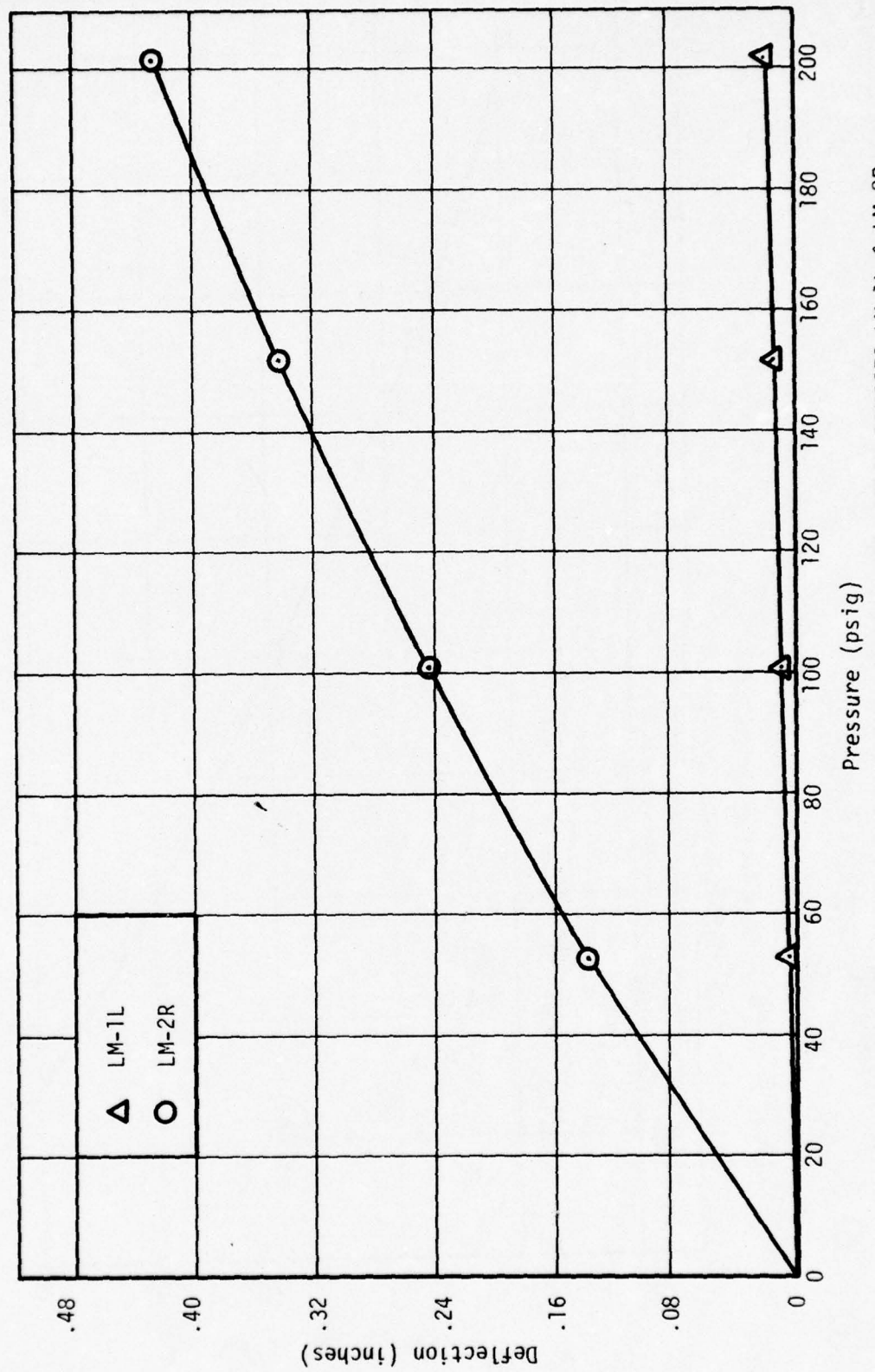


FIGURE C-4. DEFLECTION VS INTERNAL PRESSURE FOR DEFLECTION DEVICES LM-1L & LM-2R



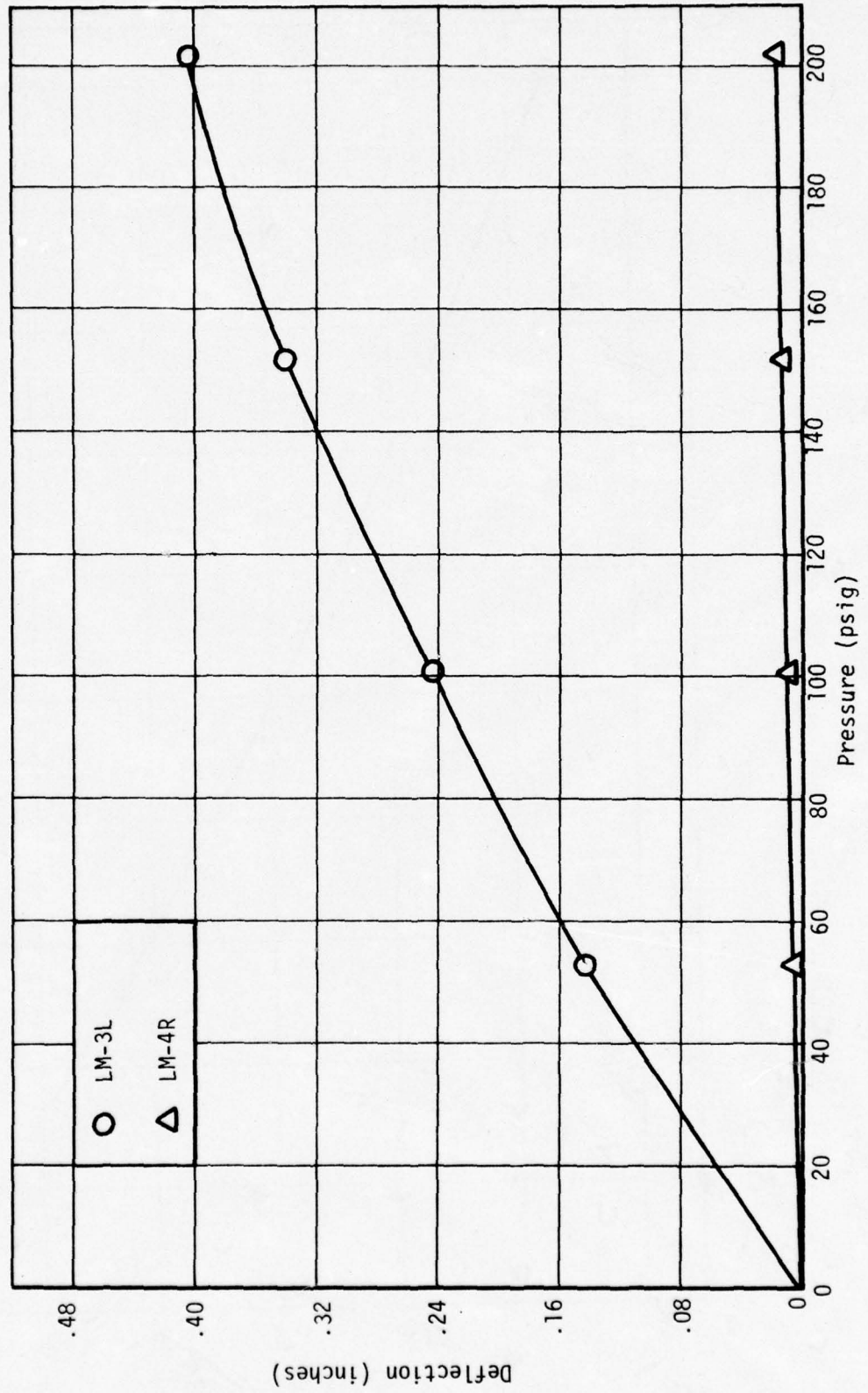


FIGURE C-5. DEFLECTION VS INTERNAL PRESSURE FOR DEFLECTION DEVICES LM-3L & LM-4R

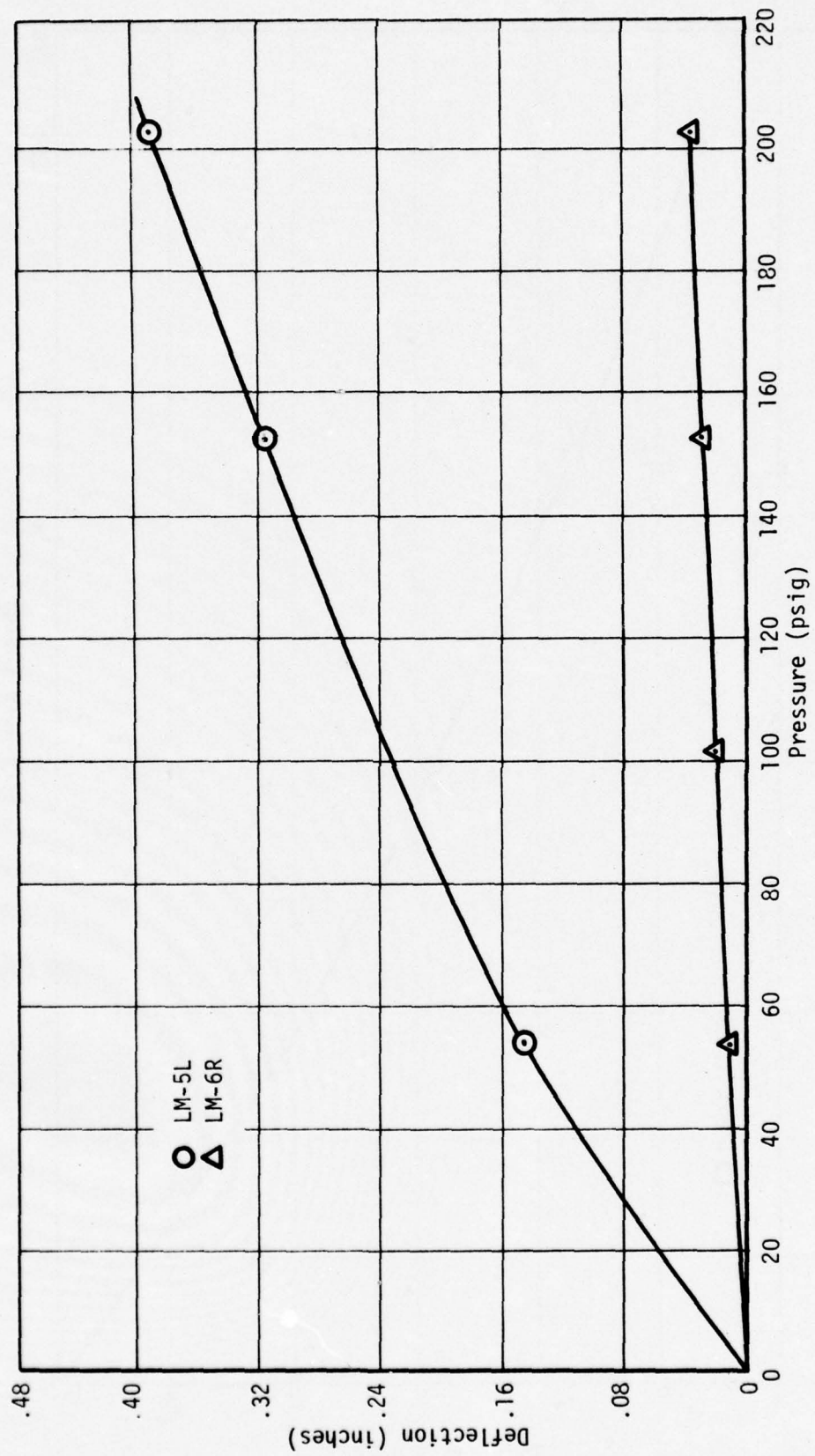


FIGURE C-6. DEFLECTION VS PRESSURE FOR DEVICES LM-5L & LM-6R

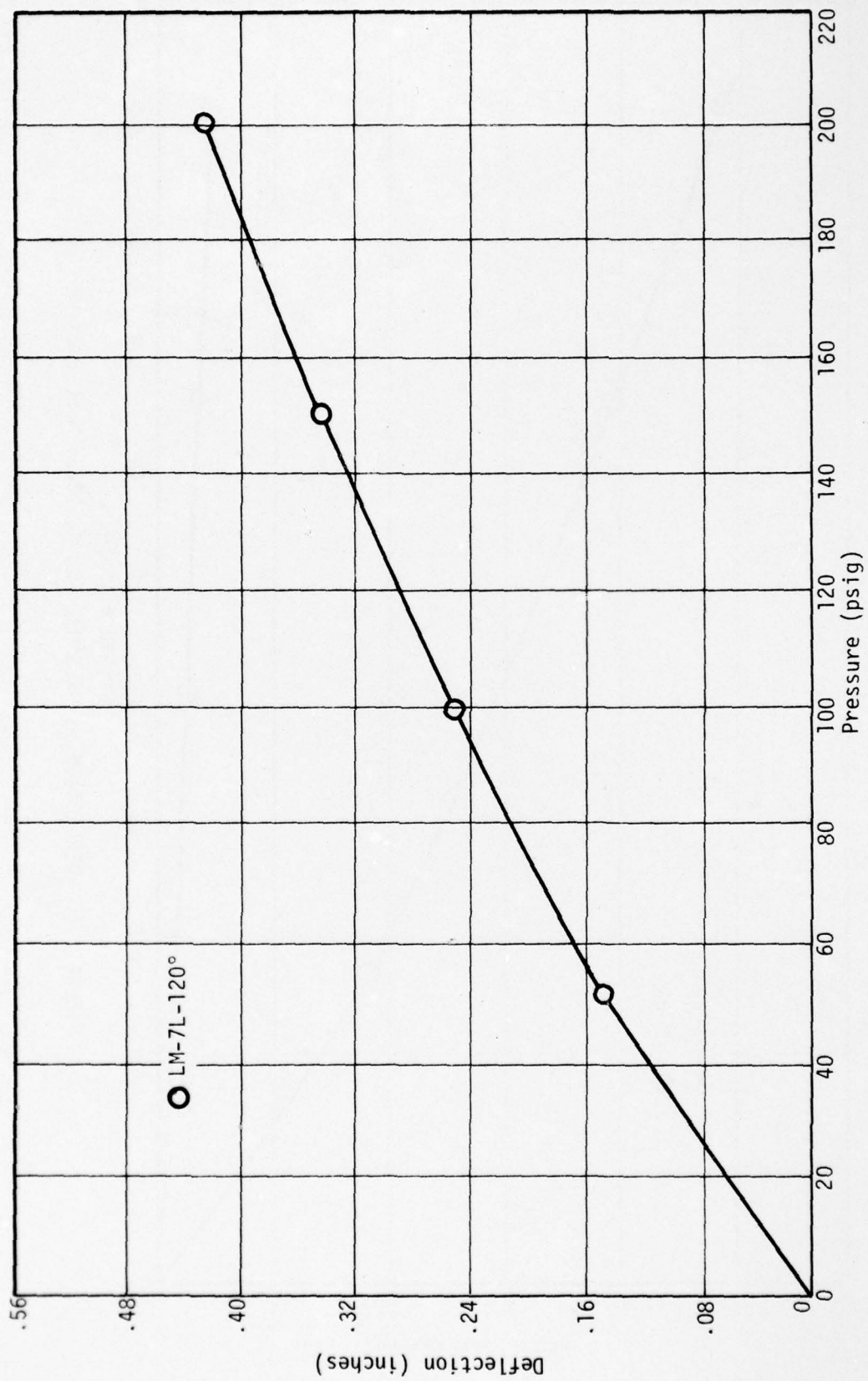


FIGURE C-7. DEFLECTION VS PRESSURE FOR DEVICE LM-7L



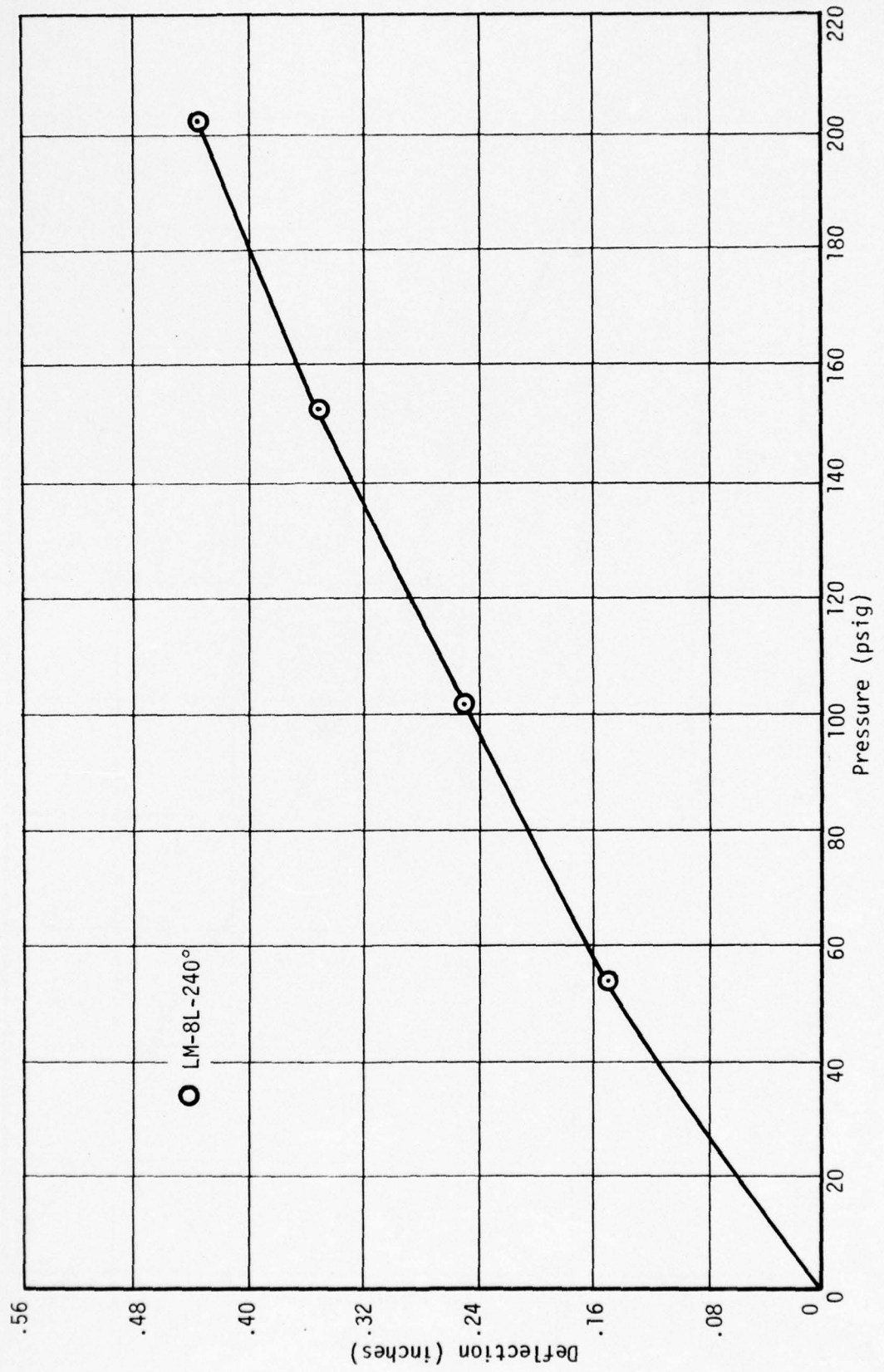


FIGURE C-8. DEFLECTION VS PRESSURE FOR DEVICE LM-8L

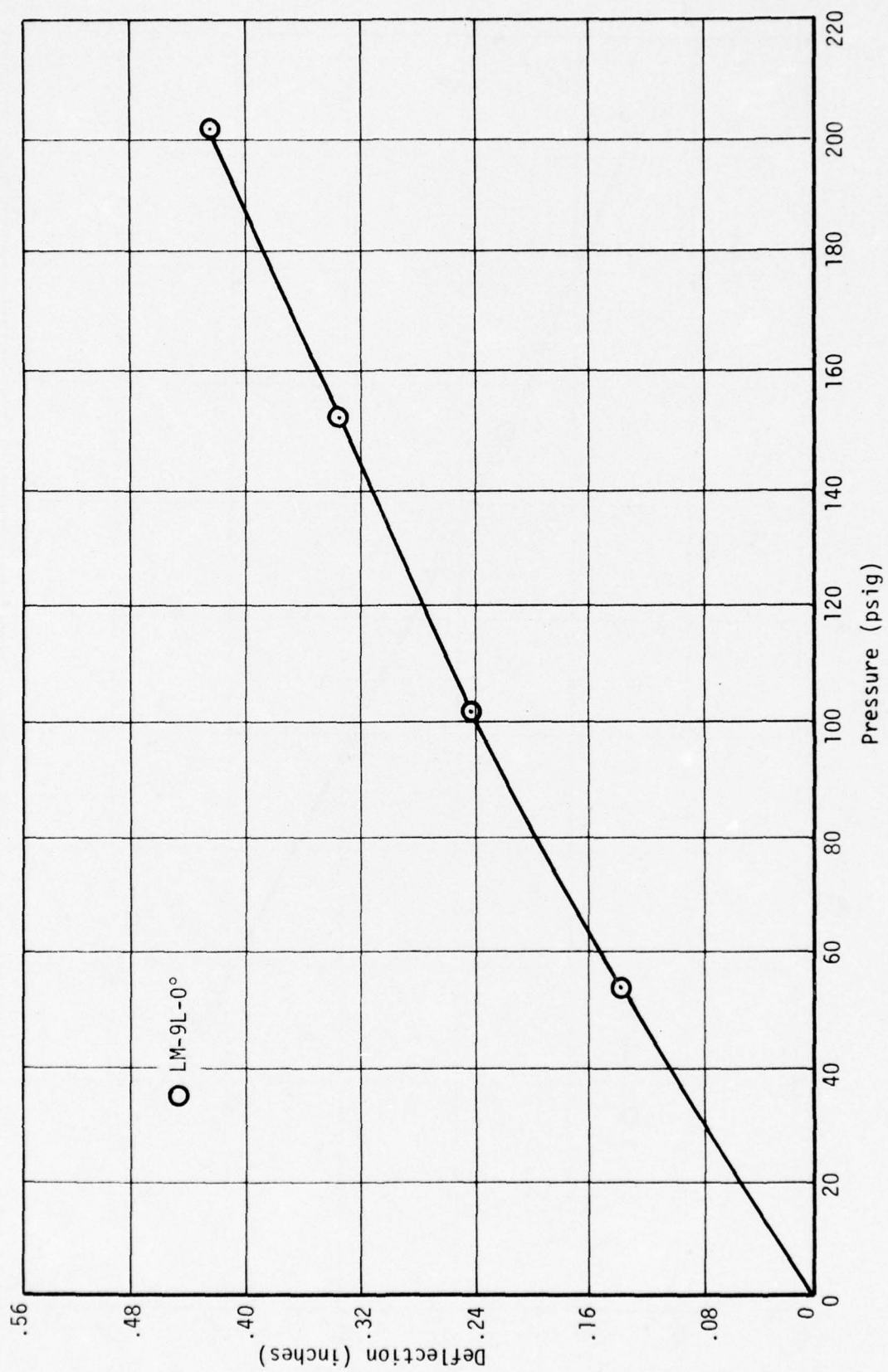


FIGURE C-9. DEFLECTION VS PRESSURE FOR DEVICES LM-9L

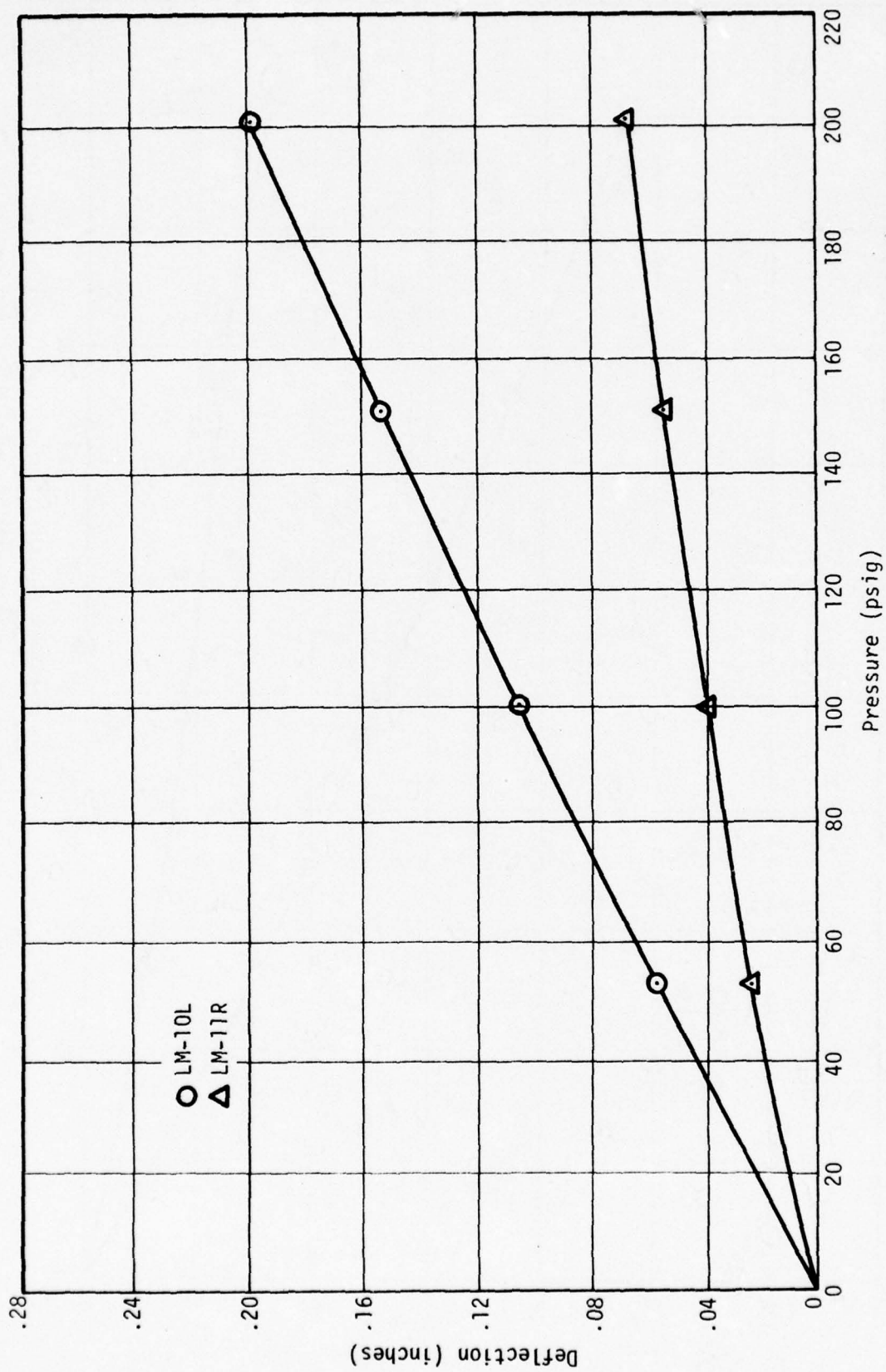


FIGURE C-10. DEFLECTION VS PRESSURE FOR DEVICES LM-10L AND LM-11R



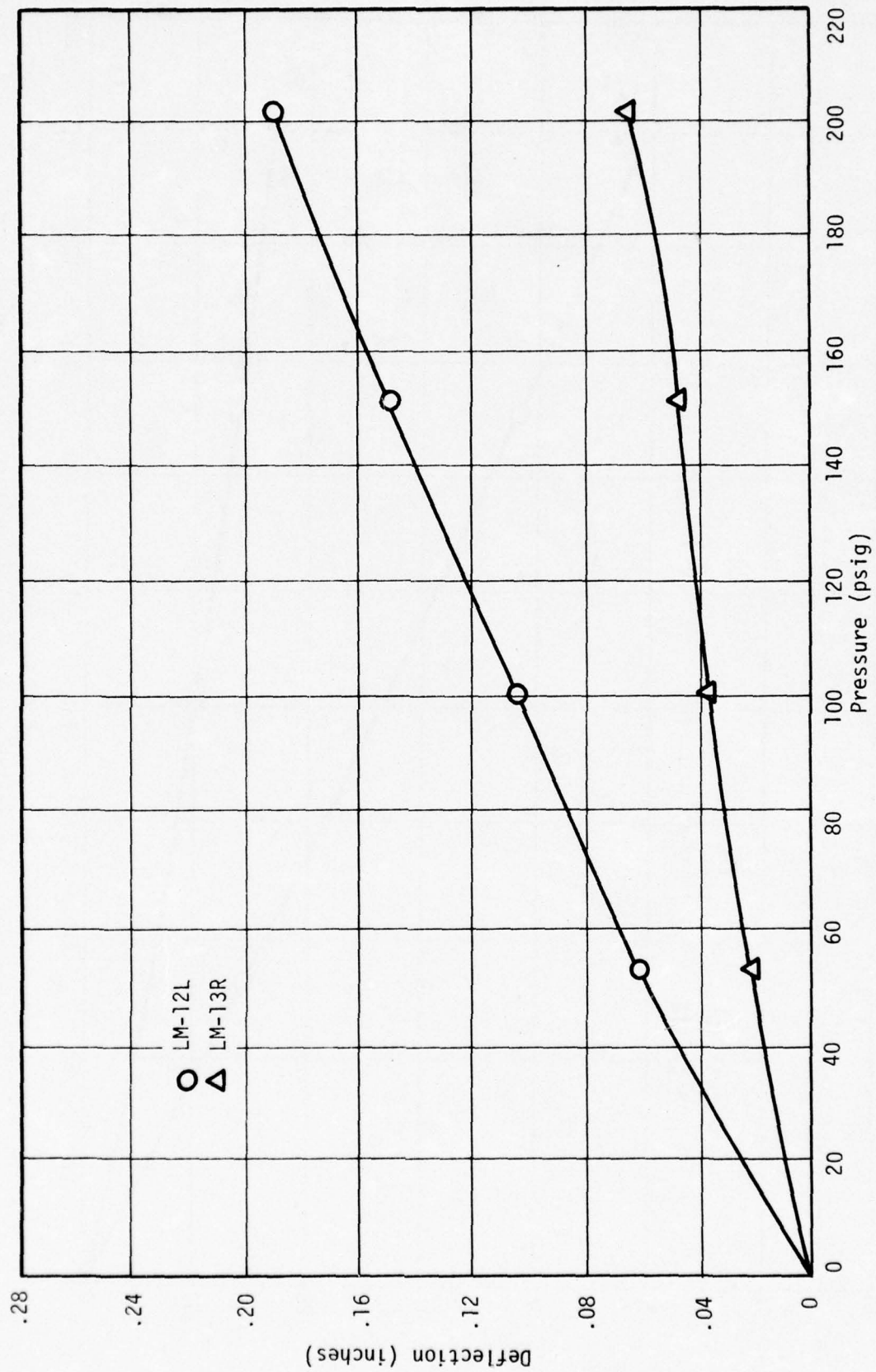


FIGURE C-11. DEFLECTION VS PRESSURE FOR DEVICES LM-12L AND LM-13R

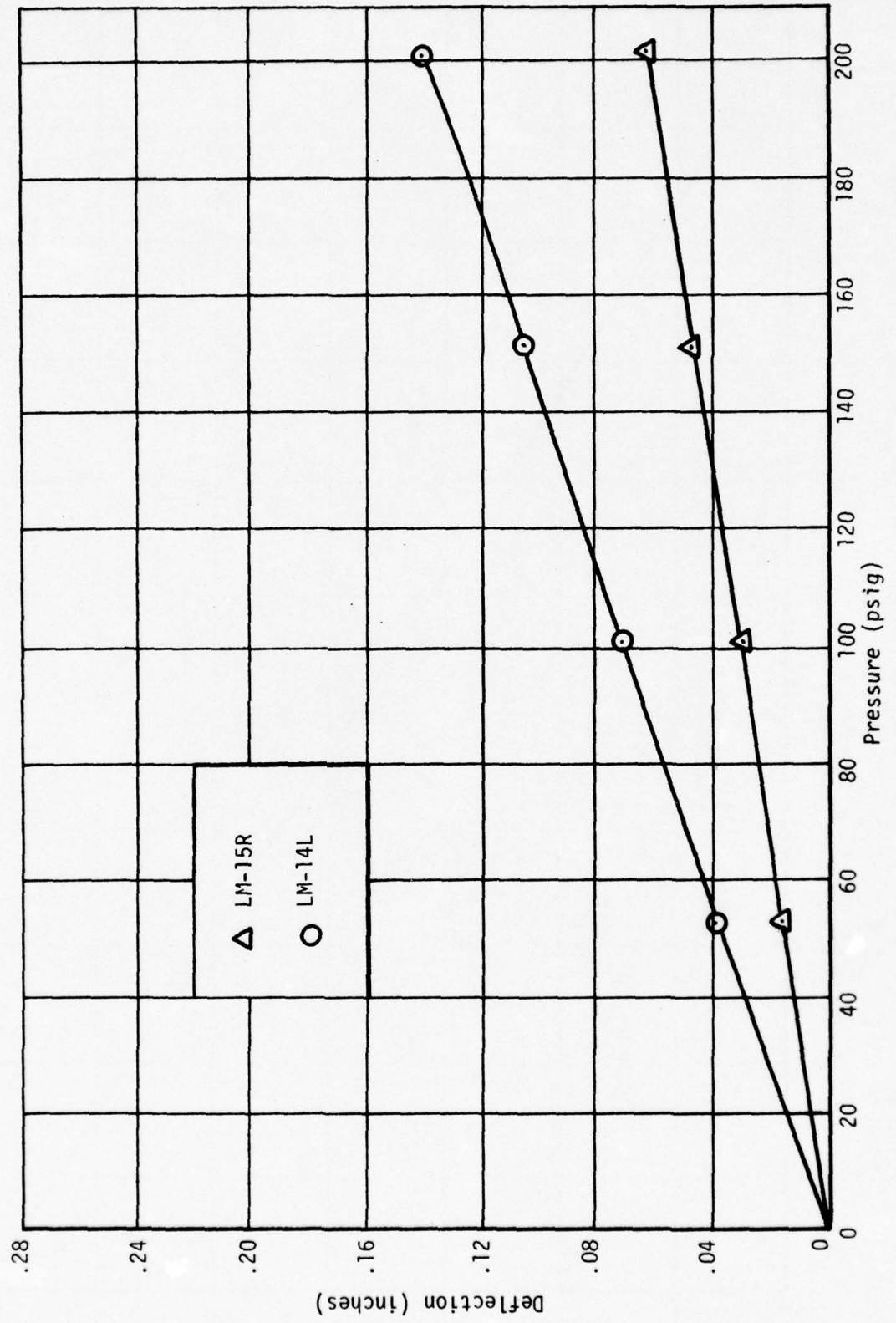


FIGURE C-12. DEFLECTION VS INTERNAL PRESSURE FOR DEFLECTION DEVICES LM-14L & LM-15R

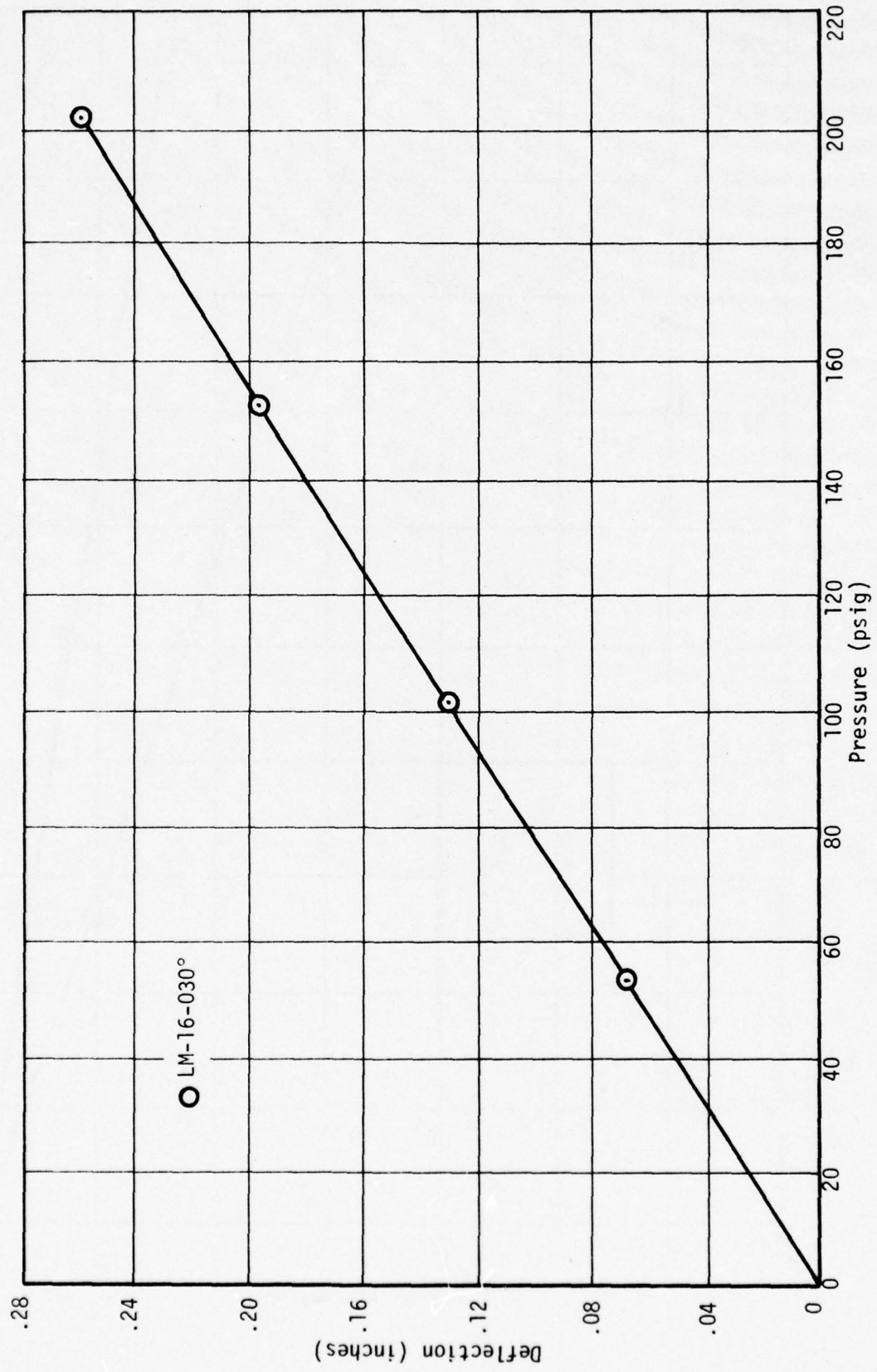


FIGURE C-13. DEFLECTION VS PRESSURE FOR DEVICE LM-16



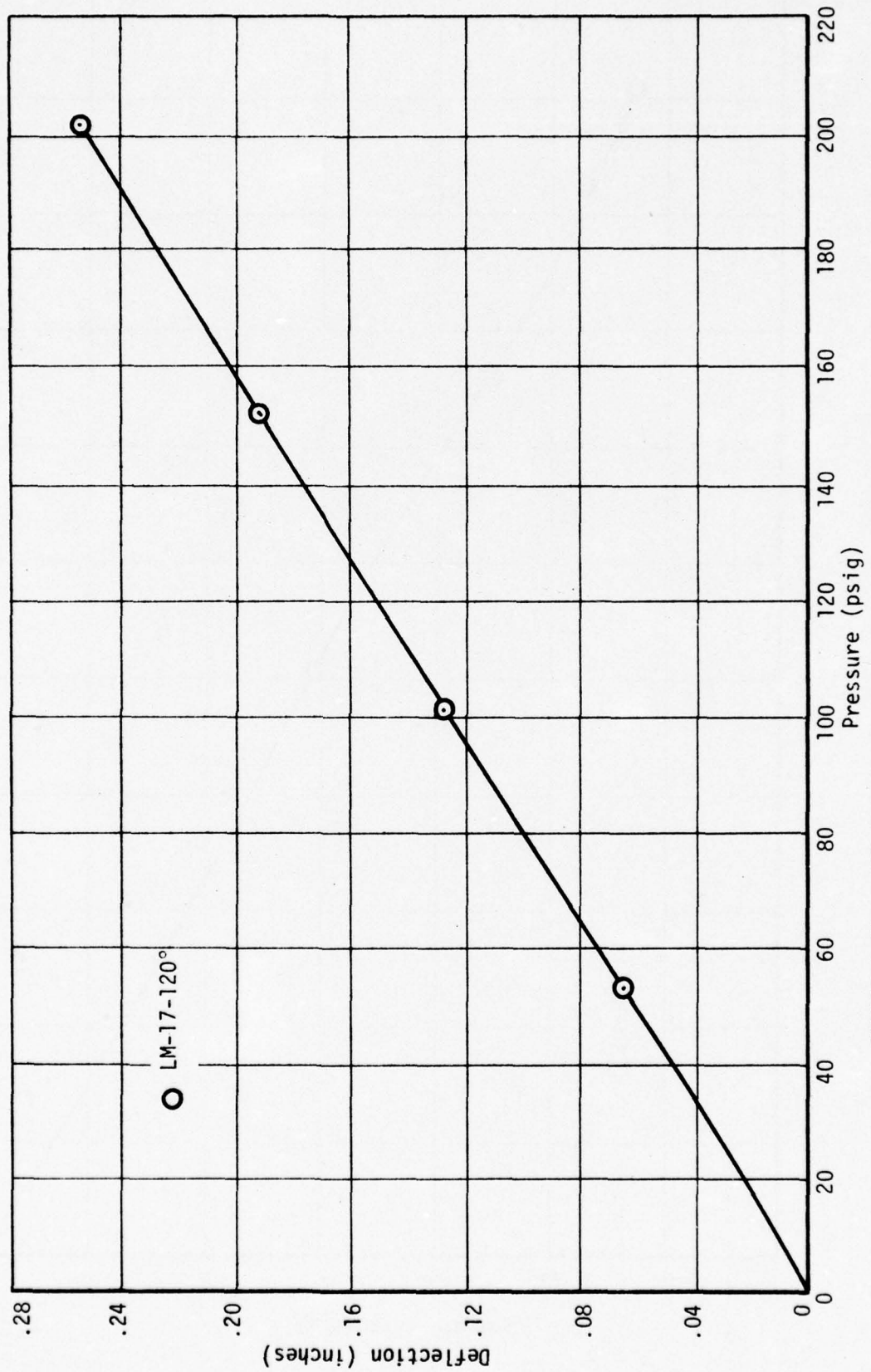


FIGURE C-14. DEFLECTION VS PRESSURE FOR DEVICES LM-17

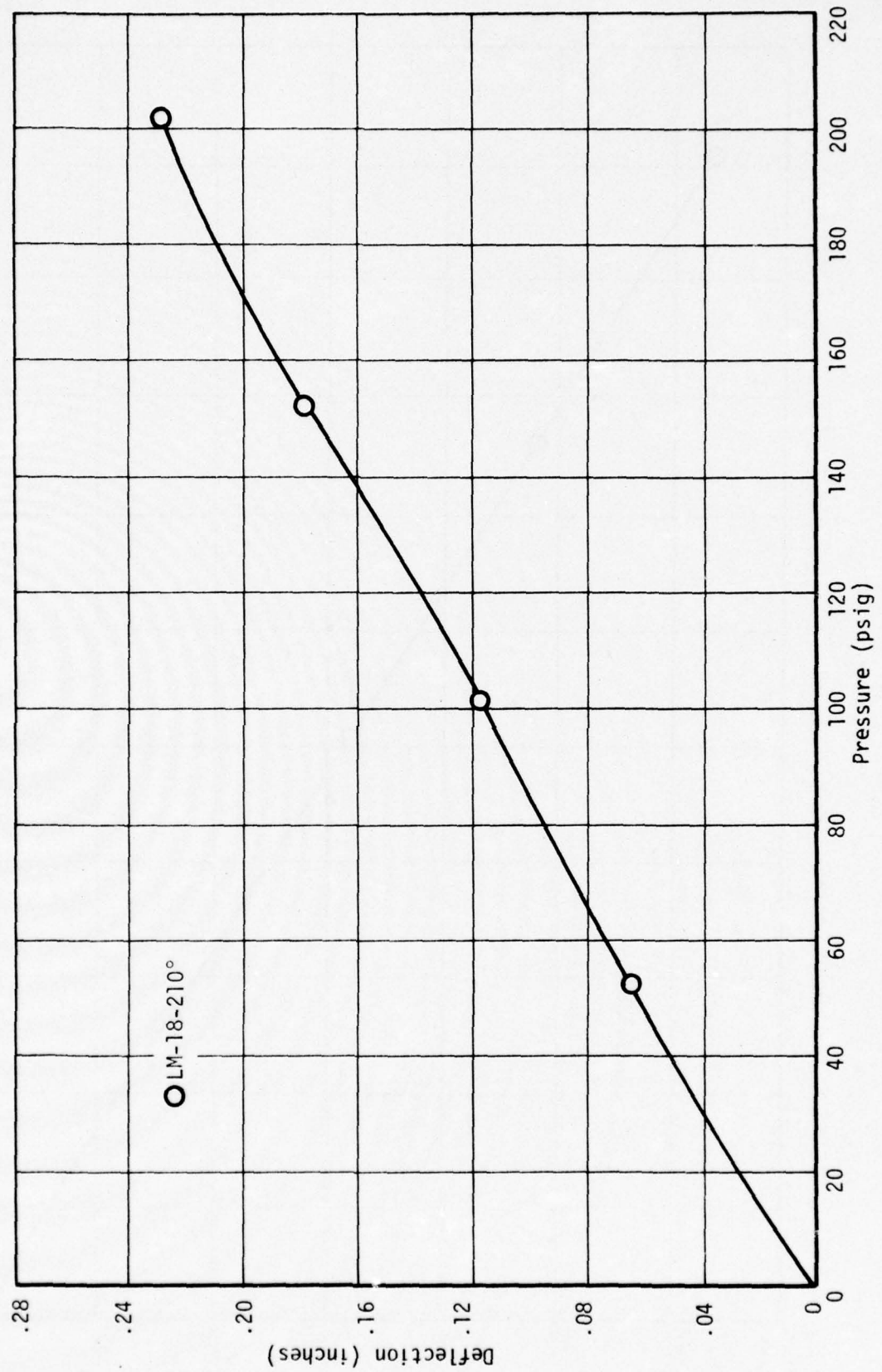


FIGURE C-15. DEFLECTION VS PRESSURE FOR DEVICE LM-18

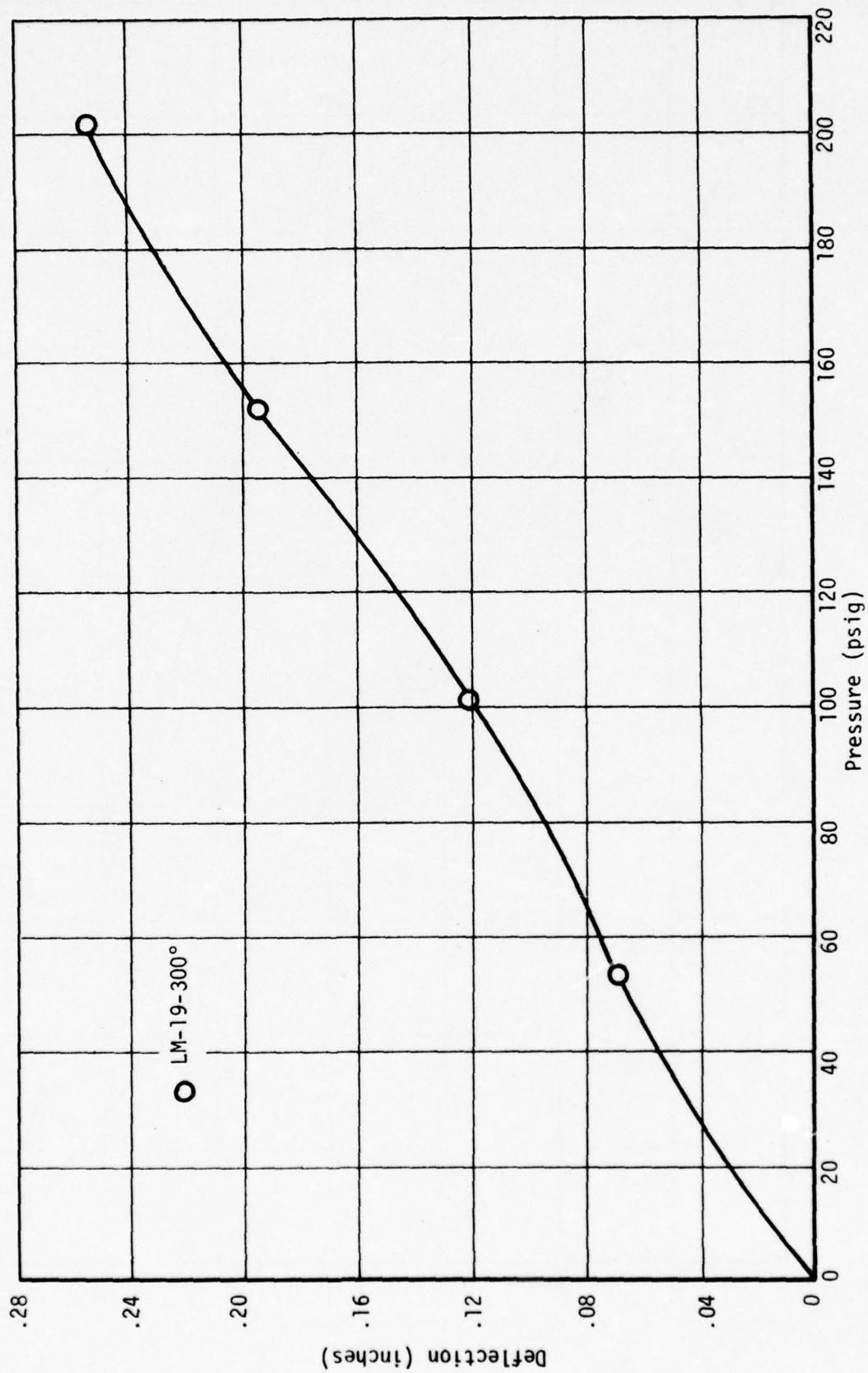


FIGURE C-16. DEFLECTION VS PRESSURE FOR DEVICE LM-19



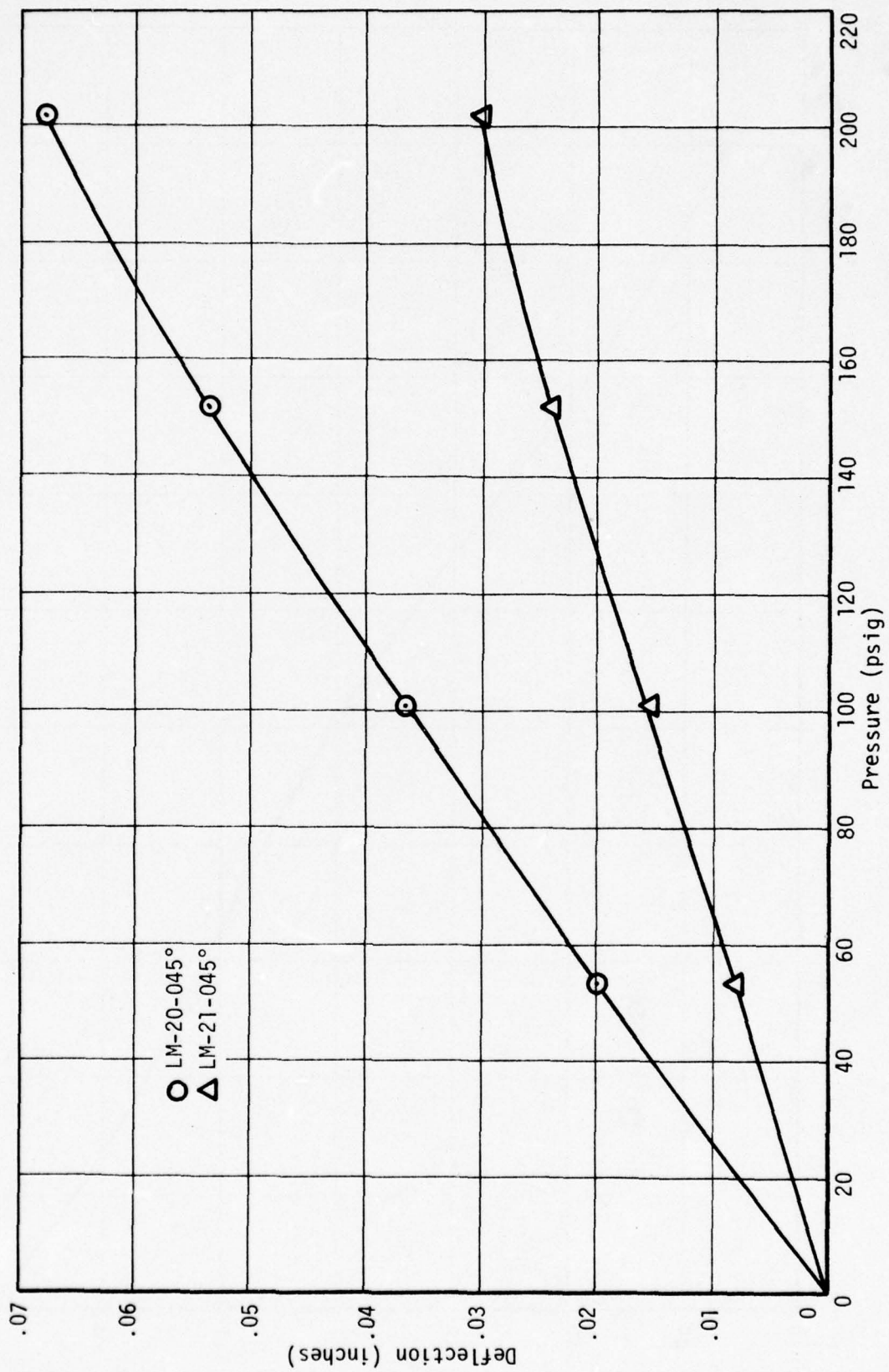


FIGURE C-17. DEFLECTION VS PRESSURE FOR DEVICES LM-20 AND LM-21

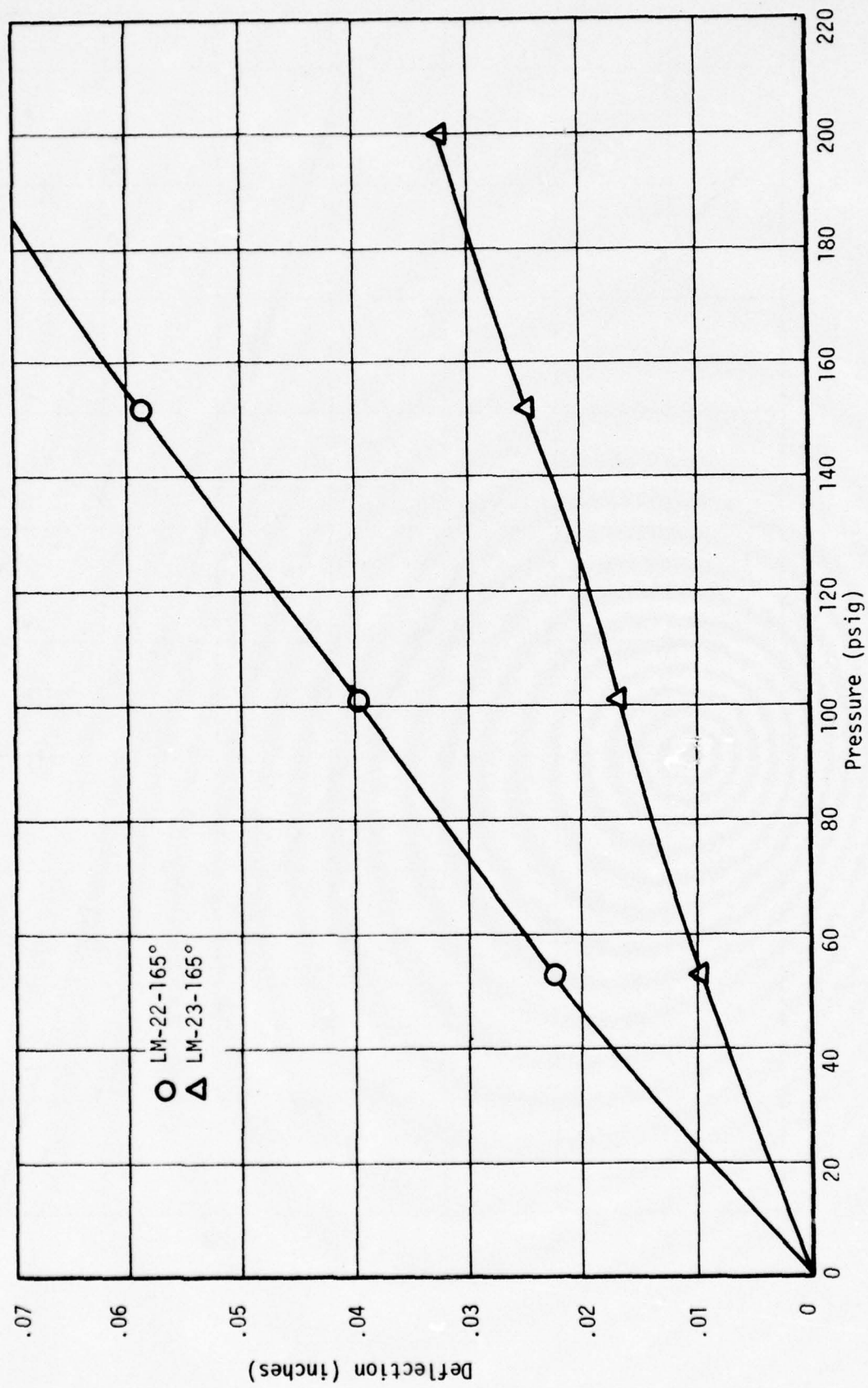


FIGURE C-18. DEFLECTION VS PRESSURE FOR DEVICES LM-22 AND LM-23

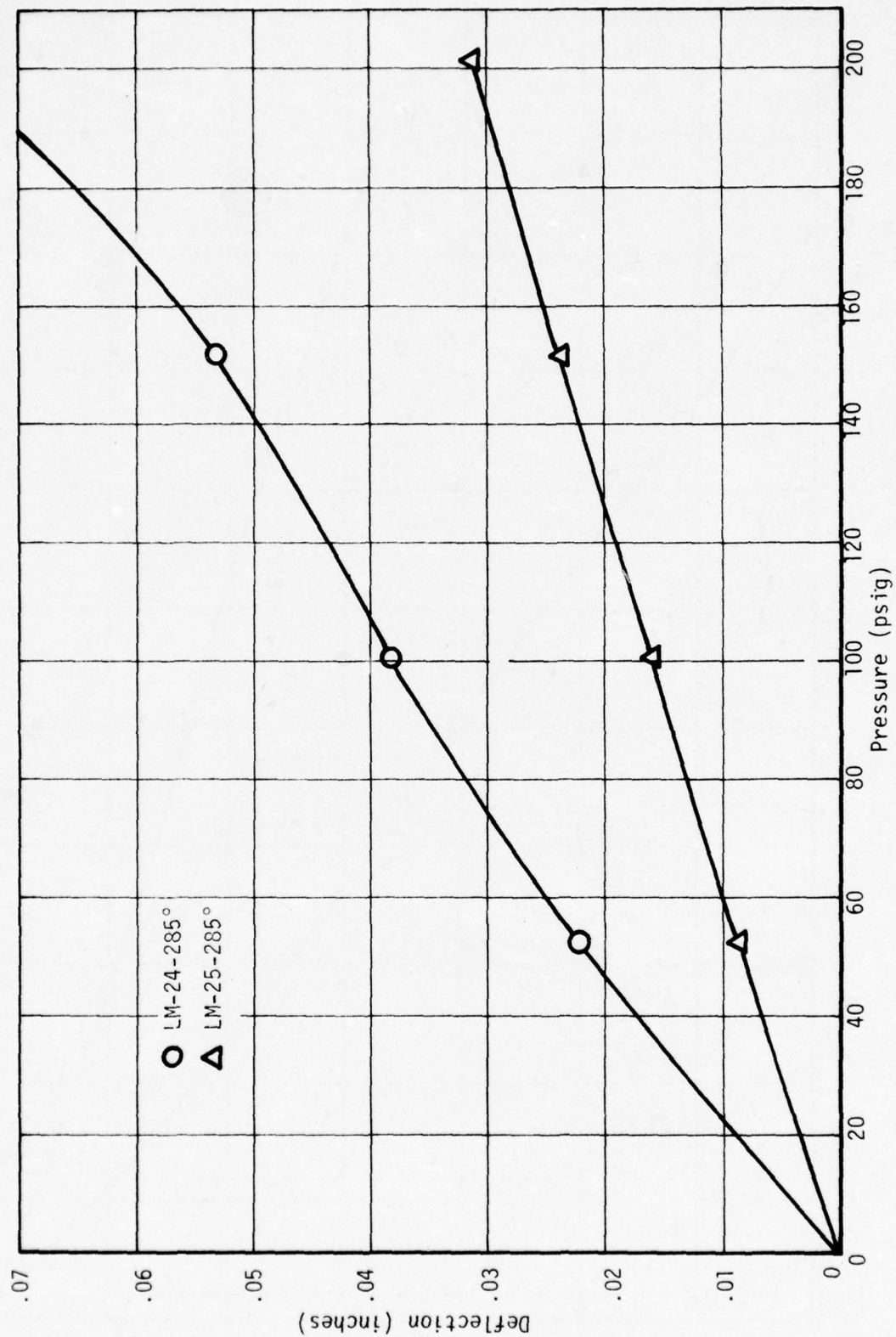


FIGURE C-19. DEFLECTION VS PRESSURE FOR DEVICES LM-24 & LM-25



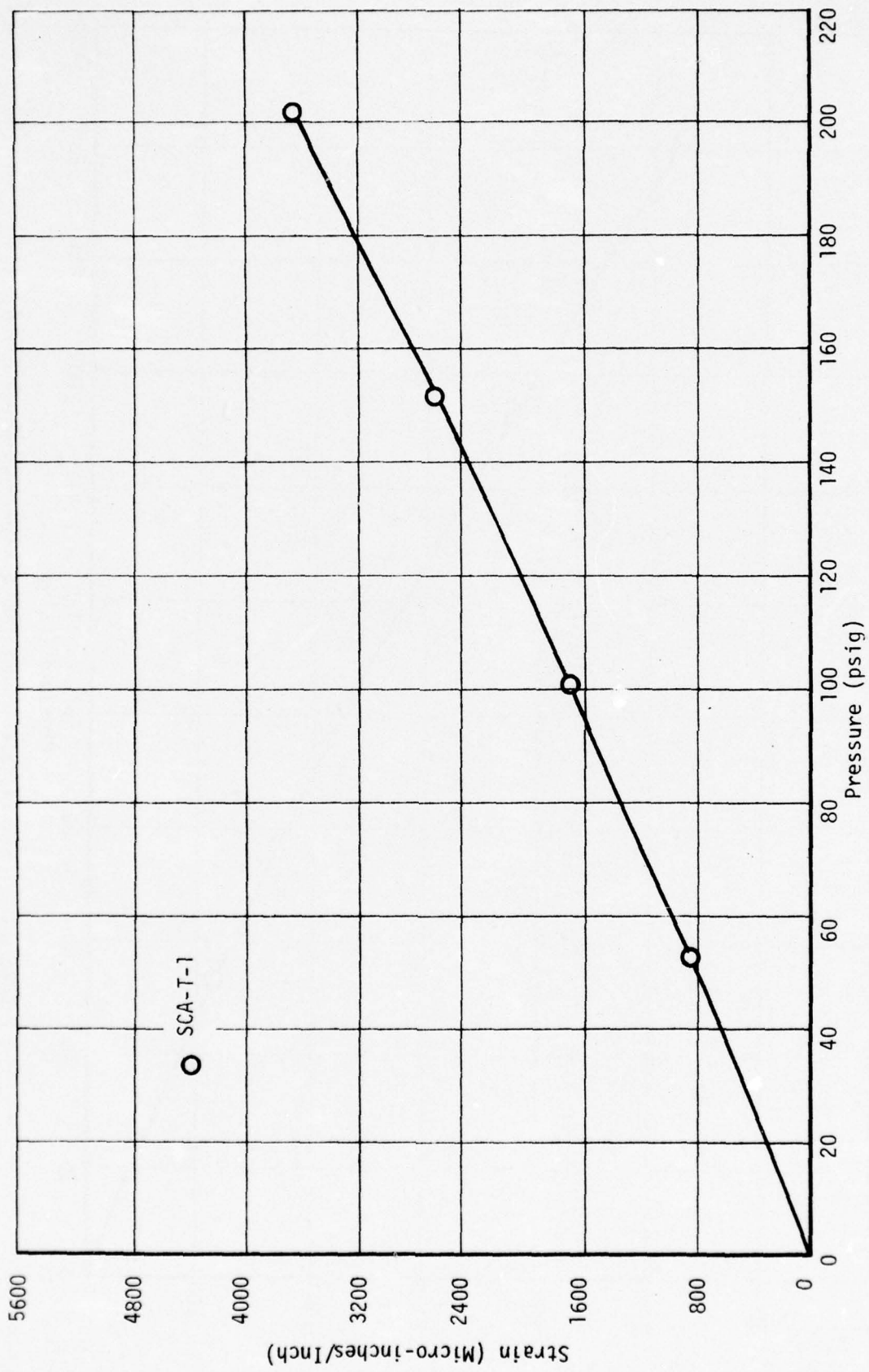


FIGURE C-20. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-1

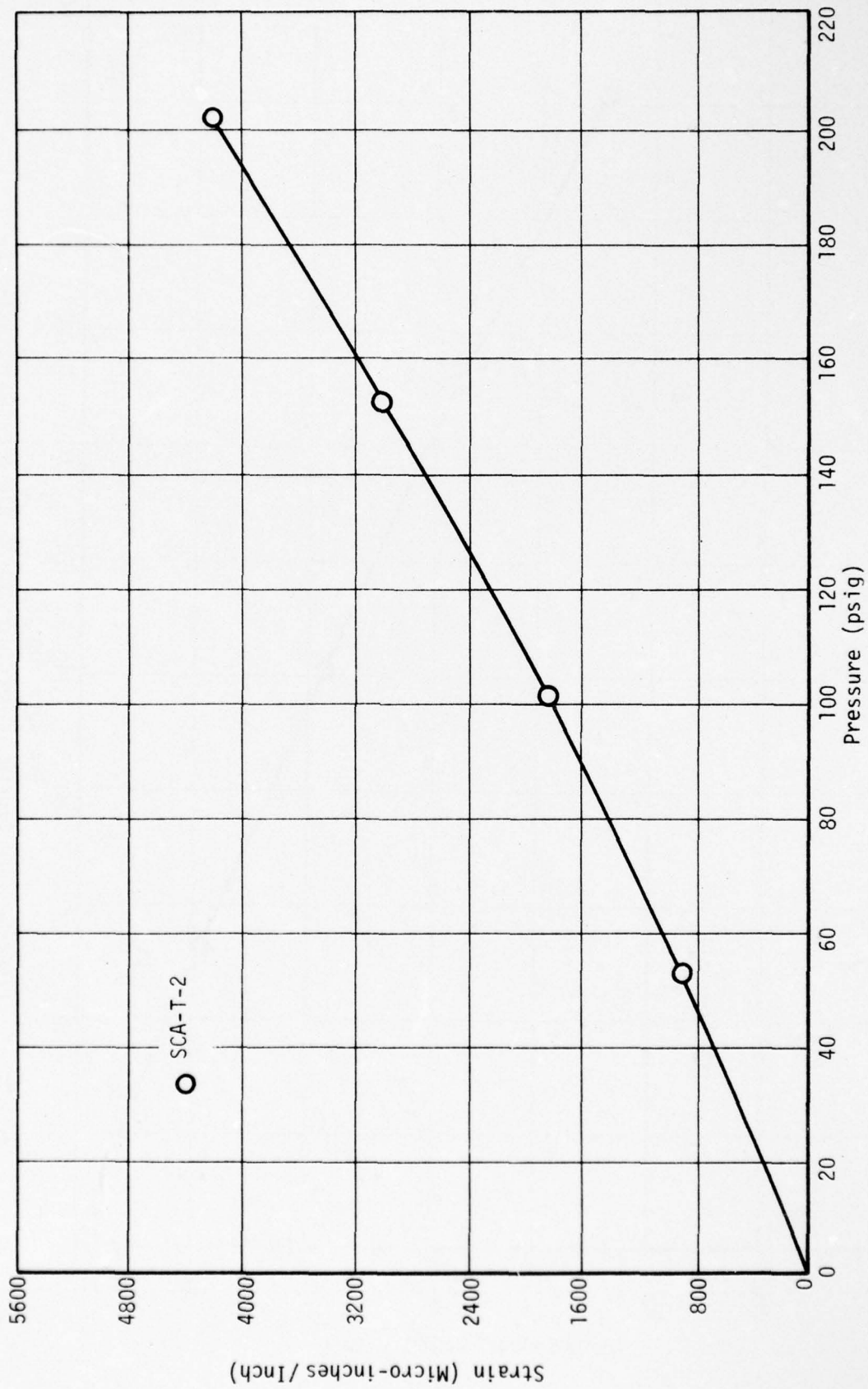


FIGURE C-21. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-2

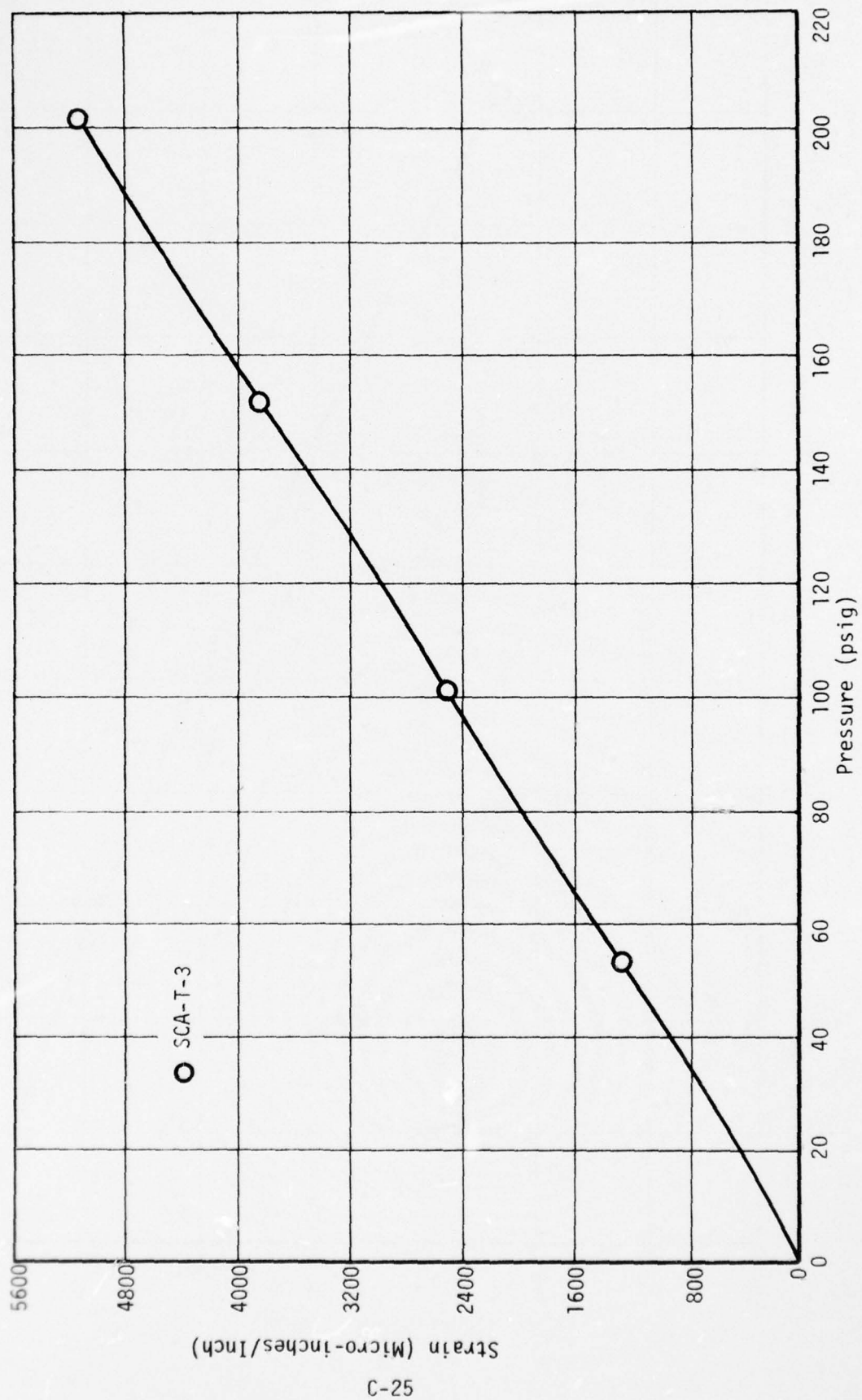


FIGURE C-22. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-3



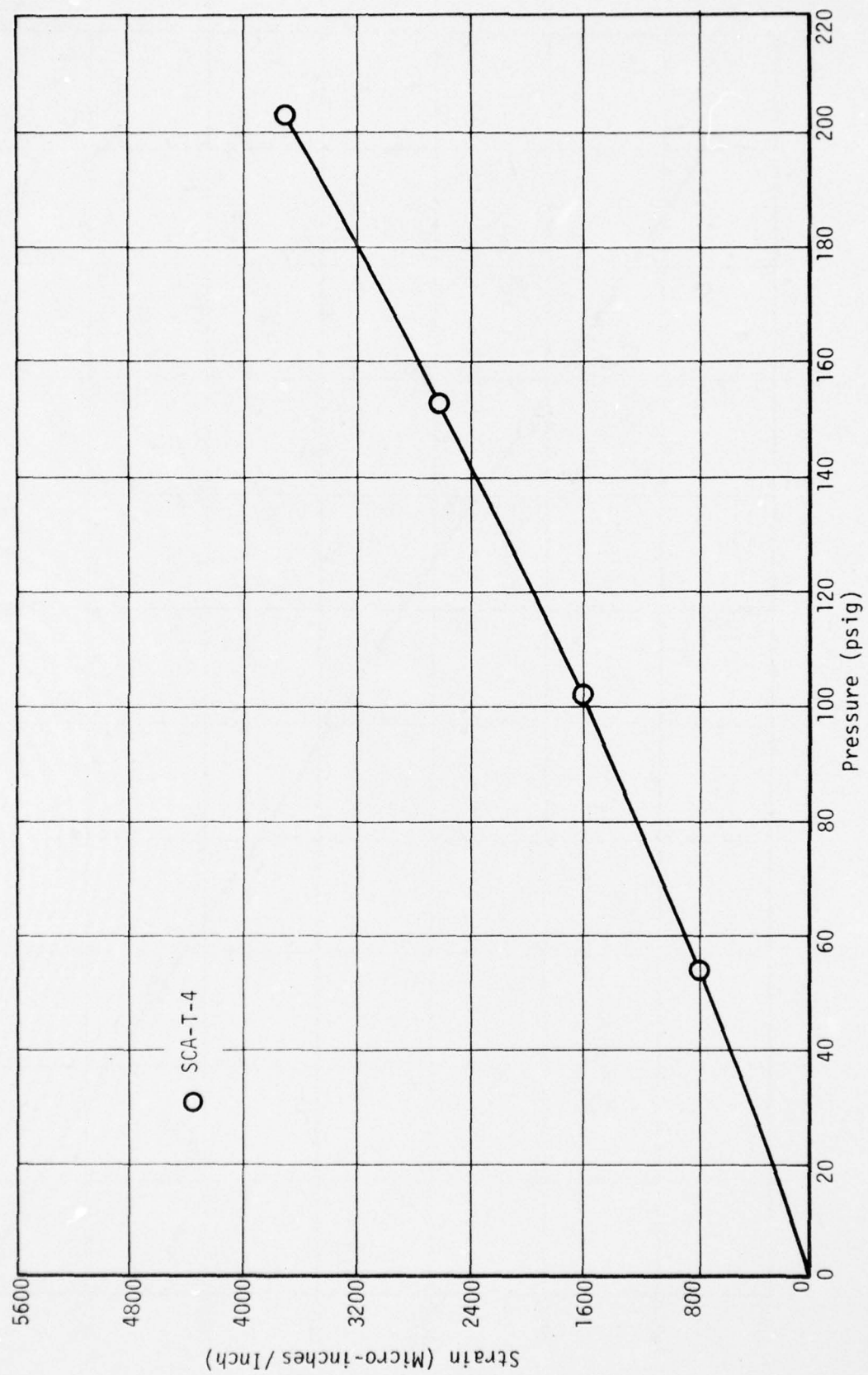


FIGURE C-23. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-4

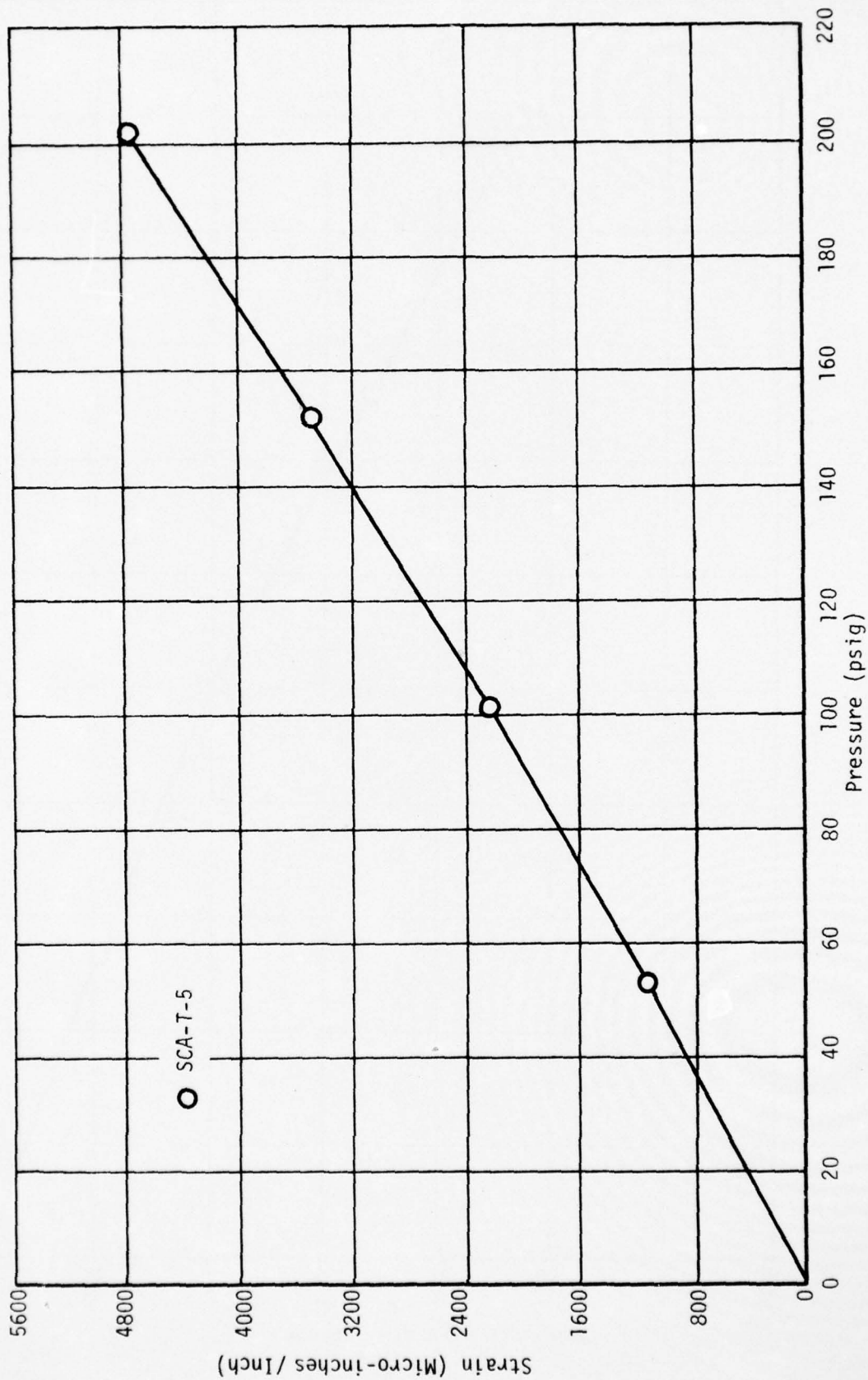


FIGURE C-24. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-5

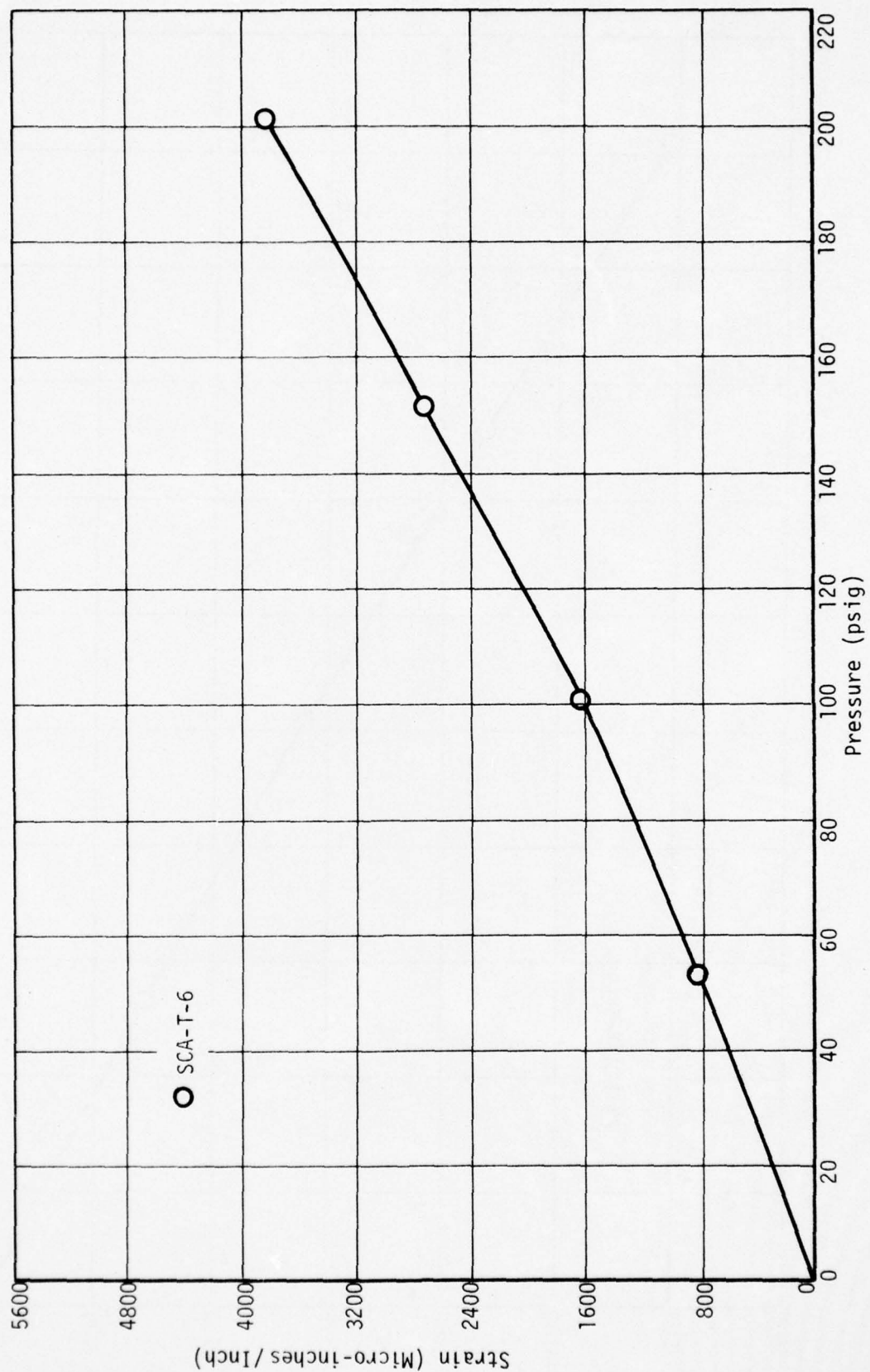


FIGURE C-25. STRAIN VS PRESSURE FOR STRAIN GAGE NO. SCA-T-6



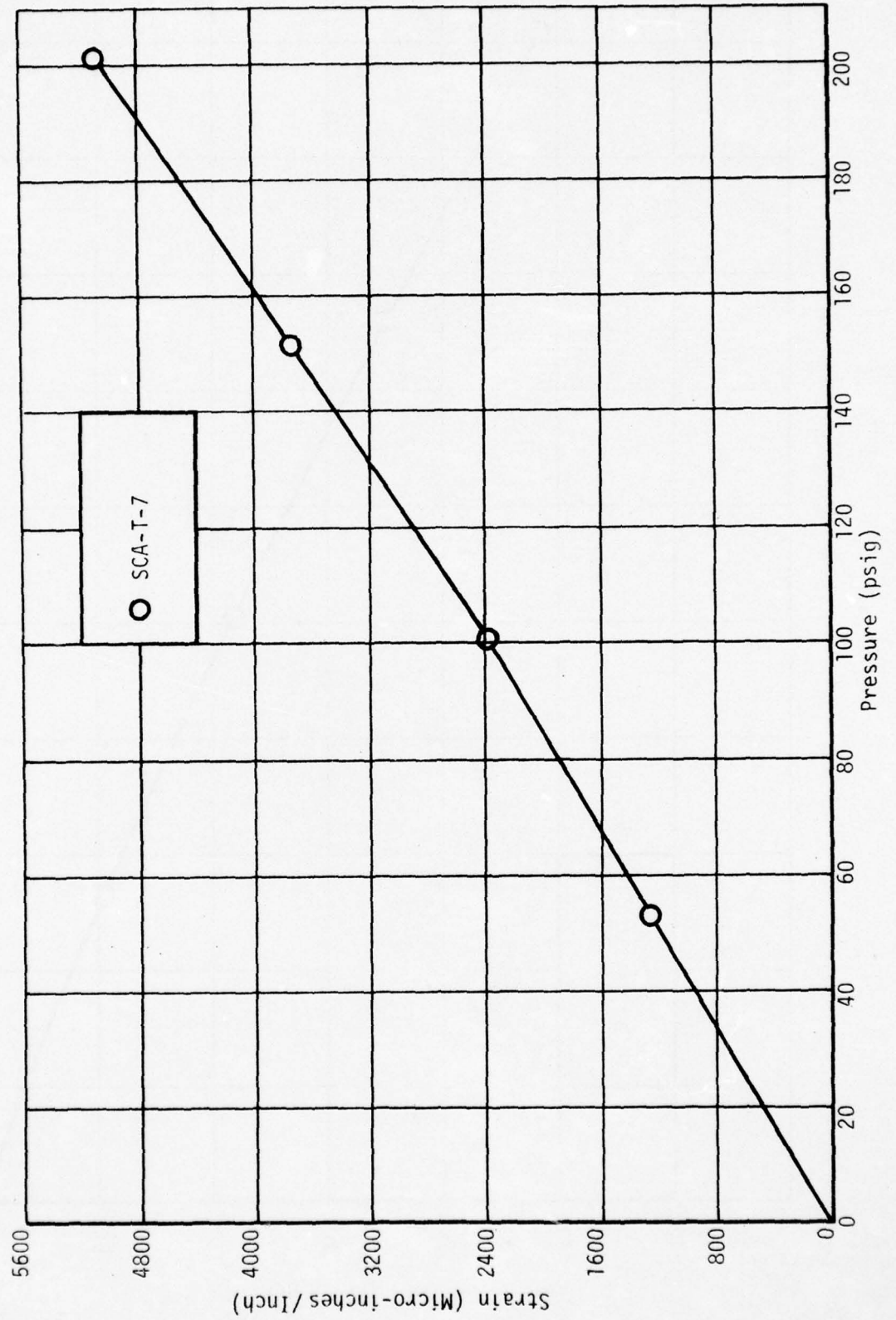


FIGURE C-26. STRAIN VS PRESSURE FOR GAGE NO. SCA-T-7

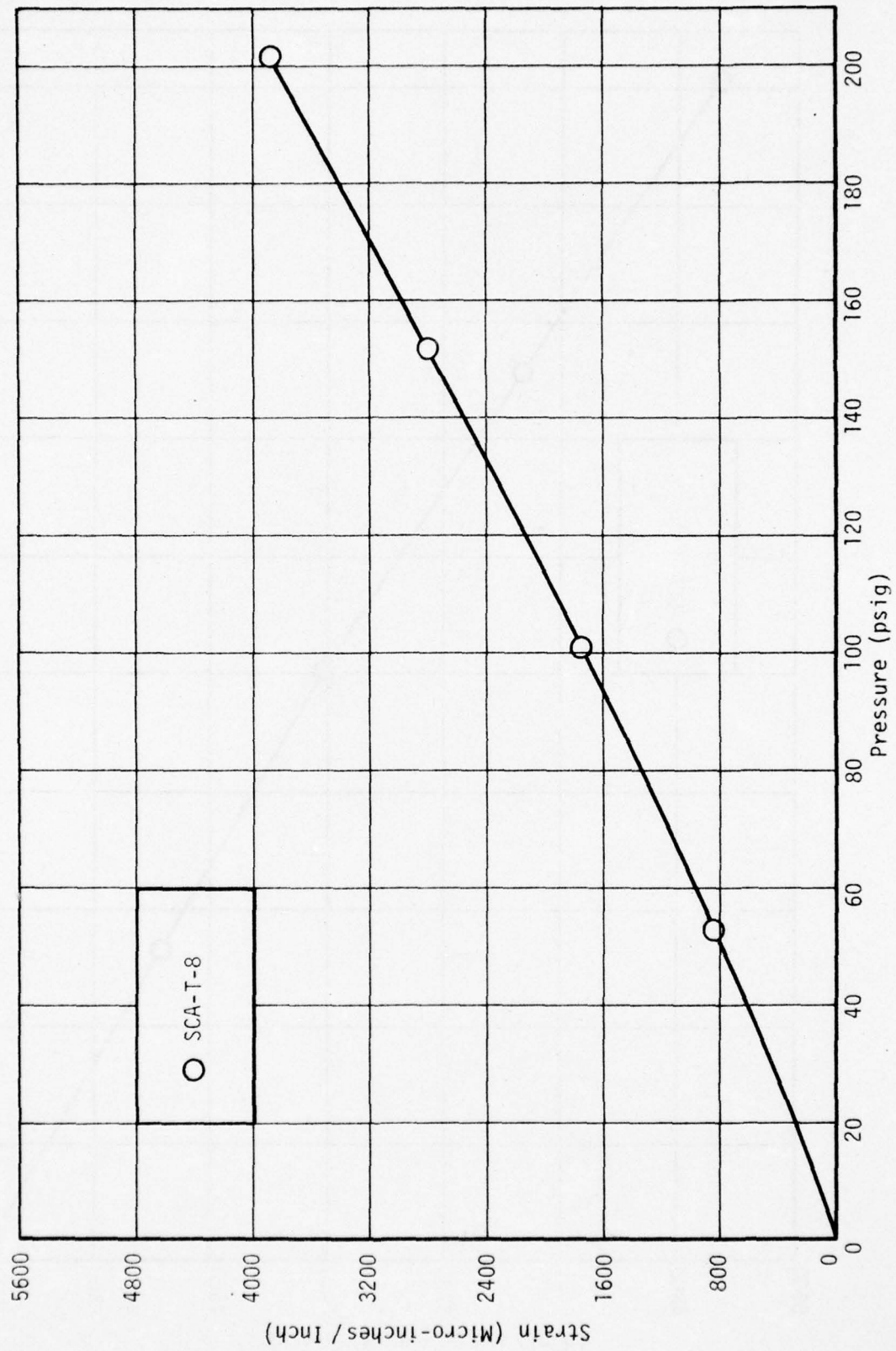


FIGURE C-27. STRAIN VS PRESSURE FOR GAGE NO. SCA-T-8

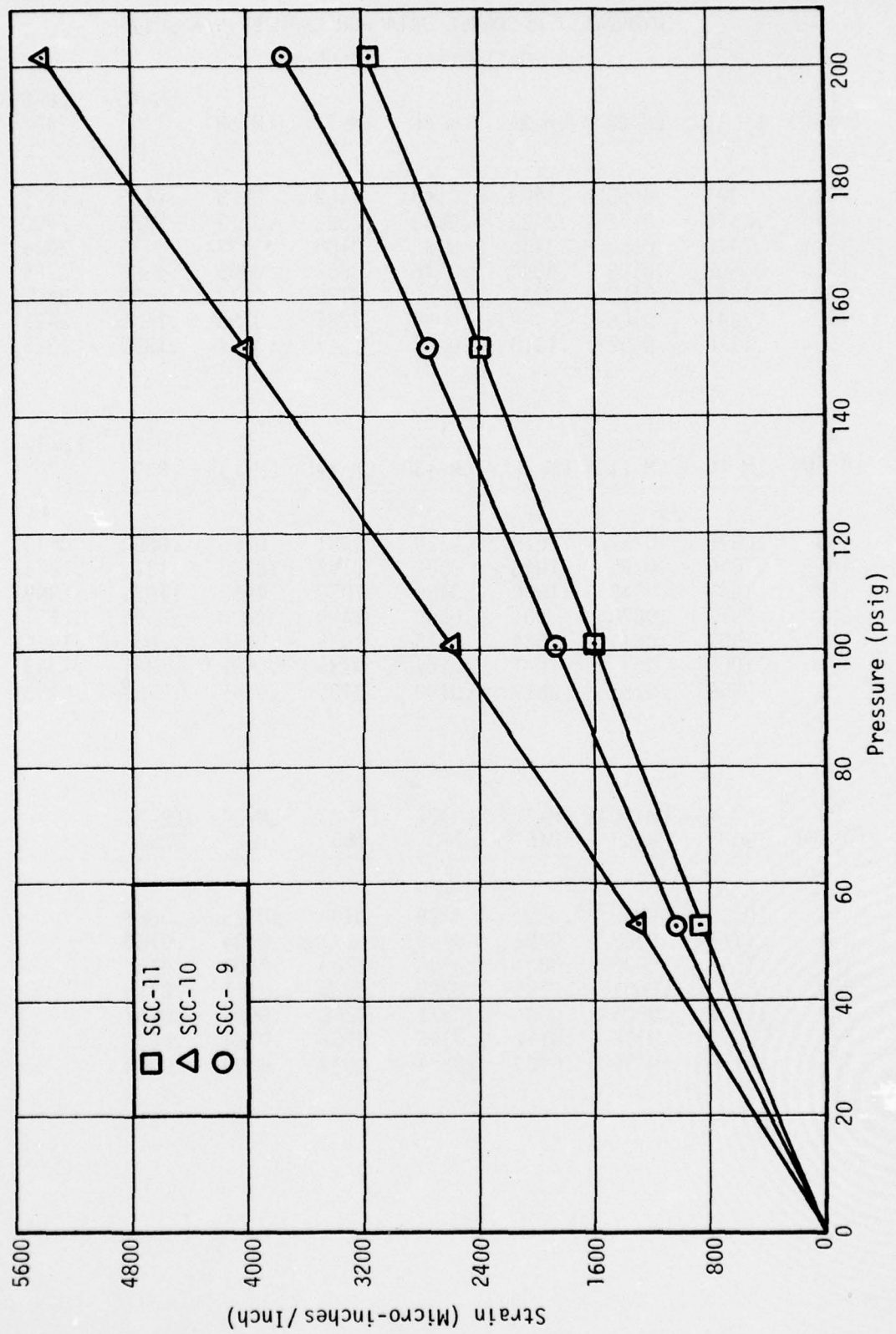


FIGURE C-28. STRAIN VS PRESSURE FOR CIRCUMFERENTIAL GAGES: SCC-9, SCC-10, & SCC-11



TABLE C-1

## HYDROTEST RESPONSE DATA FOR CHAMBER S/N 30114

## Deflections - Inches

PC (psig)	LM-1L	LM-2R	LM-3L	LM-4R	LM-5L	LM-6R	LM-7L- 120°	LM-8L- 240°	LM-9L- 0°
53	.1381	.0065	.1409	.0048	.1452	.0128	.1475	.1475	.1375
101	.2428	.0112	.2432	.0089	.2333	.0215	.2475	.2500	.2400
152	.3428	.0158	.3405	.0132	.3159	.0297	.3475	.3428	.3333
202	.4262	.0198	.4048	.0176	.3886	.0365	.4300	.4225	.4238
151	.3381	.0152	.3381	.0125	.3136	.0279	.3425	.3405	.3310
100	.2341	.0105	.2409	.0080	.2286	.0193	.2425	.2405	.2350
50	.1286	.0052	.1318	.0040	.1357	.0107	.1350	.1325	.1475

PC (psig)	LM-10L	LM-11R	LM-12L	LM-13R	LM-14L	LM-15R	LM-16- 030°	LM-17- 120°	LM-18- 210°
53	.0575	.0252	.0619	.0220	.0381	.0165	.0690	.0643	.0643
101	.1050	.0395	.1048	.0381	.0714	.0300	.1310	.1262	.1167
152	.1523	.0548	.1500	.0486	.1050	.0465	.1976	.1909	.1786
202	.1977	.0678	.1905	.0668	.1400	.0620	.2595	.2524	.2286
151	.1523	.0543	.1524	.0515	.1075	.0455	.2024	.1954	.1810
100	.1068	.0391	.1071	.0357	.0738	.0291	.1381	.1333	.1214
50	.0550	.0239	.0619	.0190	.0405	.0145	.0738	.0738	.0667

PC (psig)	LM-19- 300°	LM-20- 045°	LM-21- 045°	LM-22- 165°	LM-23- 165°	LM-24- 285°	LM-25- 285°
53	.0690	.0200	.0083	.0228	.0100	.0223	.0087
101	.1214	.0367	.0157	.0400	.0171	.0384	.0160
152	.1952	.0535	.0241	.0590	.0250	.0586	.0241
202	.2548	.0681	.0304	.0745	.0328	.0767	.0316
151	.1976	.0525	.0241	.0571	.0250	.0586	.0232
100	.1357	.0362	.0167	.0396	.0167	.0384	.0152
50	.0738	.0195	.0100	.0224	.0089	.0209	.0078

TABLE C-2

## HYDROTEST RESPONSE DATA FOR CHAMBER S/N 30114

PC (psig)	Strain - Microinches/Inch							
	SCAT-1	SCAT-2	SCAT-3	SCAT-4	SCAT-5	SCAT-6	SCAT-7	SCAT-8
53	848	896	1269	783	1125	800	1242	836
101	1695	1858	2504	1600	2233	1617	2367	1736
152	2657	3032	3841	2617	3488	2717	3726	2780
202	3668	4238	5127	3701	4760	3834	5099	3873
151	2706	3097	3959	2734	3619	2784	3857	2828
100	1728	1956	2741	1750	2461	1767	2586	1800
50	831	945	1506	867	1320	850	1344	868

PC (psig)	Strain - Microinches/Inch		
	SCC-9	SCC-10	SCC-11
53	1015	1306	864
101	1861	2581	1597
152	2741	4012	2380
202	3553	5411	3146
151	2758	4012	2396
100	1895	2581	1630
50	1015	1275	864

TABLE C-3

## SUMMARY RESPONSE DATA FOR CHAMBER S/N 30114

Hoop Strains at Barrel Section (Micro-inches/inch)		Pressure, psig				
		0	53	101	152	202
Forward Equator		0	1015	1861	2741	3553
Mid Barrel		0	1306	2581	4012	5411
Aft Equator		0	864	1597	2380	3146
Fwd Dome Deflections (inches)						
7.1 Radius	Longitudinal	0	.1381	.2428	.3428	.4262
	Radial	0	.0065	.0112	.0158	.0198
10.5 Radius	Longitudinal	0	.1409	.2432	.3405	.4048
	Radial	0	.0048	.0089	.0132	.0176
14.3 Radius	Longitudinal	0	.1452	.2333	.3159	.3886
	Radial	0	.0128	.0215	.0297	.0365
Aft Dome Deflection (inches)						
14.3 Radius	Longitudinal	0	.0525	.1050	.1523	.1977
	Radial	0	.0252	.0395	.0548	.0678
17.0 Radius	Longitudinal	0	.0619	.1048	.1500	.1905
	Radial	0	.0220	.0381	.0486	.0668
19.9 Radius	Longitudinal	0	.0381	.0714	.1050	.1400
	Radial	0	.0165	.0300	.0405	.0620



APPENDIX D

CALIBRATION DATA OF THE STRESS GAGES USED IN MOTOR NO. 1

## CALIBRATION DATA OF THE STRESS GAGES USED IN MOTOR NO. 1

## A. NORMAL STRESS GAGE CALIBRATION DATA

The normal stress transducers, Konigsberg Instruments, Inc. Models P14EB-SC-150 and P14EB-SC-450 were calibrated and temperature compensated by the supplier at the rated full scale pressure ranges of 150 and 450 psig, respectively, and at +30°, +80° and +130°F. They were then encapsulated with inert liner material (IBT-115) and the final bench pressure calibration and temperature compensation were performed. The manufacturer's bench calibration results for the normal stress gages are contained in the documentary files. However, a typical normal stress gage calibration data sheet is given in Table D-1. The ASPC code designation which corresponds to Konigsberg Instrument normal gage serial numbers are given in Table D-2. Table D-3 summarizes the normal gage sensitivity and zero readings at three different temperatures, 30°, 77° and 130°F. These values represent the gage calibration parameters which were originally used for the reduction of data taken on Motor No. 1.

After receipt at ASPC the gages were subjected to further calibrations, both in the laboratory and after installation in the motor.

To insure that the gages were operating in the tensile mode and that the gages were connected properly, they were calibration tested under vacuum. That is, the gages were placed in a vacuum jar, one at a time, at a temperature of 77°F + 3°F and tested in pressure increments (5 psi) to  $10^{-2}$  Torr. Figure D-1 contains typical vacuum calibration curves for Gages N4-1 and N4-2. Similar calibration curves for the rest of the normal gages are in the documentary files.

## B. SHEAR STRESS GAGE CALIBRATION DATA

The shear gages, designated Konigsberg Instruments Model H-2A, were fabricated by Konigsberg Instruments, Inc.. The ASPC shear gage code designation which corresponds to Konigsberg's Instrument serial numbers is given in Table D-4.

The fabrication and calibration procedures used were discussed extensively in the STV Report, AFRPL-TR-75-7.

The shear gages were cast and molded into the test fixture shown in Figure D-2. Then calibrated in an Instron Tensile Machine with a wrap-around, temperature conditioning chamber. The outputs of the shear gages were recorded directly onto an X-Y plotter. The first calibration series consisted of applying the load at a constant displacement rate onto the shear fixture, which contained 4 to 6 individual shear gages. Then the tests were repeated with the shear fixture inverted. Then constant strain rate calibration in the initial and "inverted" position were conducted at temperatures of 30°, 77°, and 130°F. Figures D-3 and D-4 provide the constant strain rate calibration curves for shear gage Nos. 10 and 12. The resulting calibration parameters are summarized in Table D-3.

To determine the effects of normal stress on gage response, constant strain rate calibrations were performed on the shear gages at 77°F by gripping the wooden test fixture and pulling normal to it. To assure that the gages

would perform properly for the pressurization tests, constant strain rate shear tests with a superimposed hydrostatic pressure of 25 psi were performed. The normal response and the superimposed hydrostatic calibrations for shear gages No. 10 and 12 are given in Figures D-5 and D-6. All shear calibration curves are contained in the documentary files.



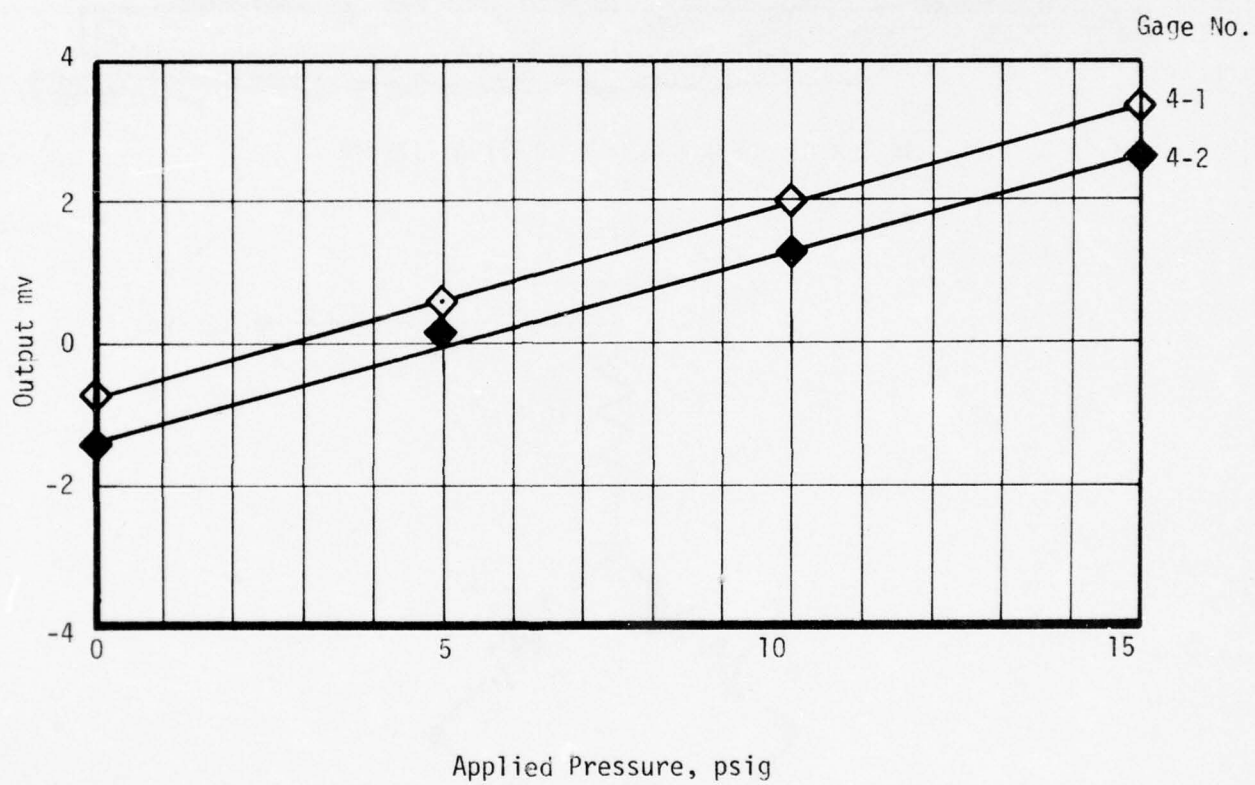


FIGURE D-1. PRESSURE CALIBRATION RESULTS FOR 450 PSI GAGES  
NOS. 4-1 AND 4-2 INSTALLED IN FULL SCALE  
MOTOR NO. 1; PRIOR TO CASTING

AD-A032 637

AEROJET SOLID PROPULSION CO SACRAMENTO CALIF  
FLEXIBLE CASE-GRAIN INTERACTION IN BALLISTIC WEAPON SYSTEMS. VO--ETC(U)  
OCT 76 K W BILLS, S W JANG, H LEEMING

F/G 21/9.2

F04611-72-C-0055

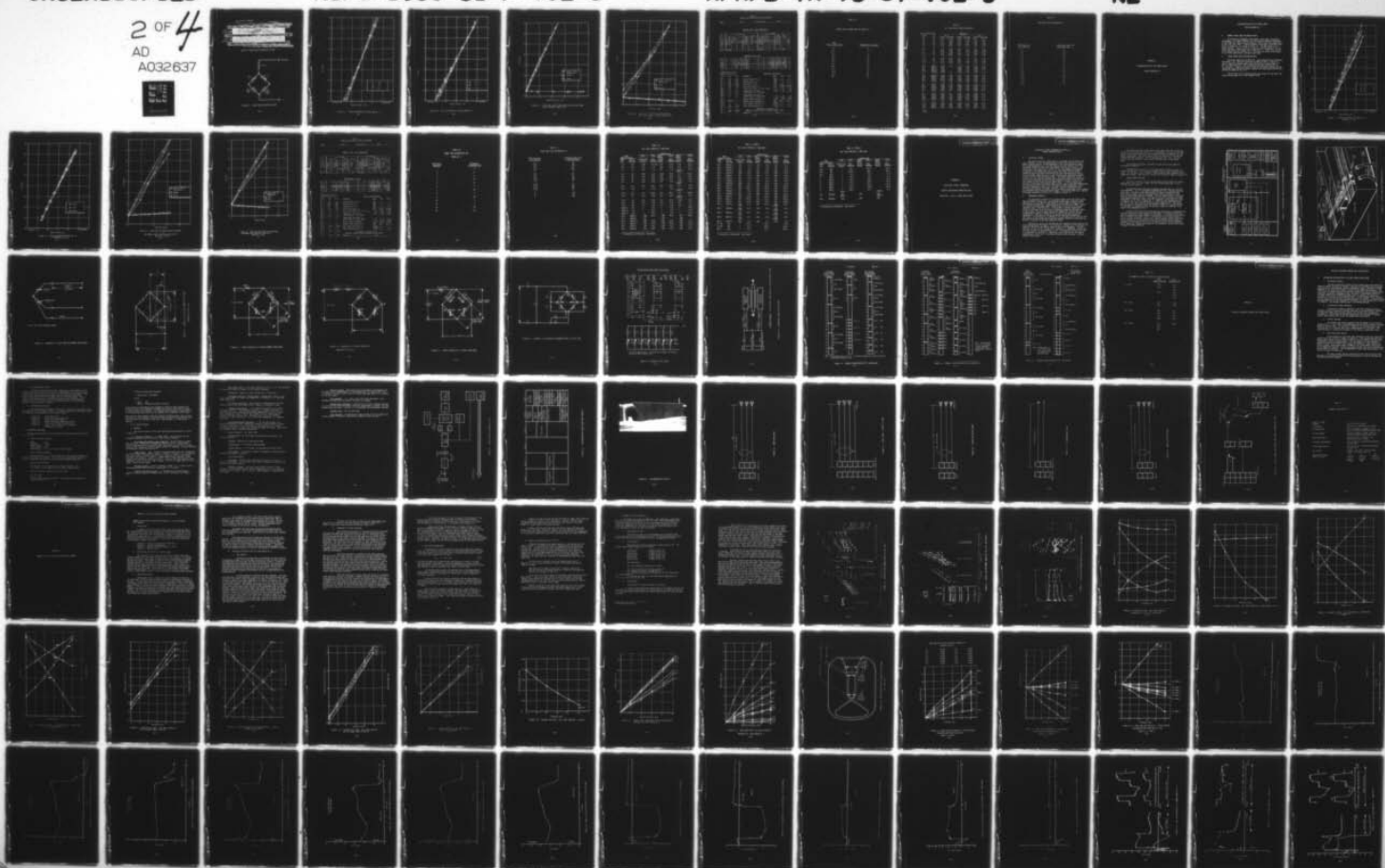
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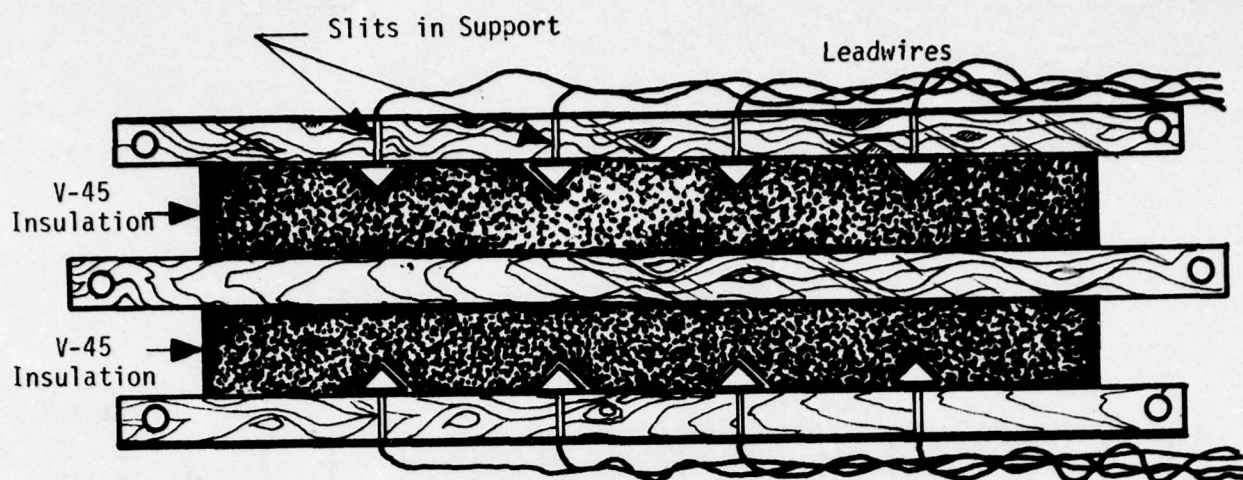
ASPC-1953-81-F-VOL-3

AFRPL-TR-76-57-VOL-3

NL

2 OF 4  
AD  
A032637





SKETCH OF SHEAR GAGE CALIBRATION FIXTURE

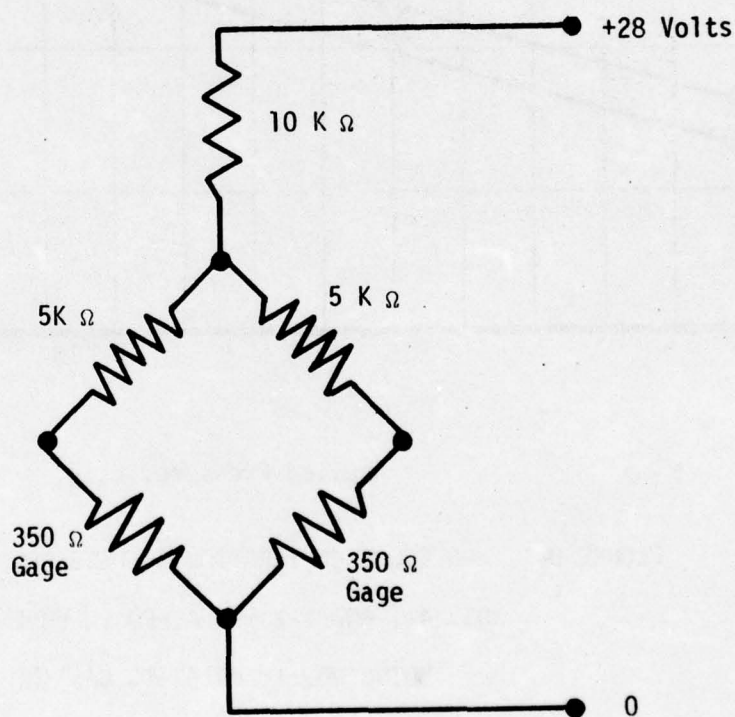


FIGURE D-2. SHEAR GAGE ELECTRICAL CIRCUIT



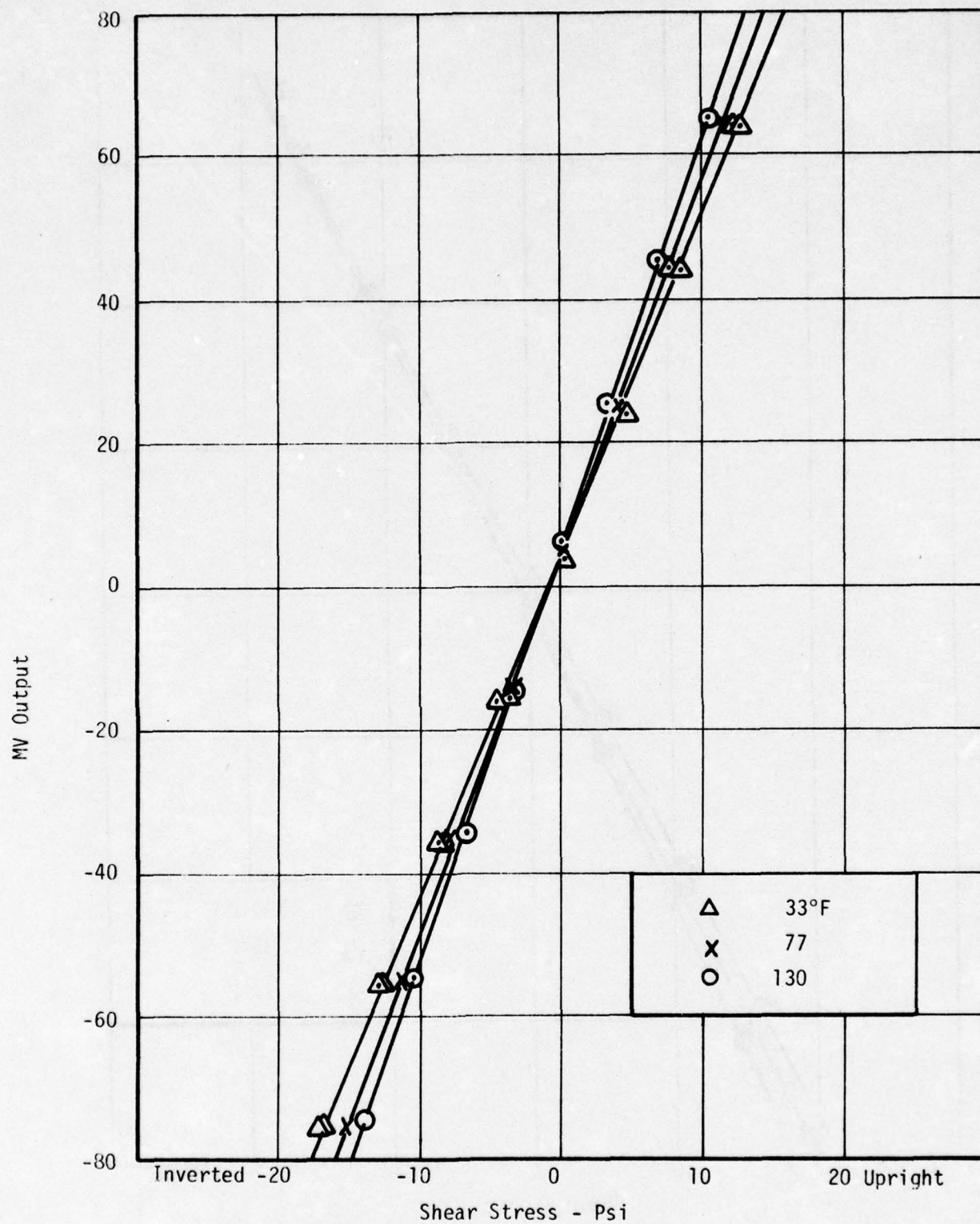


FIGURE D-3. SHEAR CALIBRATION OF SHEAR GAGE NO. 10  
D-7

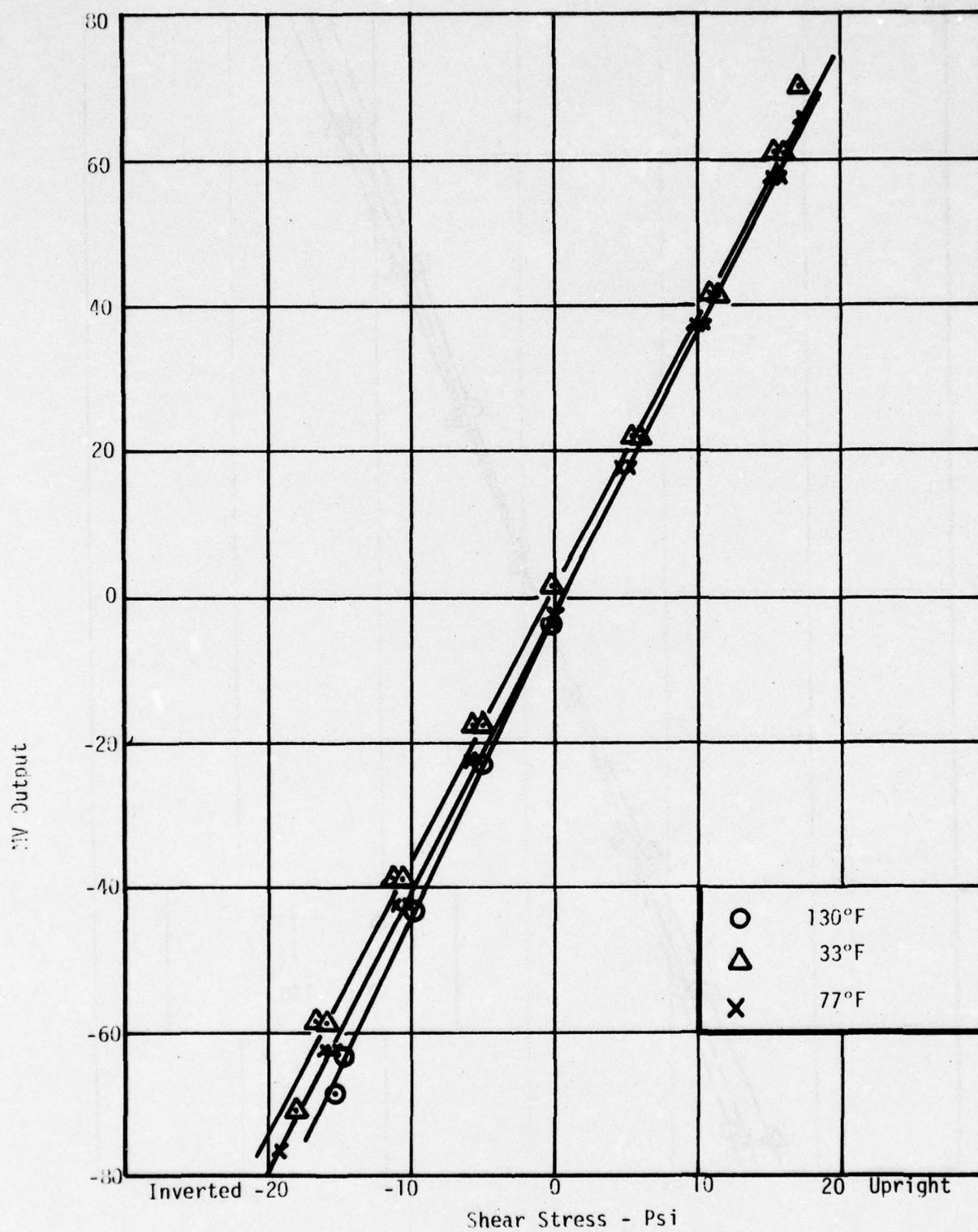


FIGURE D-4. SHEAR CALIBRATION OF SHEAR GAGE NO. 12

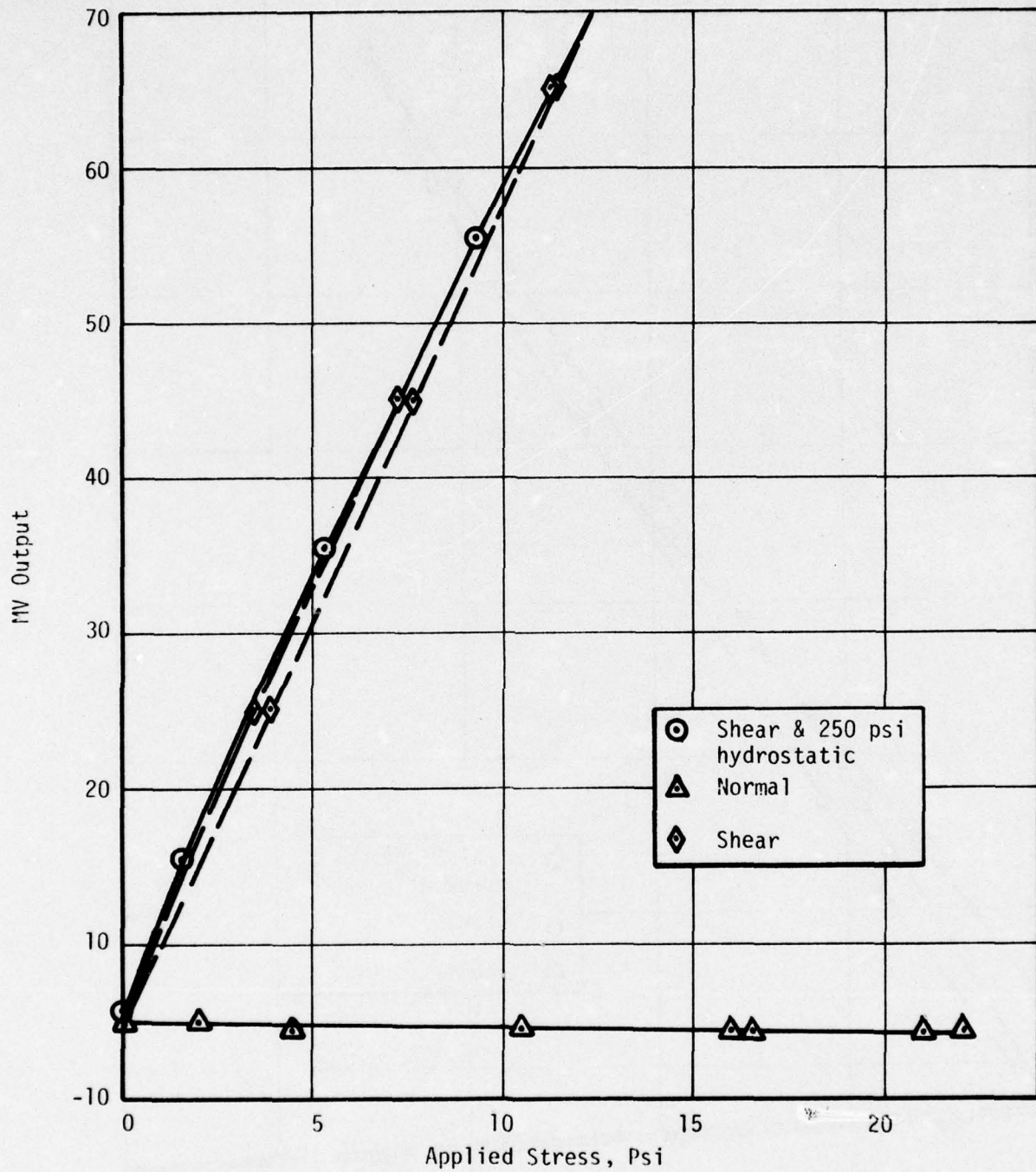


FIGURE D-5. SHEAR GAGE CALIBRATION UNDER PRESSURE AND NORMAL STRESS RESPONSE; GAGE NO. 10



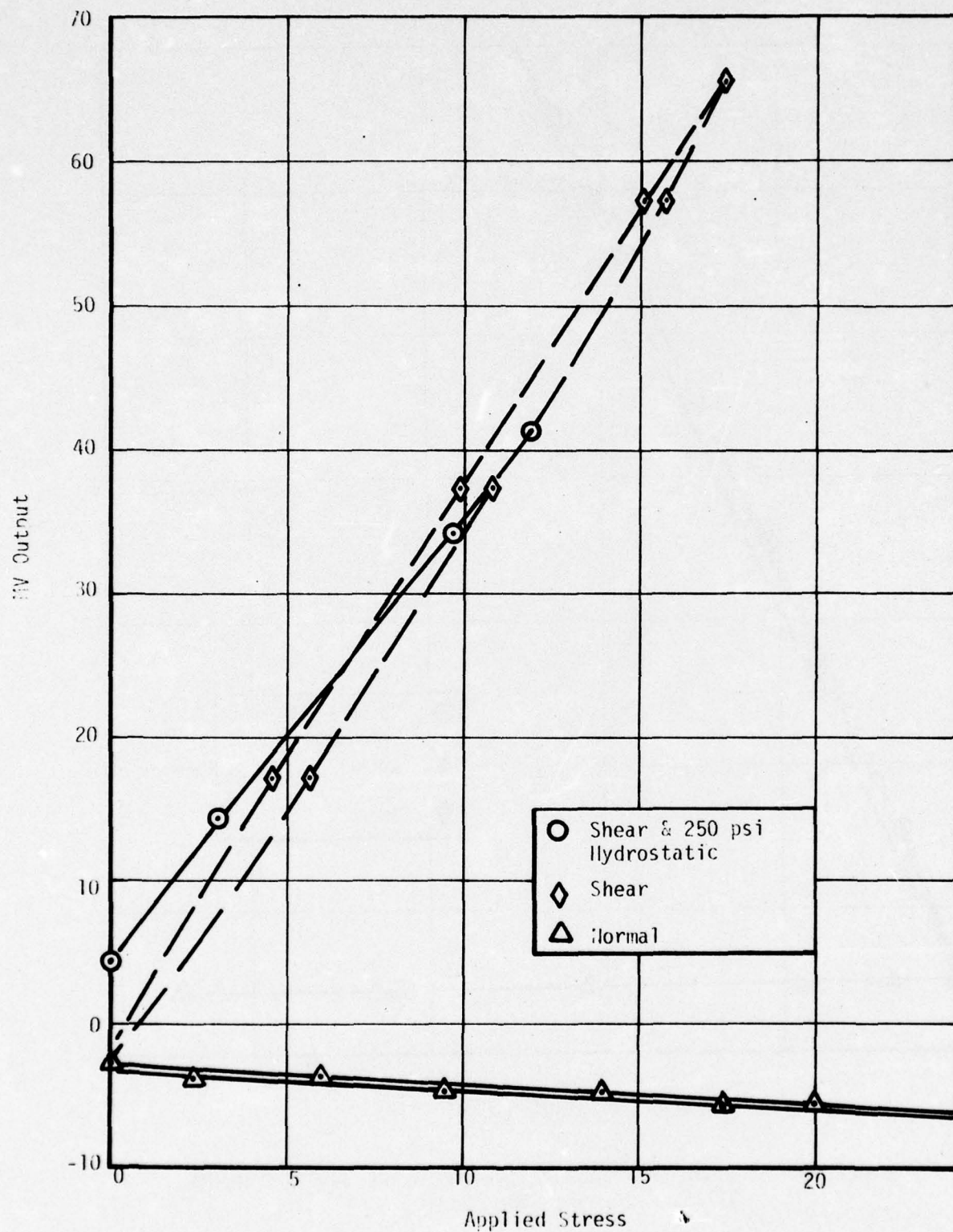


FIGURE D-6. SHEAR GAGE CALIBRATION UNDER PRESSURE  
AND NORMAL STRESS RESPONSE GAGE NO. 12

TABLE D-1  
VENDOR OBSERVATIONS OF GAGE CALIBRATIONS

Model \_\_\_\_\_ Serial \_\_\_\_\_ Calibrated By \_\_\_\_\_ Date \_\_\_\_\_

AMBIENT TEMP. STATIC ERROR BAND

PSI	1 Half Bridge					2 Half Bridge				
	Eo (mv)			Theo.	Dev'n.	Eo (mv)			Theo.	Dev'n
0	+ .90	+ .84	+	+ .90	- .06	+2.10	+2.01	+	+2.10	- .09
112.5	31.52	31.42	-	31.65	- .23	32.84	33.73	-	32.67	+ .17
225	62.44	62.37	62.38	62.40	- .03 + .04	63.64	63.53	63.70	63.24	+ .40
337.5	92.03		-	93.15	- .12	94.32		-	93.81	+ .51
450										
		↑					↑			
F.S.O.	123.00				%.27	122.28				%.60

TEMPERATURE TEST DATA

PSI °F	30	80	130			30	80	130
0	+1.24	+ .53	+1.08			- .18	- .29	- .34
150	41.78	41.53	41.68			40.88	40.47	40.21
F.S.O.	40.54	41.00	40.60			41.06	40.76	40.55
Δ F.S.O.	- .46	0	- .40			+ .30	0	- .21
Δ 0	+ .71	0	+ .55			+ .11	0	- .05

Compensation & Bridge  
Completion Resistors

Type	1 HB	2 HB
R <sub>1</sub> Ω	5.0K	5.0K
R <sub>2</sub> Ω	500	500
R <sub>3</sub> Ω	500	500
R <sub>4</sub> Ω	5.0K	5.0K
R <sub>B1</sub> Ω	0	0
R <sub>B2</sub> Ω	13.2	60
R <sub>B3</sub> Ω	0	0
R <sub>B4</sub> Ω	0	0
R <sub>Z1</sub> Ω	∞	∞
R <sub>Z2</sub> Ω	22.89K	∞
R <sub>Z3</sub> Ω	∞	∞
R <sub>Z4</sub> Ω	1.5K	1.5K
R <sub>S</sub> Ω	∞	∞
R <sub>P</sub> Ω	4.997	4.998
R <sub>L</sub> Ω		

Summarized Performance

Parameter	Goal	1 HB	2 HB
Pressure Range, P.S.I.	450	✓	✓
Overpressure, P.S.I.	900	✓	✓
Excitation, V.D.C.	28	✓	✓
Full Scale Output, mv	120	123.00	122.28
Static Error Band, + % F.S., B.S.L.	2.0	.11	.24
Temperature Zero Shift Error Band, + % F.S./100°F	2.0	.29	.07
Temperature Span Shift Error Band, + % Read./100°F	2.0	.56	.62
Nominal Input Impedance, Ω	5.0K	4.25K	4.25K
Nominal Output Impedance, Ω	900	813	817
Temp. Range-Calibrated, °F	30-130	✓	✓
Temp. Range-Operating, °F	0-150	✓	✓

Konigsberg Instruments, Inc.  
2000 East Foothill Blvd., Pasadena, Calif. 91107

Approved: \_\_\_\_\_ Date: \_\_\_\_\_

TABLE D-2

## NORMAL GAGE DESIGNATIONS FOR MOTOR NO. 1

<u>ASPC</u> <u>Normal Gage Numbers</u>	<u>Konigsberg Instruments</u>
N-1	12
N-2	1
N-3	2
N-4	10
N-5	14
N-6	15
N-7	16
N-8	17
N-9	18
N-10	19
N-11	20
3D	25



TABLE D-3  
FULL SCALE MOTOR #1 GAGE CALIBRATIONS

Gage Identification		Temperature					
Motor No.	KI No.	30°F		77°F		130°F	
		mv/psi	Zero Rdg.	mv/psi	Zero Rdg.	mv/psi	Zero Rdg.
S-1	23	2.67	-3.50	2.84	-2.90	3.15	-3.81
S-2	22	2.47	0.30	2.51	0.67	2.65*	3.61
S-3	10	4.13	0.00	4.59	-0.42	5.01	-2.05
S-4	40	3.39	-1.00	4.16	-6.35	3.20	6.93
S-5	18	4.62	-6.50	5.14	-10.79	5.73	-9.90
S-6	17	2.78	1.00	2.82	6.11	2.63	2.35
S-7	15	3.03	4.00	3.01	3.09	3.22	4.08
S-8	41	5.16	1.00	5.71	-4.11	4.92*	-6.80
S-9	16	4.78	-15.00	4.59	-4.10	5.06	-2.23
S-10	12	3.08	14.00	3.20	7.66	3.39	4.50
S-11	24	3.45	16.00	3.87	7.79	4.00	-1.41
S-12	21	3.33	5.50	3.61	1.13	3.32	-1.00
S-13	20	3.18	2.00	3.58	9.04	3.97	9.00
S-14	11	4.28	-0.70	4.70	-3.72	5.34	-4.78
N1-1	450/12	0.274	7.50	0.277	-5.56	0.277	-9.08
N1-2	450/12	0.270	?	0.280	-7.51	0.280	-4.74
N2-1	450/1/5	0.273	-	0.275	1.25	0.276	0.40
N2-2	450/1/5	0.268	-	0.269	-11.67	0.270	-
N3-1	450/2/5	0.269	-1.50	0.272	-5.37	0.274	-11.64
N3-2	450/2/5	0.267	5.80	0.270	5.00	0.270	6.77
N4-1	450/10	0.275	3.00	0.277	-0.76	0.276	4.78
N4-2	450/10	0.274	2.50	0.273	-1.48	0.270	-0.44
N5-1	150/14	0.810	-3.00	0.810	-4.80	0.815	3.52
N5-2	150/14	0.750	-4.00	0.754	-5.85	0.760	-4.65
N6-1	150/15	0.808	-5.50	0.812	-7.50	0.813	0.59
N6-2	150/15	0.805	-1.50	0.806	-3.50	0.810	3.24
N7-1	150/16	0.808	-3.50	0.810	-3.65	0.810	10.11
N7-2	150/16	0.803	2.20	0.806	+0.50	0.808	11.46
N8-1	150/17	0.820	4.0	0.827	2.52	0.830	14.87
N8-2	150/17	0.835	-6.5	0.840	2.50	0.840	-26.72
N9-1	150/18	0.807	3.30	0.810	4.36	0.815	5.20
N9-2	150/18	0.807	-7.00	0.805	-10.52	0.806	-3.72
N10-1	150/19	0.815	-3.00	0.810	-3.42	0.806	3.27
N10-2	150/19	0.825	-5.50	0.820	-7.56	0.825	-1.46
N11-1	150/20	0.825	5.00	0.825	+3.10	0.825	13.54
N11-2	150/20	0.823	-9.00	0.820	-14.43	0.816	-8.76
3DN-1	450/25	0.273	2.00	0.273	-0.19	0.273	-2.38
3DN-2	450/25	0.275	-3.00	0.275	4.33	0.275	0.67

\*144°F  
110°F

TABLE D-4

## SHEAR GAGE CODE FOR MOTOR NO. 1

<u>ASPC Gage Code</u> <u>Shear Gage No.</u>	<u>Konigsberg Gage Code</u> <u>Shear Gage S/N</u>
S-1	23
S-2	22
S-3	10
S-4	40
S-5	13
S-6	17
S-7	15
S-8	41
S-9	16
S-10	12
S-11	24
S-12	21
S-13	20
S-14	11

APPENDIX E

CALIBRATION DATA OF THE STRESS GAGES

USED IN MOTOR NO. 2



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## CALIBRATION DATA OF THE STRESS GAGES

### USED IN MOTOR NO. 2

#### A. NORMAL STRESS GAGE CALIBRATION DATA

As discussed in Appendix D the normal stress gage, Konigsberg Instruments, Inc. Models P14EB-SD-150 and P143B-SC-450 were calibrated at the rated full scale pressure ranges of 150 and 450 psig respectively at +30, +80, and +130°F. A typical normal stress gage calibration data sheet for Motor No. 2 is shown in Table E-1. The data sheet is for normal gage N6. Tables E-2 and E-3 show the code designations for the normal and shear stress gages used in Motor No. 2. Table E-4 lists the zero readings and the sensitivity of the shear and normal stress gages at the three temperatures previously mentioned.

#### B. SHEAR STRESS GAGE CALIBRATION DATA

The shear gages used in Motor No. 2 were fabricated by Konigsberg Instruments, Inc. and were calibrated by ASPC and HL&A. Figures E-1 and E-2 show the constant shear rate calibration for shear gages S-2 and S-3 (SN 26 and 27) in the inverted and upright positions at 33, 77, and 130°F. Figures E-3 and E-4 show the calibration curves for the same shear gages during the normal response and superimposed hydrostatic calibration tests at  $77 \pm 3^\circ\text{F}$ .

The balance of the calibration data and curves for the shear and normal stress gages are in the document files.

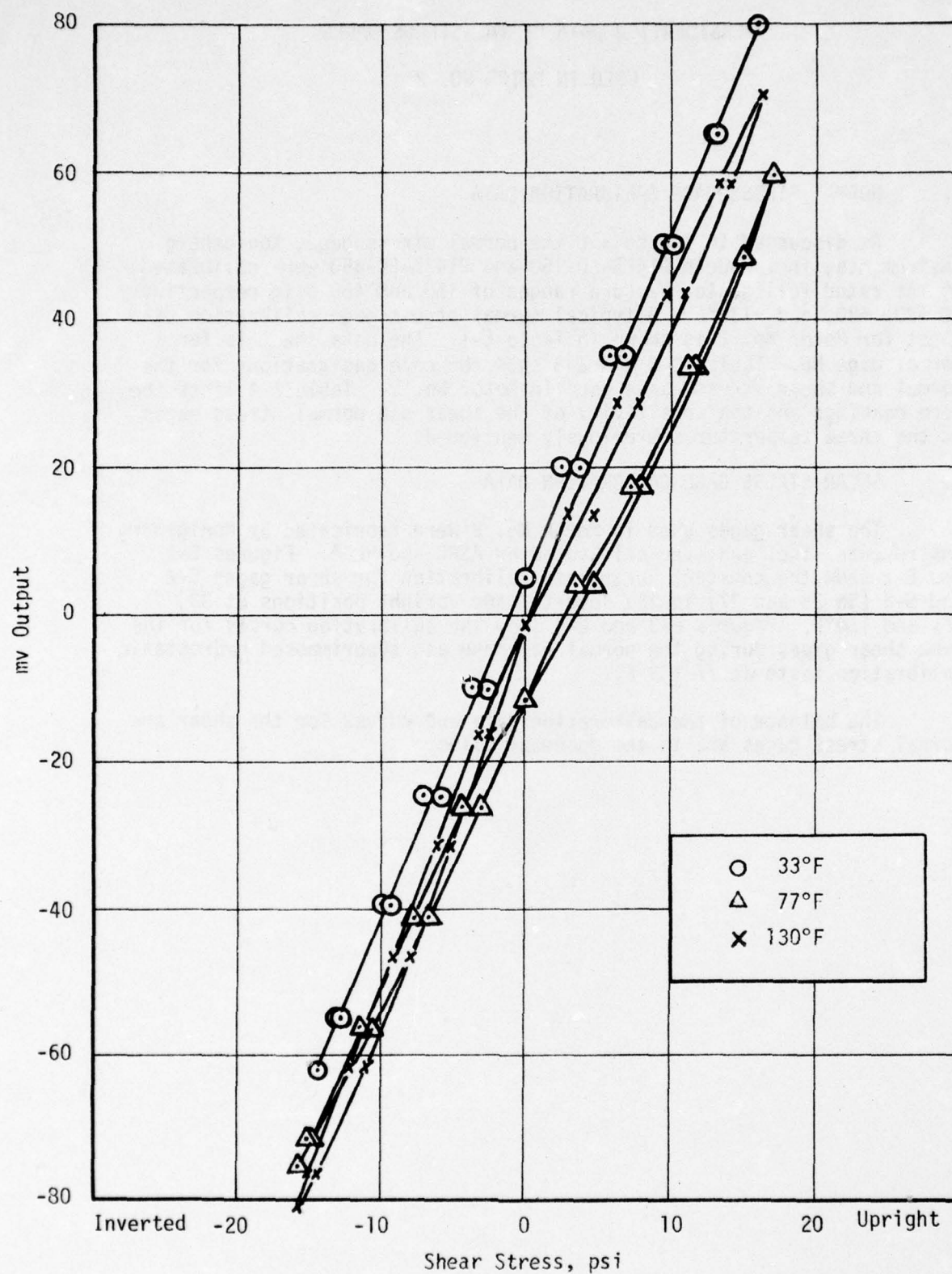


FIGURE E-1. SHEAR CALIBRATION OF GAGE NO. 26  
MOTOR NO. 2, SH2

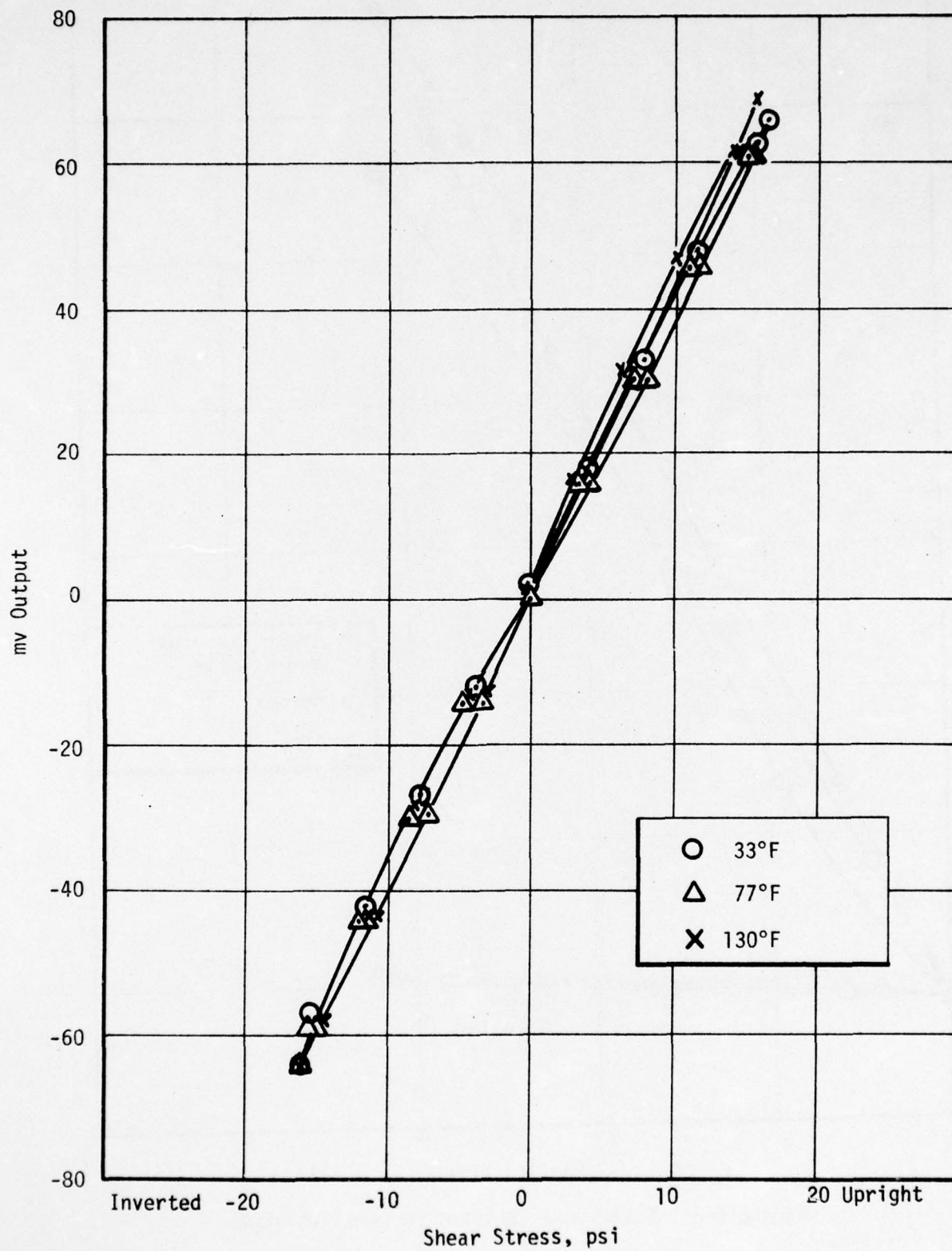


FIGURE E-2. SHEAR CALIBRATION OF GAGE NO. 27  
MOTOR NO. 2, SH3



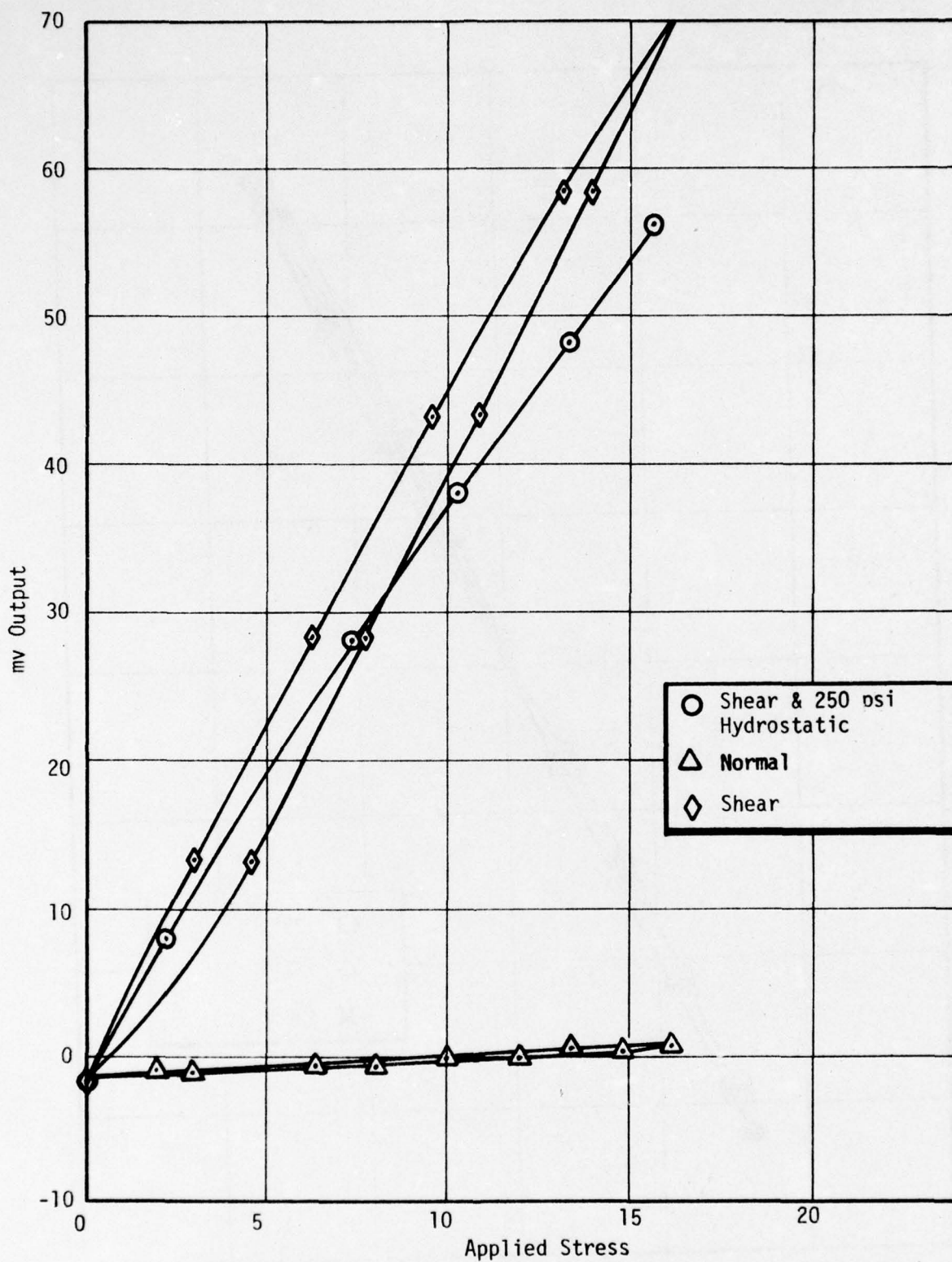


FIGURE E-3. SHEAR GAGE CALIBRATION UNDER PRESSURE

AND NORMAL STRESS RESPONSE; GAGE NO. 26

MOTOR NO. 2, SH2

E-6

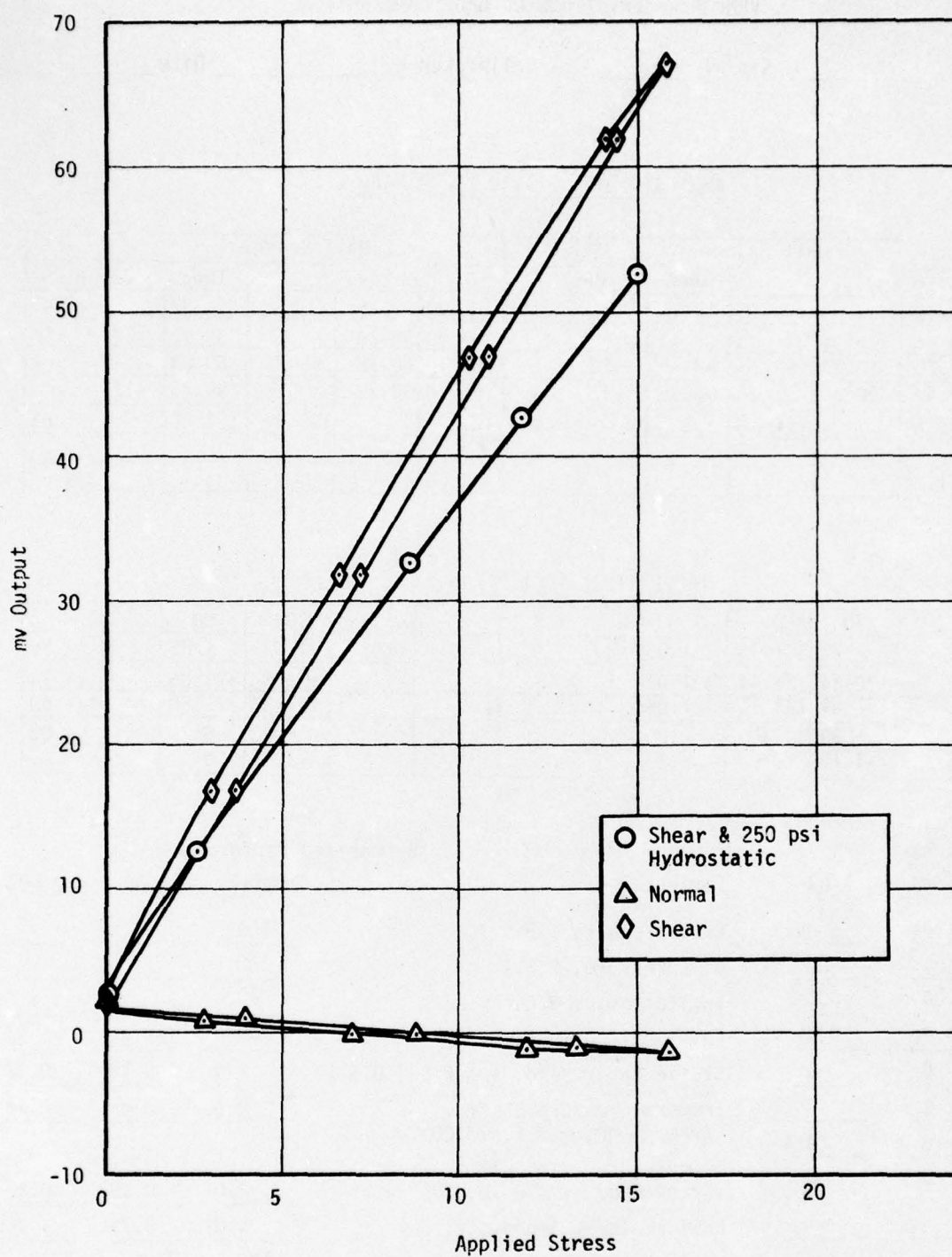


FIGURE E-4. SHEAR GAGE CALIBRATION UNDER PRESSURE  
AND NORMAL STRESS RESPONSE; GAGE NO. 27  
MOTOR NO. 2, SH2

TABLE E-1  
VENDOR OBSERVATIONS OF GAGE CALIBRATIONS

Model \_\_\_\_\_ Serial \_\_\_\_\_ Calibrated By \_\_\_\_\_ Date \_\_\_\_\_

AMBIENT TEMP. STATIC ERROR BAND

PSI	1 Half Bridge					2 Half Bridge				
	Eo (mv)			Theo.	Dev'n.	Eo (mv)			Theo.	Dev'n
0	-.16	-.03	+	-.16	+.13	-.30	-.26	+	-.30	+.04
375	29.84	29.94	-	30.14	-.30	29.90	30.00	-	30.12	-.22
75	60.22	60.29	60.28	60.45	-.23	60.38	60.38	60.36	60.53	-.17
112.5	90.57	90.66	-	90.75	-.18	90.70	90.78	-	90.95	-.25
	121.05		121.08	121.05	-.03	121.36		121.39	121.36	+.03
		+					+			
F.S.O.	121.21				Σ .43	121.66				Σ .29

TEMPERATURE TEST DATA

PSI °F	30	80	130				30	80	130
0	-.77	+.33	+.03				-1.03	+.05	-.54
150	120.84	121.44	121.09				120.80	121.65	121.14
F.S.O.	121.61	121.11	121.06				121.83	121.60	121.68
ΔF.S.O.	+.50	0	-.05				+.23	0	+.08
Δ 0	-1.10	0	-.30				-1.08	0	-.59

Compensation & Bridge  
Completion Resistors

Summarized Performance

Type	1 HB	2 HB	Parameter	Goal	1 HB	2 HB
R <sub>1</sub> Ω	5.0K	5.0K	Pressure Range, P.S.I.	150	✓	✓
R <sub>2</sub> Ω	500	500	Overpressure, P.S.I.	300	✓	✓
R <sub>3</sub> Ω	500	500	Excitation, V.D.C.	28	✓	✓
R <sub>4</sub> Ω	5.0K	5.0K	Full Scale Output, mv	120	121.21	121.66
R <sub>B1</sub> Ω	0	0	Static Error Band, + % F.S., B.S.L.	2.0	0.18	0.12
R <sub>B2</sub> Ω	0	0	Temperature Zero Shift	2.0	0.45	0.44
R <sub>B3</sub> Ω	10.3	9.4	Error Band, + % F.S./100°F			
R <sub>B4</sub> Ω	0	0	Temperature Span Shift	2.0	0.23	0.13
			Error Band, + % Read./100°F			
R <sub>Z1</sub> Ω	∞	∞	Nominal Input Impedance, Ω	5.0K	4.75K	4.95K
R <sub>Z2</sub> Ω	∞	∞	Nominal Output Impedance, Ω	900	770	770
R <sub>Z3</sub> Ω	55.0K	30.0K	Temp. Range-Calibrated, °F	30-130	✓	✓
R <sub>Z4</sub> Ω	∞	∞	Temp. Range-Operating, °F	0-150	✓	✓
R <sub>S</sub> Ω	2.0K	2.2K				
R <sub>P</sub> Ω	∞	∞				
R <sub>L</sub> Ω	5.0K	5.0K				

Konigsberg Instruments, Inc.  
2000 East Foothill Blvd., Pasadena, Calif. 91107

Approved: \_\_\_\_\_ Date: \_\_\_\_\_



TABLE E-2  
NORMAL GAGE DESIGNATIONS FOR  
MOTOR NO. 2

<u>ASPC Normal Gage Nos.</u>	<u>Konigsberg Instruments S/N</u>
1	3
2	11
3	4
4	22
5	21
6	6
7	29
8	3
9	28
10	7
11	4
26	26
27	27

TABLE E-3  
SHEAR GAGE CODE FOR MOTOR NO. 2

<u>ASPC Gage Code</u> <u>Shear Gage No.</u>	<u>Konigsberg Gage Code</u> <u>Shear Gage S/N</u>
S-1	30
S-2	26
S-3	27
S-4	42
S-5	32
S-6	29
S-7	31
S-8	33
S-9	S-B
S-10	34
S-11	13
S-12	35
S-13	25
S-14	36
A	S-A
B	S-C
C	S-D

TABLE E-4  
FULL SCALE MOTOR NO. 2 GAGE DATA

Gage Identification		Temperature					
		33°F	30°F	80°F	74°F	130°F	110°F
Motor No.	KI or HL&A No.	Sensitivity mv/psi	Zero Reading	Sensitivity mv/psi	Zero Reading	Sensitivity mv/psi	Zero Reading
S-1	30	4.35	+2.1 mv	4.6	+1.8	4.8	+2.7
S-2	26	4.0	-7.3	4.8	+1.8	4.6	+6.7
S-3	27	3.8	-1.4	4.1	-1.8	3.8 (144°F)	-2.0
S-4	42	3.05	-3.2	3.65	+4.5	3.7	+11.1
S-5	32	3.6	+0.2	3.6	+3.5	3.5	+5.2
S-6	29	3.6	+8.3	3.85	+1.3	3.75	-4.3
S-7	31	4.0	-6.5	3.9	0.0	3.75 (144°F)	-1.2
S-8	33	2.0	+10.0	2.0	+6.5	2.0	+4.5
S-9	S-B	*	-0.5	*	0.0	* (144°F)	-1.8
S-10	34	2.6	+42.0	2.4	+37.5	2.2	+35.0
S-11	13	3.2	+5.8	3.0	-0.5	3.5 (144°F)	-3.5
S-12	35	2.6	+4.0	2.3	+1.2	2.45	+1.4
S-13	25	2.8	-2.4	3.1	-2.6	3.2	-1.7
S-14	36	2.1	-0.7	1.9	-1.0	1.6	-1.8
Shear A	S-A	*	-3.7	*	-1.5	*	+0.4
Shear C	S-C	*	+0.2	*	0.0	*	-1.0
Shear D	S-D	*	+0.1	*	-1.0	*	-0.3
N1-1	450/3-1	.268	-19.2	.269	-19.2	.271	-20.2
N1-2	450/3-2	.271		.272		.272	
N2-1	450/11-1	.269	-12.0	.271	-8.6	.269	-7.2
N2-2	450/11-2	.272	-14.9	.274	-14.8	.272	-15.3
N3-1	450/4-1	.275	-10.2	.274	-4.0	.272	-4.3
N3-2	450/4-2	.273	-10.3	.273	-9.0	.274	-11.2

\* Viscoelastic calibration. See curves.



TABLE E-4 (CONT.)  
FULL SCALE MOTOR NO. 2 GAGE DATA

Gage Identification		Temperature					
		33°F	30°F	80°F	74°F	130°F	110°F
Motor No.	KI or HL&A No.	Sensitivity mv/psi	Zero Reading	Sensitivity mv/psi	Zero Reading	Sensitivity mv/psi	Zero Reading
N4-1	450/22-1	.270	-3.0	.270	-2.5	.270	-1.3
N4-2	450/22-2	.272	-2.1	.272	-1.6	.274	-0.3
N5-1	150/21-1	.823	-7.5	.821	-8.4	.818	-8.6
N5-2	150/21-2	.820	-18.4	.821	-17.7	.826	-17.8
N6-1	150/6-1	.811	+2.0	.807	+1.7	.807	+0.9
N6-2	150/6-2	.812	+5.9	.811	+5.1	.811	+3.4
N7-1	150/29-1	.816	+3.7	.814	+3.2	.816	+4.0
N7-2	150/29-2	.820	-5.1	.816	-4.3	.818	-3.8
N8-1	150/3-1	.825	+18.7	.819	+18.5	.813	+17.3
N8-2	150/3-2	.818	+20.3	.818	+21.0	.812	+20.7
N9-1	150/28-1	.811	-8.0	.813	-10.5	.811	-11.5
N9-2	150/28-2	.801	-9.5	.806	-10.7	.807	-11.2
N10-1	450/7-1	.270	+18.0	.271	+16.7	.268	+18.0
N10-2	450/7-2	.272	+20.0	.271	+18.7	.274	+20.5
N11-1	150/4-1	.841	+16.8	.839	+15.2	.841	+14.8
N11-2	150/4-2	.806 (-75)	+15.3	.806	+15.5	.805 (+180)	+15.2
N26-1	450/26-1	.268 (-75)	-10.9	.269	-8.0	.270 (+180)	-6.3
N26-2	450/26-2	.270 (-75)	-10.7	.270	-7.5	.270 (+180)	-6.3
N27-1	450/27-1	.270 (-75)	-0.7	.266	+2.8	.267 (+180)	+4.3
N27-2	450/27-2	.268	+2.6	.266	-1.7	.270	-1.8
BI-7SH	3D5	*	+13.0		+10.1		+7.6
BI-5SH	3D5	*	+7.0		+20.7		+25.7
6A-D	3D5	*	-12.2		+11.0		+23.0

\* Viscoelastic calibration. See curves.

TABLE E-4 (CONT.)  
FULL SCALE MOTOR NO. 2 GAGE DATA

Gage Identification Motor No.      KI or HL&A No.		Temperature					
		33°F	30°F	80°F	74°F	130°F	110°F
		Sensitivity mv/psi	Zero Reading	Sensi- tivity mv/psi	Zero Reading	Sensi- tivity mv/psi	Zero Reading
6B-D	3D5	*	+5.1		+18.7		+25.0
5A-D	3D5	*	+23.8		+28.7		+30.6
7A-D	3D5	*	-3.8		+0.4		+1.6
2+	3D6-4	*	+7.0		-1.8		-11.8
2-	3D6-3	*	-10.3		-20.5		-13.0
3+	3D6-6	*	-9.0		-19.0		-29.0
3-	3D6-5	*	-13.8		-27.6		-37.5
1+	3D6-2	*	-13.5		-18.0		-24.5
1-	3D6-1	*	+16.3		+11.3		+3.0
		(0°F)				(150°F)	
N35	150/35-1	.761 (0°F)		.759		.762 (150°F)	
N36	150/36-1	.792		.804		.812	

\* Viscoelastic calibration. See curves.

APPENDIX F

ELECTRICAL SYSTEM, TRANSDUCER

CIRCUITS AND BRIDGE COMPLETION UNIT

(Motor Nos. 1 and 2, except when noted)



ELECTRICAL SYSTEM, TRANSDUCER CIRCUITS, &  
BRIDGE COMPLETION UNIT

## A. ELECTRICAL SYSTEM

The total electrical system used to instrument and to monitor the grain stresses, strain, and temperatures for the Flex Case/Grain Interaction Program consists of the transducers, the interconnecting network, the bridge completion unit, and the recording system (Figure F-1). The transducer system includes the normal gages, shear gages, thermocouples, event gages, clip gages, and L.V.D.T.'S. The interconnection network is made up of the connecting wires between the gages and the bridge completion unit with their terminations and the interconnection cables from the bridge completion unit to the switching and terminating network in the trailer to the recording system. The bridge completion system consists of four aluminum junction boxes, mounted directly onto the motor skirt, which contains the half-bridge completion networks (printed circuit boards) for the shear and the normal gages. It also contains the necessary terminations and feedthroughs for the event gages and thermocouples. The recording system consists of a 25 channel scanner/programmer, power supplies, temperature strip chart recorder, digital recorder/printer, an integrating digital voltmeter, auxiliary measuring equipment, oscillograph, and alternate strip charts and power supply. A pictorial view for a typical stressgage connected to the B.C.U. is shown in Figure F-2.

## B. TRANSDUCER CIRCUITS

The electrical circuits of a typical normal stress and shear stress gage are shown in Figure F-3. As previously mentioned, a typical stress gage consists of two 500 ohm semiconductor strain gages arranged into a half-bridge configuration. Three wires exit from the stress gage. Figure F-4 shows the normal gage, model P14EB-SC-150, connected to its bridge completion unit. When the B.C.U. is connected, a typical wheatstone bridge is balanced by artificially introducing balancing resistors, RB's, in the bridge. The full-scale ranging and temperature connections are accomplished by means of adjusting the  $R_1$  and  $R_7$  resistors at various temperatures. Resistors  $R_1$  and  $R_4$  are the 5K ohm bridge completion resistors. The resistor  $R_5$  is generally 2K to 5K ohms, and is used to limit the current to the bridge and to provide constant current to the bridge to maintain linearity of the semi-conductor bridge.

Resistors  $R_2$  and  $R_3$  are the 500 ohm semi-conductor strain gages of the normal stress gage. The input is 28 volts D.C., and the full-scale output is generally 125 mv for the 150 psi and the 450 psi normal gages. A brief analysis of this circuit (Figure F-5), assuming  $R_5 = 2000$  ohms, showed that the input current is approximately 5.1 ma. The dropping resistor ( $R_5$ ) has a voltage drop of 12.0 volts. Therefore, the voltage across the gage is approximately 1.50 volts. The heat dissipated from each gage is approximately 3 milliwatts.

The typical circuit of the shear stress gage, model H2A, connected to the B.C.U. is shown in Figure F-6. The equivalent circuit for the shear gage is a simplified version of the wheatstone bridge as compared with the normal stress gage. The strain gage circuit only consists of the 5K ohms bridge completion resistors, the two dropping resistors (two - 5K ohms) and the two semi-conductor strain gages, 500 ohms each, in the shear gage.

The voltage and current parameters of the shear stress gage are presented in Figure F-7.

The electrical circuit for a clip gage and a 3D gage are shown in Figures F-8 and F-9. The clip gage includes 4 active semi-conductor strain gages. The 3D gage circuit includes 6 semi-conductor gages connected into a common half-bridge. The electrical circuit for an LVDT is shown in Figure F-10.

#### C. BRIDGE COMPLETION UNIT

The B.C.U. consists of four junction boxes used to house all termination points from the stress gages and their bridge completion circuits (printed circuit boards).

The junction boxes used were purchased from Zero Mfg. Co. (Part No. Z120-192-64EE). It is a rectangular aluminum box with a separate moisture seal cover. Terminal strips were mounted in place to accept the input leads from the gages and the interconnections from the printed circuit board of the bridge completion network. All the output leads are fastened onto separate terminal strips. The identification of terminals and interconnections of the four J boxes with the respective stress gages are shown in Figures F-11 to F-14 for Motor No. 1. They show the interconnection for the 0°, 90°, 180° and 270° junction boxes, respectively. The assignment of gages to respective boxes are shown in Table F-1. Identification of terminals and interconnections of the four J boxes of Motor No. 2 are given in the documentary file.

Printed circuit boards were designed and fabricated to accommodate the required completion bridge network for the gages. The primary objectives were to standardize bridge completion connections, to minimize wiring errors, and to minimize poor solder joints. The printed circuit board layout to scale is given in Volume I (Figure 38). Due to various loading conditions, it was designed to be permanently horizontally mounted and was purposely designed to eliminate the usual type of printed circuit board connections with a socket for connections because of its tendencies to work loose under stringent loading conditions.

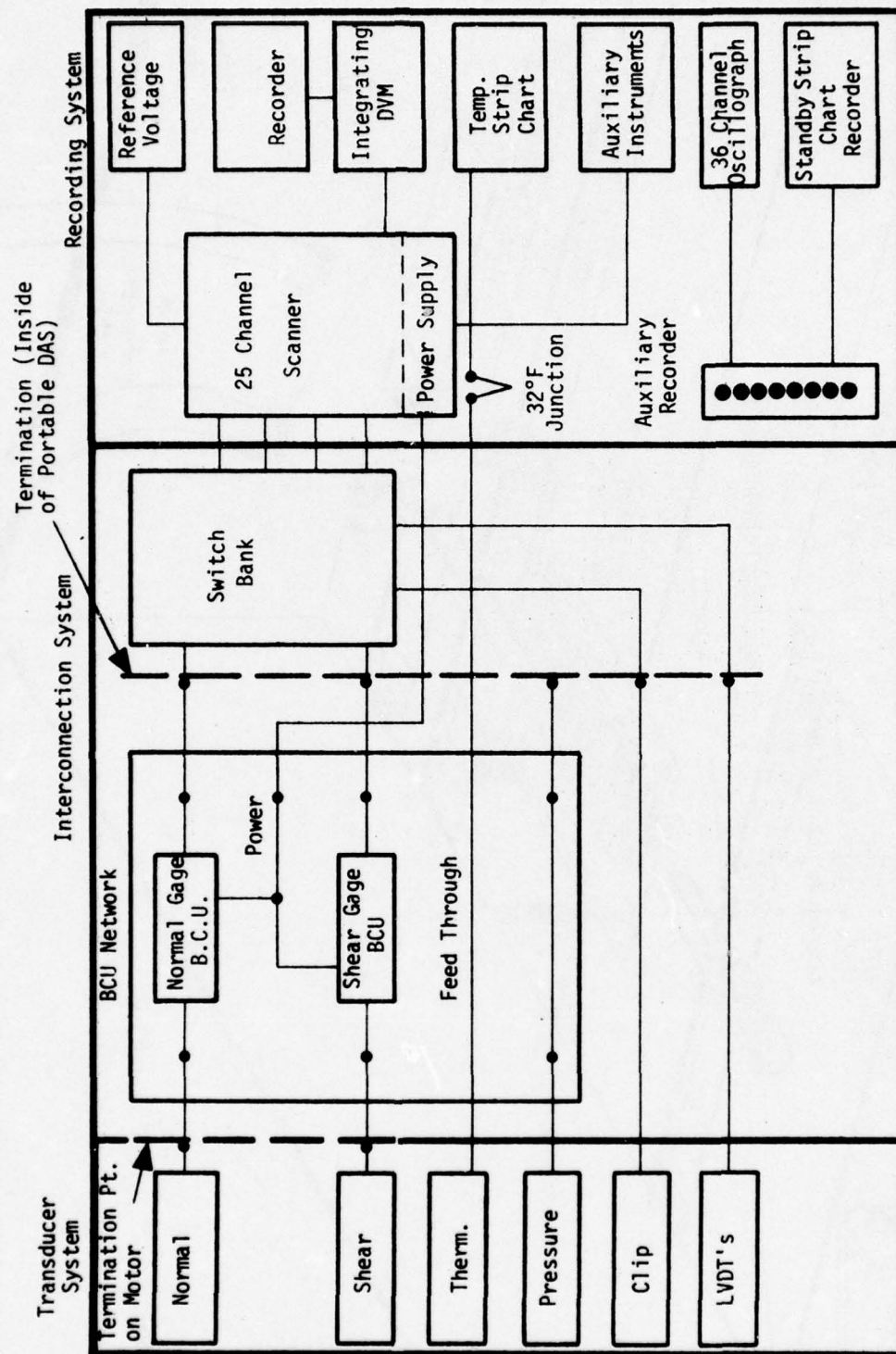


FIGURE F-1. TOTAL ELECTRICAL SYSTEM FOR FLEX CASE INSTRUMENTATION



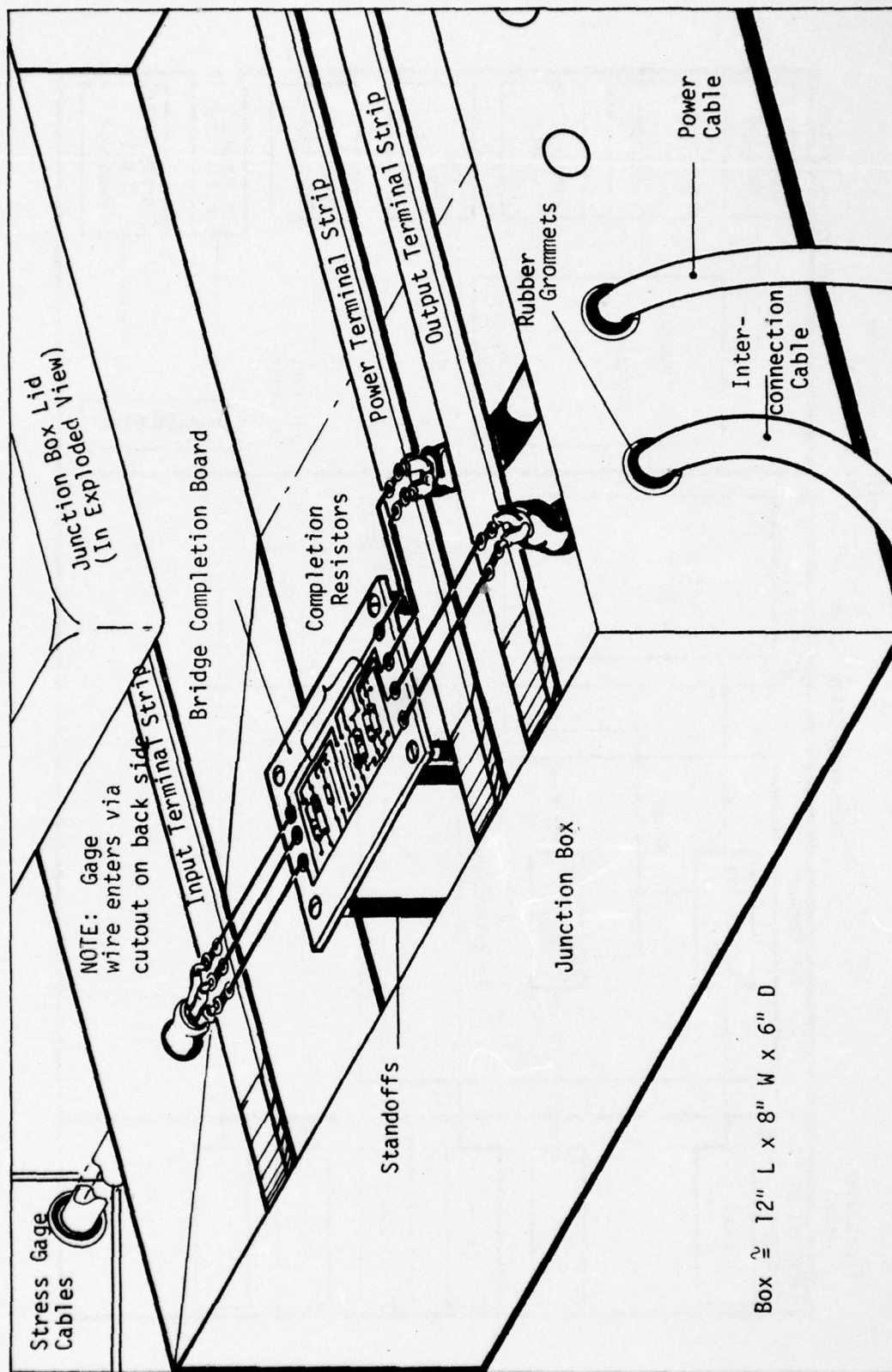
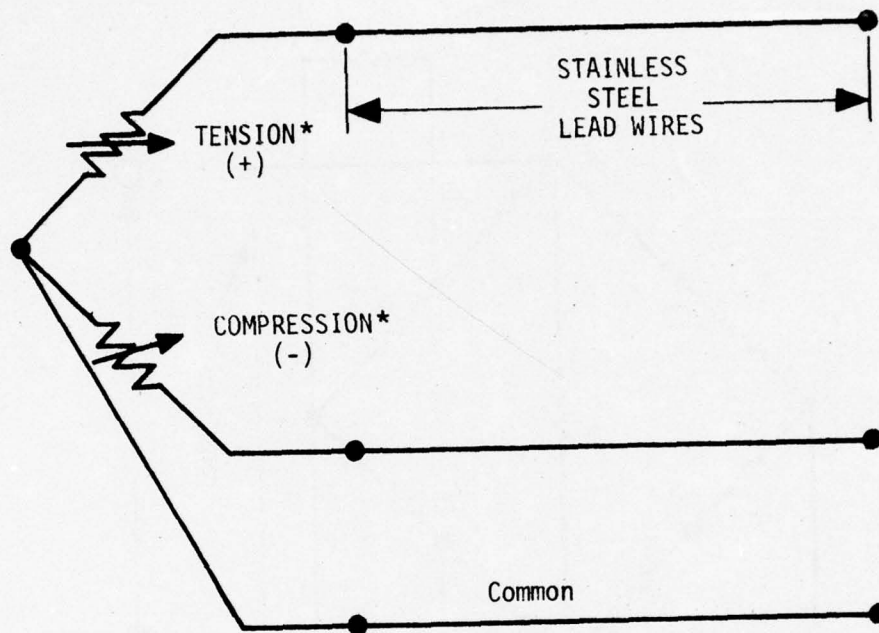


FIGURE F-2. TYPICAL STRESS GAGE CONNECTION TO B.C.U.



\* Two - 500  $\Omega$  semi-conductor gages.

FIGURE F-3. SCHEMATIC OF TYPICAL SHEAR AND NORMAL STRESS GAGES

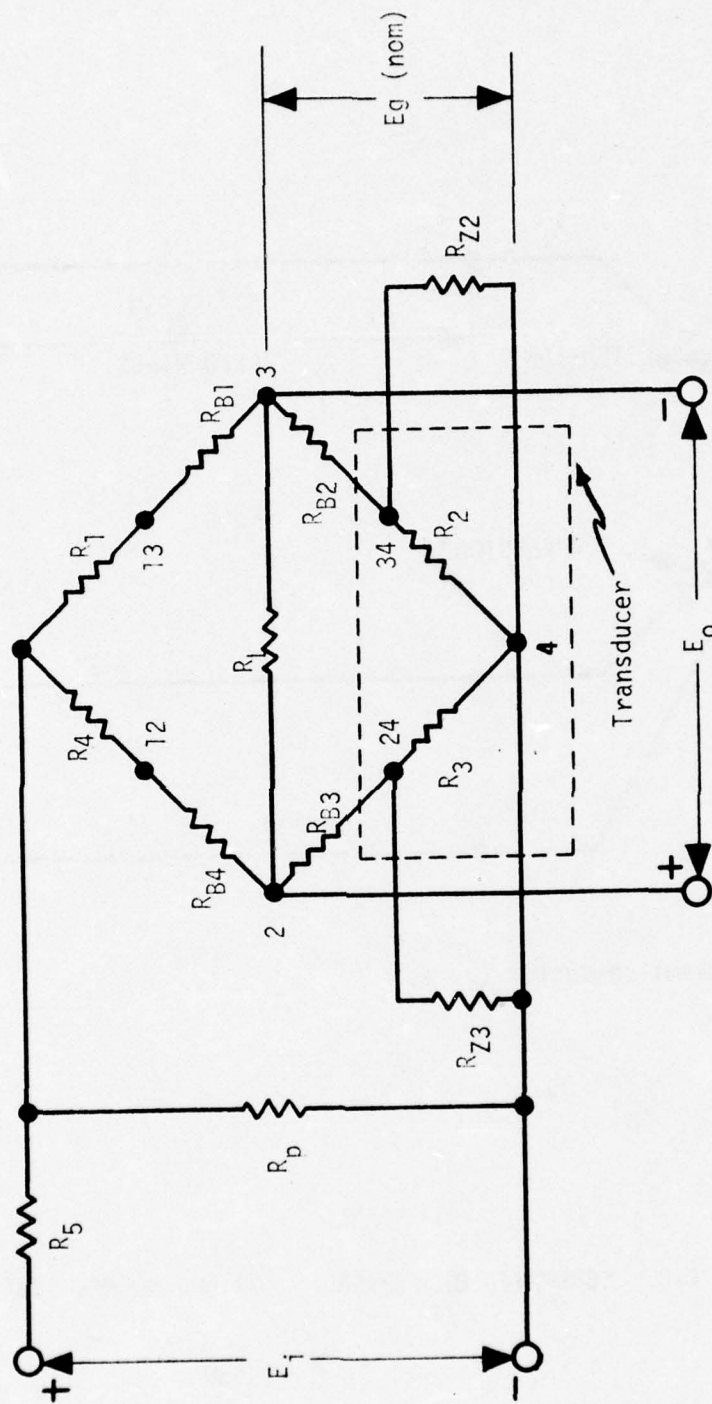


FIGURE F-4. SCHEMATIC OF A TYPICAL NORMAL STRESS GAGE

CONNECTED TO THE B.C.U.



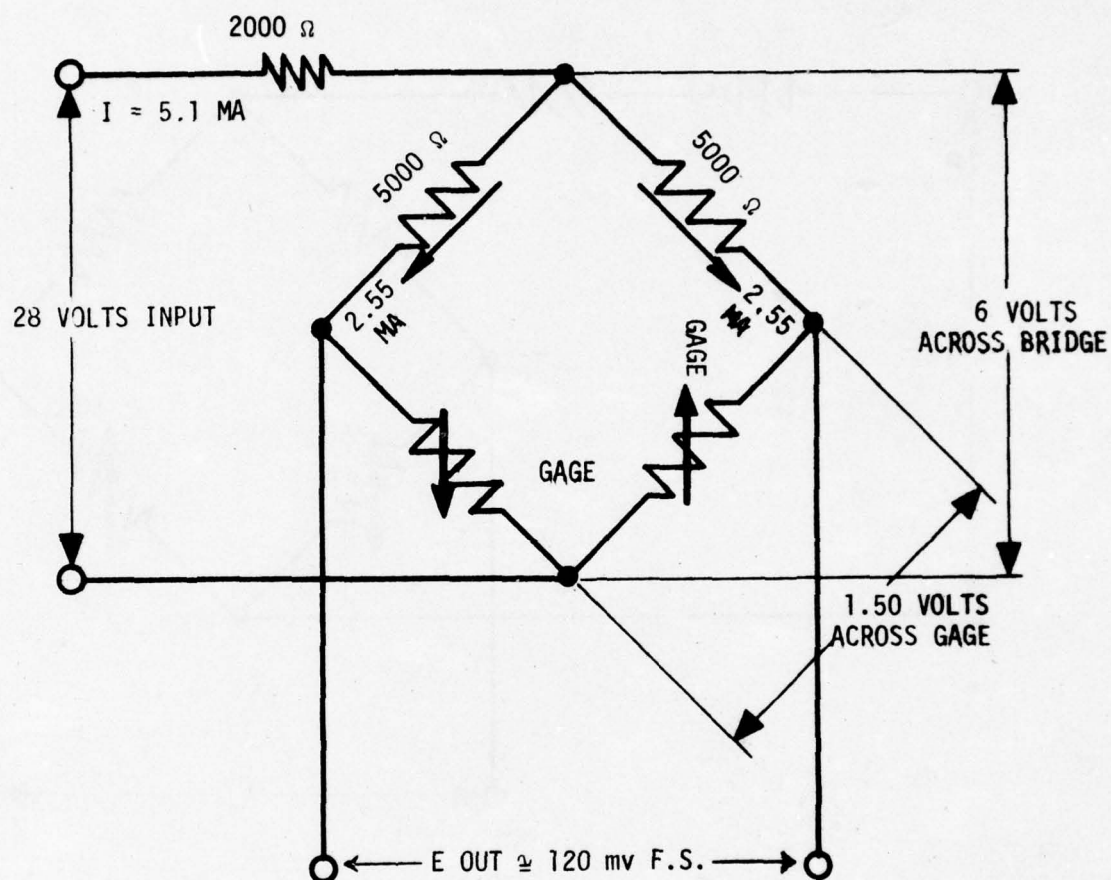


FIGURE F-5. CIRCUIT ANALYSIS OF A TYPICAL NORMAL STRESS GAGE

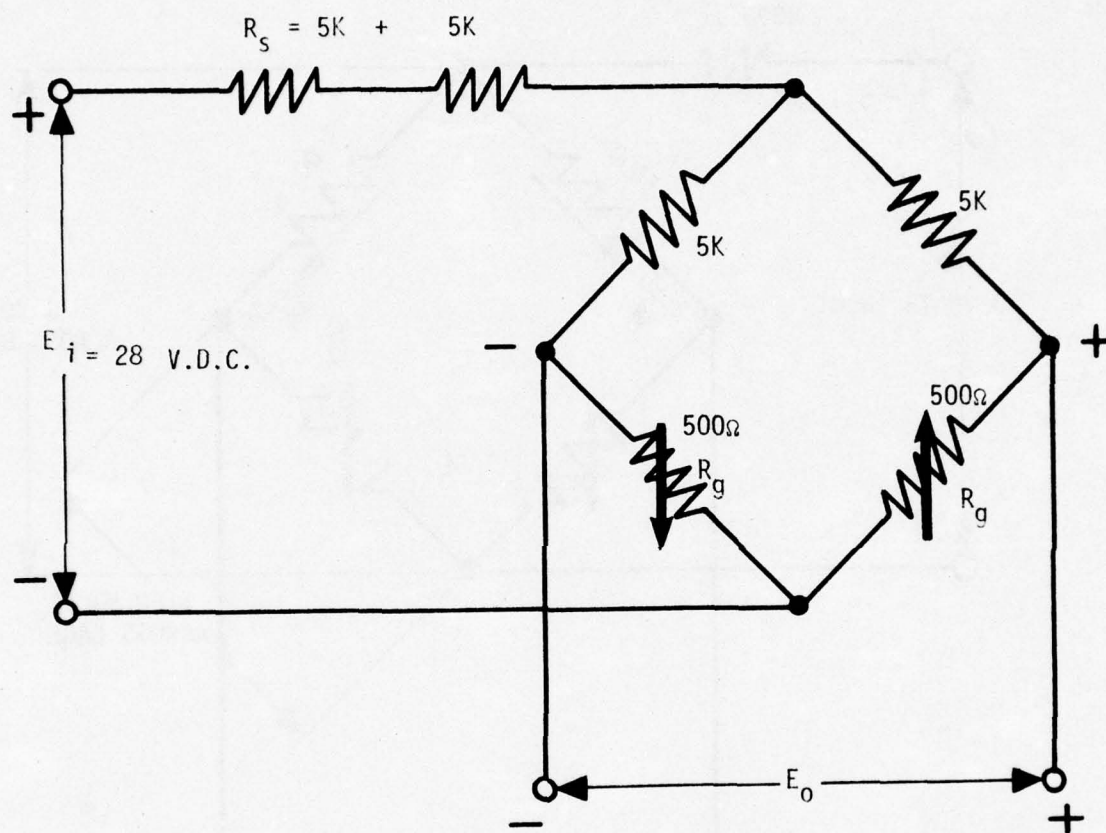


FIGURE F-6. SCHEMATIC OF A TYPICAL SHEAR GAGE  
CONNECTED TO THE B.C.U.

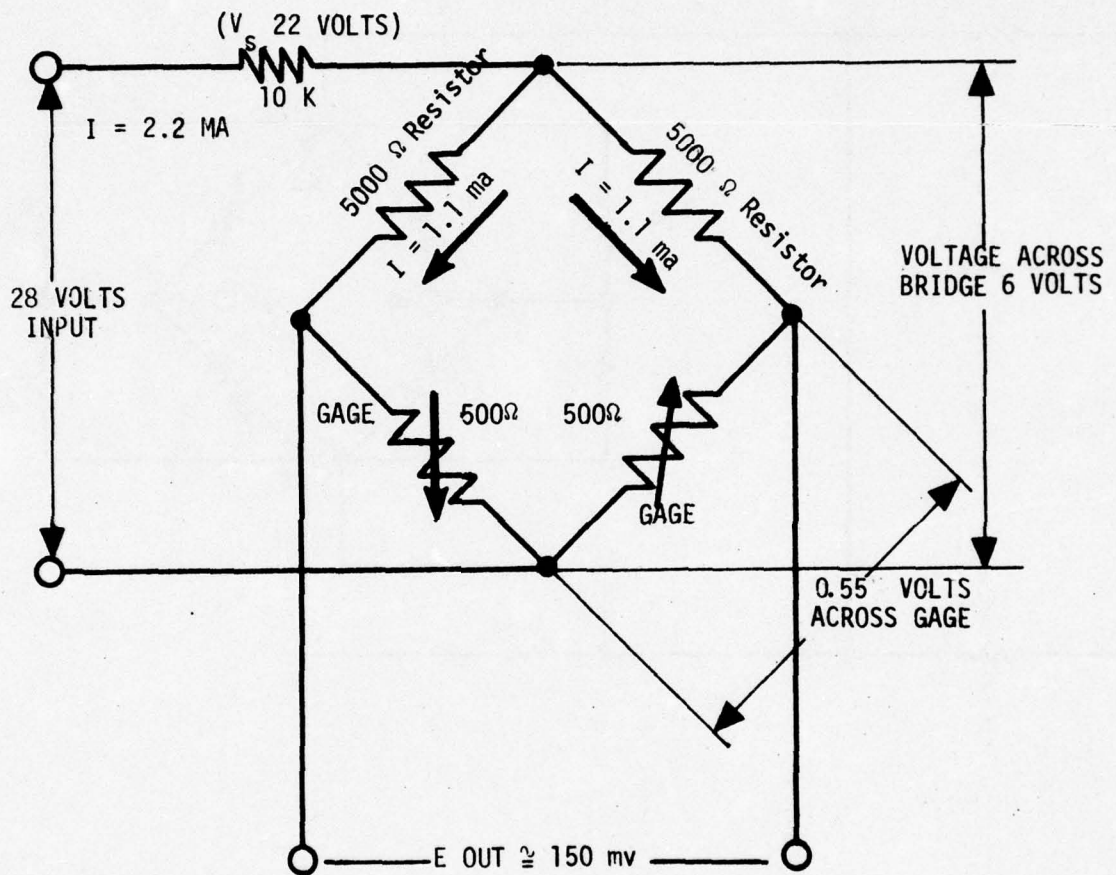


FIGURE F-7. CIRCUIT ANALYSIS OF A TYPICAL SHEAR GAGE



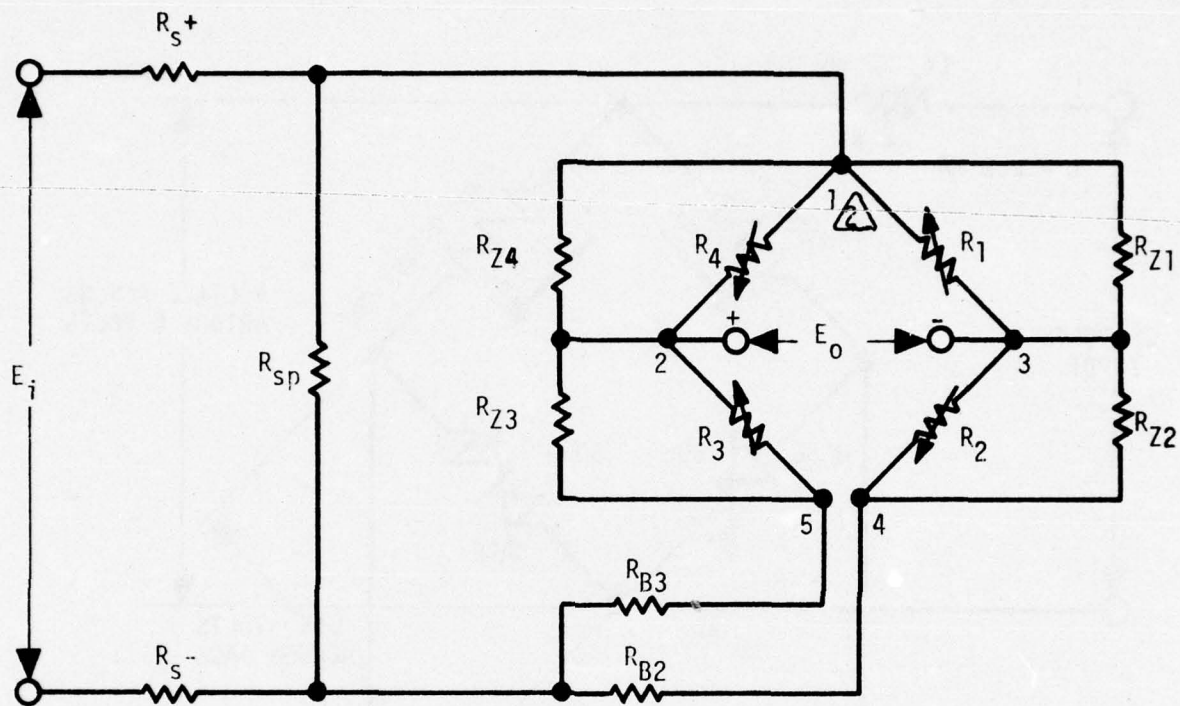


FIGURE F-8. SCHEMATIC OF KONIGSBERG INSTRUMENTS MODEL J1C CLIP GAGE

# 3D Gage Employs Three Gage Circuit Boards

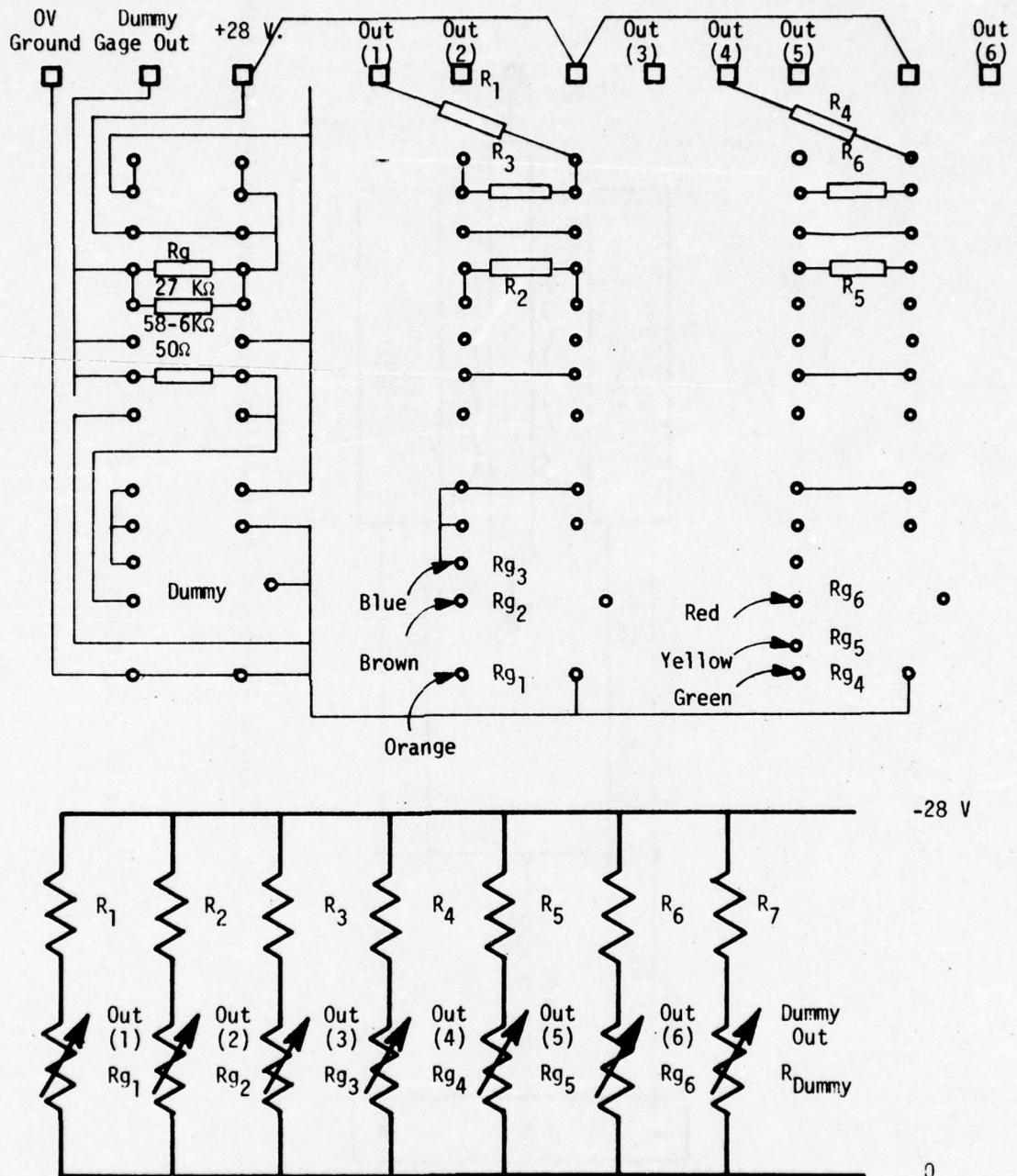
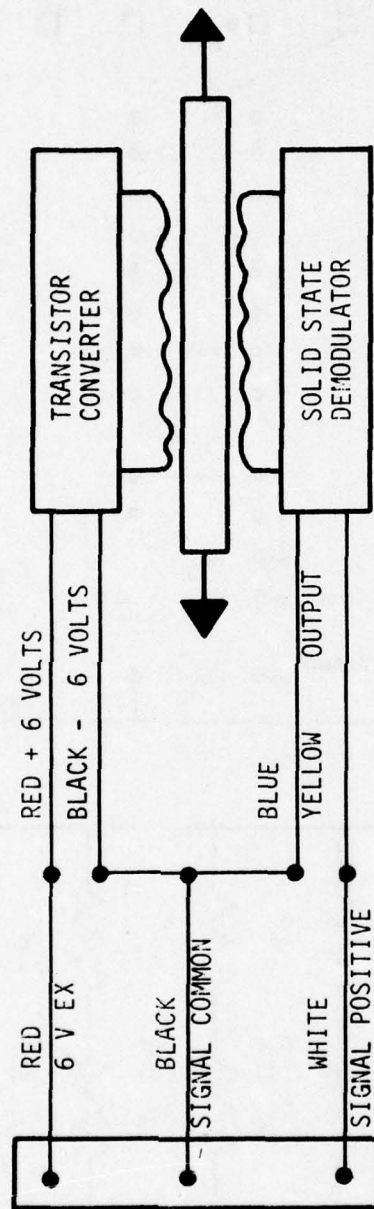


FIGURE F-9. SCHEMATIC OF THE 3D GAGE



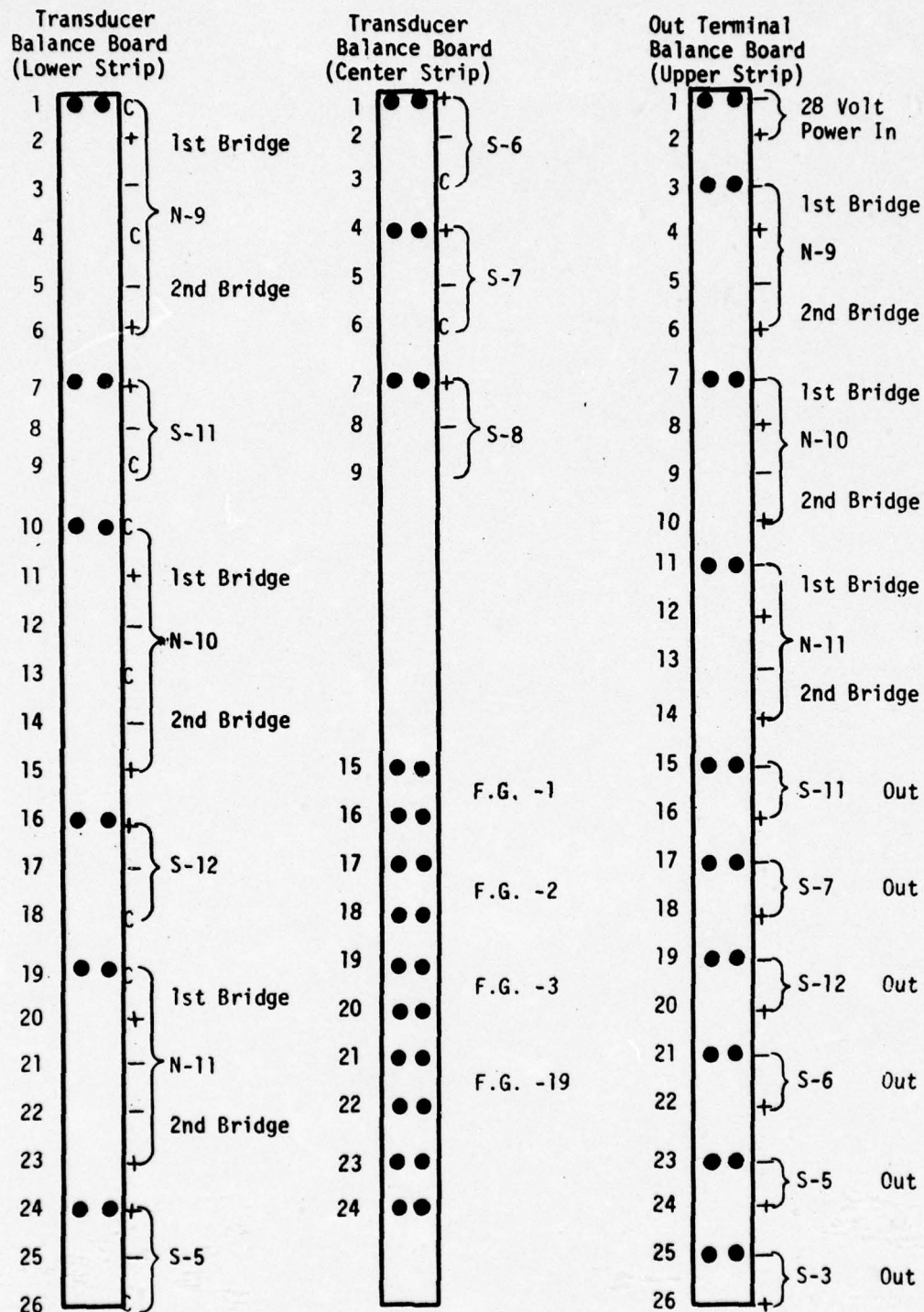
STANDARD PHASING: Yellow lead + when probe is inserted toward leads.

FIGURE F-10. ELECTRICAL CIRCUIT FOR AN LVDT



0° Terminal

Motor No. 1

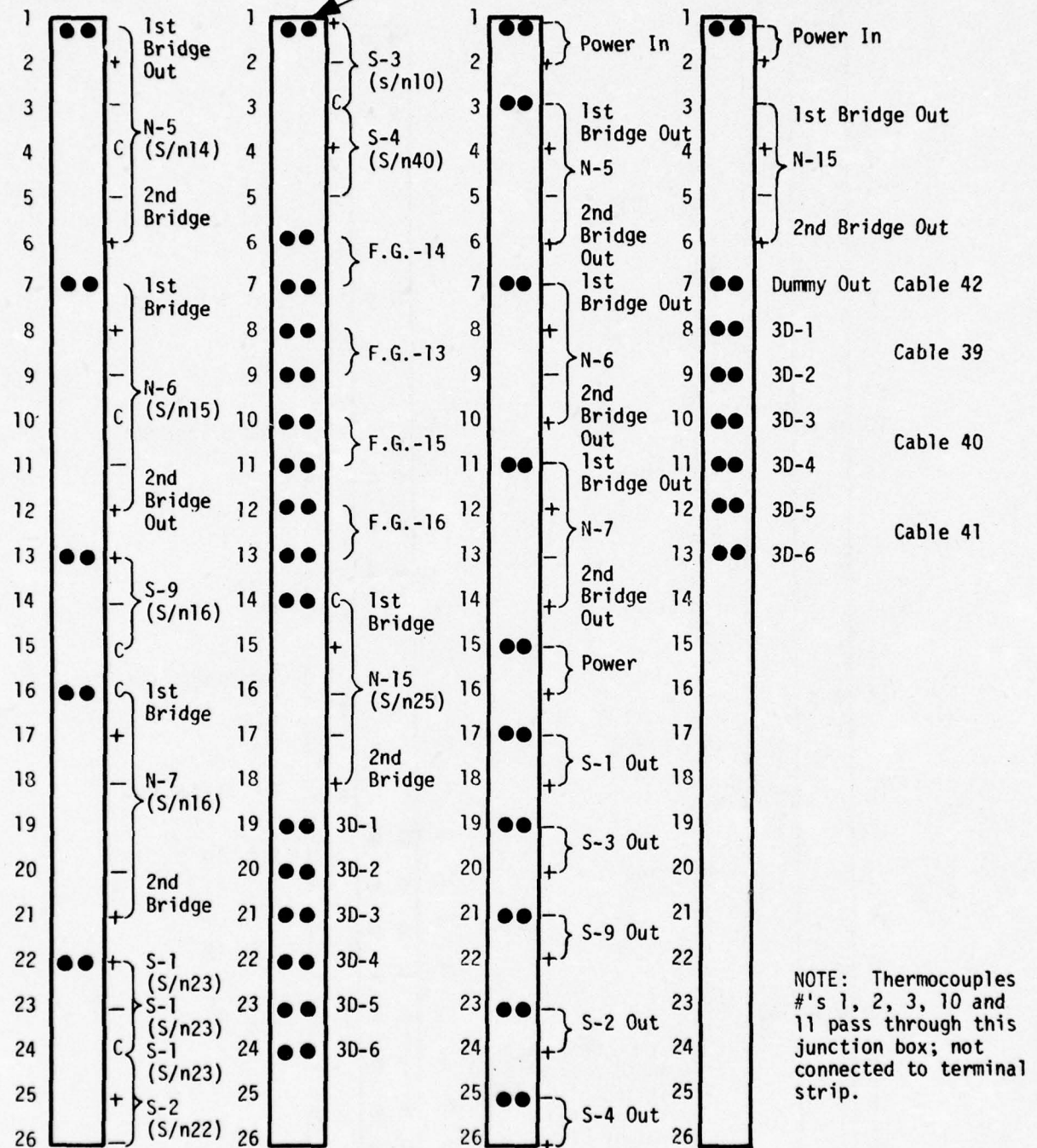


NOTE: Thermocouples Nos. 5, 6, 7, 7 & 18 pass through this junction box; not connected to terminal strip.

FIGURE F-11. TERMINAL IDENTIFICATION FOR 0° JUNCTION BOX

180° Terminal

Motor No. 1

Transducer  
Balance Board  
(Lower Strip)Transducer  
Balance Board

NOTE: Thermocouples #1's 1, 2, 3, 10 and 11 pass through this junction box; not connected to terminal strip.

FIGURE F-13. TERMINAL IDENTIFICATION FOR 180° JUNCTION BOX

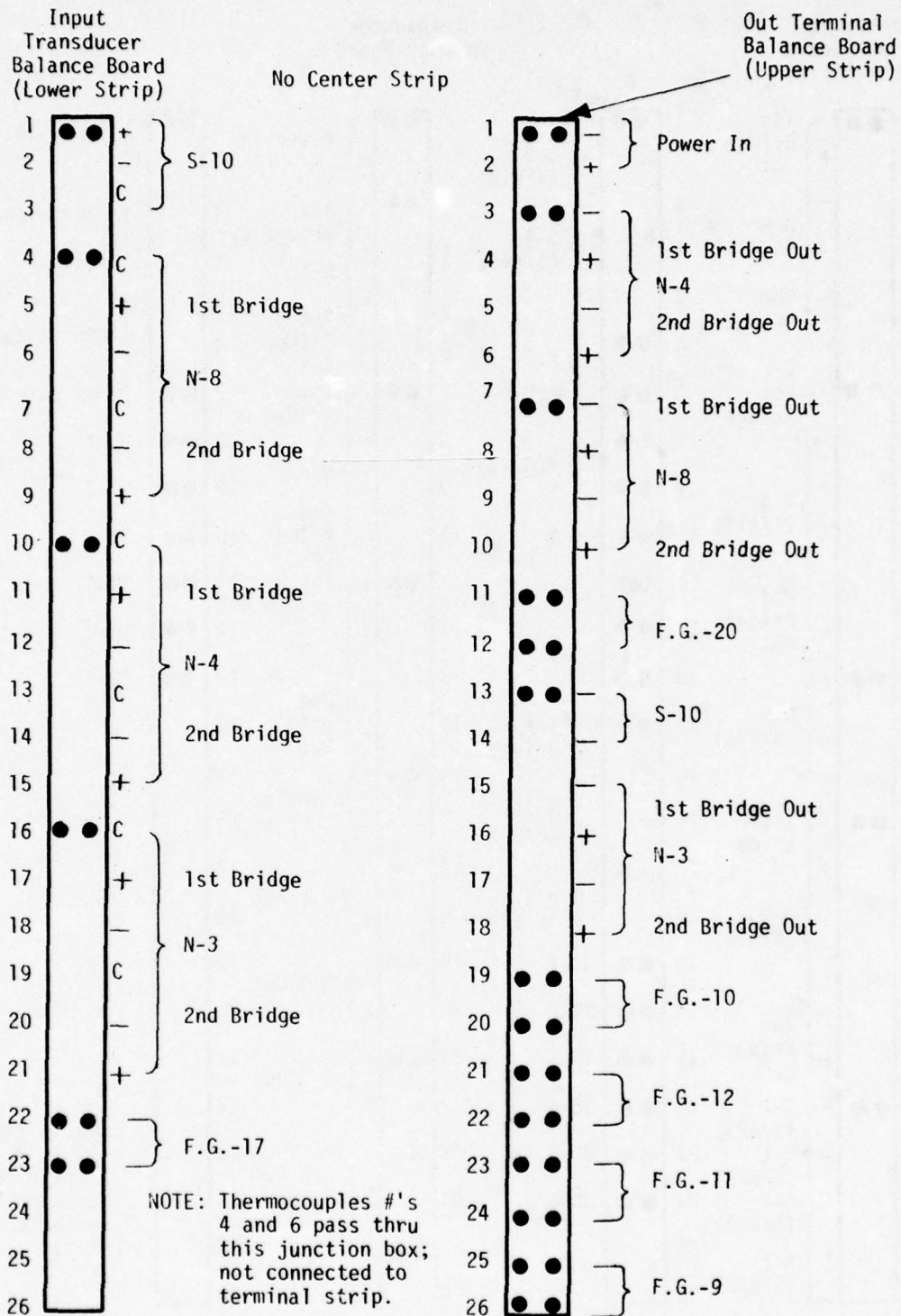


FIGURE F-14. TERMINAL IDENTIFICATION FOR 270° JUNCTION BOX



TABLE F-1

## ASSIGNMENT OF GAGES TO RESPECTIVE JUNCTION BOXES

	ASPC NORMAL GAGE NO.	ASPC SHEAR GAGE NO.
0° J-Box	N-9	S-5
	N-10	S-6
		S-7
	N-11	S-8
		S-11
		S-12
90° J-Box	N-1	S-13
	N-2	S-14
180° J-Box	N-5	S-1
	N-6	S-2
	N-7	S-4
	N-15	S-9
270° J-Box	N-3	S-10
	N-4	
	N-8	

APPENDIX G

DESIGN OF ORIGINAL PORTABLE DAS INSTALLATION

## DESIGN OF ORIGINAL PORTABLE DAS INSTALLATION

## A. DESIGN AND INSTALLATION OF THE DATA ACQUISITION SYSTEM

## 1. 50 Channel Scanner

The data acquisition system for the Flexible Case tests which are shown in Figures 39 and 40 of Volume I consisted of the HP2010 series systems which utilizes a 2401 Hewlett Packard Integrating Digital Voltmeter (S/N A9C 2500020) as the digitizing element, along with the 2901A Input Scanner. The 2010 series is characterized by exceptional common mode and superimposed noise rejection, selectable integration line, and built in programming capability. The Dymec 2901A Input Scanner/Programmer scans 25 channels of inputs and programs all functions of associated systems. It was expanded to 50 channels with an auxiliary system which includes two banks of switches. System functions and measurement delay are programmed individually for each channel with a built-in pinboard. The maximum scanning rate is 12 channels/second.

## 2. Integrating Digital Voltmeter

The analog to digital converter on the 2401 integrating digital voltmeter features floated and guarded input and is average reading, yielding an effective common mode rejection better than 140 DB at all frequencies, including DC. Since they are average reading and fully guarded, it greatly reduces superimposed noise errors and common mode noise errors.

## 3. Digital Recorder

An HP Model 562A Digital Recorder (AGC-247-00863), the recording system used, is a solid state electromechanical device which provides a printed record of digital data. Its accuracy is identical to the input device used. Printing rate is 5 lines/sec and it has column capacity of 11 digits. The system accuracy is rated at DC accuracy of .01% rdg or  $\pm$  .005% full scale.

## 4. Data Acquisition System

A flow diagram of the stress gage connected to the data acquisition system inside the trailer is shown in Figure G-1. This system scans multiple analog input signals, converts them to digital form and visually displays and permanently records the measurements. All instruments are rack-mounted as shown in Figure 39 of Volume I and include the following: Dymec Scanner and Printer System, integrating digital voltmeter, CEC oscillograph, Brown Multi-point Temperature Recorder, a patch and junction panel for parallel monitoring, and two banks of switches. The instrumentation layout when rack-mounted in the trailer is given in Figure G-2.

The input scanner and the integrating D.V.M. and the printer were periodically calibrated by PATO Calibration Laboratory using the voltage substitution calibration method.



## 5. Instrumentation Trailer

A large van-type utility trailer (Figure G-3), approximately 30 feet long by 8 feet wide, was used to house all instrumentation for monitoring and recording all tests in the various phases of the Flexible Case Program. Power to the trailer was provided by a 3 phase 440 volt step-down transformer. Power was then routed to a 30 amp power panel with circuit breaker. Output cables, which were attached to the junction boxes, were routed through an opening in the floor of the trailer to termination points. They were then patched to their respective channels.

## 6. Interconnecting Network

The interconnecting network for the D.A.S. consists of individual cables and connectors for each type of sensor. The wiring diagrams for the various cables and connectors are presented in the following figures:

Figure G-4	Event Gage Cable
Figure G-5	Strain Gage Single Leg Cable
Figure G-6	Bridge Excitation Cable
Figure G-7	Copper Constantan Thermocouple Cable
Figure G-8	Stress-Strain Cable for Full Bridge
Figure G-9	Typical Stress Gage Wiring From Panel Termination to D.V.M.

## B. OPERATIONAL PROCEDURES

The operating procedure for data sampling of the data acquisition system is as follows:

### 1. Digital Voltmeter Switches

Range	- .1V
Function	- Volt
Sample Period	- .1 sec
Power Switch	- ON
Sampling Rate	- Set at pen mark on face of panel

### 2. Input Scanner/Programmer

Selector Switches - All depressed but #5 - This deselects Channel #5 and #30 from scanning or printout. Press the reset switch which selects the scanner to home and then press the function switch labeled Single Scan.

#### Channel Selector Panel

All switches in the UP position to acquire Channels 1-25  
All switches in the DOWN position to acquire Channels 26-50

### 3. Harrison 28V power supplies in rack on right

P.S. #1 - ON  
P.S. #2 - ON  
Voltage should read 28V on meters - don't adjust unless reading is off by more than .1 V.

## 24 Point CC Temperature Recorder

Speed Switch - High Mode

### Printer

Power - ON

Record - ON when printout is desired

Data consists of approximately 50 permanently connected input channels plus 10 more that must be independently plugged into place of the first 10 inputs to acquire these extra channels. All channels have correlation identification in the layout book plus the extra ten channels have ID markers on the cables.

Record the 50 input channels using the selector switching panel, note the time and date plus the temperature from the 24 point recorder on the printout paper. Also, mark the time and date on the temp. recorder when it is started and stopped.

### C. D.A.S. SPECIFICATIONS

#### 1. General

The specifications for the total Data Acquisition System are listed in Table G-1.

2. Integrating Voltmeter - H.P. Model 2401C. Specifications for the integrating voltmeter used in the D.A.S. are listed as follows:

DC voltage measurements, noise rejection: Overall effective common mode rejection: 140 dB at all frequencies 160 dB at dc (0.1 second sample period); superimposed noise rejection; more than 20 dB at 55 Hz for 0.1 second sample period, increases 20 dB per decade increase in frequency, infinite rejection at frequencies evenly divisible by 10.

Input circuit: Type: Floated and guarded signal pair, may be operated up to 500 V above chassis ground; ranges: 5 from 0.1 to 1000 V f.s., selection by front-panel switch or remote circuit closure to ground, polarity sensed automatically; over-ranging: To 300% f.s. except 1000 V range; overload: Range automatically switched to 1000 V at 310% f.s., reset by next read command; input impedance: 10 M  $\Omega$  on 10, 100, 1000 V ranges, 1 M  $\Omega$  on 1 V range, 100 k  $\Omega$  on 0.1 V range, <150 pF on all ranges.

Absolute accuracy:  $\pm 0.01\%$  of reading  $\pm 0.005\%$  f.s.  $\pm 1$  digit at 25°C; temperature coefficient  $\pm 0.001\%$  of reading per °C, 10 to 40°C.

Internal calibration source:  $\pm V$  standard for self-calibration; maintains rated accuracy for 6 months after initial calibration to 0.002% at 25°C.

Measurement speed: Fixed sample periods of 0.01, 0.1 or 1 sec selected by front-panel switch or remote circuit closure to ground.

Resolution: Depends on sample period; max  $1\mu$  V per digit.

Autoranger (optional) voltage ranges: Automatically selects range from 5 input ranges of standard instrument (0.1 V to 1000 V f.s.) .34 ms max range change time.

DC voltage integration: Input signal is integrated over selected sample period; using fixed sample period, integral is average of input.

Frequency measurements: 5 Hz to 300 kHz, optionally to 1.2 MHz; gate time 0.01, 0.1, 1 sec or manual; accuracy:  $\pm 1$  count + time base accuracy; time base: Stability at constant temperature ( $\pm 5^\circ\text{C}$ ) is  $\pm 2/10^6$ /week, temperature effect  $\pm 100/10^6$  over range 10 to  $50^\circ\text{C}$ , provisions for external time base; display time: Variable from 0.2 to 7 sec, or held until reset; input sensitivity: 0.1 to 100 V rms; impedance: 1 M $\Omega$  shunted by 150 pF.

Period measurements (optional): 1, 10, and 100 periods; 5 Hz to 10 kHz; display is in ms; resolution referred to single period; 1 period, 100  $\mu\text{sec}$ ; 10 periods, 10  $\mu\text{s}$ ; 100 periods, 1  $\mu\text{sec}$ ; accuracy is  $\pm 1$  count + time base accuracy + trigger error divided by number of periods. Sensitivity and impedance same as frequency measurements.

### 3. Digital Recorder - H.P. Model 562A

Specifications for the digital recorder used in the D.A.S. are listed as follows:

Accuracy: Identical to input device used.

Printing rate: 5 lines per second, maximum.

Column capacity: To 11 columns (12 available on special order).

Print wheels: 12 positions, numerals 0 through 9, a minus sign and a blank; other symbols available.

Input Requirements:

Data input: Parallel entry, BCD (4-2-2-1, 8-4-2-1, 2-4-2-1) or 10-line, see Options; (1) state must differ from "0" state by at least 4 Volts but by no more than 75 Volts.

Reference voltages: BCD codes require both "0" and "1" state references; 10-line codes require reference voltage for "0" state; reference voltages may not exceed  $\pm 150$  V to chassis; input impedance is approximately 270 k ohms.



Hold-off signals: Both polarities are available simultaneously for BCD codes and are diode-coupled; 10 mA maximum load +15 V open circuit from 1 k source, -5 V open circuit from 2.2 k source (160 msec hold-off is provided for 10-line codes).

Print command: + or - pulse, 4.5 to 20 volts amplitude, 1 V/ $\mu$ s minimum rise time, 20  $\mu$ s or greater in width, ac coupled.

Analog output (optional): (from 4-2-2-1 or 8-4-2-1 boards) accuracy is + 0.5% of full scale or better; 100 mV for potentiometer recorder; 50 k ohm minimum load resistance; 1 mA into 1.5 k ohm maximum for galvanometer recorder.

Transfer time: 2 ms for BCD codes.

Paper required: HP, folded paper tape (15,000 prints per packet with single spacing) HP Stock No. 560A-131A or standard 3-inch roll tape.

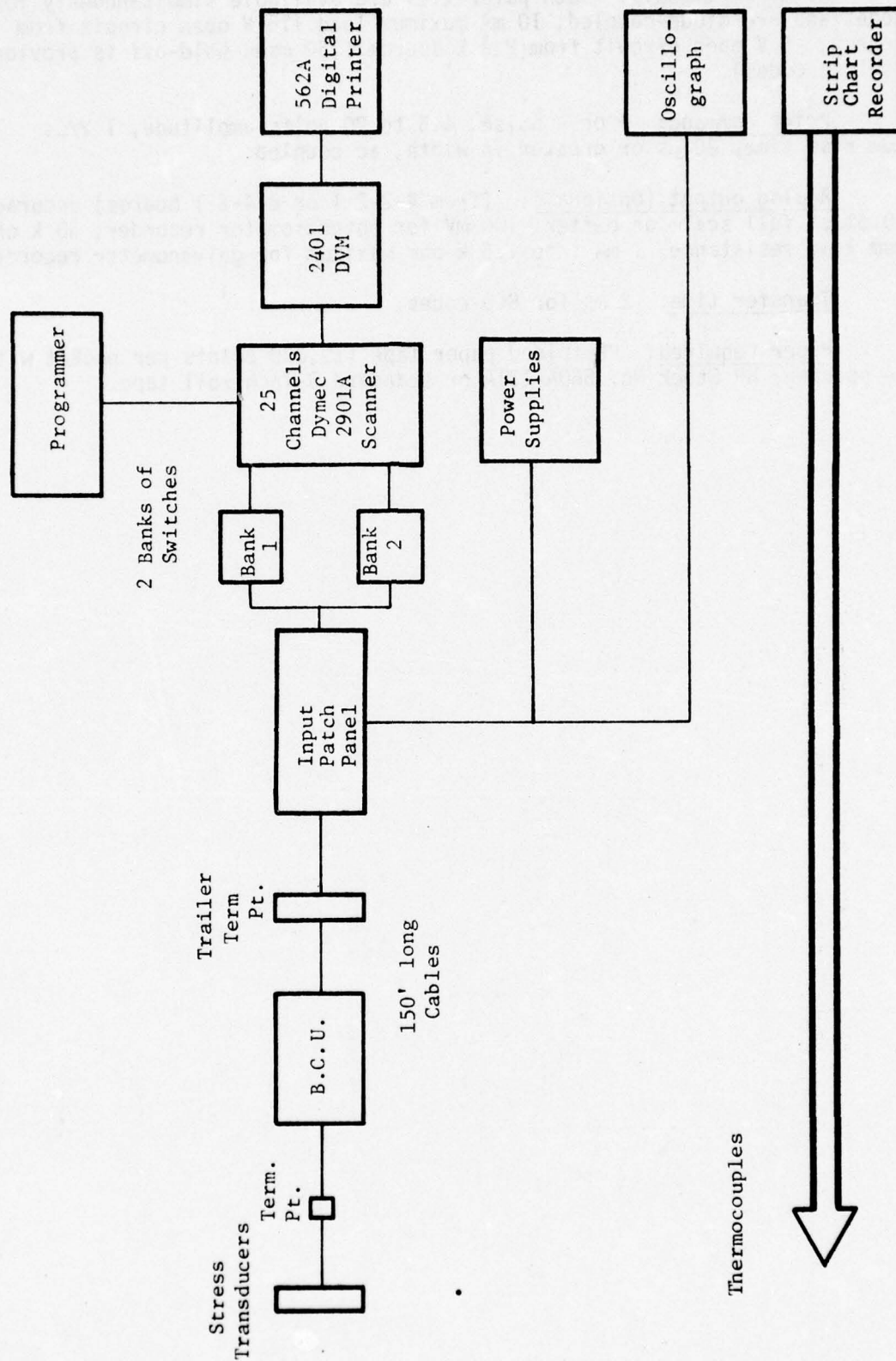


FIGURE G-1. BLOCK DESIGN OF PORTABLE D.A.S.

BAY 2	BAY 3	BAY 4	BAY 5	BAY 6	BAY 7
				Digital Voltmeter 2401 A	CC 24 Point Recorder
				Scanner 2901 A	Oscillograph #1
				Printer HP 5562A	Oscillograph #2
			Strip Chart	Switching 1-25 to 26-50	Parallel Strain Input 1-60
				Parallel LVDT	28V DC Supply
Oscillograph #1	Oscillograph #2				28V DC Supply
					28V DC Output
					Strain Gage Input 1 - 60
					Parallel Events
				LVDT DC Supply	Thermocouples 1-24
				LVDT 1-8	Events 1-20
				Blank	Blank

FIGURE G-2. INSTRUMENTATION LAYOUT WHEN RACK MOUNTED



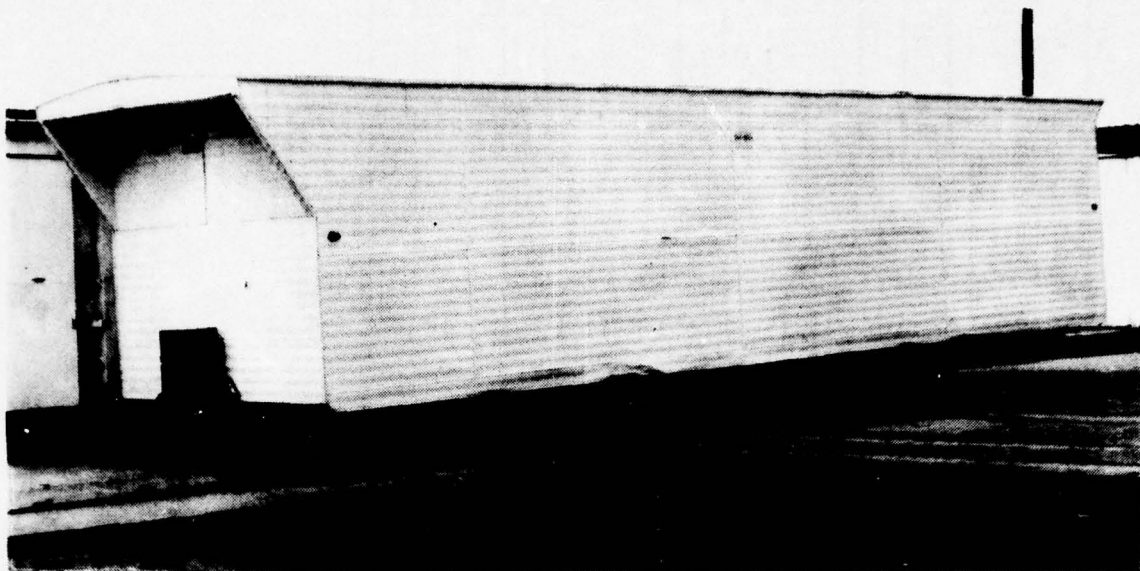


FIGURE G-3. INSTRUMENTATION TRAILER

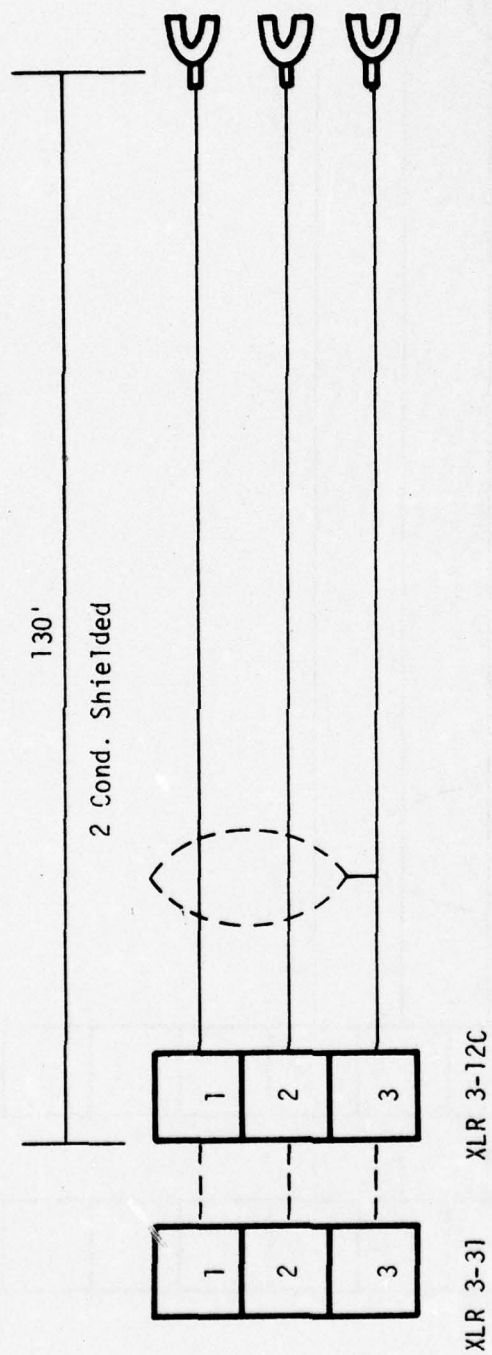
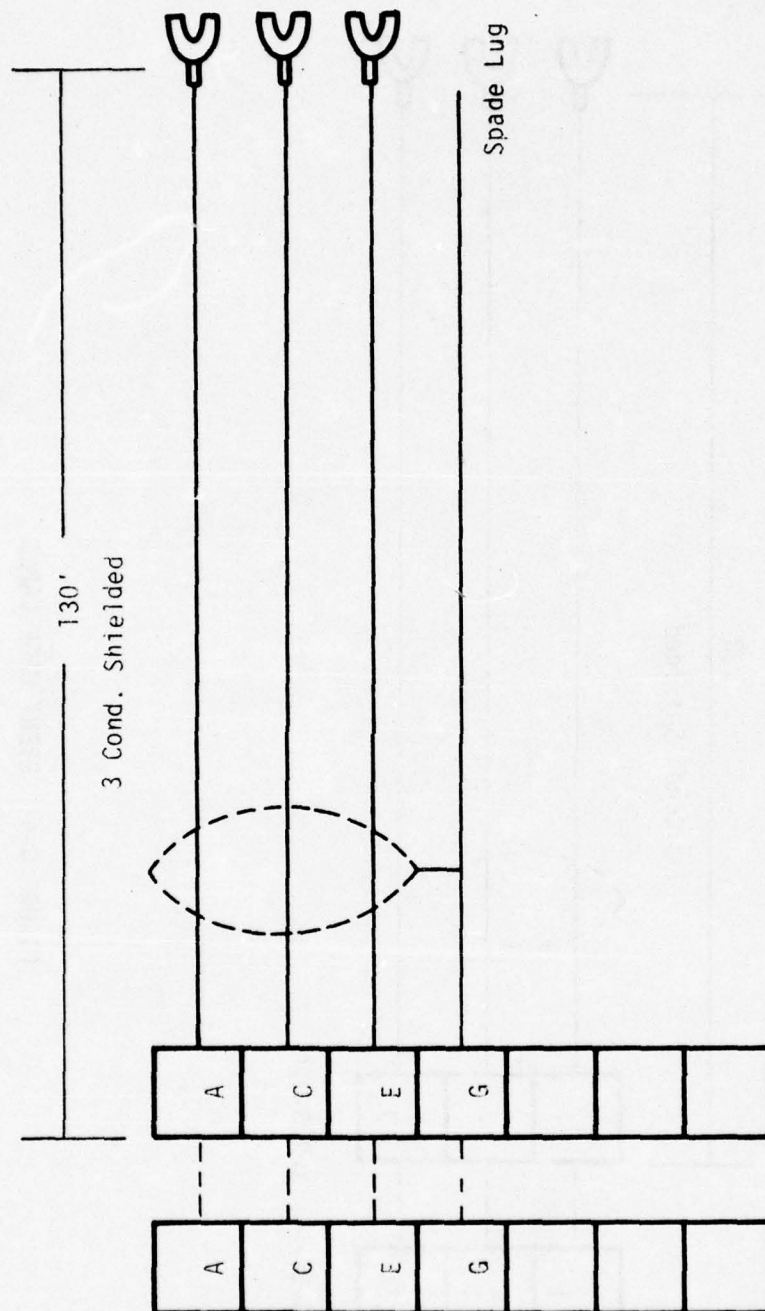


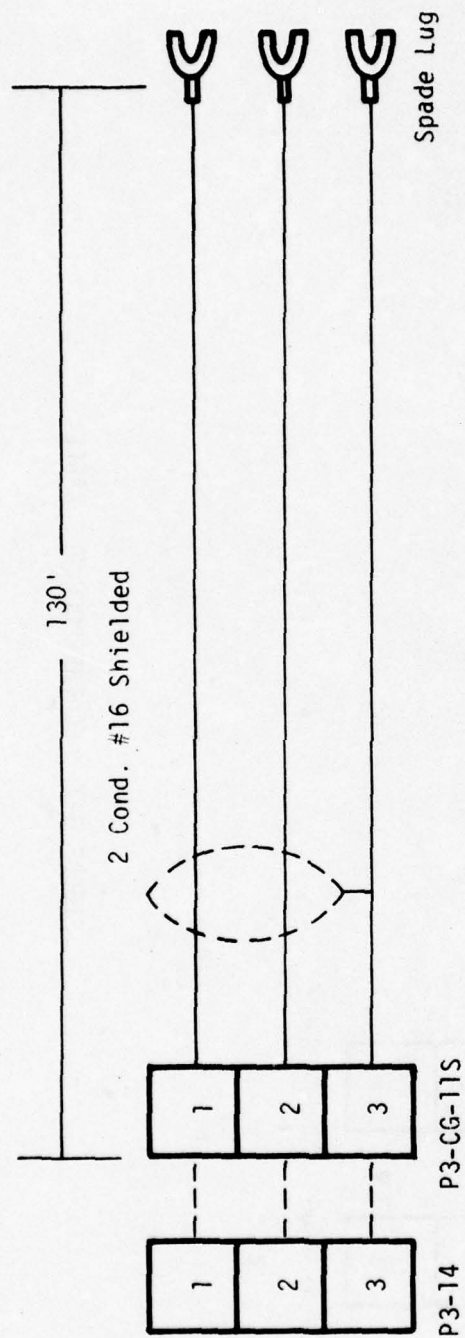
FIGURE G-4. EVENT GAGE CABLE



CA00R 16S-15 CA06R 16A-1P

FIGURE G-5. STRAIN GAGE - SINGLE LEG CABLE





G-13

FIGURE G-6. BRIDGE EXCITATION CABLE

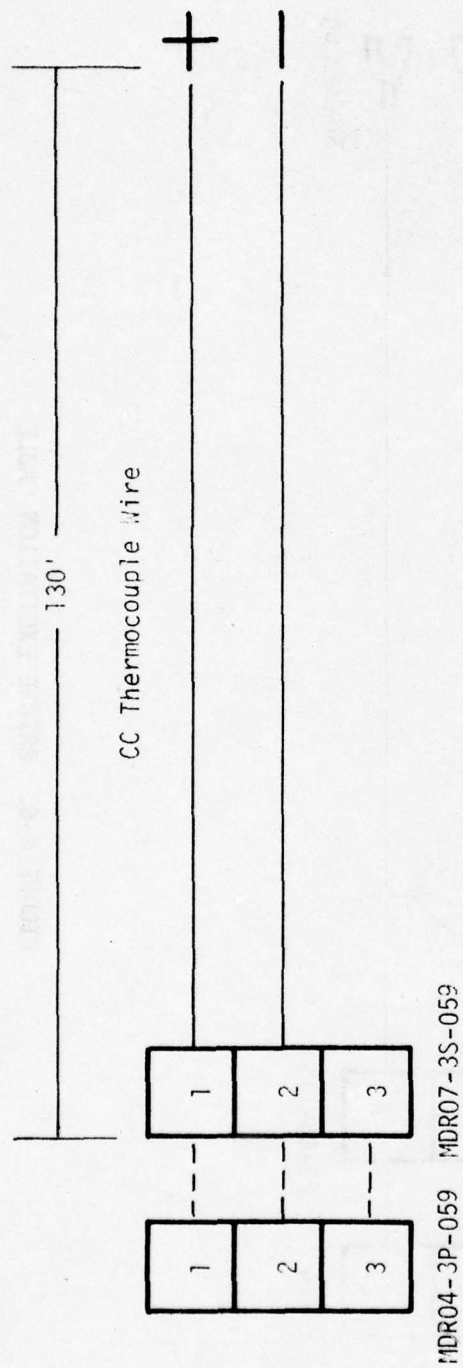


FIGURE G-7. CC THERMOCOUPLE CABLE

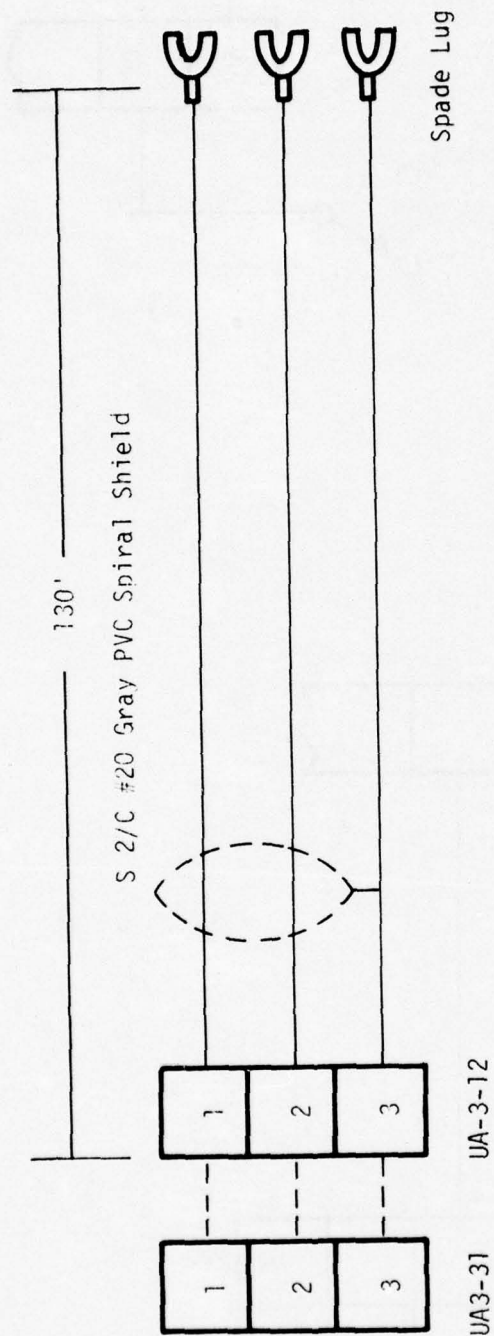


FIGURE G-8. CABLE FOR FULL BRIDGE



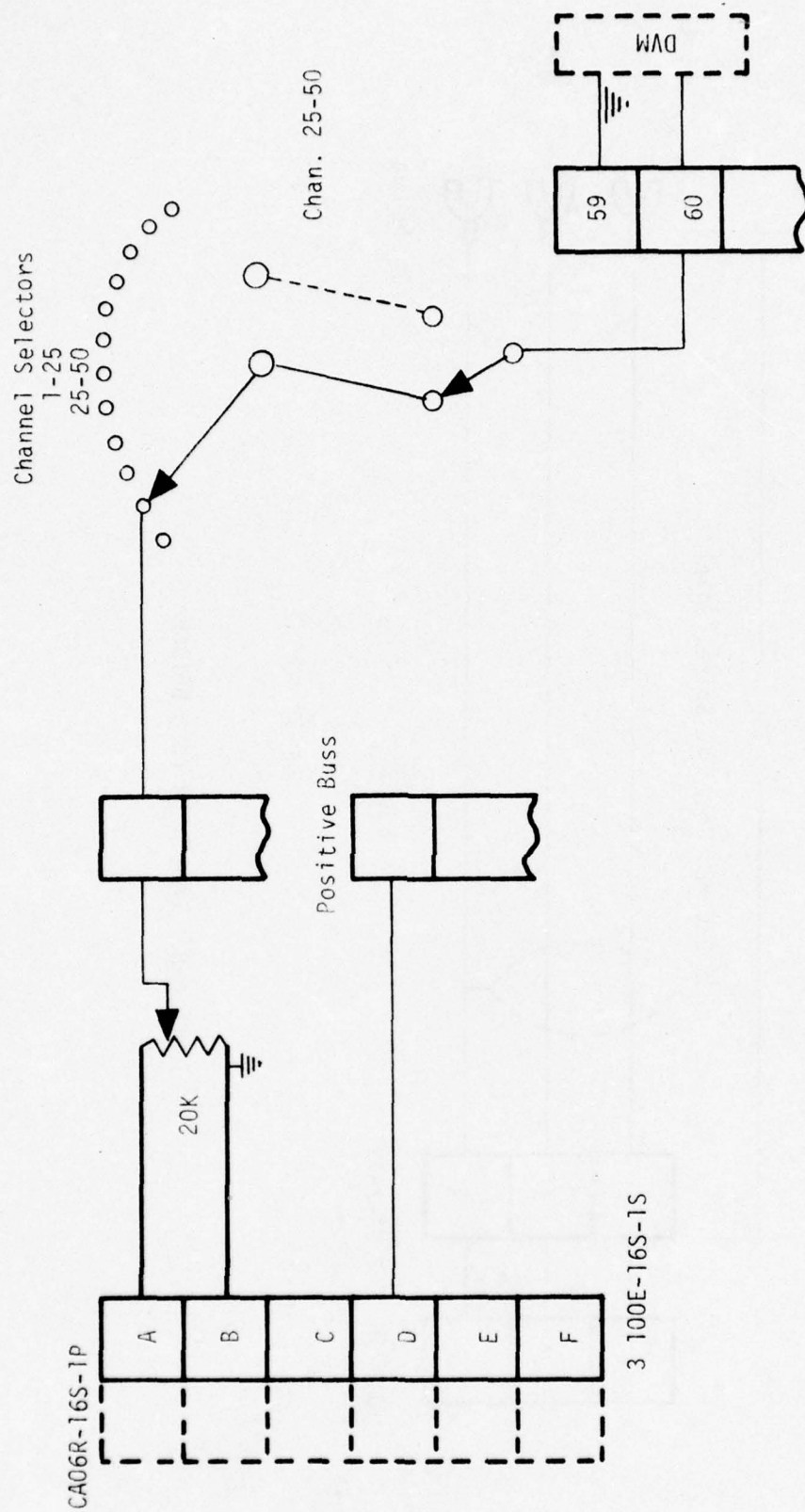


FIGURE G-9. TYPICAL STRESS GAGE WIRING FROM PANEL TERMINATION TO DVM

TABLE G-1

## GENERAL SPECS FOR D.A.S.

Number of input channels	Up to 25 3-wire inputs		
Programming	Built-in; all DVM functions, delays		
DC measurement	100 mV to 1000V, 5 ranges, $\pm 750$ V max., resolution determined by gate time		
AC measurement	100 mV to 1000V, 5 ranges, 750V max., resolution determined by gate time		
Ohms measurement	100 $\Omega$ to 10M $\Omega$ , 6 ranges, .001 resolution on 100 $\Omega$ range.		
Frequency measurement	5 Hz to 300 kHz, resolution determined by gate time.		
Period measurement	5 Hz to 10 kHz, resolution determined by gate time.		
DC accuracy	100 mV: .01% rdg $\pm$ .01% fs; other ranges: .01% rdg $\pm$ .005 % fs		
Measurement speed (max. dc volts)	6 chan/sec Digital Printer	9 chan/sec Punched Tape	54 chan/min Typewriter

APPENDIX H

SUMMARY OF TEST DATA FROM THE ORIGINAL PROGRAM



## SUMMARY OF TEST DATA FROM THE ORIGINAL PROGRAM

## A. SUMMARY OF GAGE DATA OBTAINED FROM MOTOR NO. 1 IN THE ORIGINAL TEST PROGRAM

## 1. Introduction

A complete log of the normal stress gage output data for Motor No. 1 beginning with the initial laboratory calibrations through the end of the thermal cycling tests, was prepared and is discussed in Section 3. This log represents the output from 24 independent bridges, since each of the 12 normal stress gages has two. In addition to the data log presentation, the following major tests and the resulting normal stress and shear gage data are discussed in detail as follows:

- Section 4 - Pressure Calibration of Motor No. 1
- Section 5 - Cure and Cooldown Data
- Section 6 - Aging of Propellant Grain at 110°F
- Section 7 - Thermal Cycle

## 2. Description and Location of Stress Gages

The 12 normal stress gages and the 14 shear stress gages were located in Motor No. 1 as shown in Figures 22 to 27 of Volume I. N<sub>1</sub> through N<sub>4</sub> were the 450 psi normal gages located on the aft dome between the equator and the nozzle boss. N<sub>5</sub> through N<sub>11</sub> were the 150 psi normal gages placed at selected locations in the barrel section of the motor. D2N is the normal gage for the 3-D gage located near the forward equator. Shear gages S-1 through S-8 were located on the aft dome between the equator and the nozzle boss at the 0 to 180° plane. Shear gage S-13 is located near the forward equator and shear gages S-9, 10, 11, 12 and 14 were located on the barrel section. Shear gages S-12 and S-14 were rotated 90° to sense dynamic response.

## 3. Output Data Review

Output data from all normal stress gages measured in Motor No. 1 are tabulated in Table H-1 and plotted in Figure H-1 and Figures 43 and 44 of Volume I. The data began with the potted gage calibration made at Konigsberg Instruments in the form of zero load output vs temperature. Then all the gages were calibrated in the tensile mode at ambient temperature by means of a vacuum chamber. It was at this point that noticeable zero load output changes occurred. Following the vacuum calibrations, the potted gages were installed in the empty chamber. The first test performed on the empty chamber after installation of the gages was a 15 psig pressurization for checkout of all the gage circuits and to check the normal gage responses.

In an attempt to obtain "zero load" output data at approximately 30°F for subsequent use in translating the loaded motor data to stresses, local areas of the chamber were cooled with dry ice. The dry ice was placed inside the chamber, and the chamber was rotated to bring it in proximity to the area to be cooled. The gage temperatures were taken as indicated by the nearest internal thermocouple.

The chamber was then lined and processed through the manufacturing operations. It was finally lowered into the casting bell and the instrumentation was connected. Large offsets in the zero load readings were recorded.

The propellant was cast and the cure pressure of 15 psig was applied. All the gages were sampled periodically during the 12 day propellant cure period and during the subsequent four days cooldown period. After core extraction two pressure calibrations to 50 psig were conducted as shown in Figure H-1 and in Figures 43 and 44 of Volume I. Then data for the four months aging at 110°F and the thermal cycling between 60 and 110°F were obtained as shown in Figure H-1 and Figures 43 and 44 of Volume I.

#### 4. Pressure Calibration for Full Scale Motor No. 1

##### a. Test Results

The motor was pressurized in accordance with the Test Plan with gaseous nitrogen in 10 psig steps to 50 psig at 80°F to verify that all instrumentation was functional and that polarity was correct. In addition reference points were provided for future verification of calibration and data reduction and to establish the basis for an in-situ gage calibration from which gage sensitivity factors could be calculated. In the first pressure calibration run the sensitivity of the D.A.S. was not set to resolve the output voltages to the nearest 0.01 mv, so those data are not presented here.

A second pressure calibration run was conducted. Test data from the second run are tabulated in Table H-2 and the resulting gage sensitivities of the normal stress transducers are given in Table H-3. The results from this second run are shown graphically in Figures H-2 through H-10. Some of the outputs from the shear gages (S-1, S-2, S-7, S-8, S-14) are not linear functions of pressure. This may be attributed to the fact that the smaller shear stress levels may be more related to case deformation than to applied pressure. When the gage output signals were converted to normal and shear stresses (Figures H-11 and H-12), two of the normal stress gages N2 and N3 located at the aft end of the dome respond to only a portion of the applied pressure. Similarly, this may be partly due to local case deformation. The shear stress data from Figure H-12 ranges from a maximum at gage S-13 to a minimum at gage locations S-2, S-4, and S-14. It should be noted that these data are simple conversions of the gage output into stress and do not take into account effects due to grain attenuation, gravity, or sensitivity of gage due to normal pressure.

The post cure pressure calibration test measurements from the L.V.D.T.'s located in the bore surfaces are shown in Table H-4. A graphic illustration shows the bore deflection, Figure H-13.

b. Reanalysis of Shear Gage Data

Two different techniques for interpreting the data obtained from the shear gages during pressure tests on the full scale motor were evaluated. Because the response of the shear gages is influenced by hydrostatic pressure and the normal stress at the gage location as well as by the shear stress at that point, the best approach is to employ an equation which sums these separate effects and enables the shear component to be determined. The difficulty with this approach is that the polarity of the response of the shear sensor to the separate components must be determined prior to casting the grain otherwise it will not be known if the pressure and shear effects add or subtract. Since the polarities of the various components were not determined for the full-scale motor, this approach could not be employed.

The second approach to interpreting the shear gage data made use of the shear gage response to hydrostatic pressure measured before the grain was cast. These data were obtained with the gages mounted inside the motor case and connected to the output circuits in the same manner as for the later pressure tests on the motor after the grain was cast. During the pressure tests on the shear gages inside the motor without the grain, the gage responds to the pressure and pressure normal stress in a realistic fashion but without the additional stresses induced by the propellant grain. Consequently the response of the shear gages to pressure without the grain can be subtracted from the gage response with the grain to provide the shear stress values caused by the grain.

When this technique for shear gage data analysis is employed there should be no polarity problems. However, because the wiring was disconnected from the motor after the pressure calibration tests and subsequently re-connected for the later pressure tests of the complete motor, the problem of whether or not the pressure and shear effects add or subtract still remains. Because of this problem, having determined the sensitivities of the shear gages to pressure (see Figures D-5 and D-6) the effects of adding and subtracting this pressure term were calculated and are plotted in Figures H-14 through H-16.



The data from gages S-13 and S-9 show little difference if the pressure output is added or subtracted; but S-12 shows a marked effect. Because gage S-12 is mounted at 90° to the axis of the grain, it is likely that the smaller of the two curves is the correct one for gage S-12.

When considering the data shown in Figures H-15 and H-16, it should be remembered that gages S-1 and S-5 are at similar locations in the motor, as are gages S-2 and S-6, gages S-3 and S-7 and gages S-4 and S-8. We may, therefore use similarity in the gage outputs to define the correct curves for these gages. From this approach we may deduce that for gages S-1 and S-5, the maximum data curves are the correct ones; whereas for gages S-2 and S-6, the two curves labeled minimum are the correct ones and give almost identical data. Similarly, for gages S-3 and S-7, the maximum value curves appear correct, and for gages S-4 and S-8, the minimum curves are the correct curves.

#### 5. Cure and Cooldown Data

The gage data from the calibration tests were committed to magnetic cards for easy insertion into the memory of the HP 9810A data analysis system. To provide a better fit to the zero stress calibration data curves a second order (parabolic) equation of the type;

$$S_0 = a + bT + cT^2 \quad (H-1)$$

was fitted to each set of data and the coefficients a, b and c, for each gage were stored in the memory. Once the temperature is known, the zero load gage output reading is readily calculated and subtracted from the current gage reading in order to determine the stress or strain value.

To calculate the stresses, the operator keys in the gage number, output signal and temperature and the stress value is then printed out and simultaneously stored in the memory for future use in stress-temperature of stress-time plots.

Typical plots of the stresses calculated from the gage outputs as a function of time in days are presented in Figures H-17 through H-20 where data from shear gages S-1, S-2, S-10 and S-13 are shown, and in Figures H-21 through H-24 where the data obtained from gages N2-1, N4-2, N9-1 and N10-1 are shown.

With the single exception of shear gage S-13, which showed the highest shear stresses on thermal cooldown, the shear gages all showed very low thermally induced shear stresses of the order of 1 to 3 psi maximum. Gage S-13 located at the end of the forward boot showed a stress of 10 psi at the end of cure and cooldown. (This same gage showed the highest pressure induced shear stresses.)

Figures H-21 and H-22 show the data from 450 psi gages, which were not really intended for thermal stress measurement. However, the data appear reasonable and the correlation from one half bridge circuit to its twin output is a simple means of checking the data.

Figures H-23 and H-24 show data from 150 psi gages and these data appear to be very good. Comparison between the data from one half bridge to the other half bridge circuit is usually very good and this is particularly so in the case of gages N10-1 and N10-2, where the data are almost identical.

#### 6. Aging of Propellant Grain at 110°F

Following instrument calibration, Motor No. 1 was stored for four months at 110°F. Including the one month consumed during the preceding tests at 80°F, the motor was more than five months old at the end of the storage time. The objectives of this storage period at 110°F were to minimize the effect of post-cure hardening reactions on the stress-free temperature of the ANB-3066 propellant and to ensure that the stress-free temperature was stabilized at 110°F to provide a reliable reference condition. The aging test was conducted in conformance with AGC test plan 1826-26-TP.

The test results obtained during the lengthy post-cure of Motor No. 1 at 110°F were analyzed in Figures H-25 through H-29 and are shown as follows:

Test data for 1st week of aging test is shown in Table H-5; data for the 2nd to 6th week of aging in Table H-6, and data for the 7th to 11th week in Table H-7.

It was very difficult to detect a consistent trend from the various gage outputs. Some of the gages appear to have developed a significant thermal stress during the time spent at 70° prior to post-cure, and the subsequent heating of 110° did not reduce or eliminate this stress, this is particularly true of the normal stress gages.

However, there was little doubt that there was very little change in the stress field throughout the motor during the aging time; all the aging data being consistent and showing very small changes with time.

## 7. Thermal Cycling of Motor No. 1

The thermal cycle test for Motor No. 1 was conducted in conformance with the test plan and commenced on July 19, 1973. These measurements were intended to determine the extent of post cure hardening and the effective stress-free temperature of the grain. The motor was conditioned to five different temperatures for ten days each over the range from 60 to 110°F. A test period of at least 50 days was required.

### a. Description of Test

The motor was kept in an environmental bay at Building 4637 in the Test Area for the temperature cycling. All the instrumentation cables exited from the bay to the portable DAS in the instrumentation trailer.

### b. Test Sequence

The motor was initially stored for 12 days at  $80 \pm 5^\circ\text{F}$ . The thermal conditioning timetable was as follows:

7/19 to 7/31	12 days at $80 \pm 5^\circ\text{F}$
8/1 to 8/9	9 days at $60 \pm 5^\circ\text{F}$
8/10 to 8/24	15 days at $110 \pm 5^\circ\text{F}$
8/25 to 9/5	12 days at $80 \pm 5^\circ\text{F}$
9/6 to 9/17	12 days at $60 \pm 5^\circ\text{F}$
*9/18 to 9/20	3 days at $110 \pm 5^\circ\text{F}$

### c. Test Procedures

The test procedures used were as follows:

- (1) Store the motor for 10 days at  $80 \pm 5^\circ\text{F}$
- (2) Monitor the motor instrumentation.
- (3) Condition the motor for 10 days at  $60 \pm 5^\circ\text{F}$  and monitor all instrumentation.
- (4) Repeat the above for the conditioning temperatures of  $110 \pm 5^\circ\text{F}$ ,  $80 \pm 5^\circ\text{F}$ , and  $60 \pm 5^\circ\text{F}$ .

### d. Test Results

The thermal test data are shown in Table H-8.

Figures H-30 through H-47 present most of the thermal cycling test data obtained from the gages within full scale motor #1. Figures H-30 through H-37 show the normal stress data while Figures H-38 through H-47 show the shear gage data.

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\* Additional test cycle.



Apart from the fact that almost all the data suggest the presence of a considerable amount of stress free temperature change coupled with stress ratcheting, the data are poor because the two halves of the same normal stress gage, i.e., the two dual bridge circuits on a single diaphragm show considerably different results. This fact reveals that the problem belongs to the gage because, clearly both bridge circuits must see the same stress so that they should give the same stress values. Examination of Figures H-30 through H-37 reveals that the two stress readings initially are reasonably close to each other but show increasing differences with aging. Some gages, notably gages N7-1 and N7-2 (Figure H-36), show a consistent trend towards a bigger difference in the two sets of data from the end of cure through aging at 110°F to the temperature cycling data. Other gages such as N1-1 and N1-2 show a large difference in their stress values at the end of cure and cooldown.

An examination of the shear gage data shows less indication of problems. While some of the gages clearly have incorrect zero stress values, there does not seem to be the tendency towards a continual drift in readings with aging time. Of course there are no dual shear gage circuits in the motor so that these types of change are much more difficult to detect.

The basic problem with the data seems to be a shift in the zero stress output of the gage with aging time. This may be demonstrated by plotting the gage output for the pairs of dual gages directly in millivolts and using an arbitrary vertical shift in the readings to obtain reasonable agreement of the two sets of data. The resulting data are presented in Figures H-48 through H-53 and it will be noted that, whereas there is a great difference between the gage outputs when converted into stress using the stress free gage readings from the pre-casting calibrations, the changes in gage readings in millivolts are very much closer. In fact the agreement is very good. These data could in fact be converted into stress readings by using an arbitrary zero stress value, at 110°F for example, and using the gage sensitivities to convert the changes in output into changes in stress. This approach supposes that there is now an essentially constant zero stress gage output across the temperature range of interest, which seems to agree with the experimental gage data.

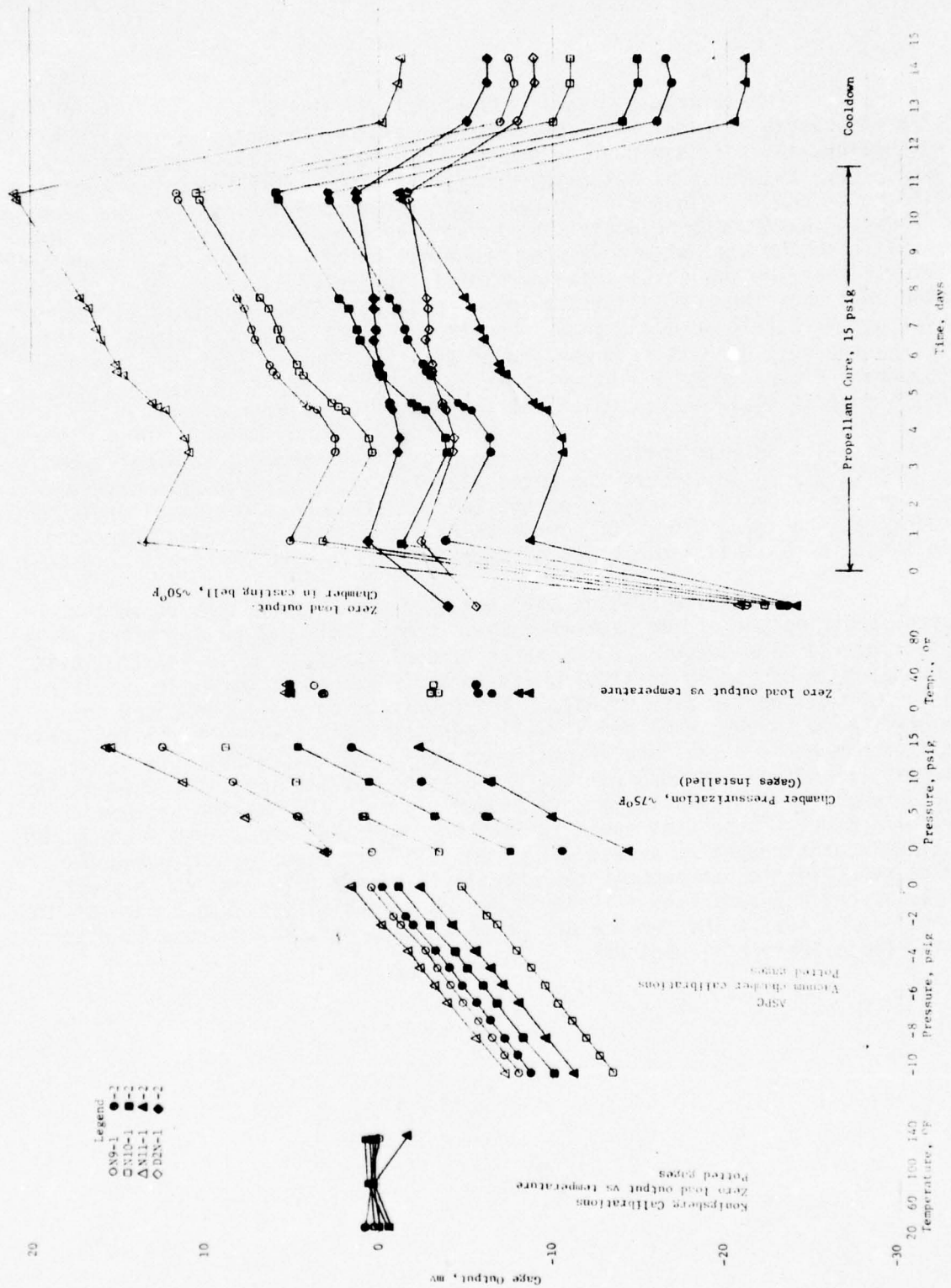


FIGURE H-1. OUTPUT LOG FOR NORMAL GAGES N9, N10, D2N

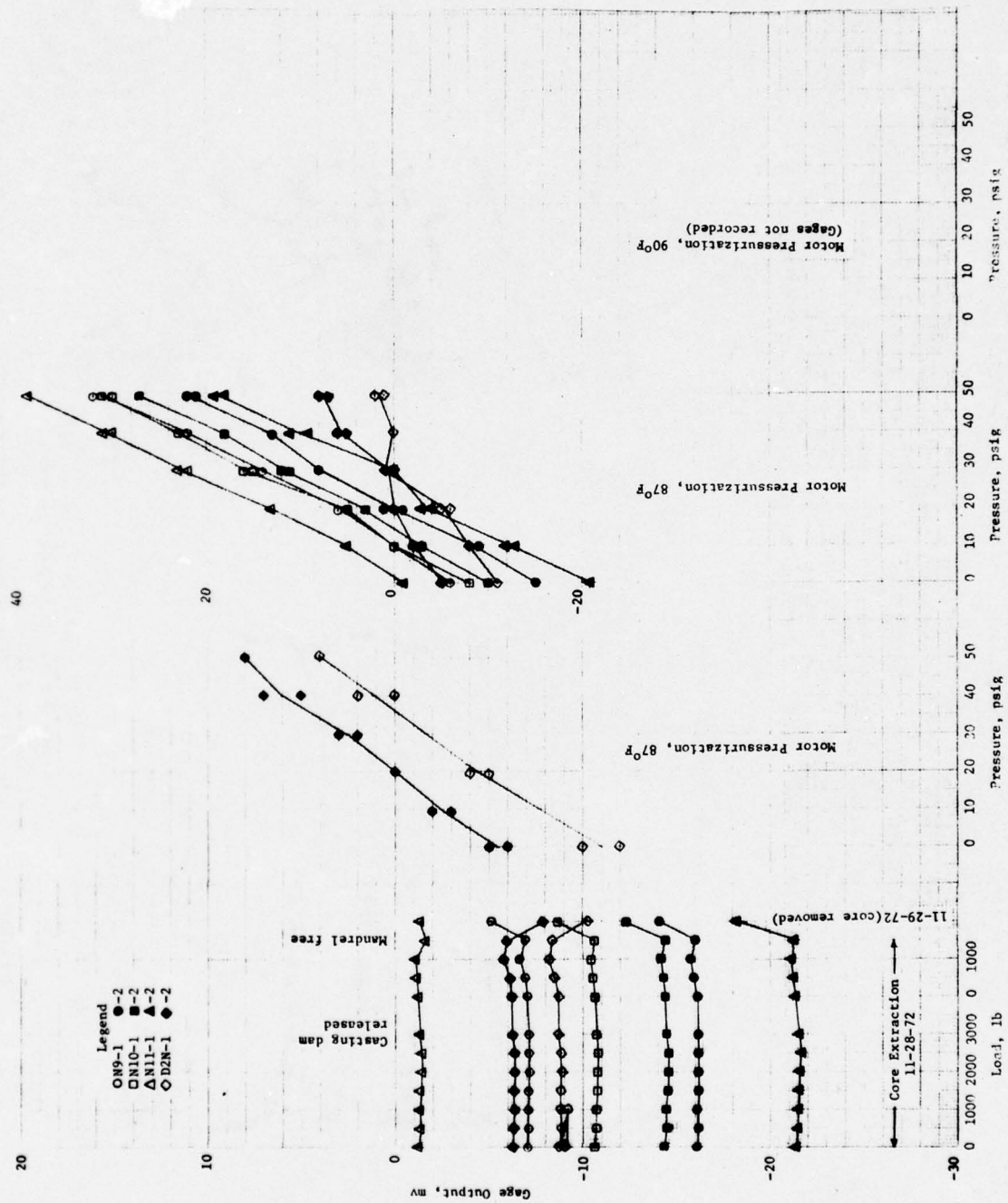


FIGURE H-1. OUTPUT LOG FOR NORMAL GAGES N9, N10, N11, D2N (CONT.)



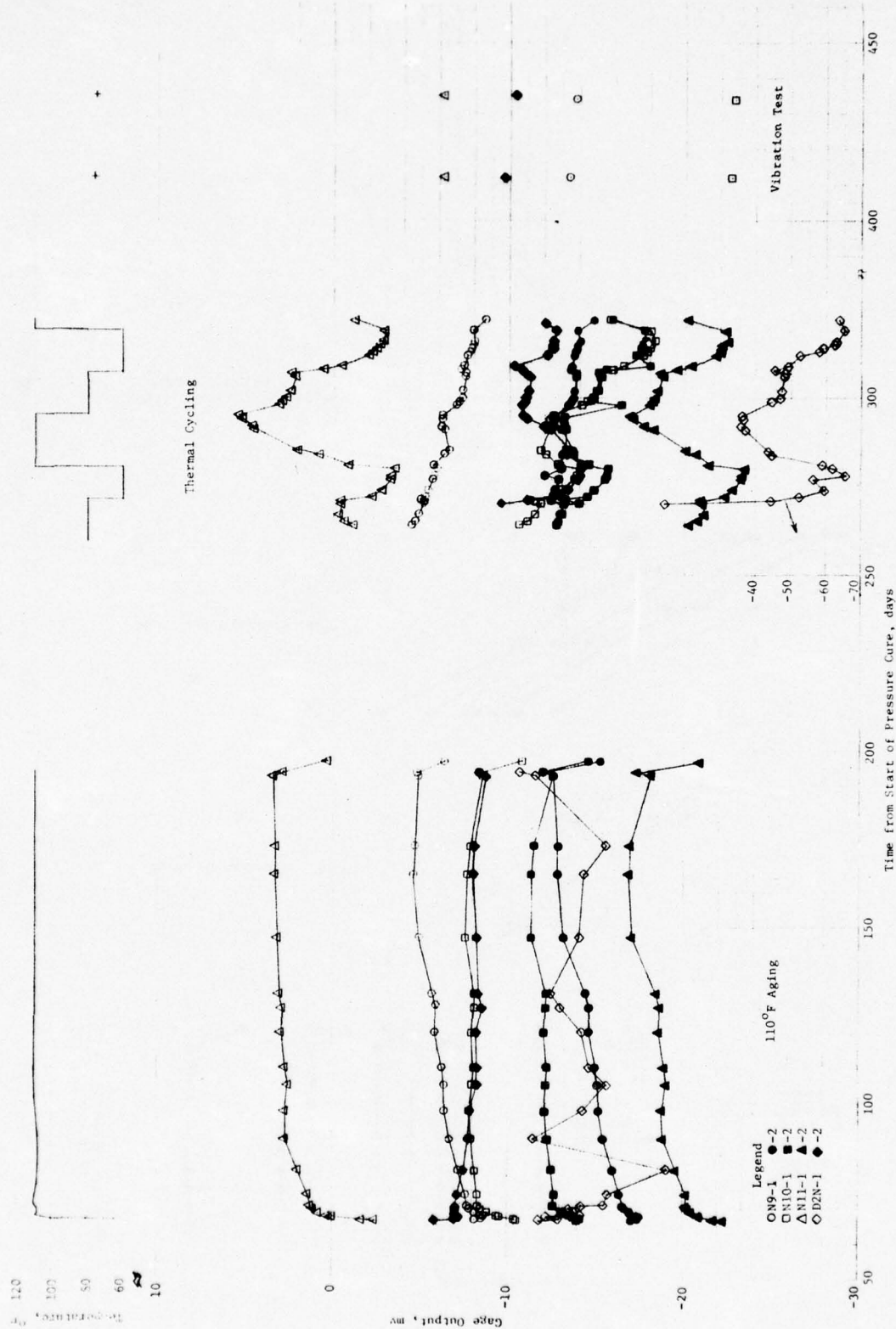


FIGURE H-1. OUTPUT LOG FOR NORMAL GAGES N9, N10, N11, D2N (CONT.)

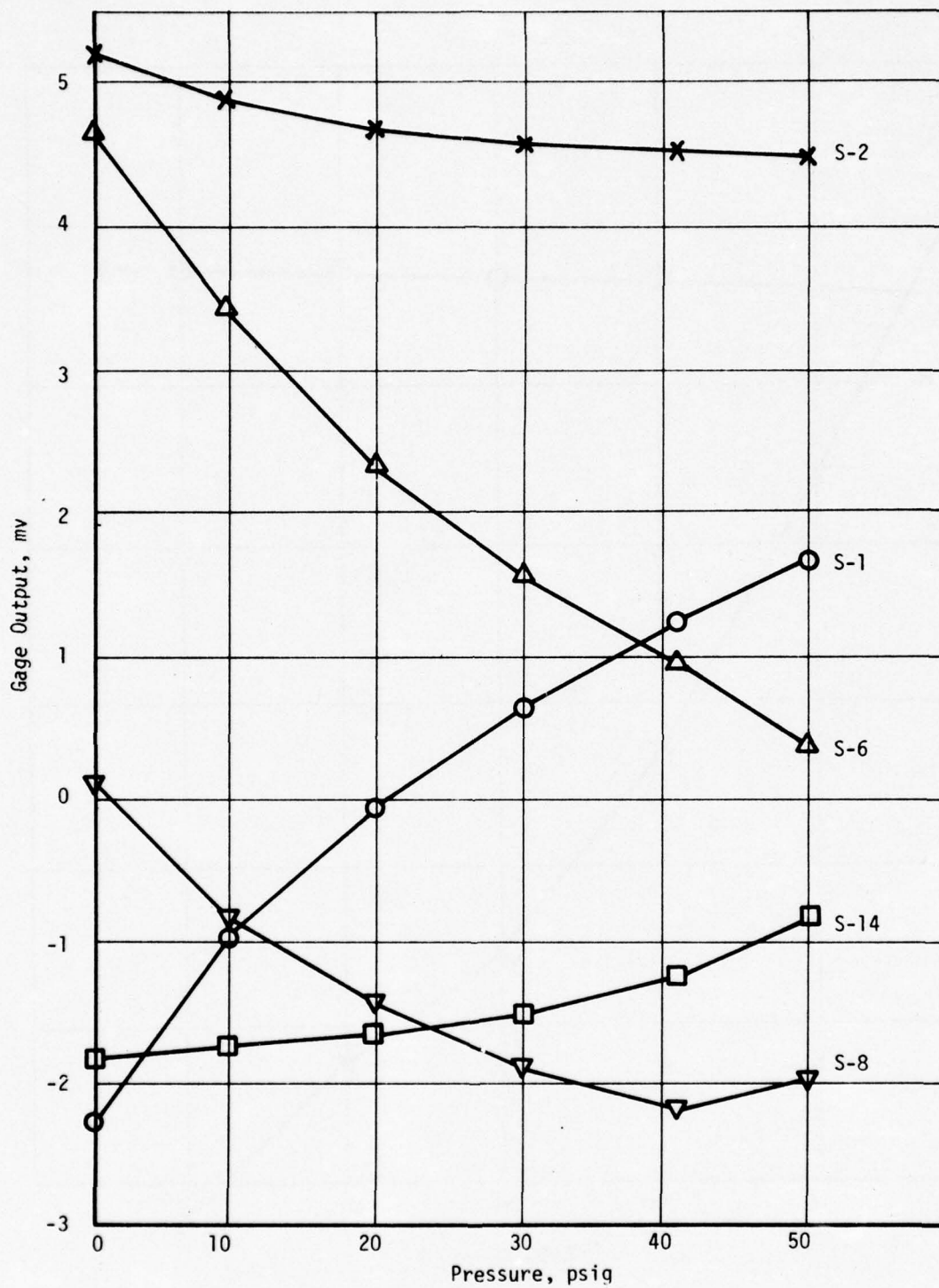


FIGURE H-2. PRESSURE TEST DATA: FULL SCALE MOTOR NO. 1  
SHEAR GAGES S-1, S-2, S-6, S-8 AND S-14  
H-13

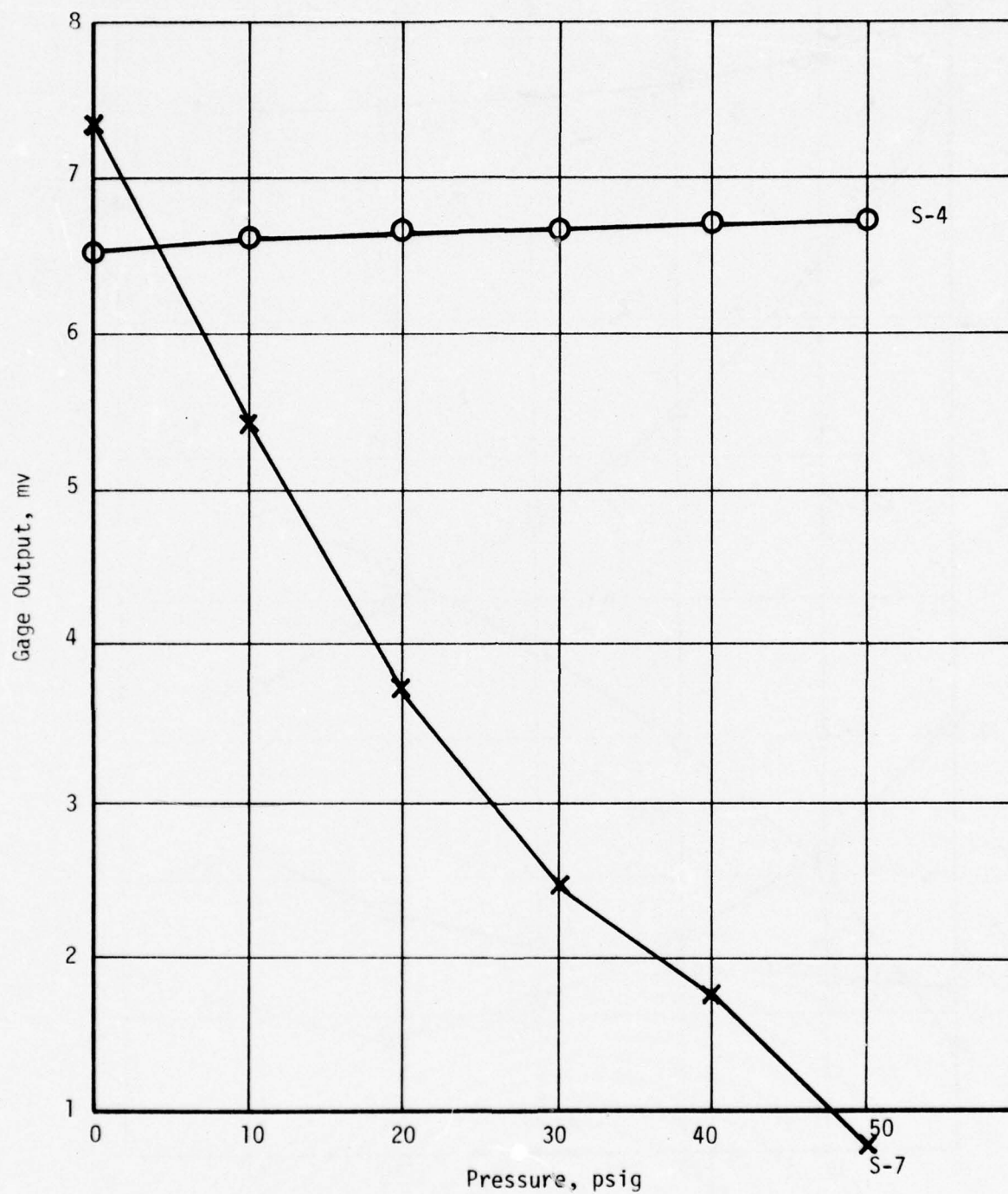


FIGURE H-3. PRESSURE TEST DATA: FULL SCALE MOTOR NO. 1 SHEAR GAGES S-4 & S-7



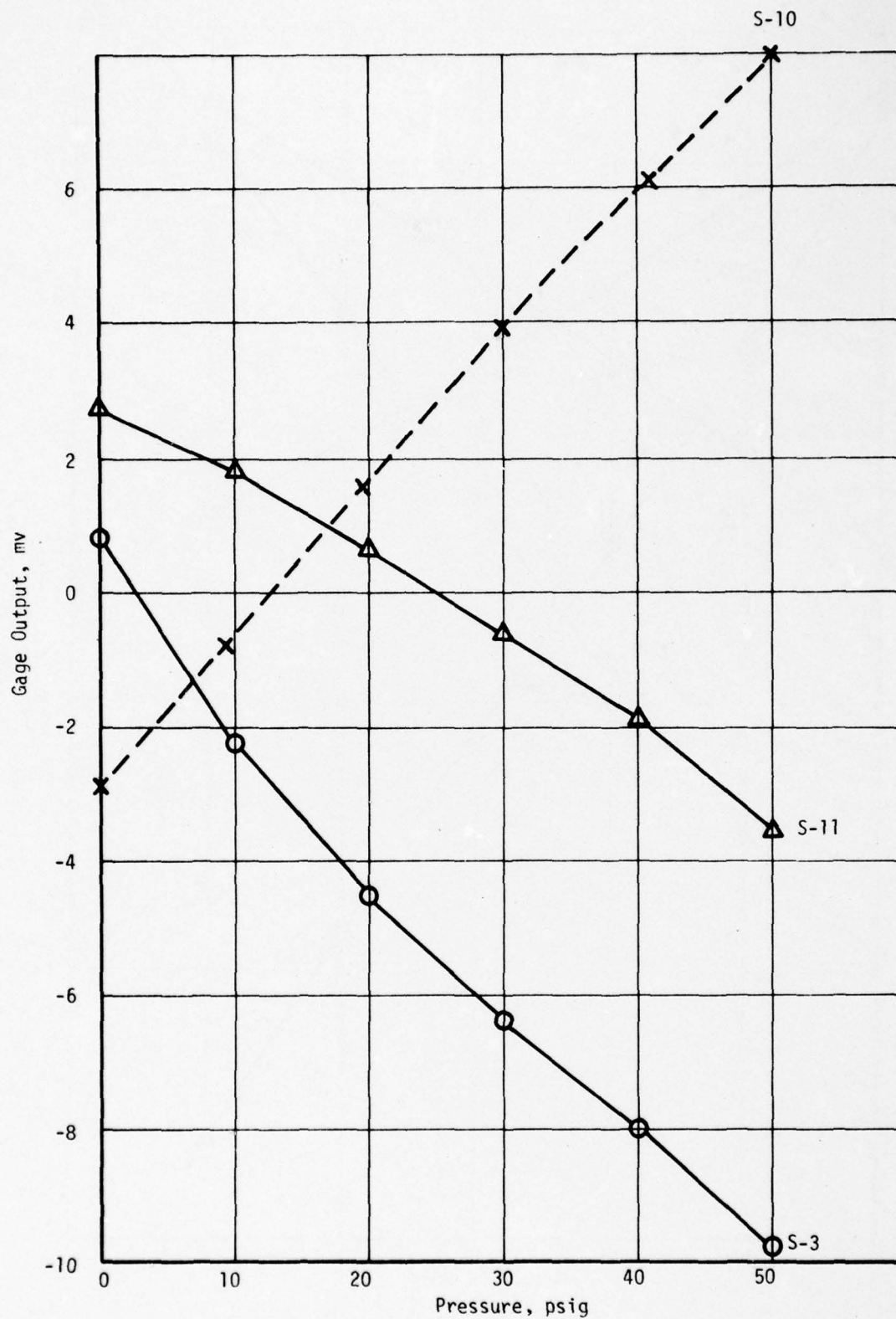


FIGURE H-4. PRESSURE TEST DATA: FULL SCALE MOTOR NO. 1, SHEAR GAGES  
S-3, S-10, AND S-11  
H-15

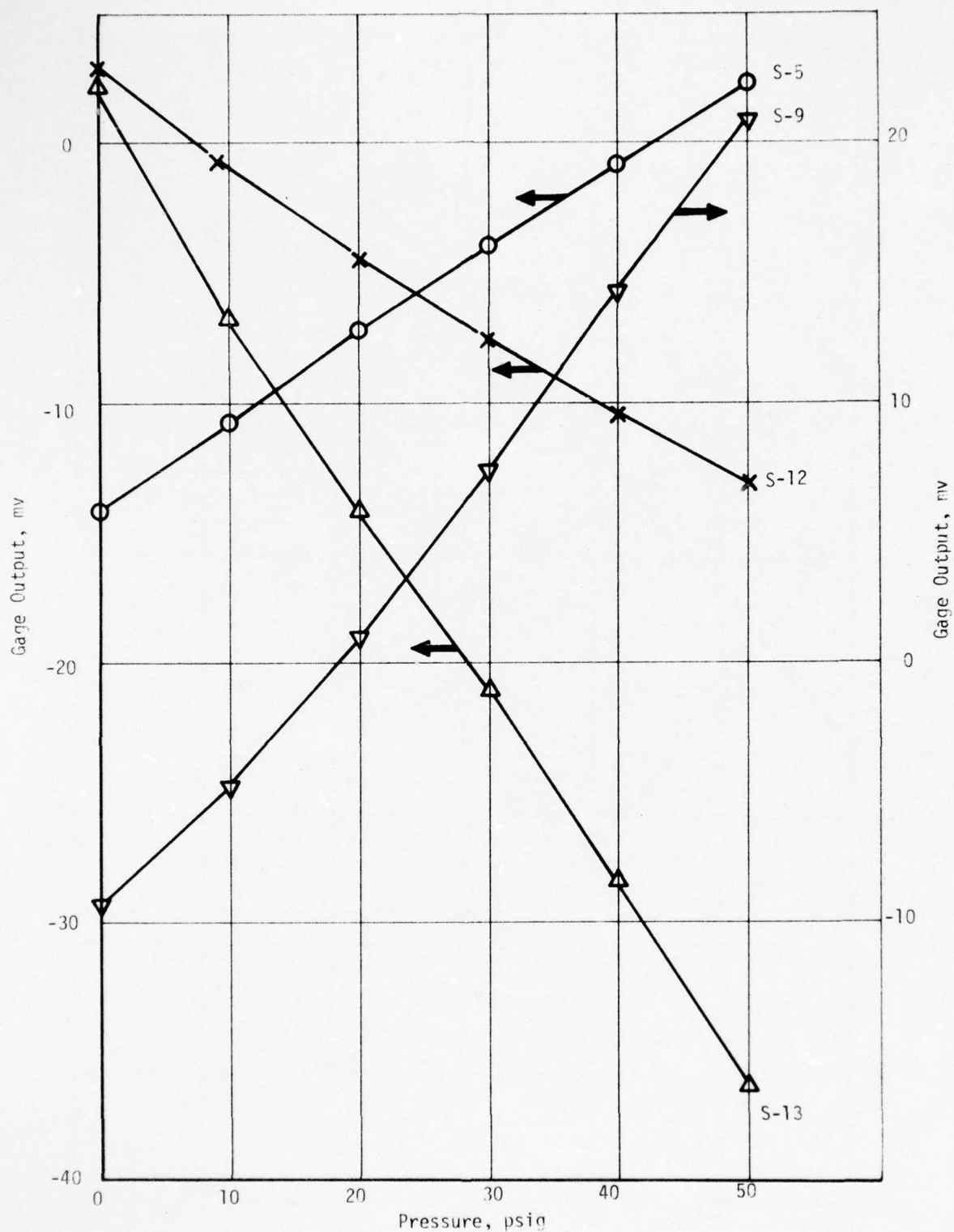


FIGURE H-5. PRESSURE TEST DATA: FULL SCALE MOTOR NO. 1, SHEAR GAGES S-5, S-9, S-12 AND S-13

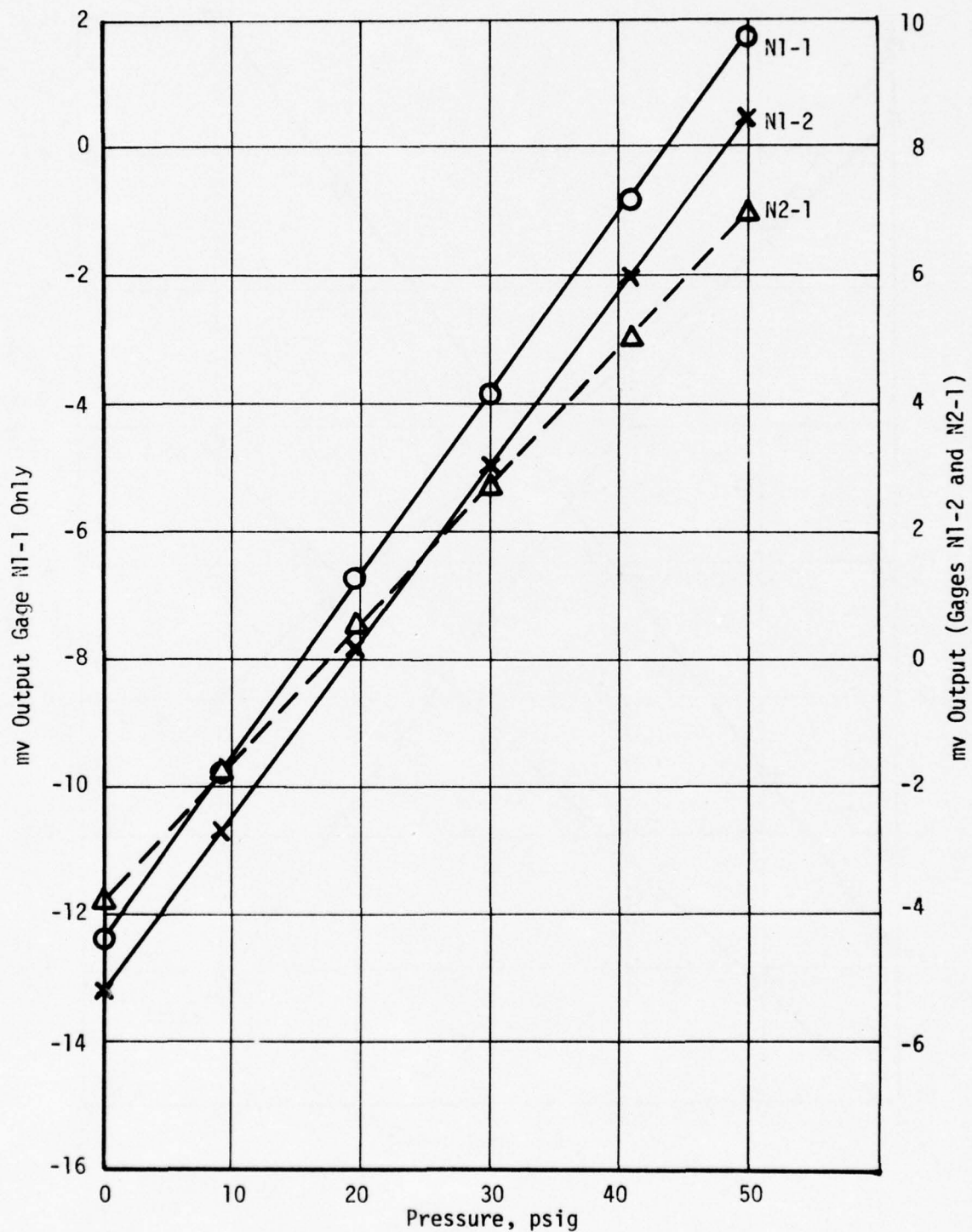


FIGURE H-6. PRESSURE TEST DATA: FULL SCALE MOTOR NO. 1  
450 PSI NORMAL GAGES N1 AND N2



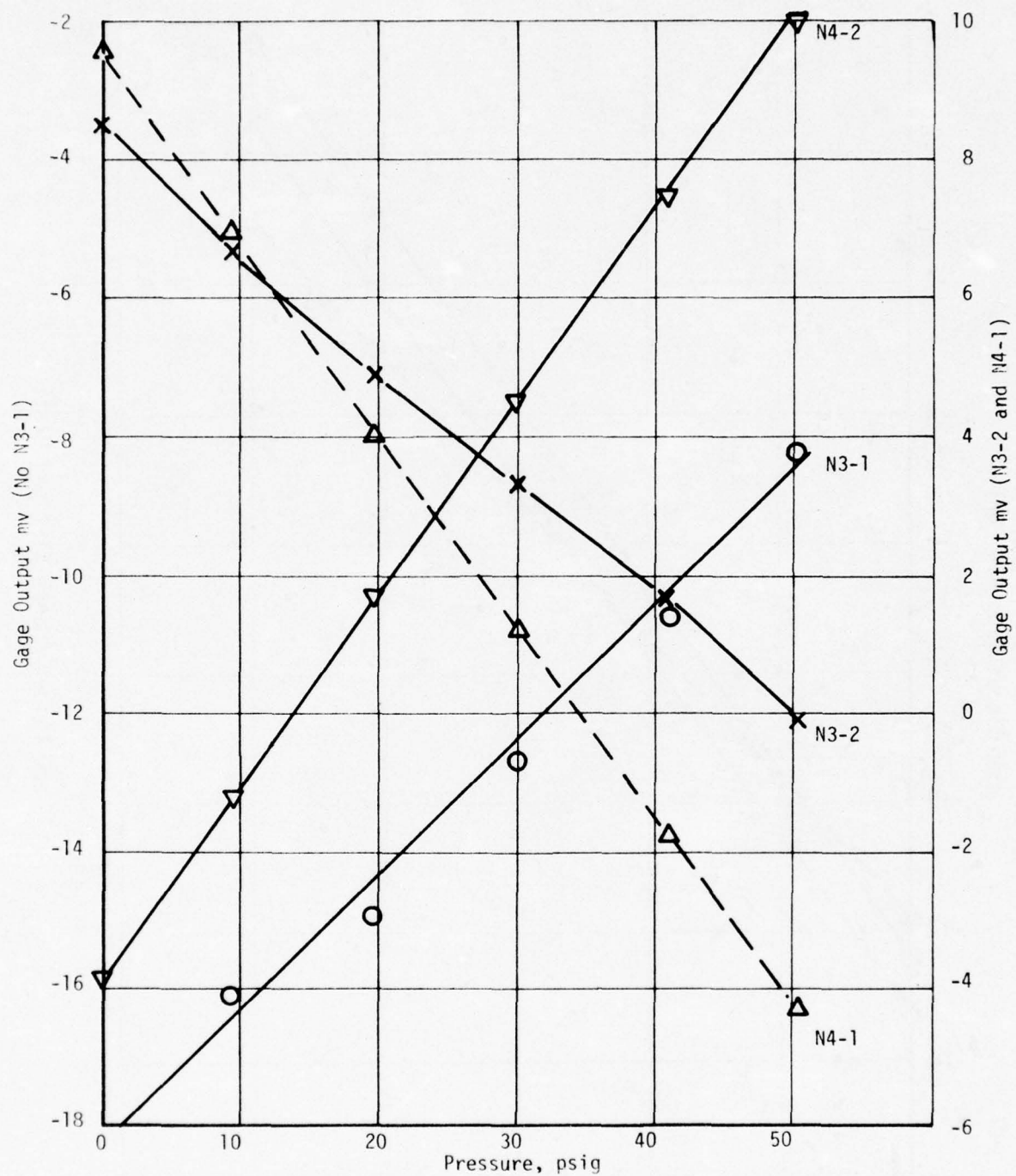


FIGURE H-7. PRESSURE TEST DATA: FULL SCALE MOTOR NO. 1: 450 PSI  
NORMAL GAGES N3 AND N4

H-18

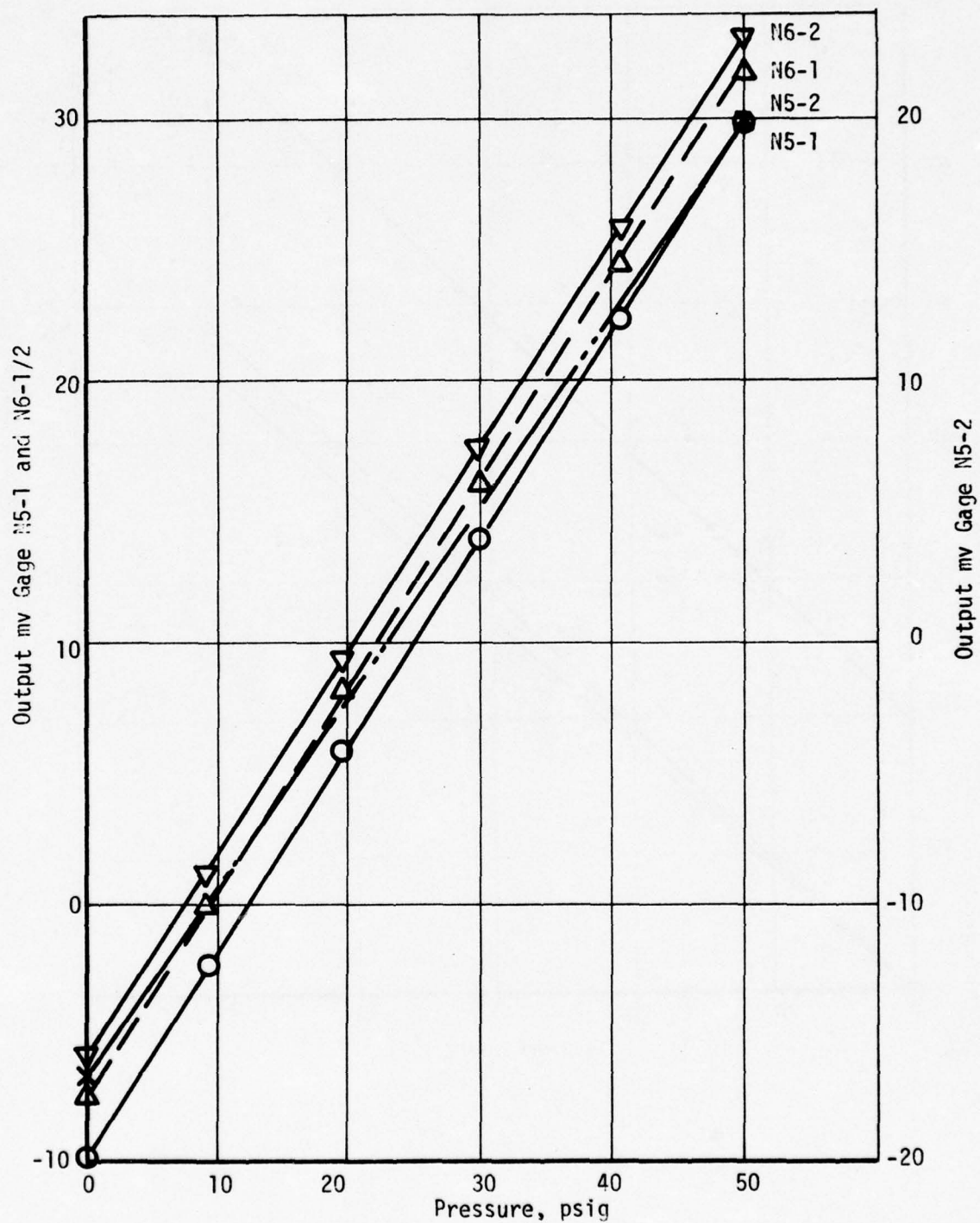


FIGURE H-8. PRESSURE TEST DATA: FULL SCALE MOTOR NO. 1  
450 PSI NORMAL GAGES N5 AND N6

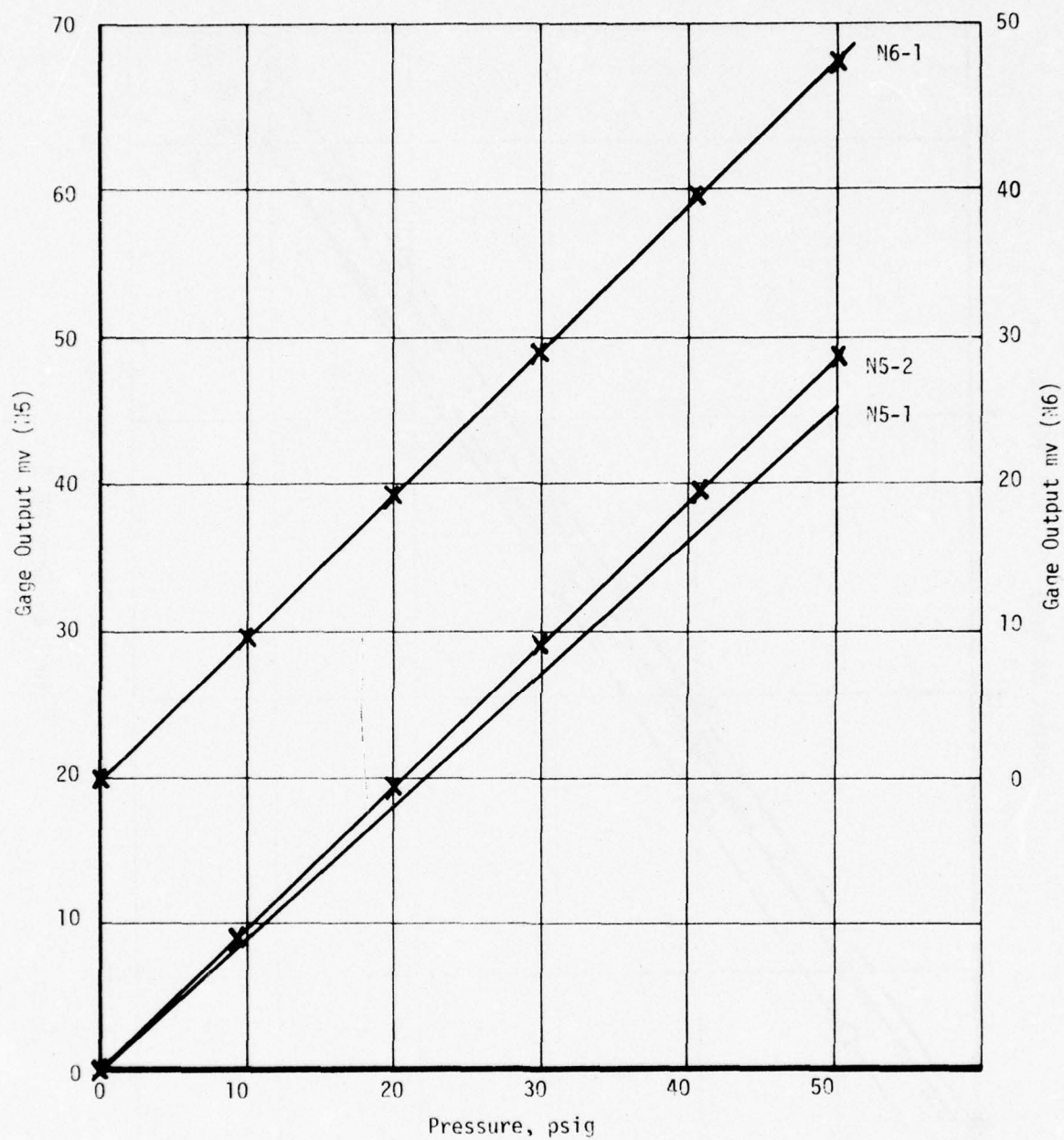


FIGURE H-9. PRESSURE TEST DATA: FULL SCALE MOTOR NO. 1  
150 PSI GAGES N5 AND N6

H-20



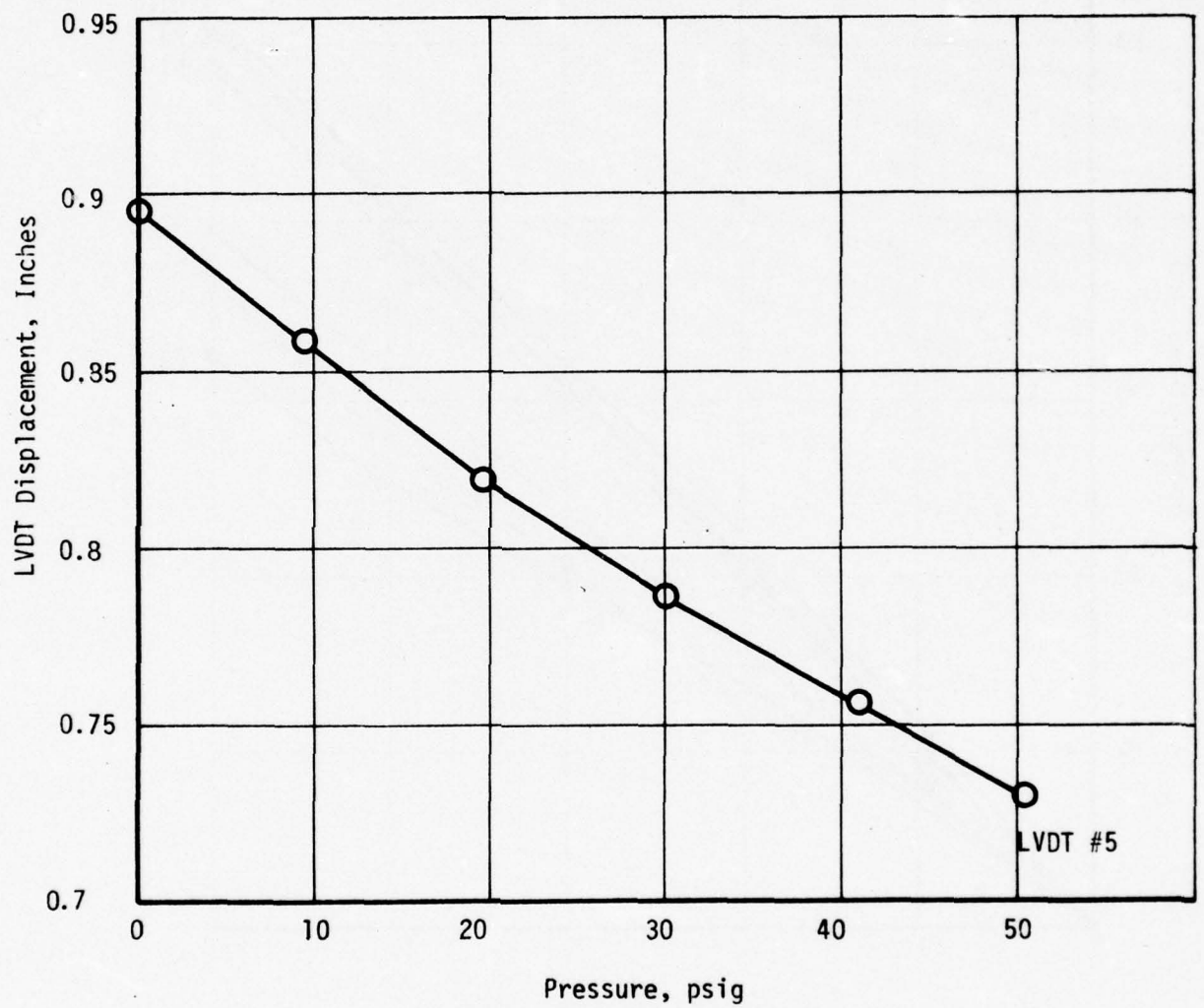


FIGURE H-10. PRESSURE TEST DATA: FULL SCALE MOTOR NO. 1 LVDT #5

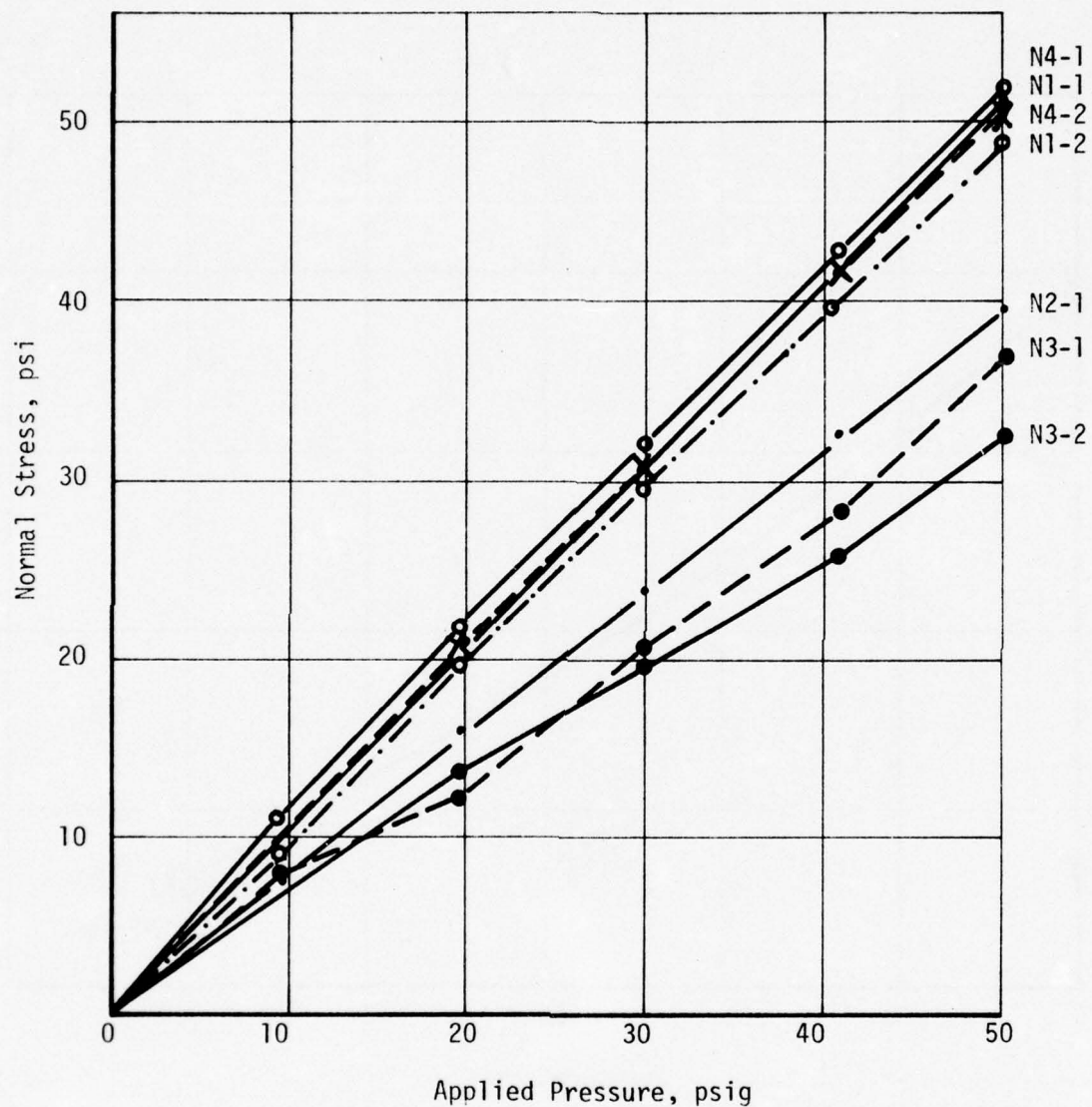


FIGURE H-11. NORMAL STRESS GAGE OUTPUT (PSI) VERSUS APPLIED PRESSURE FULL SCALE MOTOR NO. 1

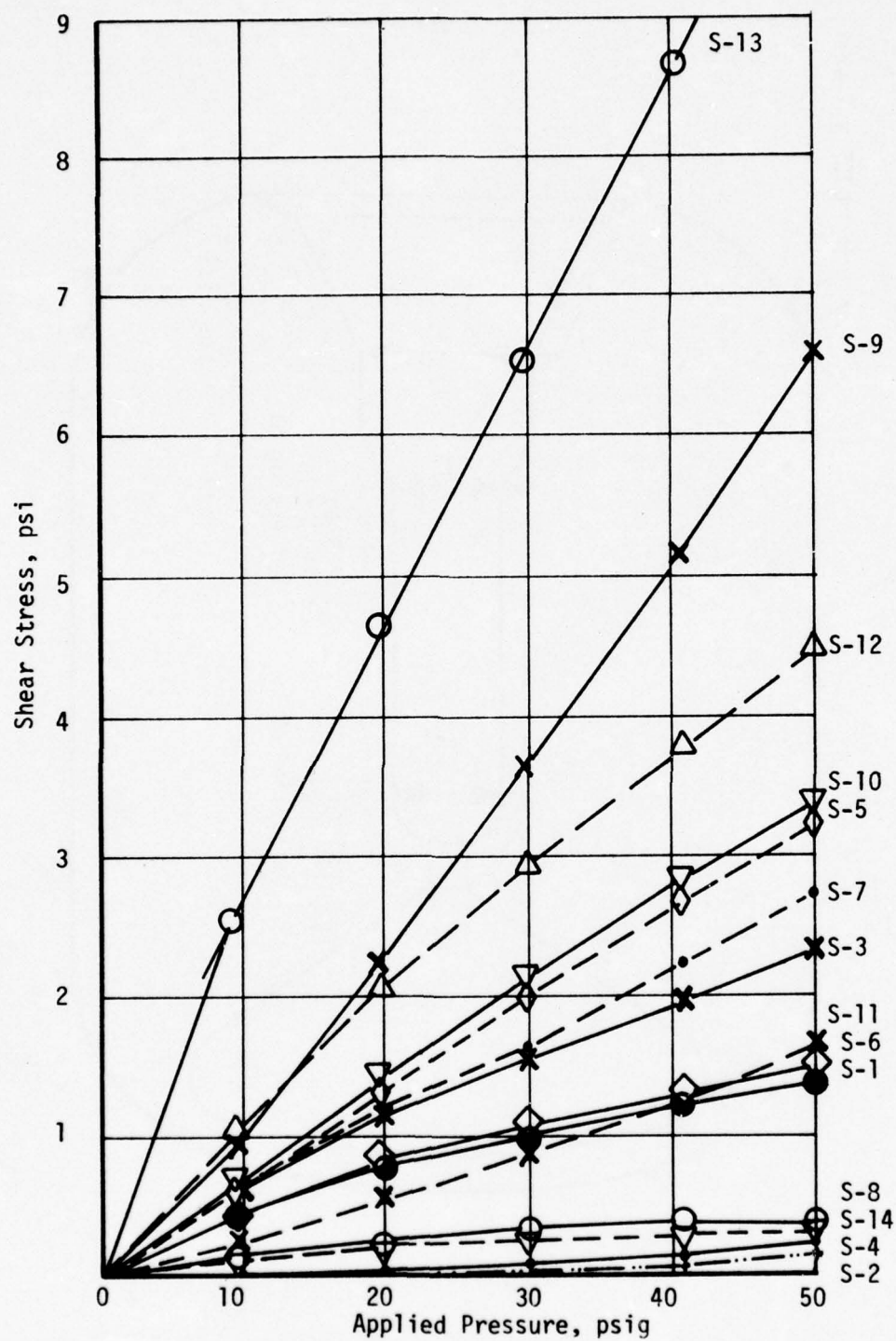


FIGURE H-12. SHEAR GAGE OUTPUT PSI VERSUS APPLIED  
PRESSURE FULL SCALE MOTOR NO. 1



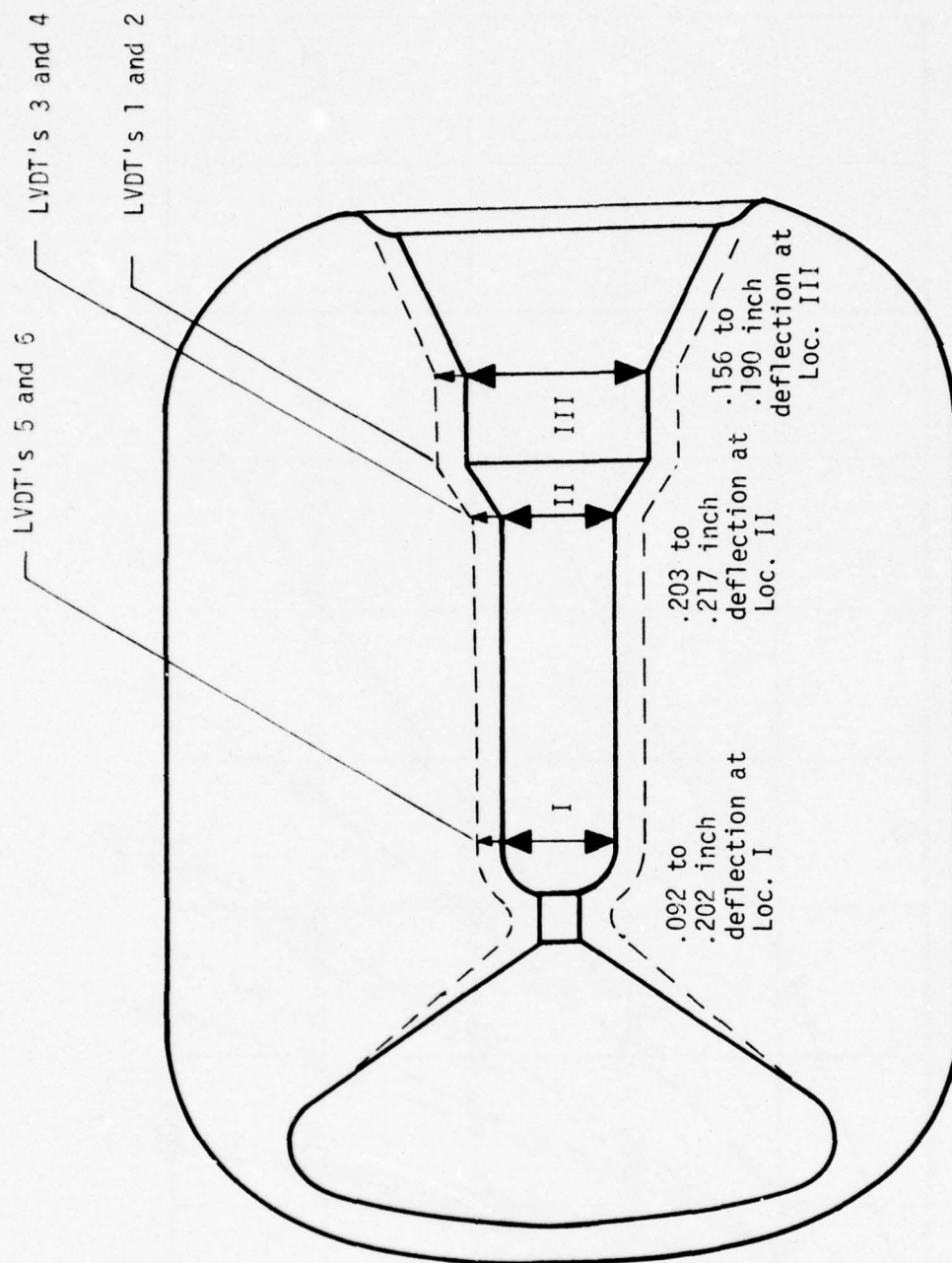


FIGURE H-13. BORE DEFLECTIONS; TEST RESULTS AT 49.45 PSIG

( $T = 80 \pm 5^{\circ}\text{F}$ )

Shear Gage Sensitivity to Pressure in Motor No. 1  
(Before Casting)

S-1	+0.270	S-8	-0.074
S-2	-0.050	S-9	+0.026
S-3	+0.280	S-10	+0.376
S-4	+0.150	S-11	-0.062
S-5	+0.386	S-12	-0.194
S-6	+0.012	S-13	-0.118
S-7	-0.078	S-14	-0.092

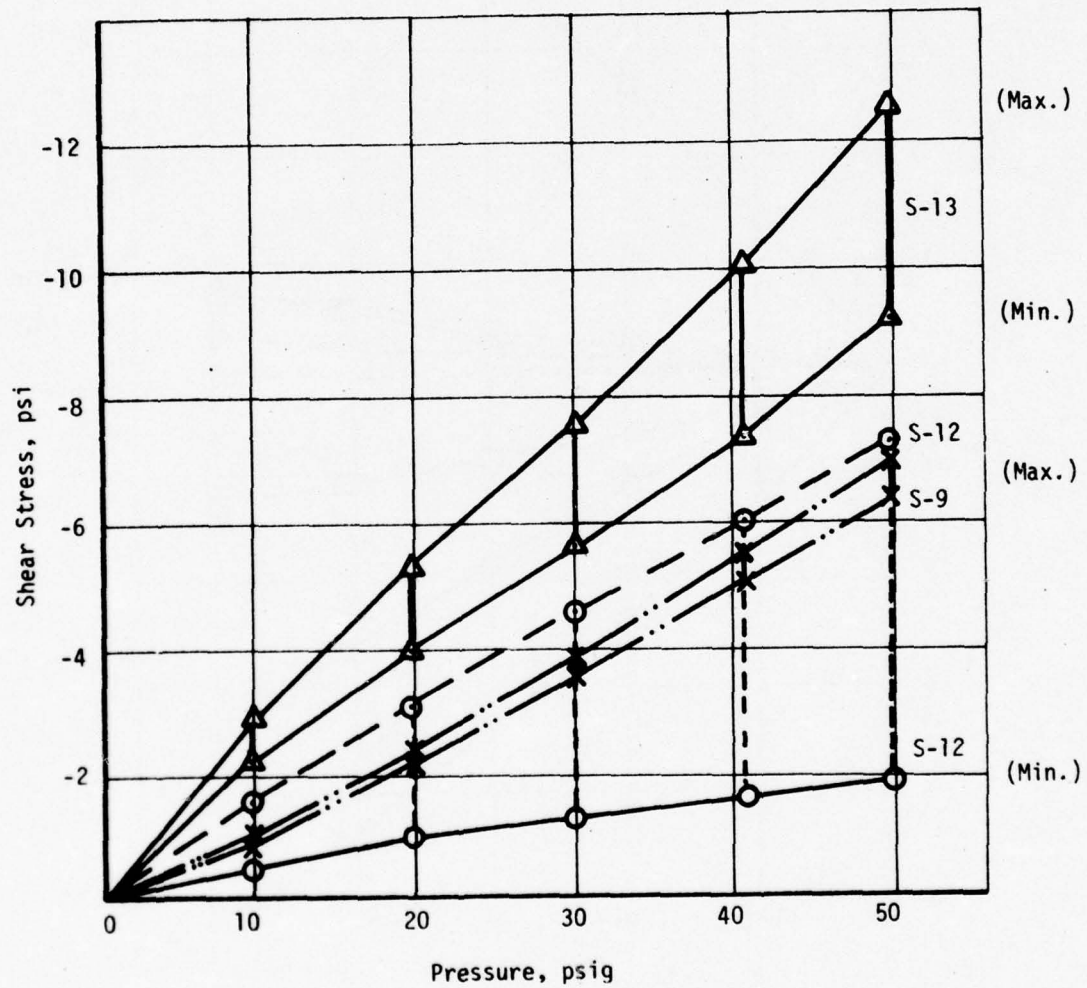


FIGURE H- 14. SHEAR STRESS DATA MOTOR NO. 1 USING PRESSURE  
CALIBRATION DATA FROM PRE-CAST TEST  
GAGES 9, 12 AND 13

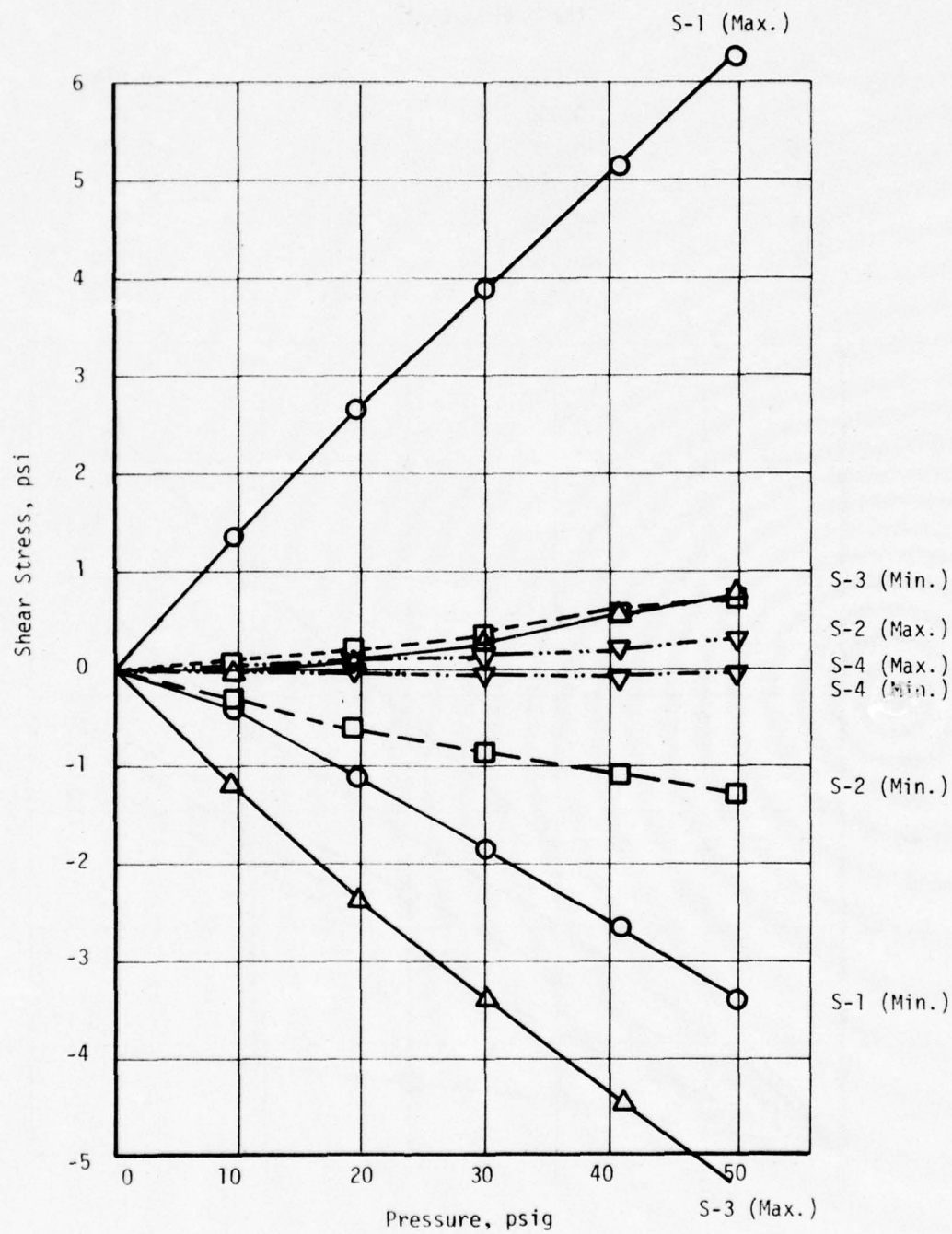


FIGURE H-15. SHEAR GAGE DATA MOTOR NO. 1 USING PRESSURE  
CALIBRATION DATA FROM PRE-CAST TEST  
GAGES 1, 2, 3 AND 4  
H-26



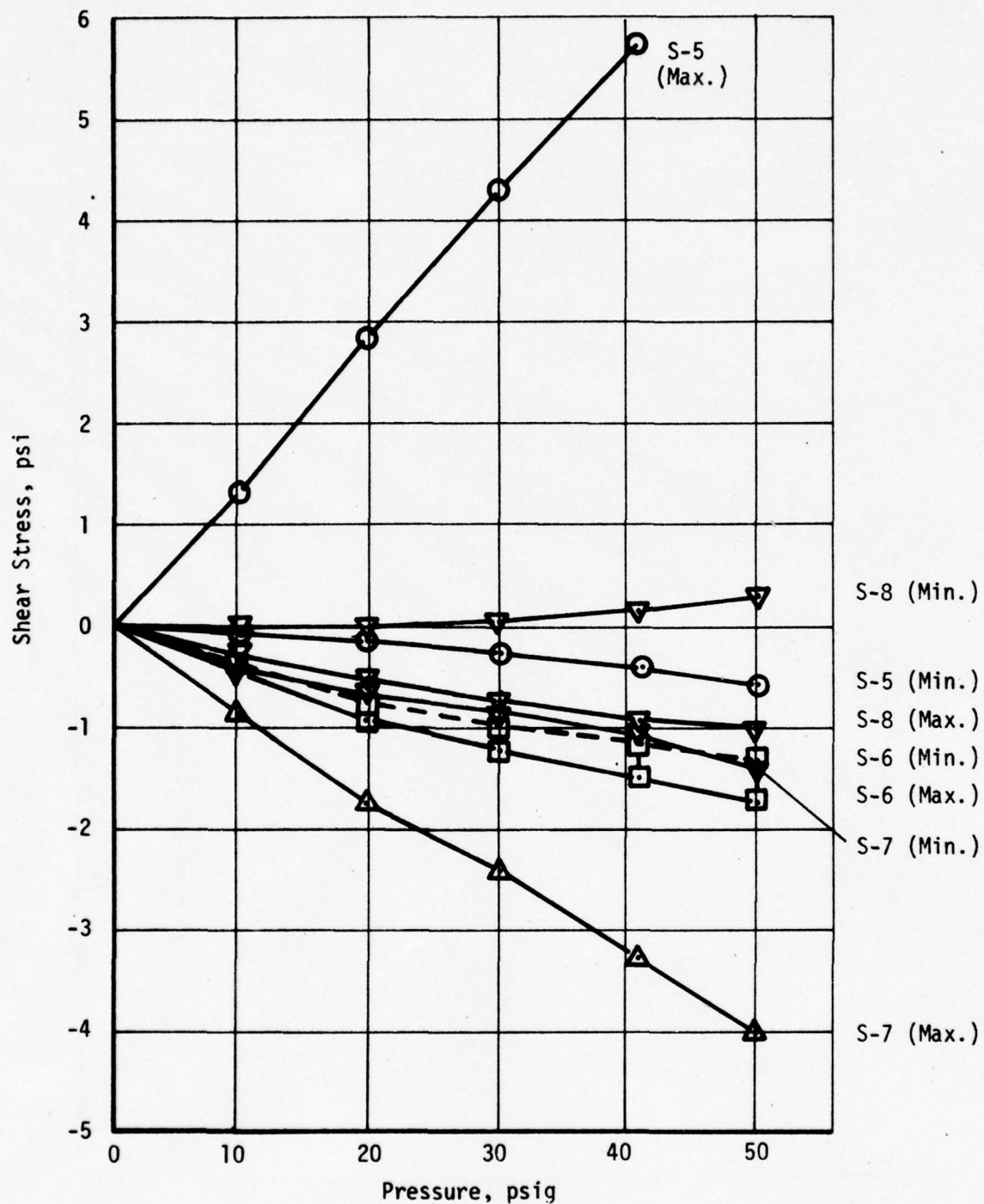


FIGURE H- 16. SHEAR GAGE DATA MOTOR NO. 1 USING PRESSURE  
CALIBRATION DATA FROM PRE-CAST TEST  
GAGES 5, 6, 7 AND 8  
H-27

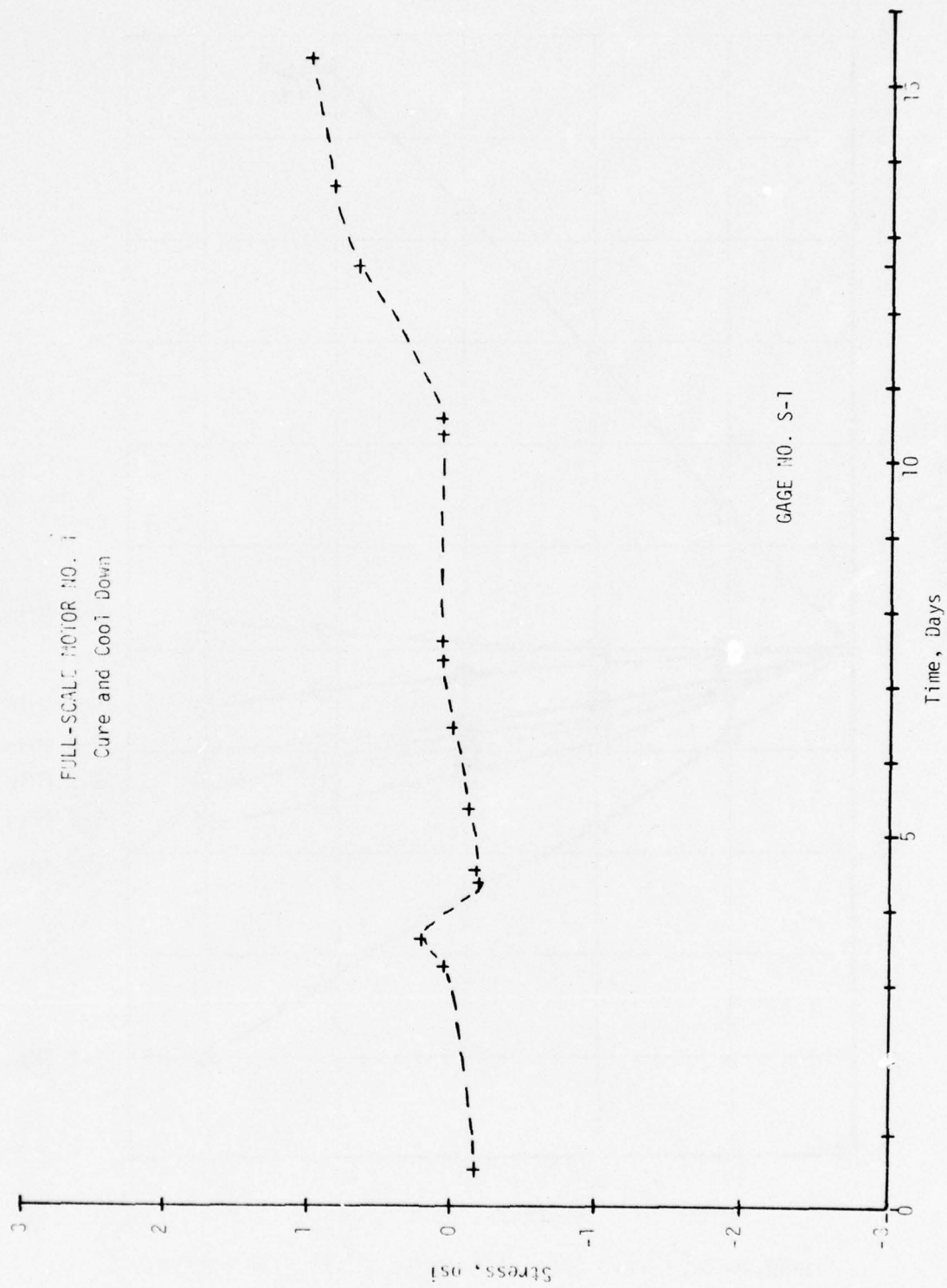
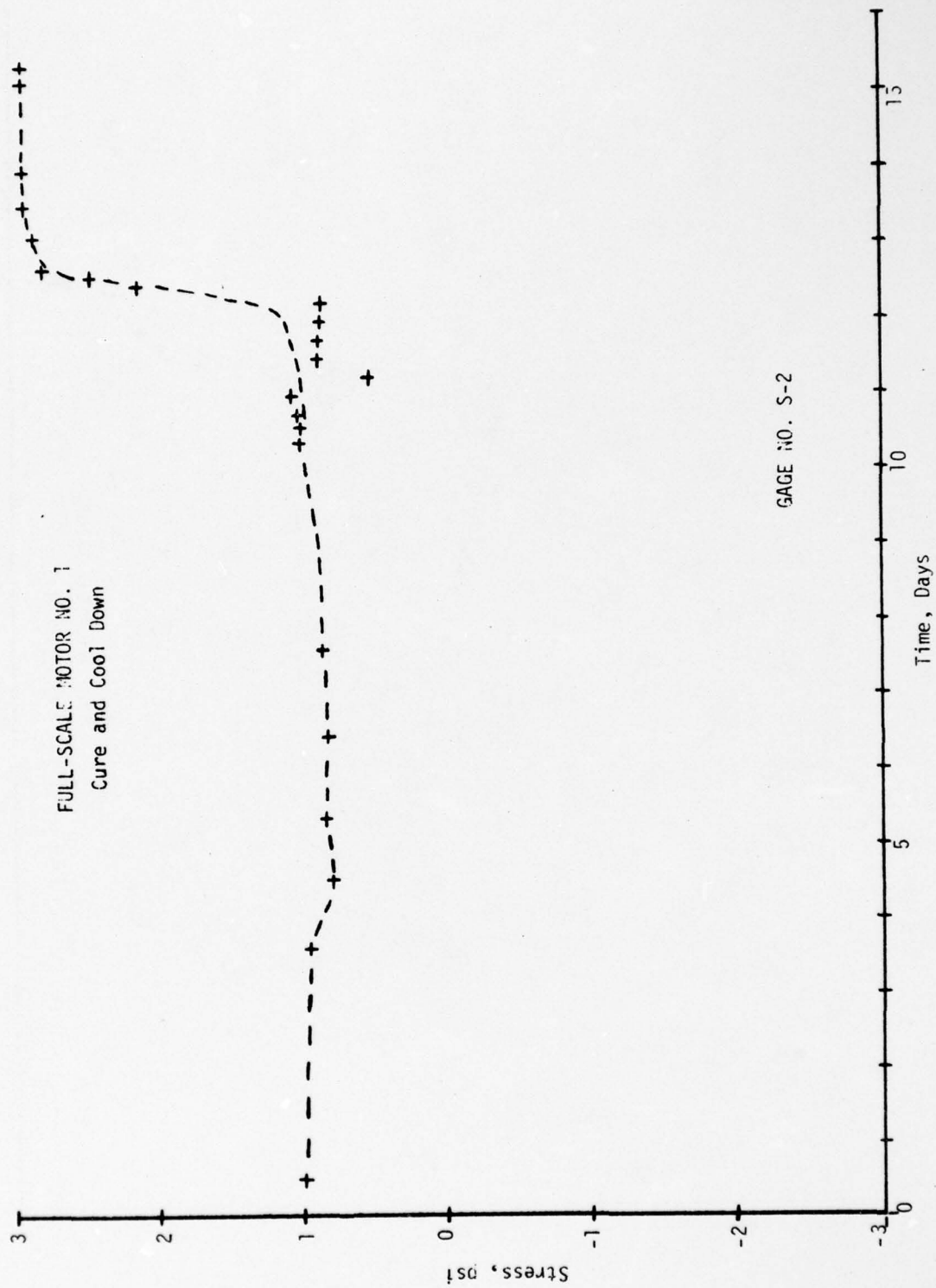


FIGURE H-1/. SHEAR STRESS DATA FROM GAGE S-1; CURE AND COOLDOWN



H-29

FIGURE H-18. SHEAR STRESS DATA FROM GAGE S-2; CURE AND COOLDOWN



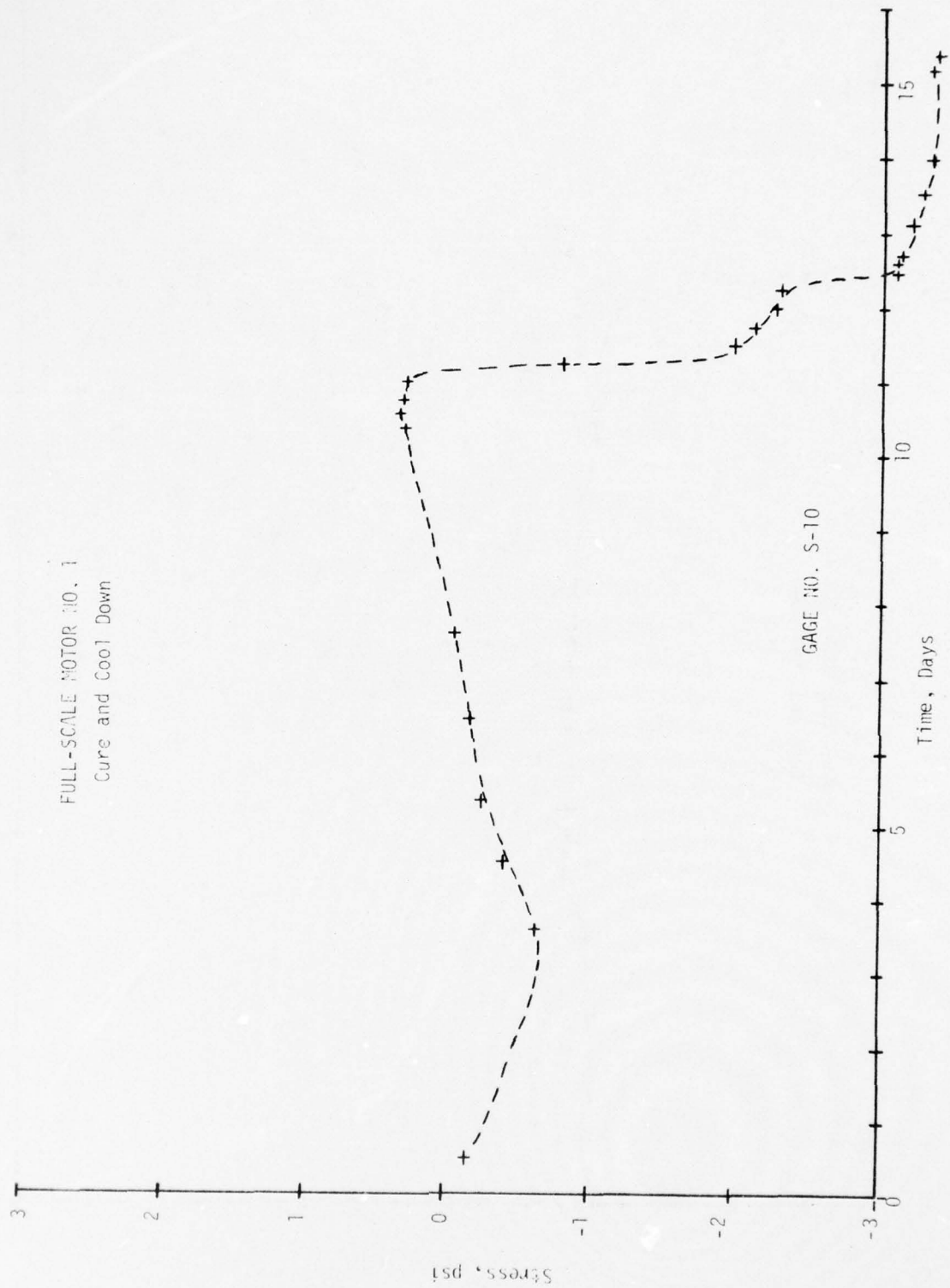


FIGURE H-19. SHEAR STRESS DATA FROM GAGE S-10; CURE AND COOLDOWN

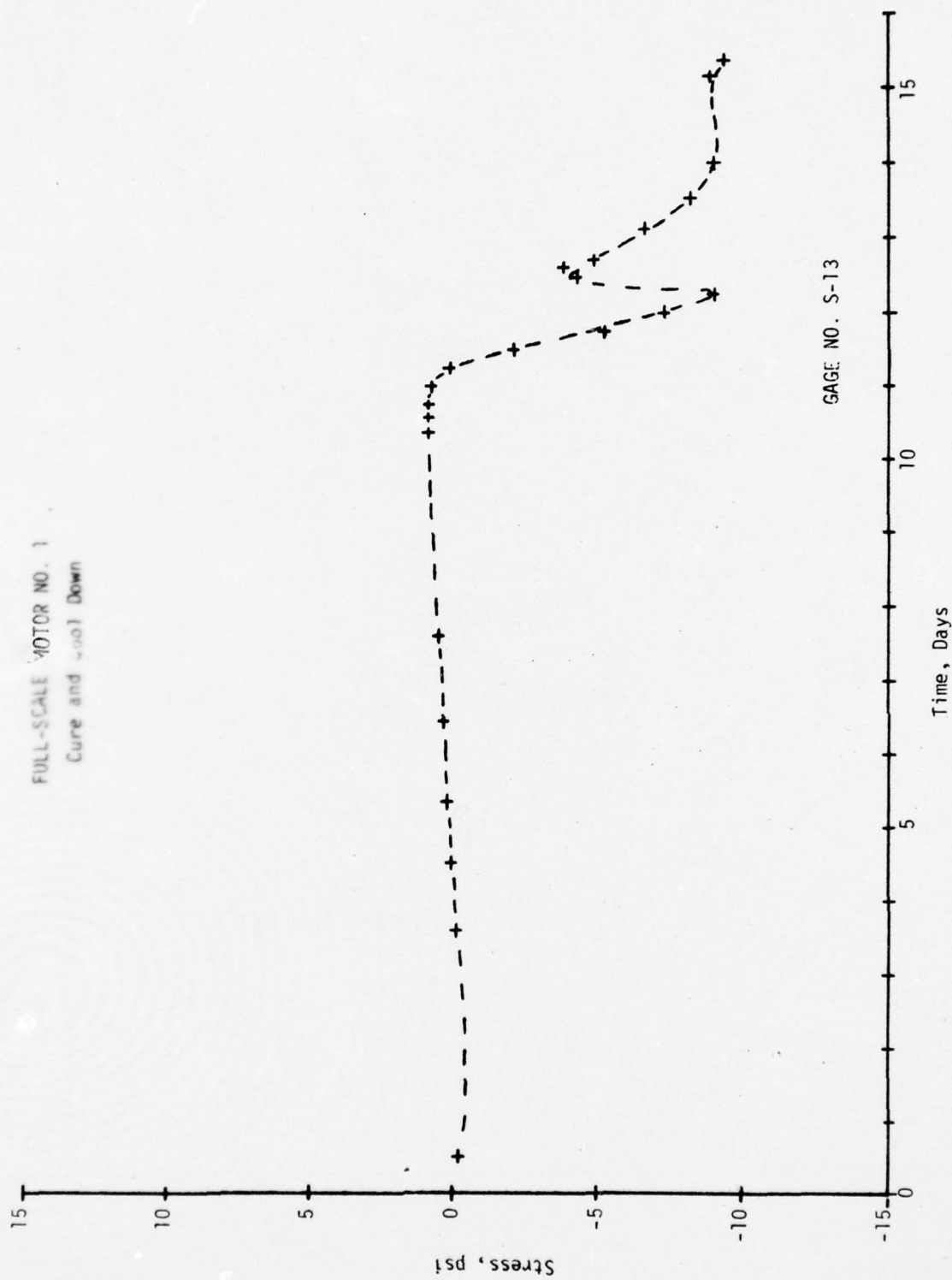


FIGURE H-20. SHEAR STRESS DATA FROM GAGE S-13; CURE AND COOLDOWN

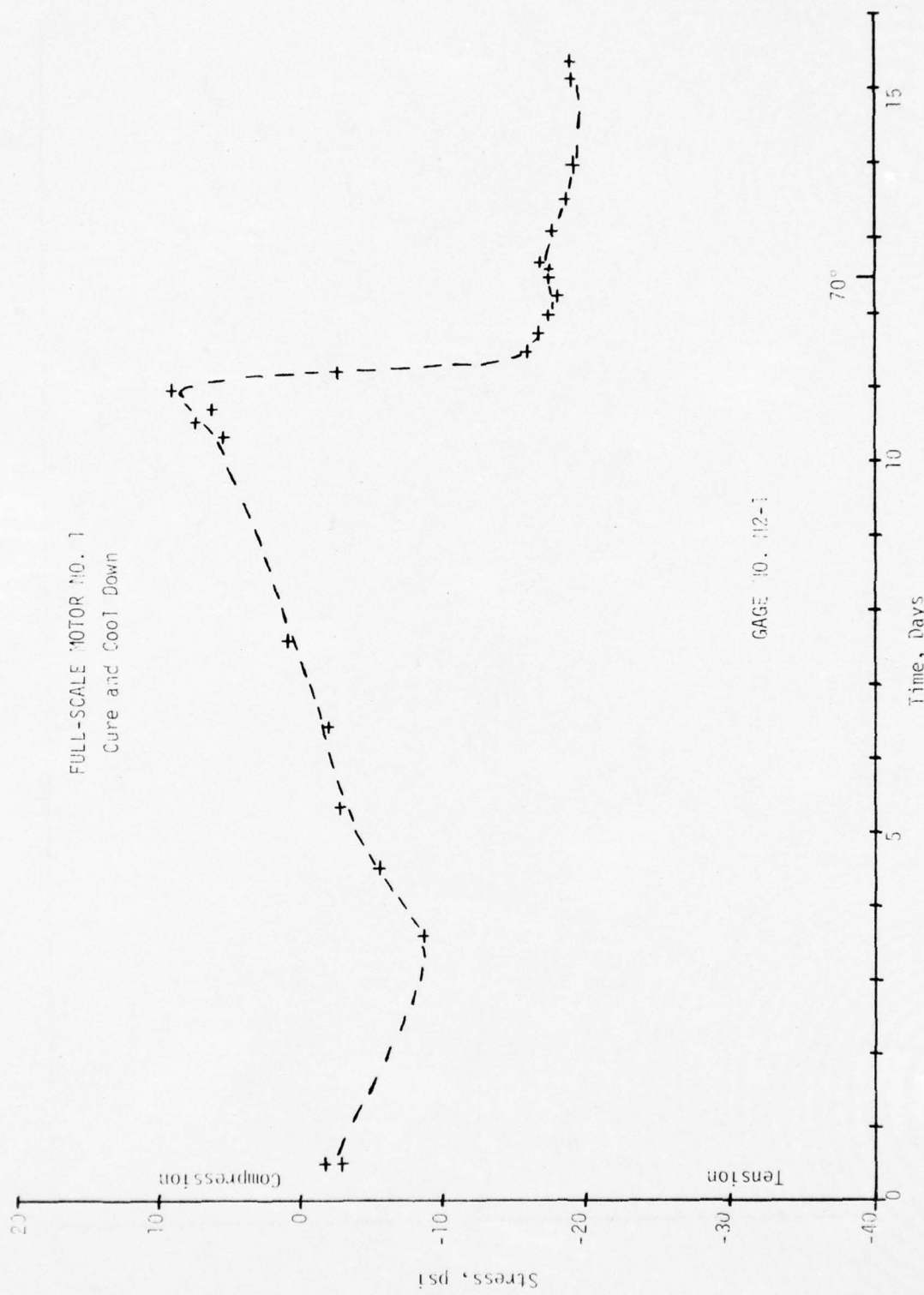


FIGURE H-21. NORMAL STRESS DATA FROM GAGE H2-1; CURE AND COOLDOWN



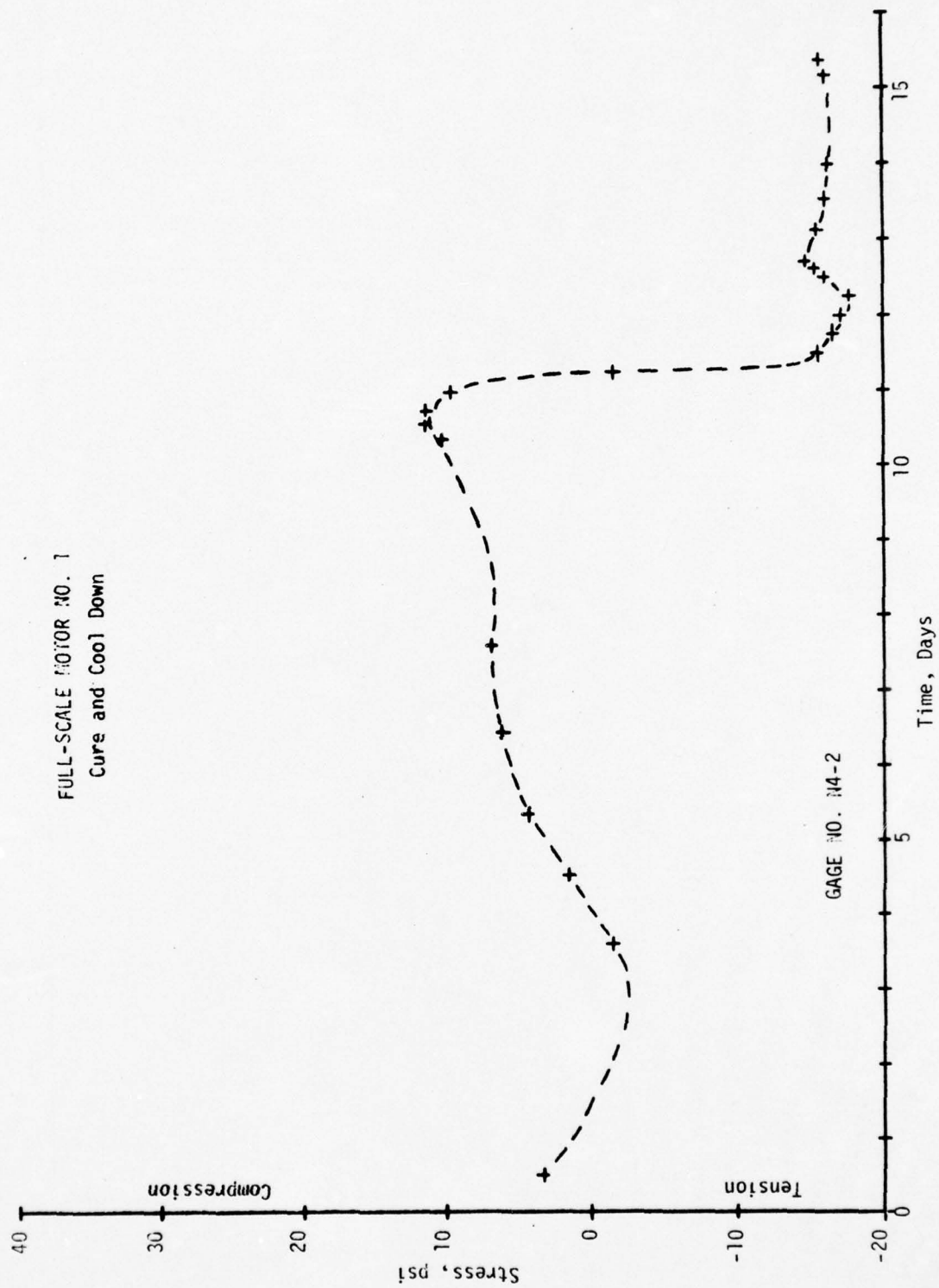


FIGURE H-22. NORMAL STRESS DATA FROM GAGE N4-2; CURE AND COOL DOWN



FIGURE H-23. NORMAL STRESS DATA FROM GAGE #19-1; CURE AND COOLDOWN

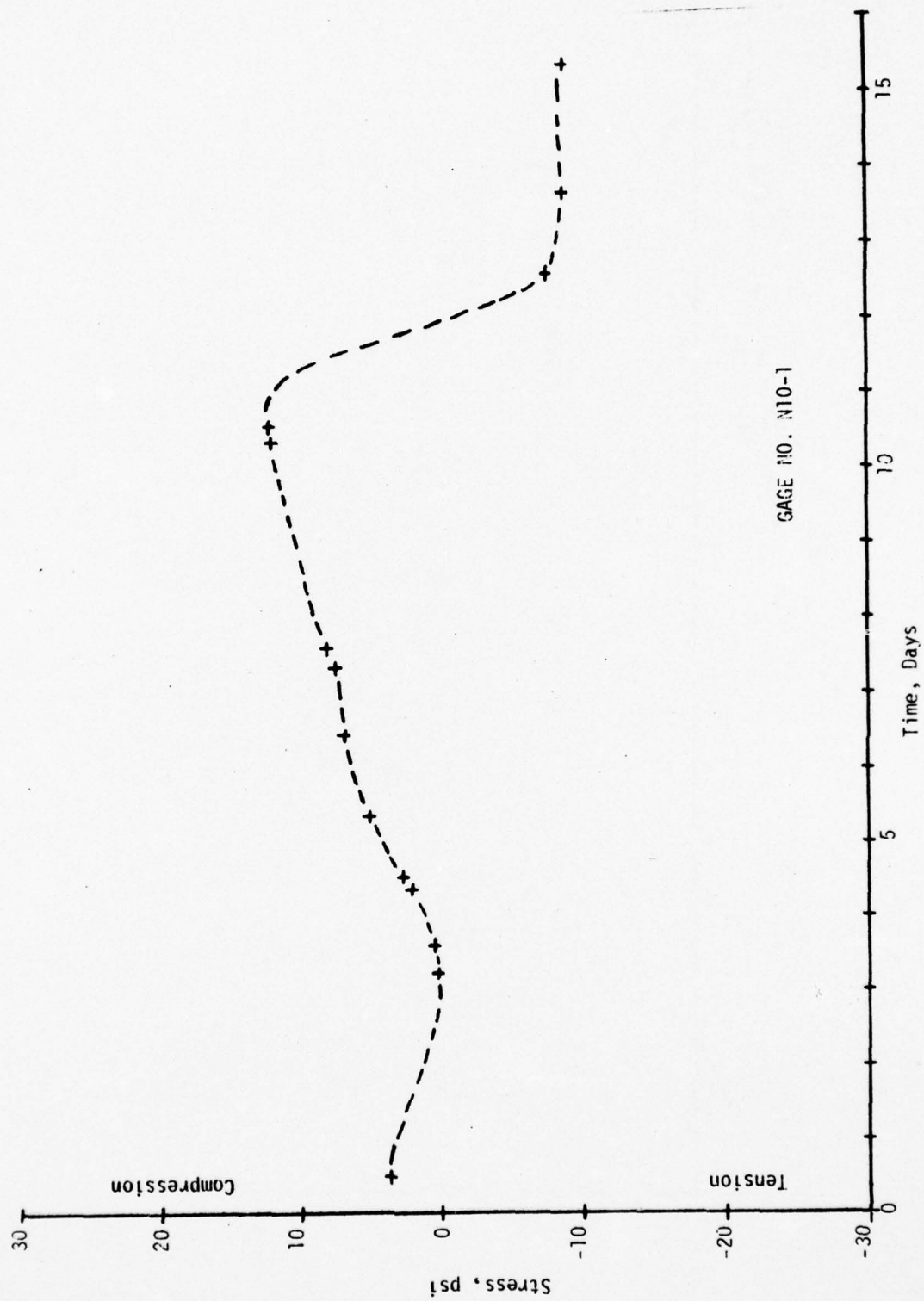


FIGURE H-24. NORMAL STRESS DATA FROM GAGE N10-1; CURE AND COOLDOWN



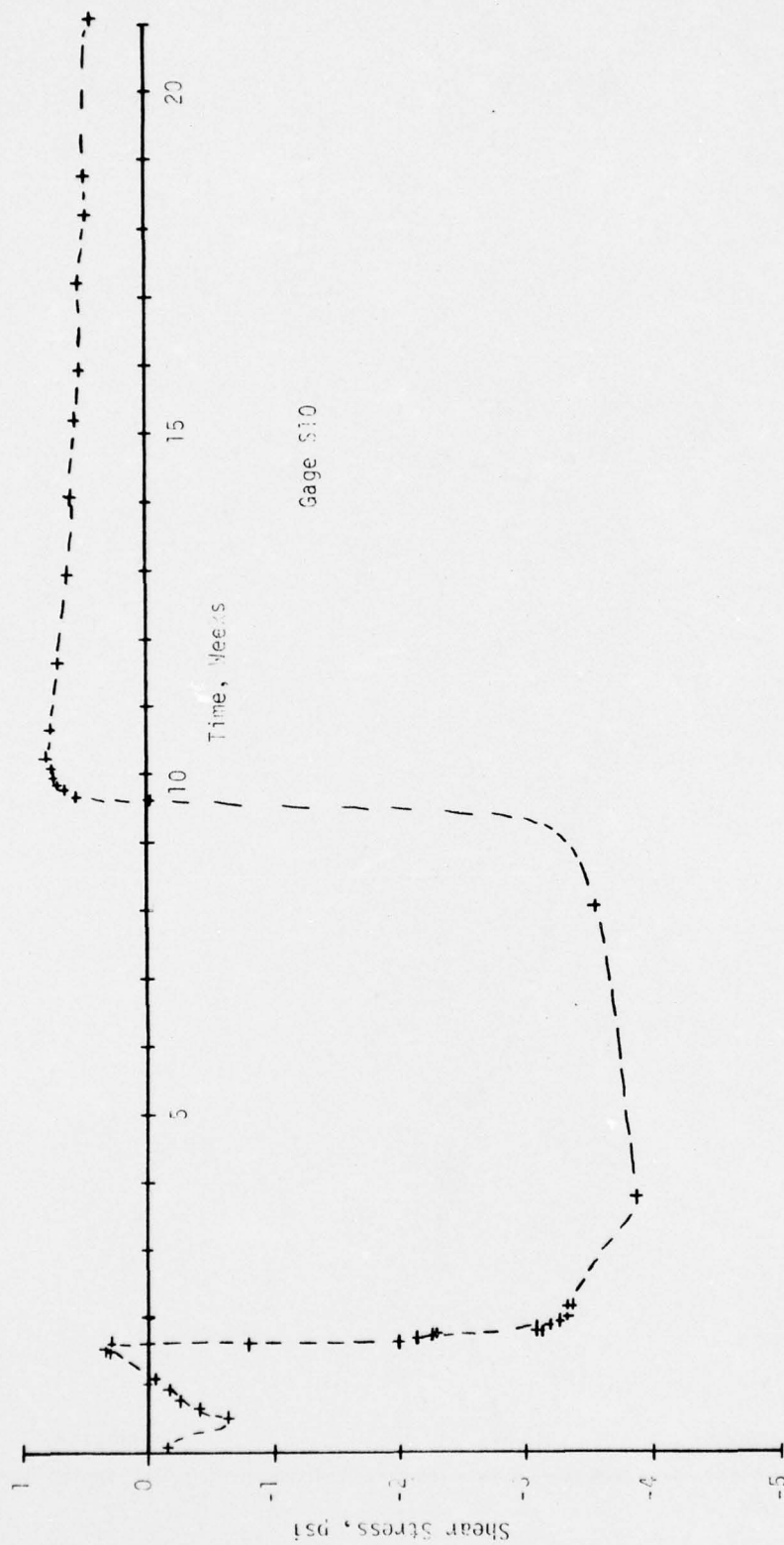
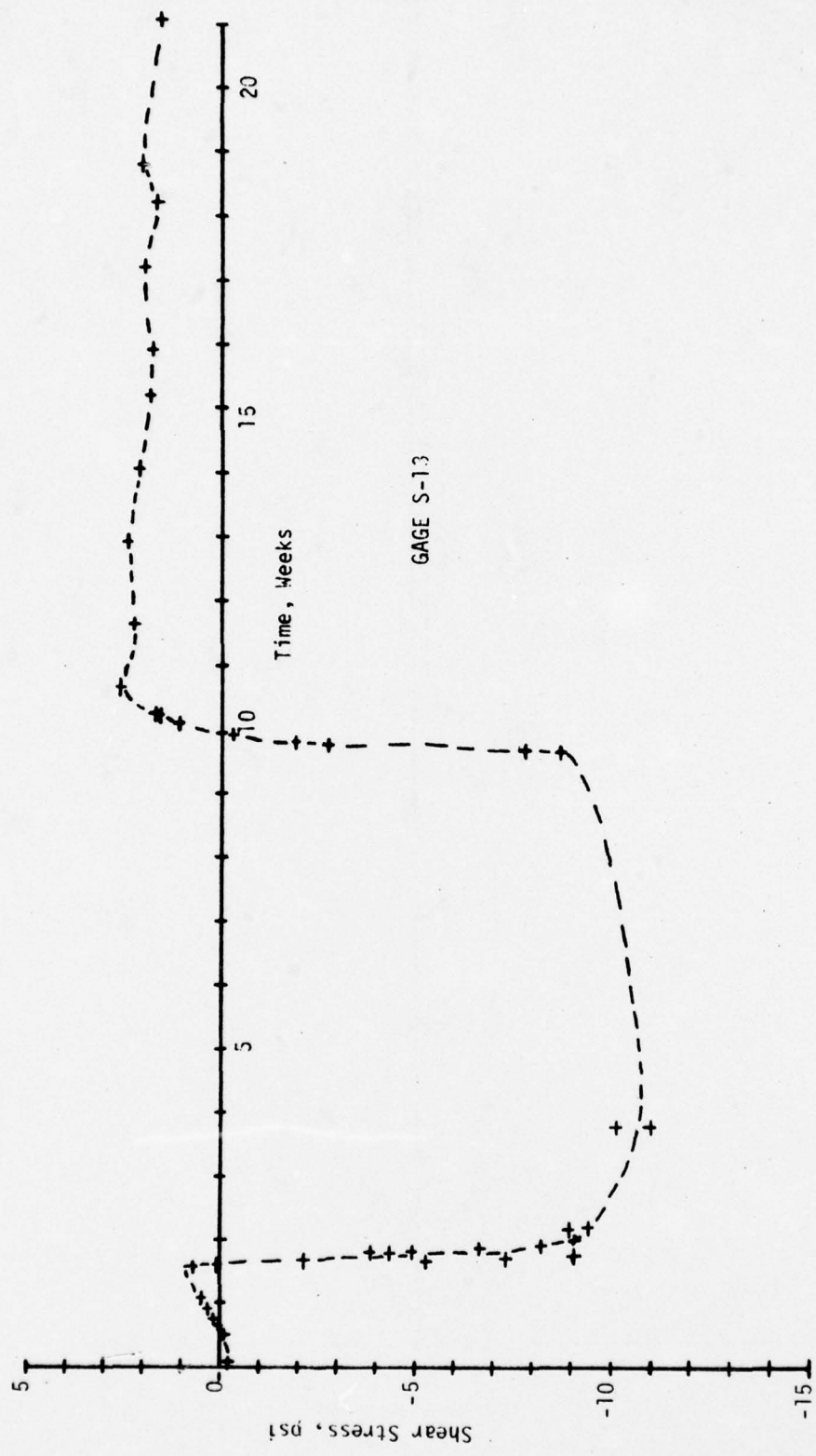
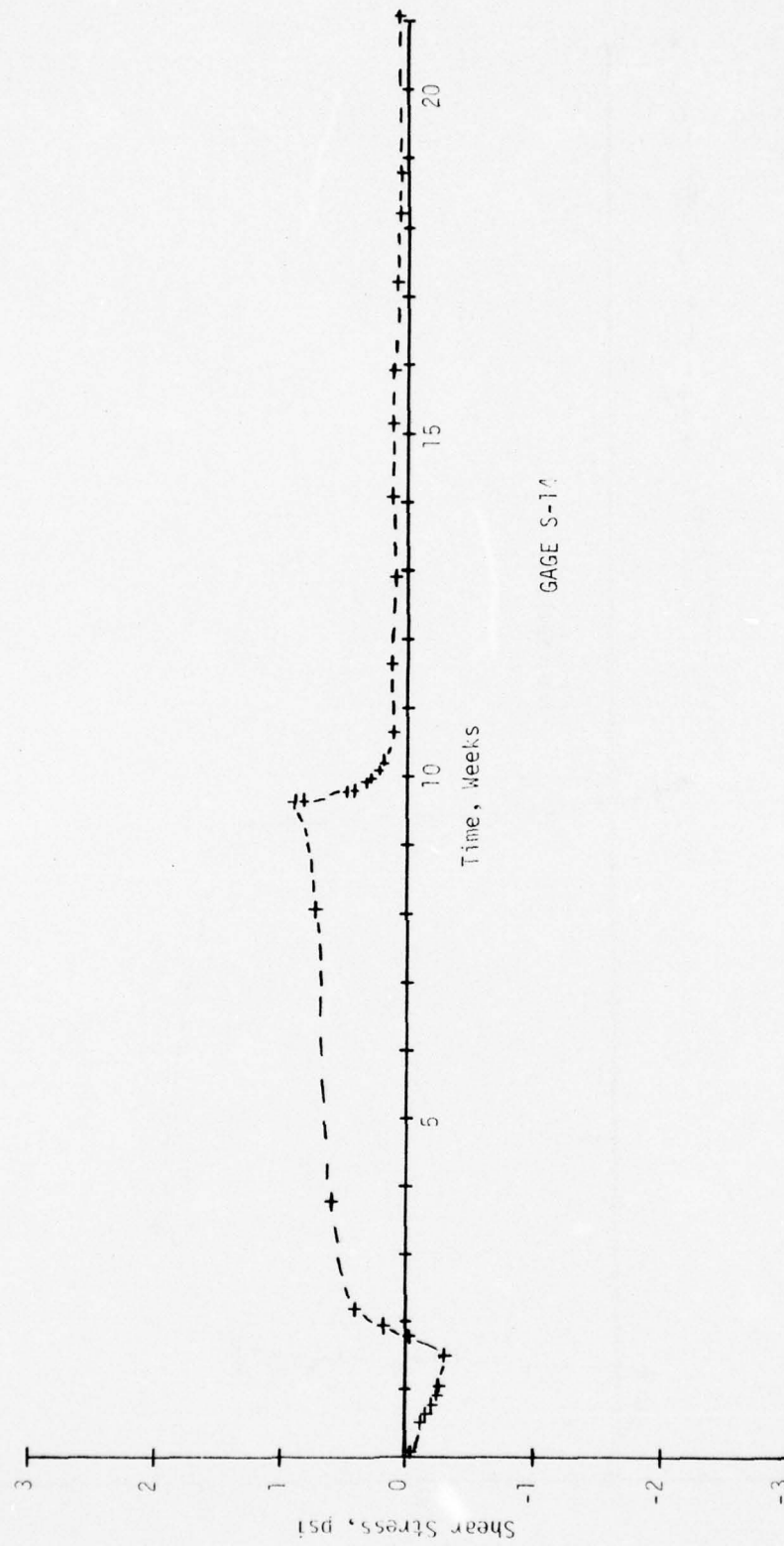


FIGURE H-25. FULL-SCALE MOTOR NO. 1, CURE AND COOL-DOWN, 110°F CONDITIONING



GAGE S-13

FIGURE H-26. FULL-SCALE MOTOR NO. 1, CURE AND COOLDOWN, 110°F CONDITIONING



GAGE S-10

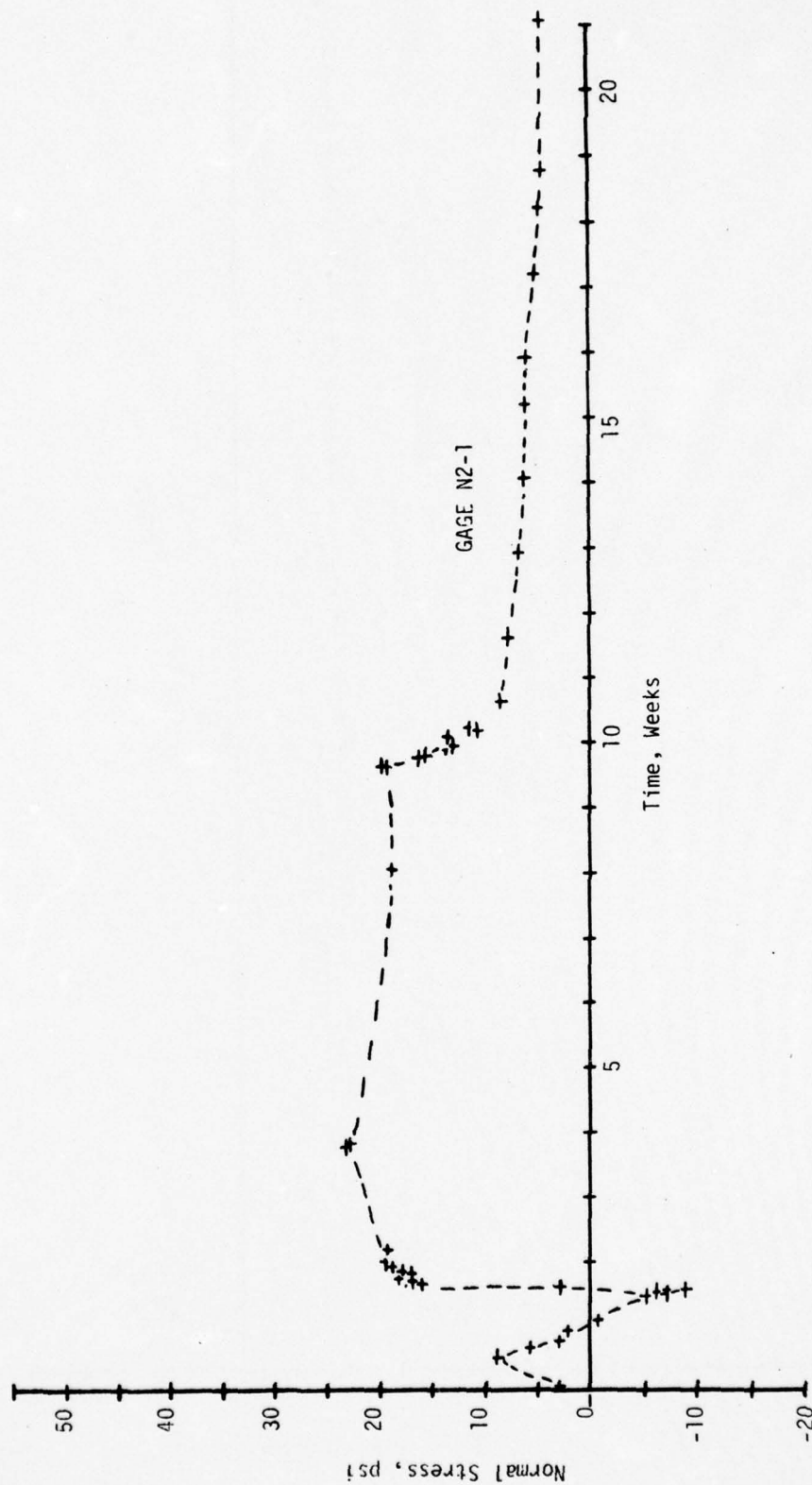
Time, Weeks

Shear Stress, psi

H-38

FIGURE H-27. FULL-SCALE MOTOR NO. 1, CURE AND COOLDOWN, 110 F CONDITIONING





H-39

FIGURE H-28. FULL-SCALE MOTOR NO. 1, CURE AND COOLDOWN, 110°F CONDITIONING

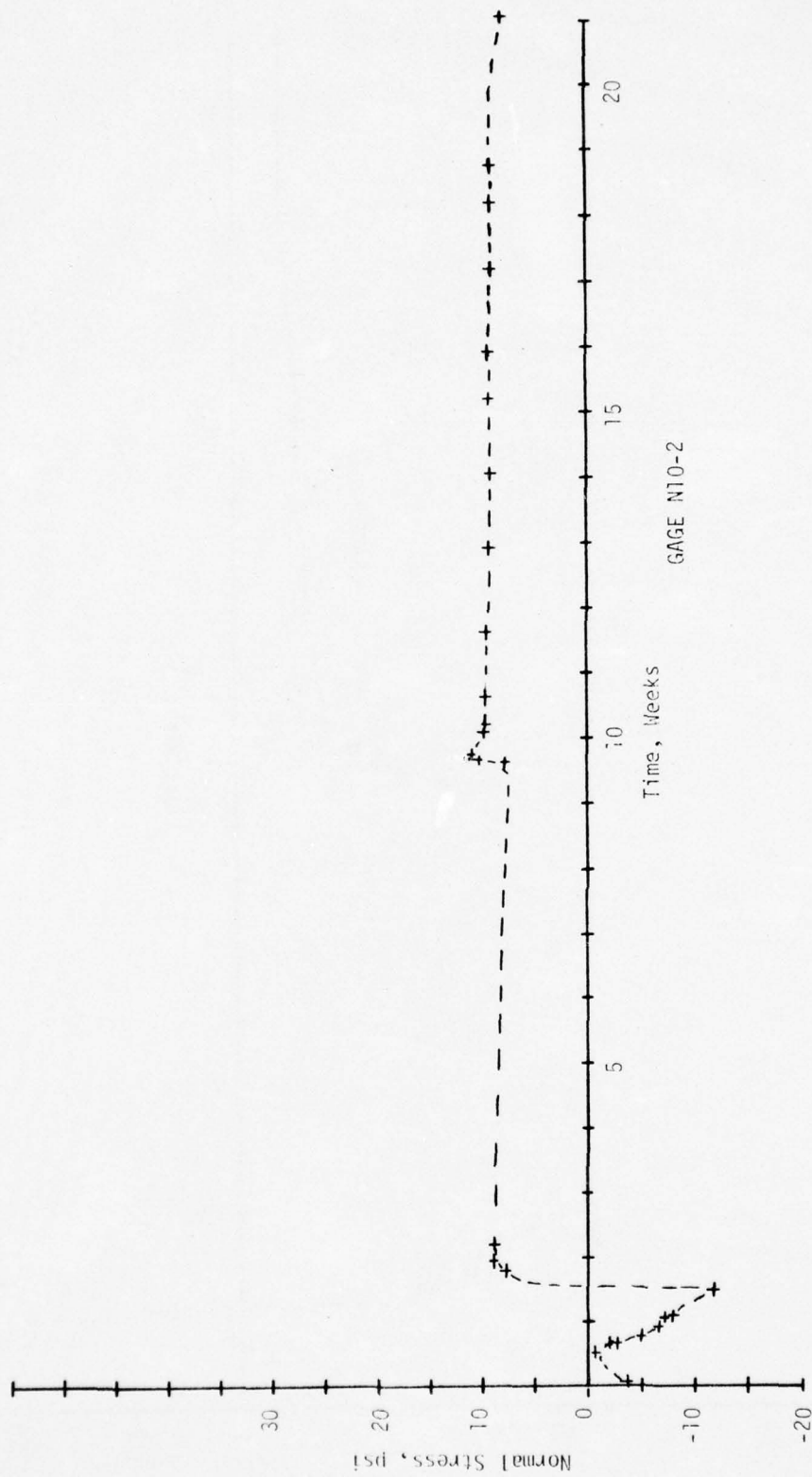


FIGURE H-29. FULL-SCALE MOTOR NO. 1, CURE AND COOLDOWN, 110°F CONDITIONING

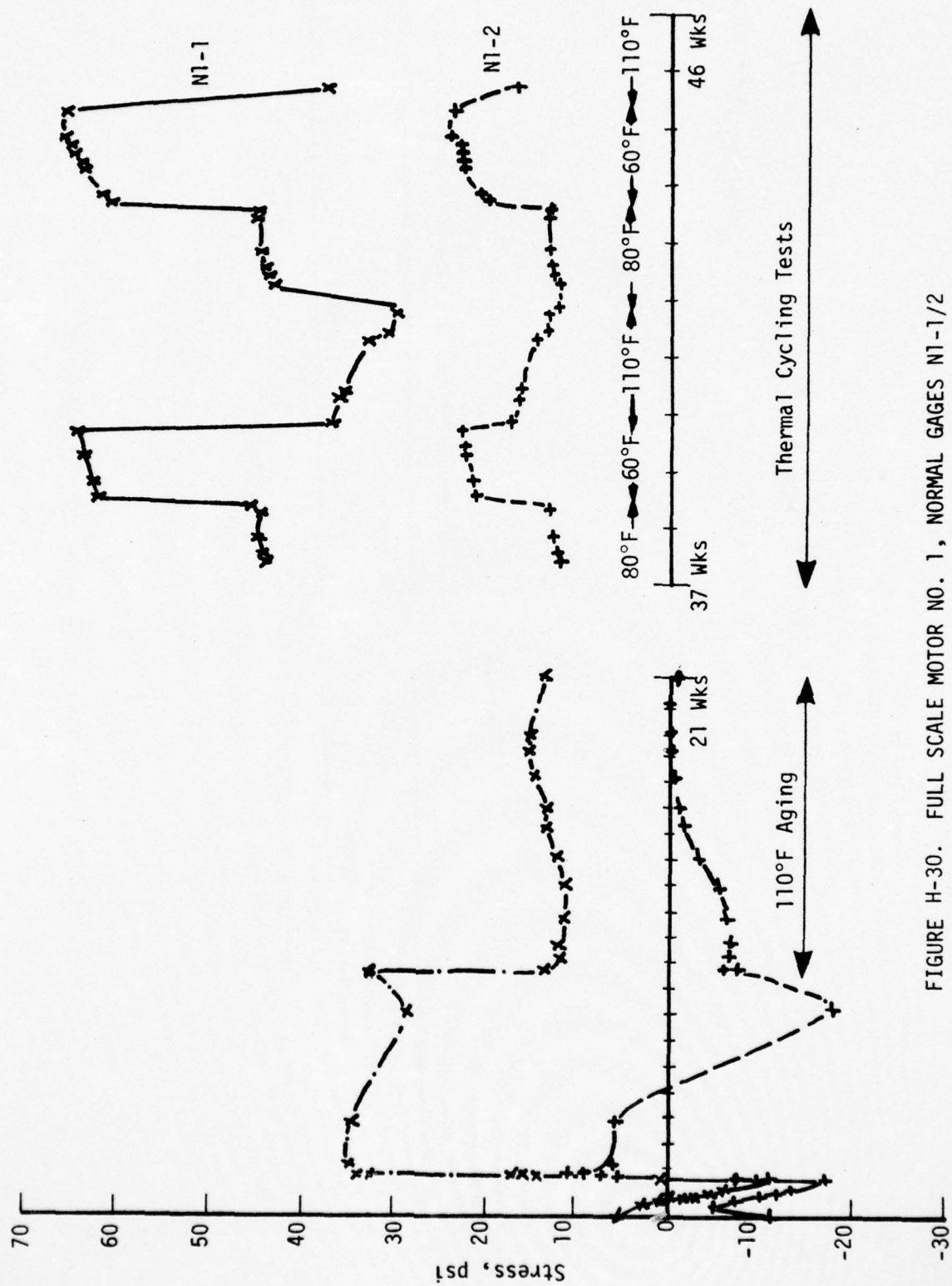


FIGURE H-30. FULL SCALE MOTOR NO. 1, NORMAL GAGES NI-1/2



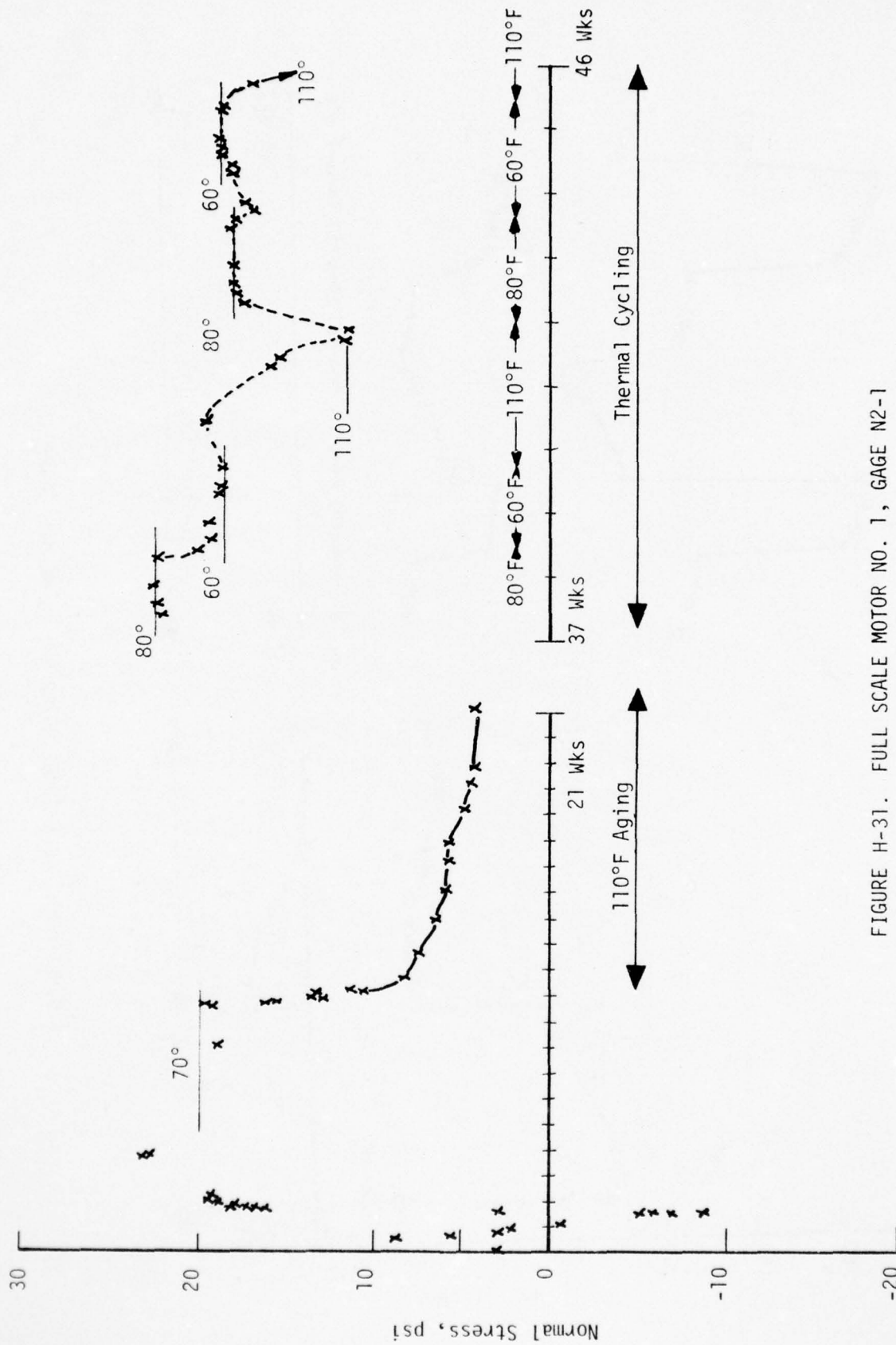


FIGURE H-31. FULL SCALE MOTOR NO. 1, GAGE N2-1

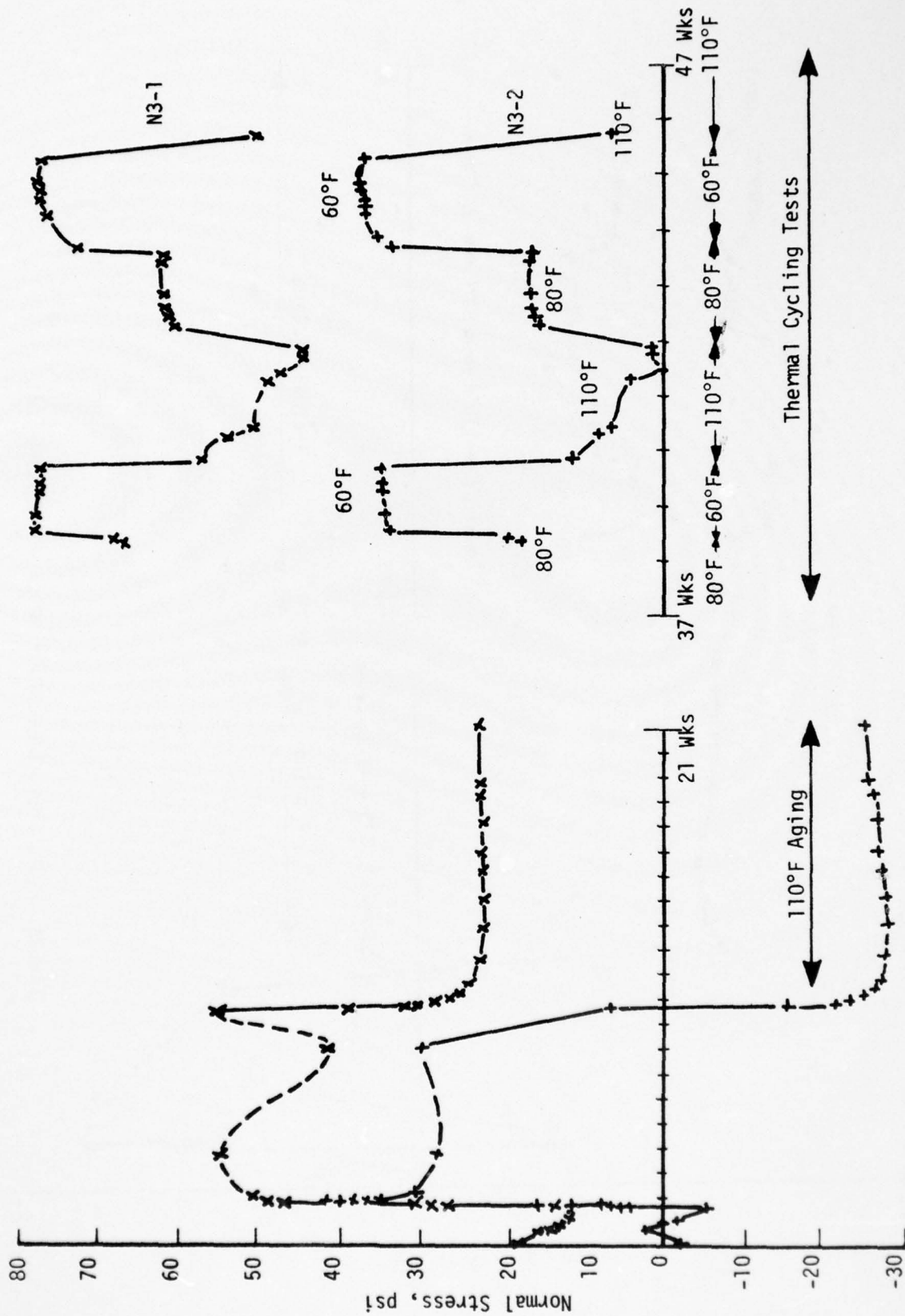


FIGURE H-32. FULL SCALE MOTOR NO. 1 - GAGES N3-1/2

AD-A032 637

AEROJET SOLID PROPULSION CO SACRAMENTO CALIF  
FLEXIBLE CASE-GRAIN INTERACTION IN BALLISTIC WEAPON SYSTEMS. VO--ETC(U)  
OCT 76 K W BILLS, S W JANG, H LEEMING

F/6 21/9.2

F04611-72-C-0055

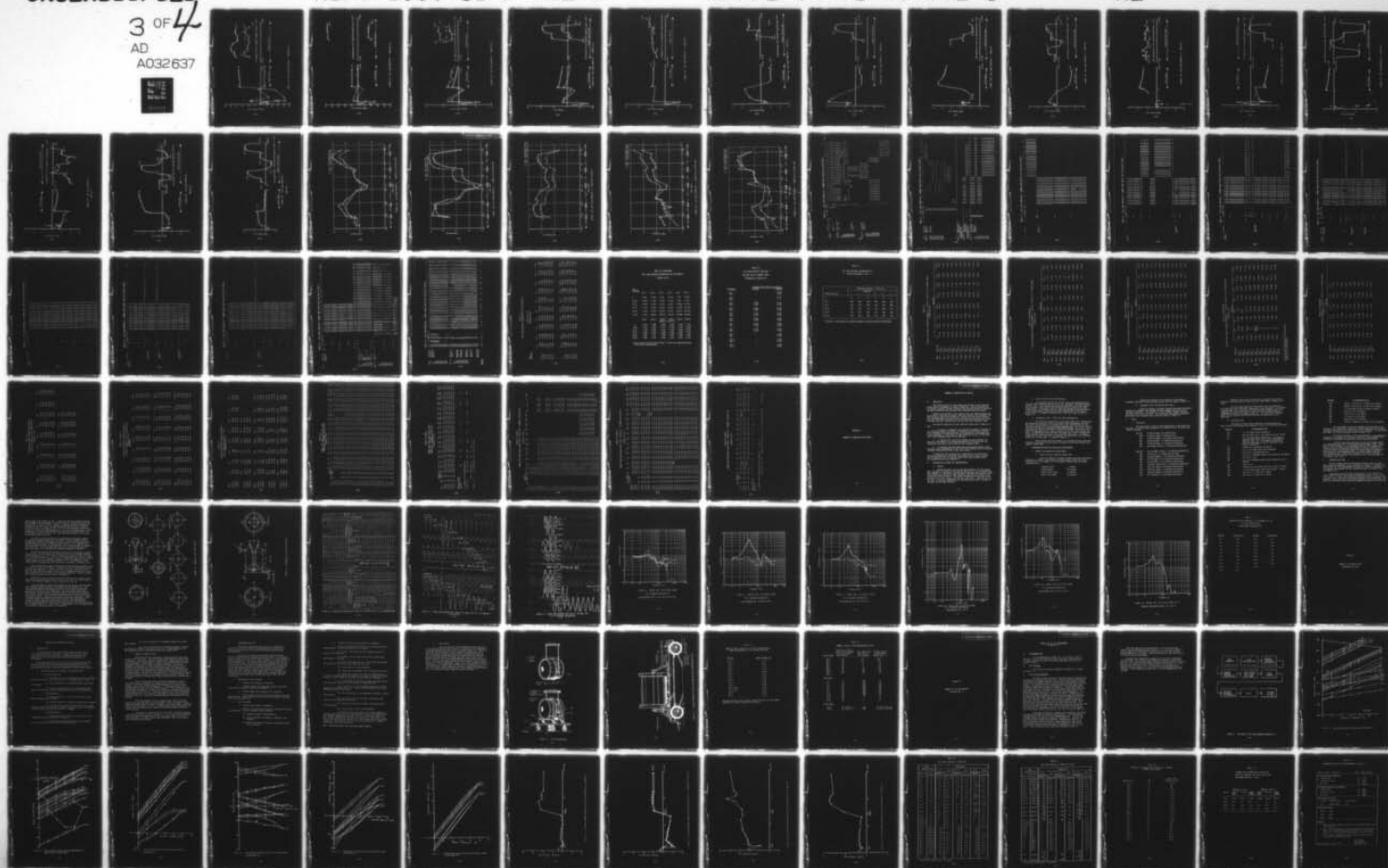
UNCLASSIFIED

ASPC-1953-81-F-VOL-3

AFRPL-TR-76-57-VOL-3

NL

3 OF 4  
AD  
A032637





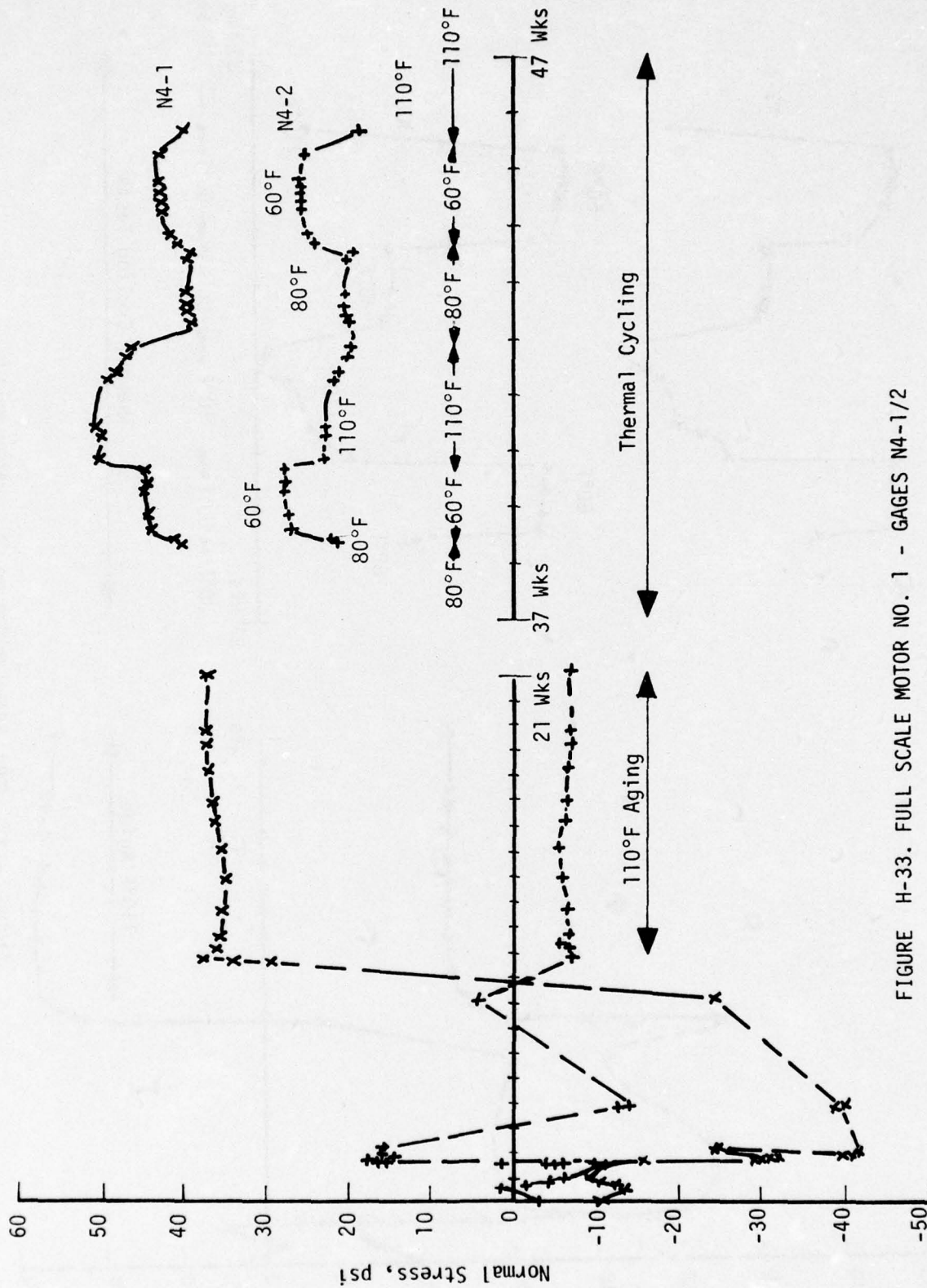


FIGURE H-33. FULL SCALE MOTOR NO. 1 - GAGES N4-1/2

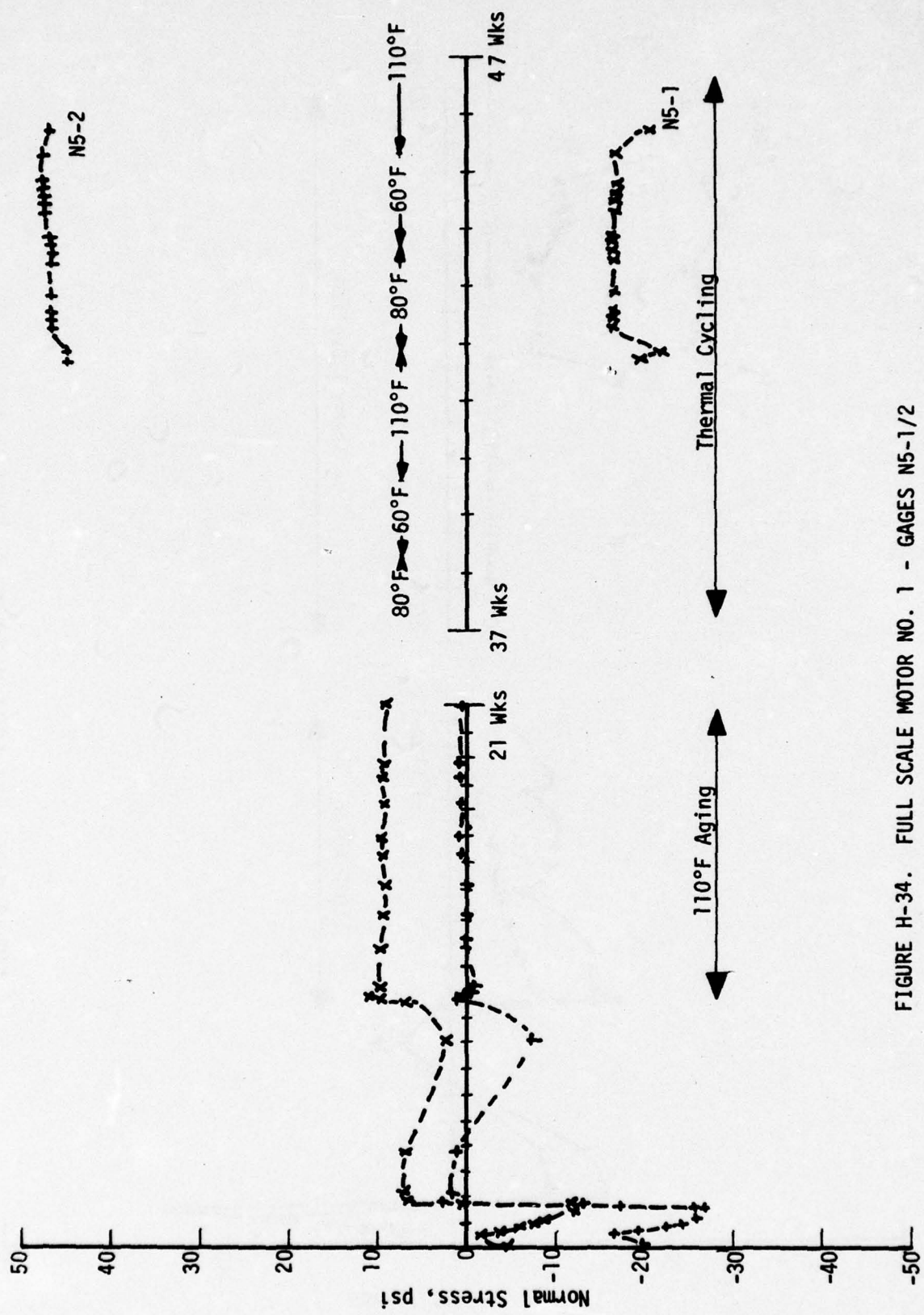


FIGURE H-34. FULL SCALE MOTOR NO. 1 - GAGES N5-1/2

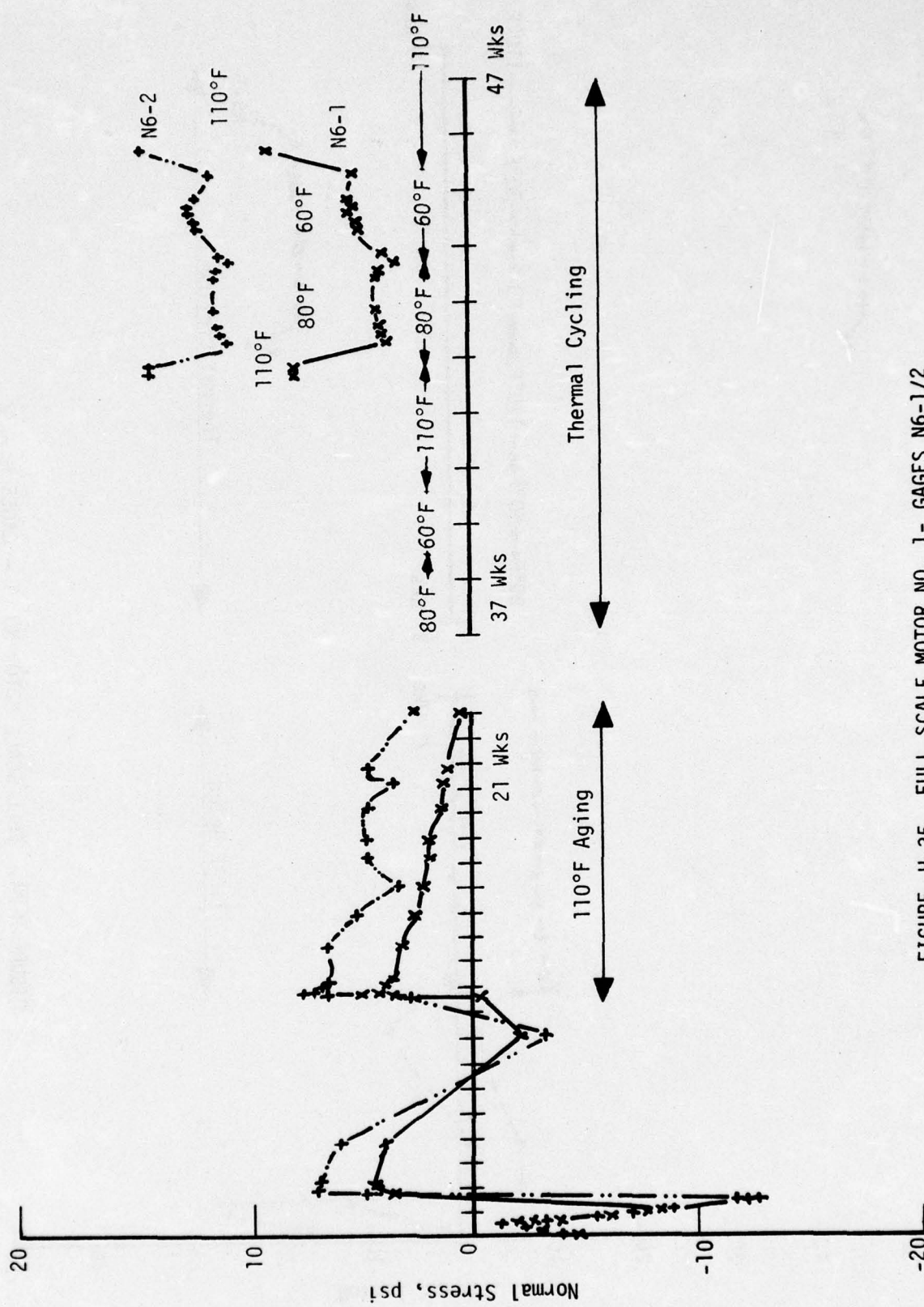


FIGURE H-35. FULL SCALE MOTOR NO. 1- GAGES N6-1/2



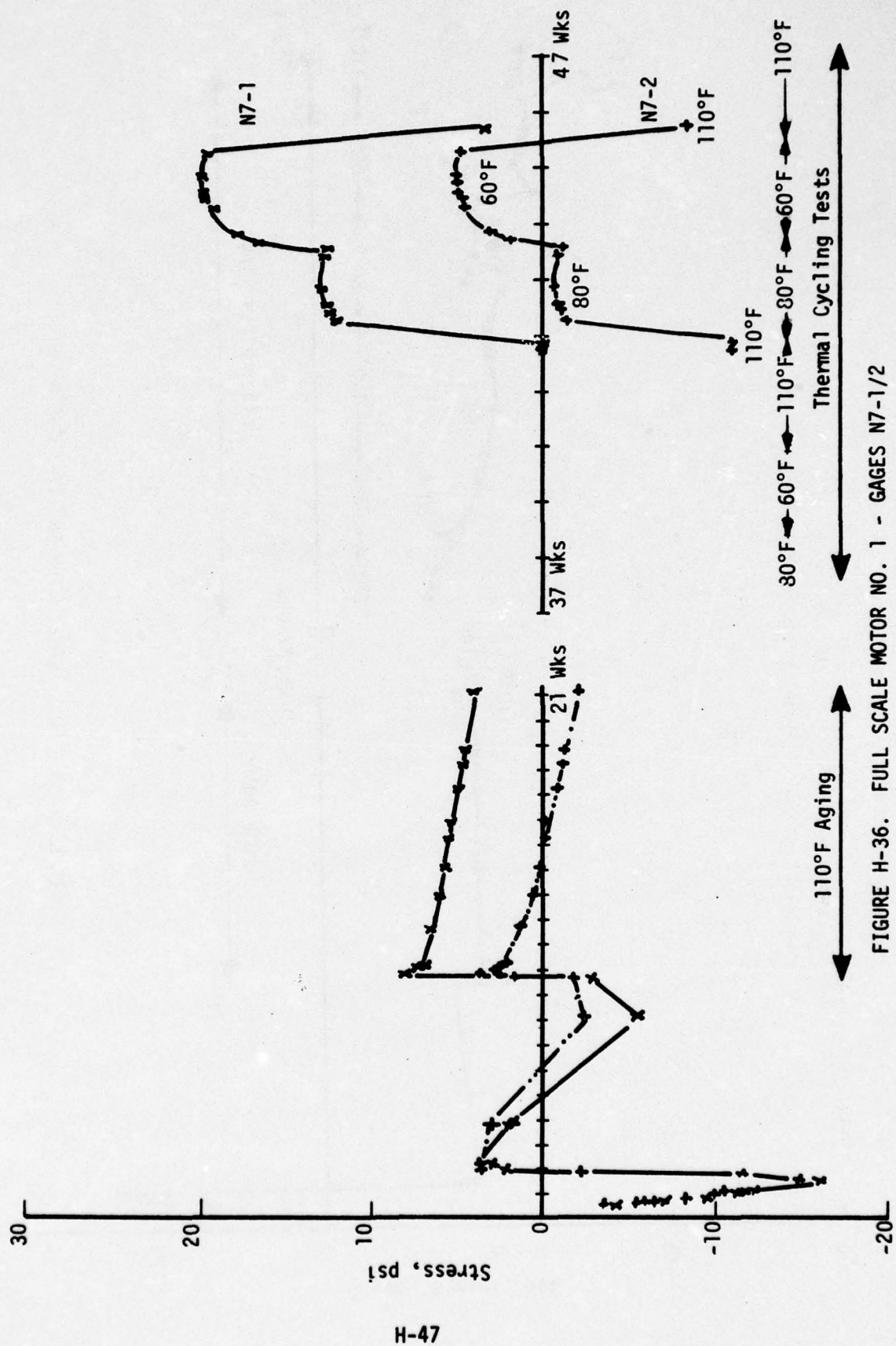


FIGURE H-36. FULL SCALE MOTOR NO. 1 - GAGES N7-1/2

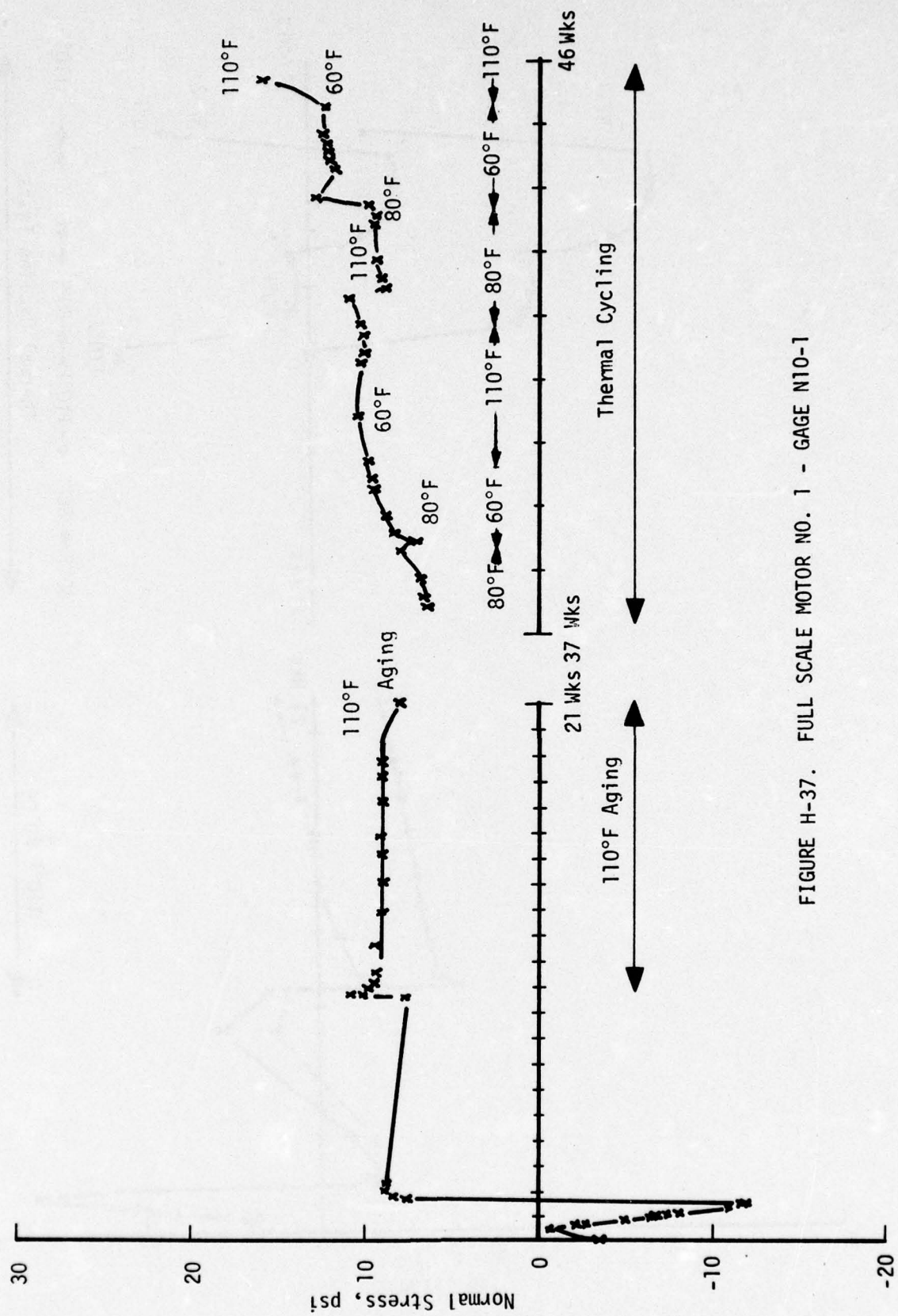


FIGURE H-37. FULL SCALE MOTOR NO. 1 - GAGE N10-1





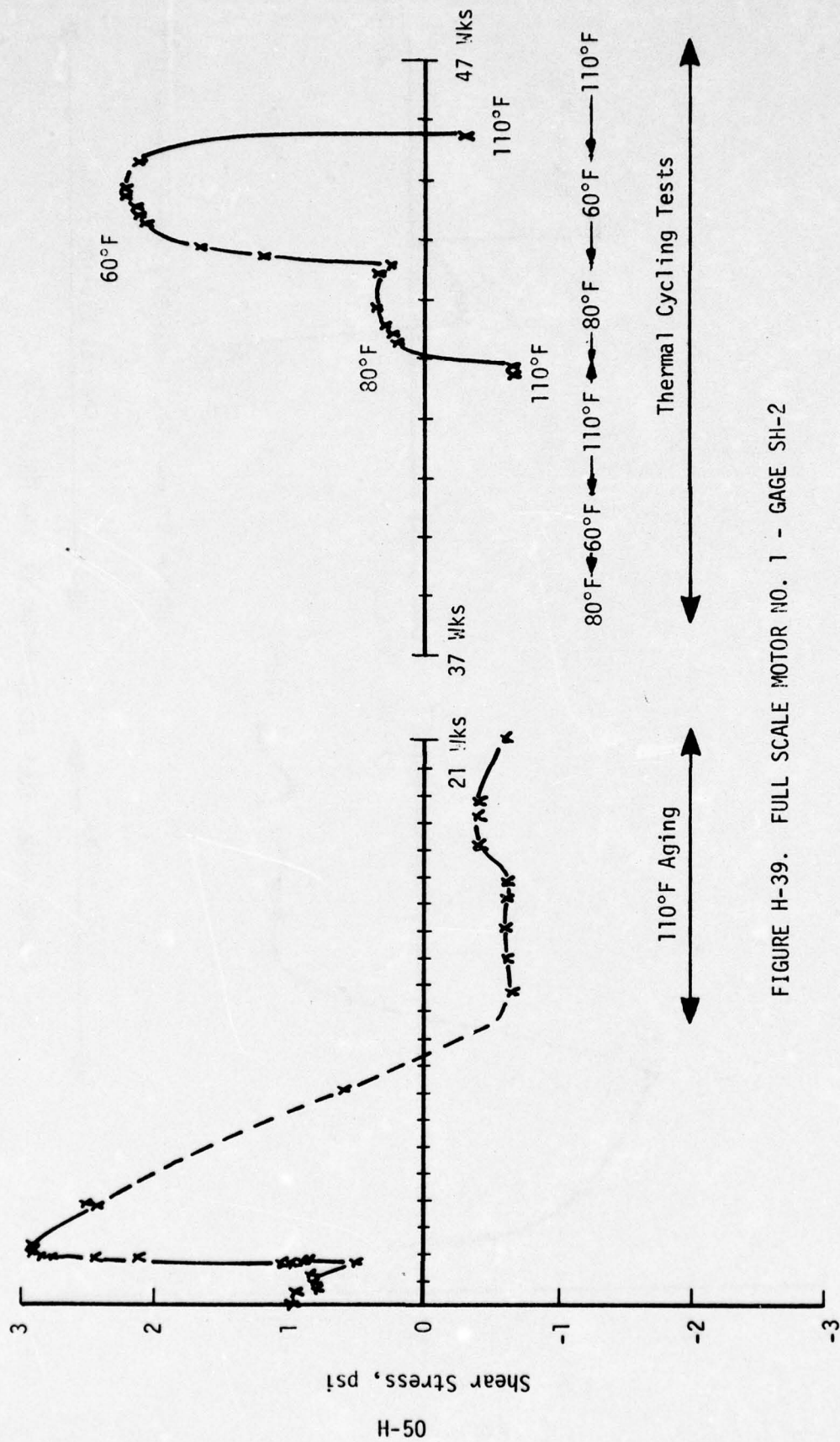


FIGURE H-39. FULL SCALE MOTOR NO. 1 - GAGE SH-2

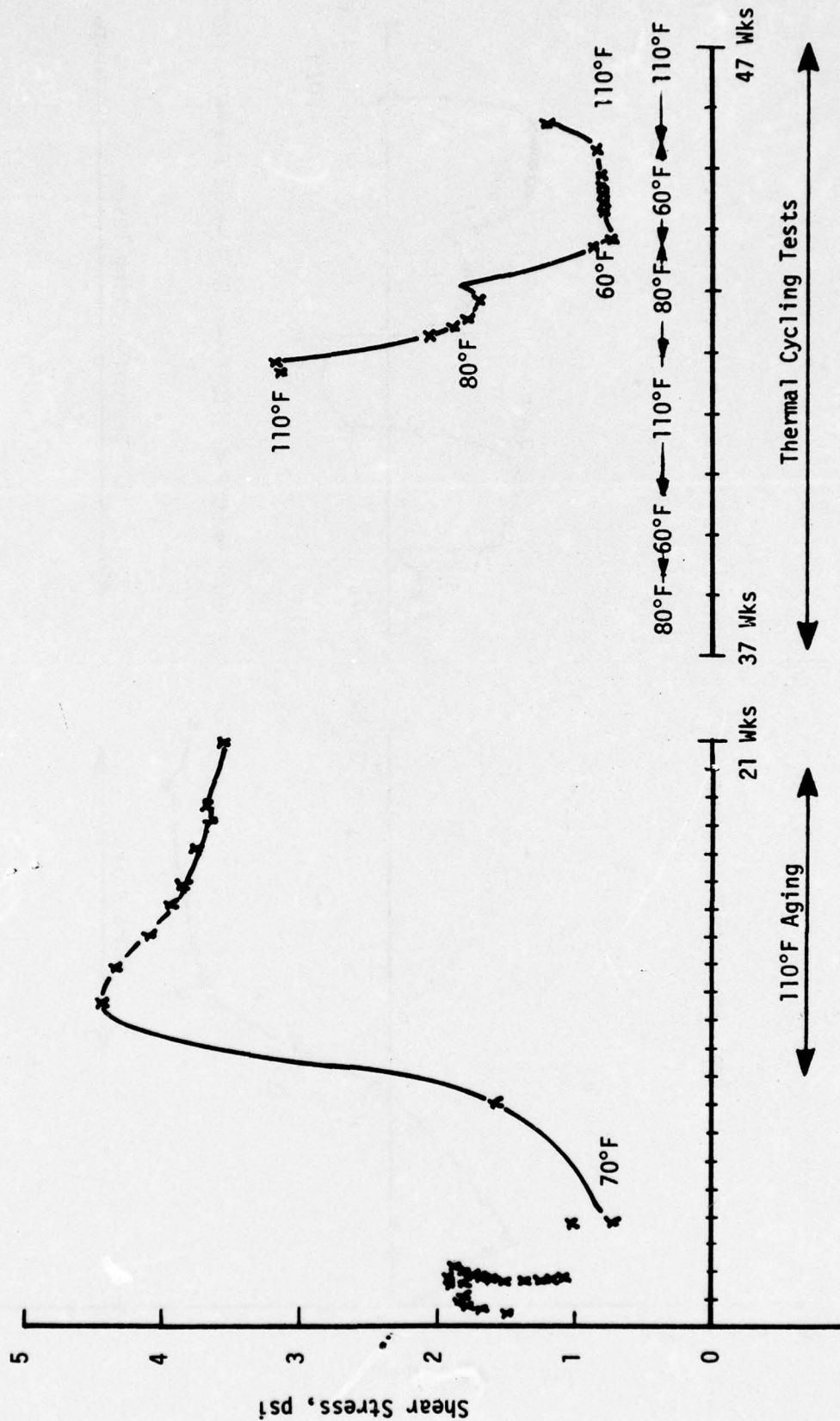


FIGURE H-40. FULL SCALE MOTOR NO. 1 - GAGE SH-4

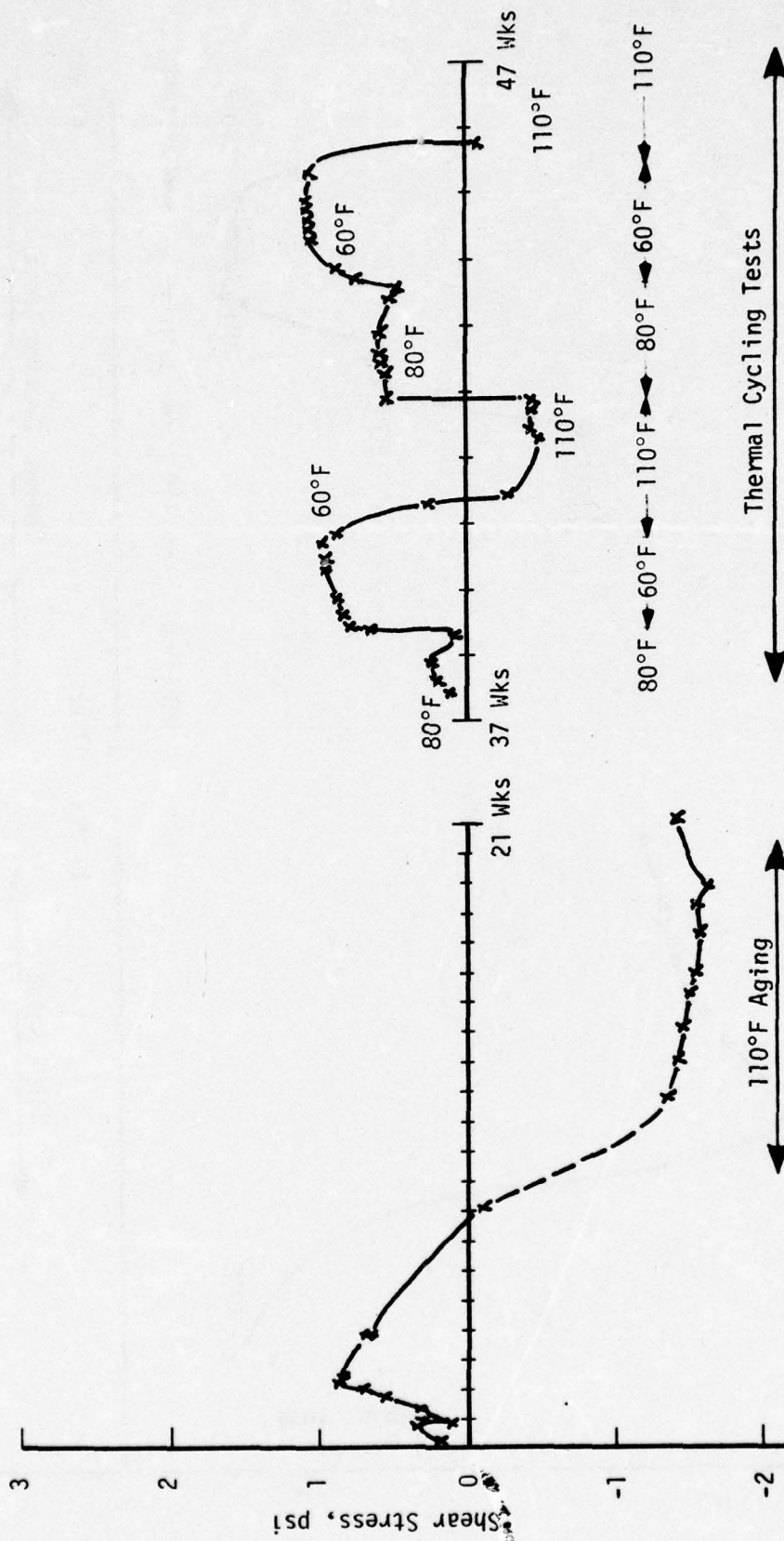


FIGURE H-41. FULL SCALE MOTOR NO. 1 - GAGE SH-5



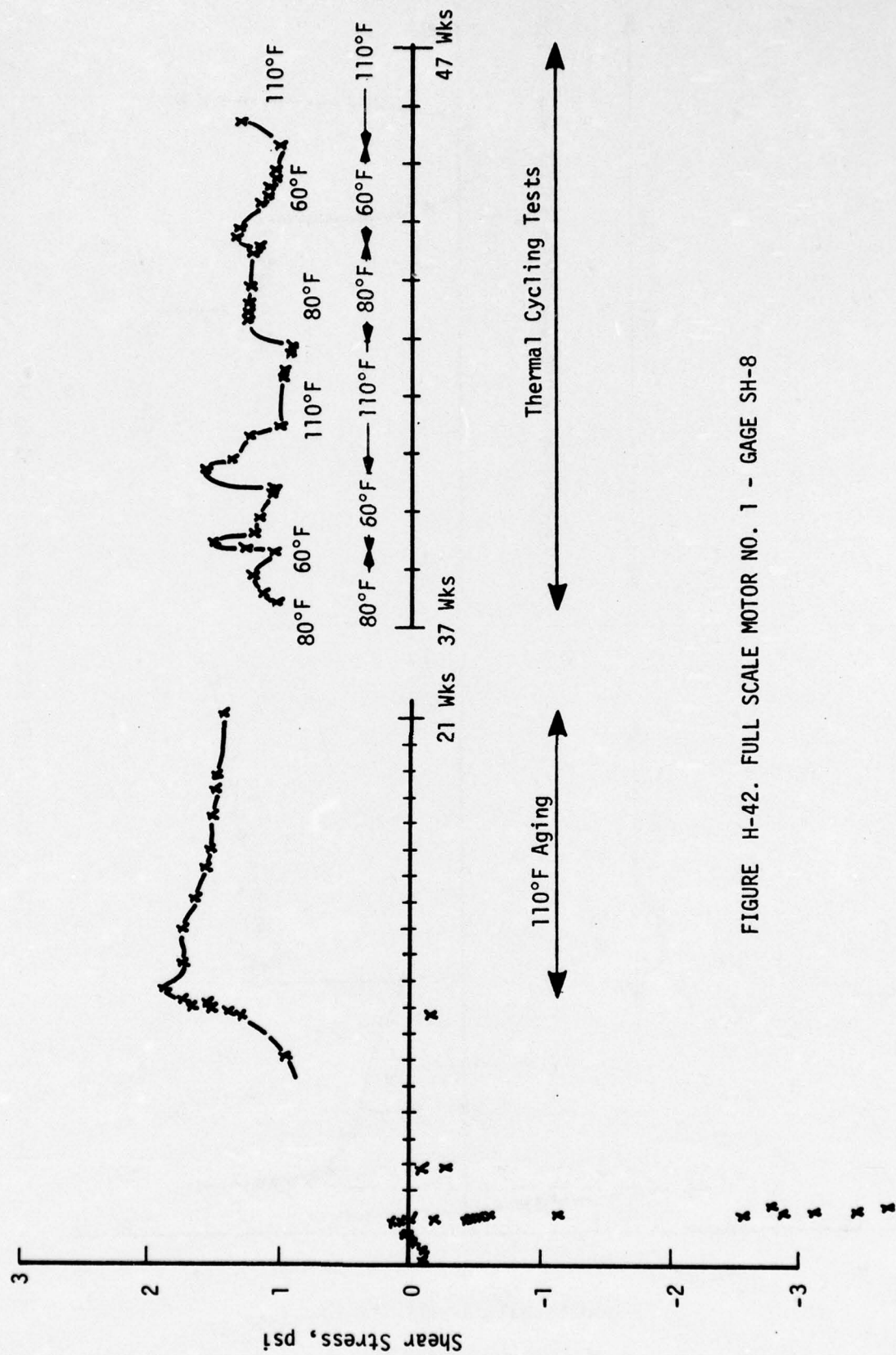
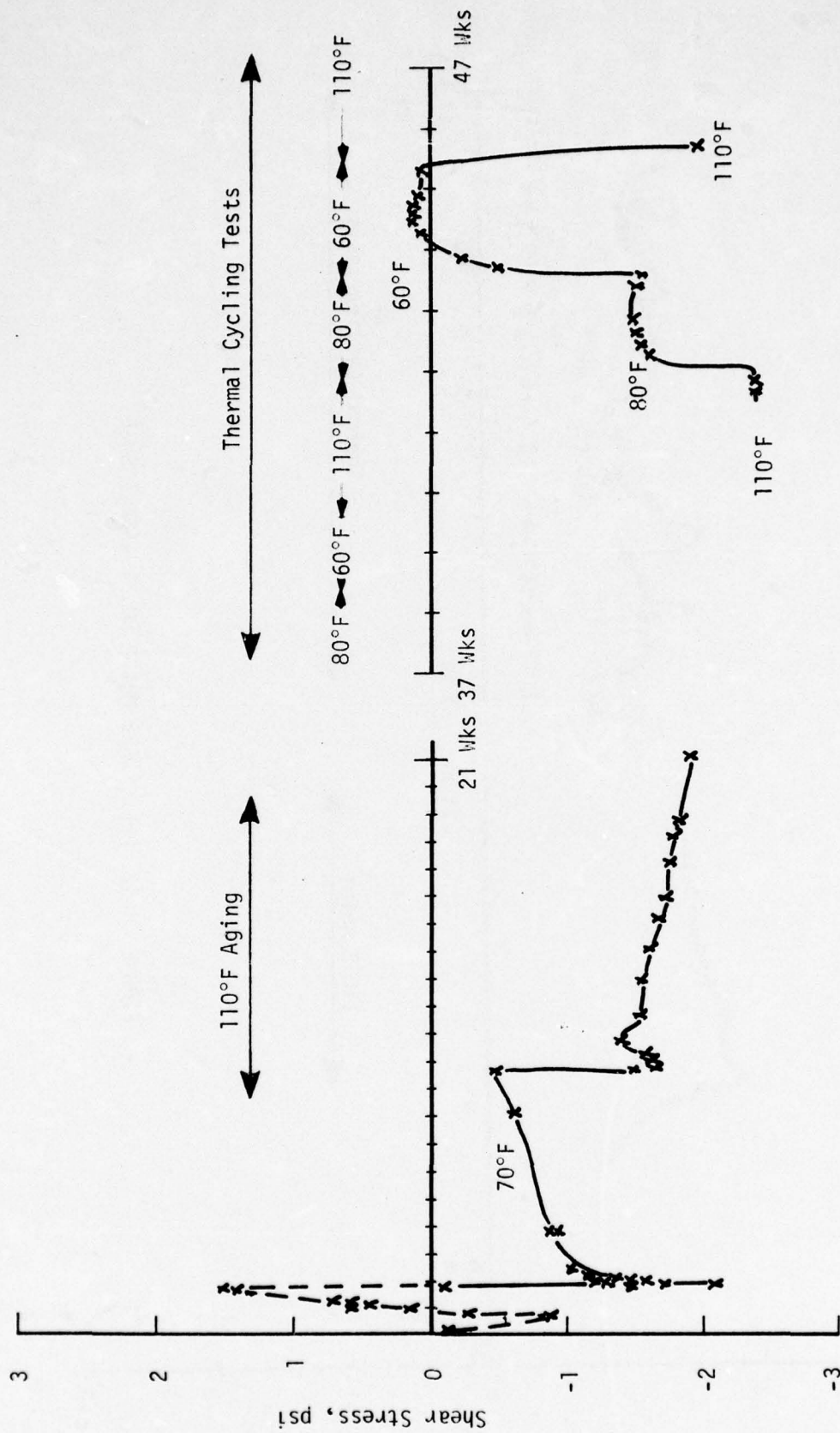
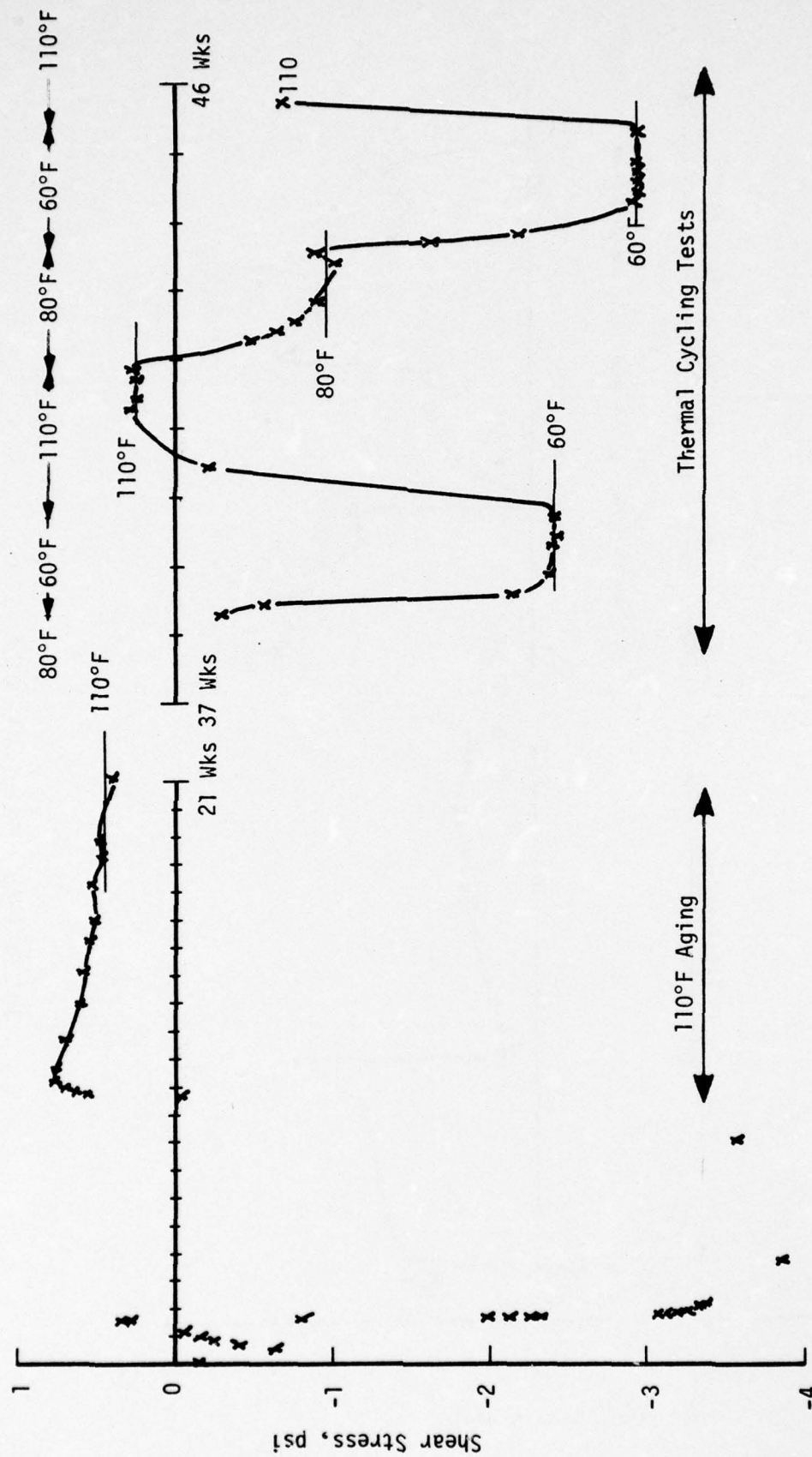


FIGURE H-42. FULL SCALE MOTOR NO. 1 - GAGE SH-8



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FIGURE H-43. FULL SCALE MOTOR NO. 1 - GAGE SH-9



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FIGURE H-44. FULL SCALE MOTOR NO. 1 - GAGE SH-10



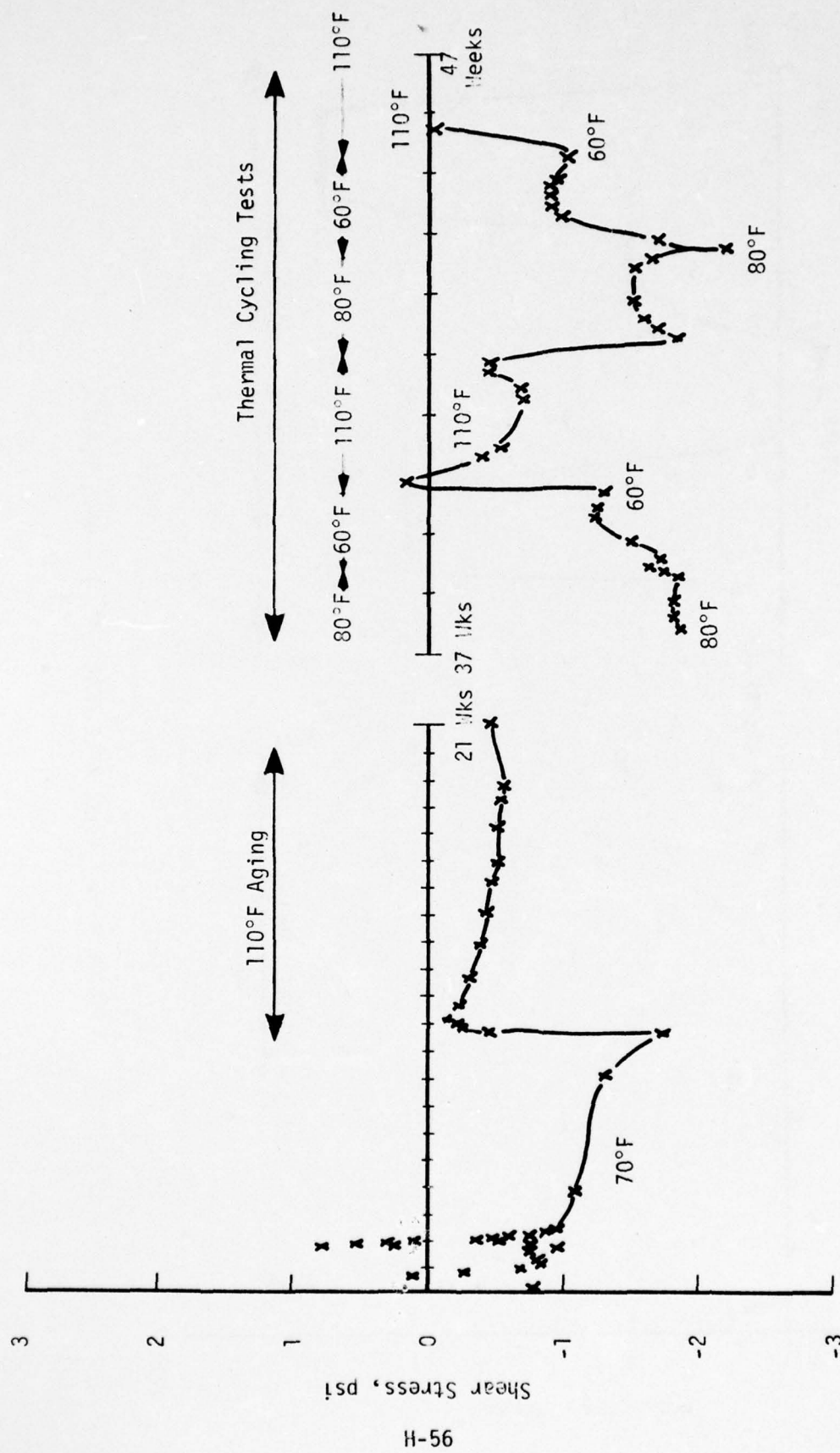


FIGURE H-45. FULL SCALE MOTOR NO. 1

GAGE SH-11

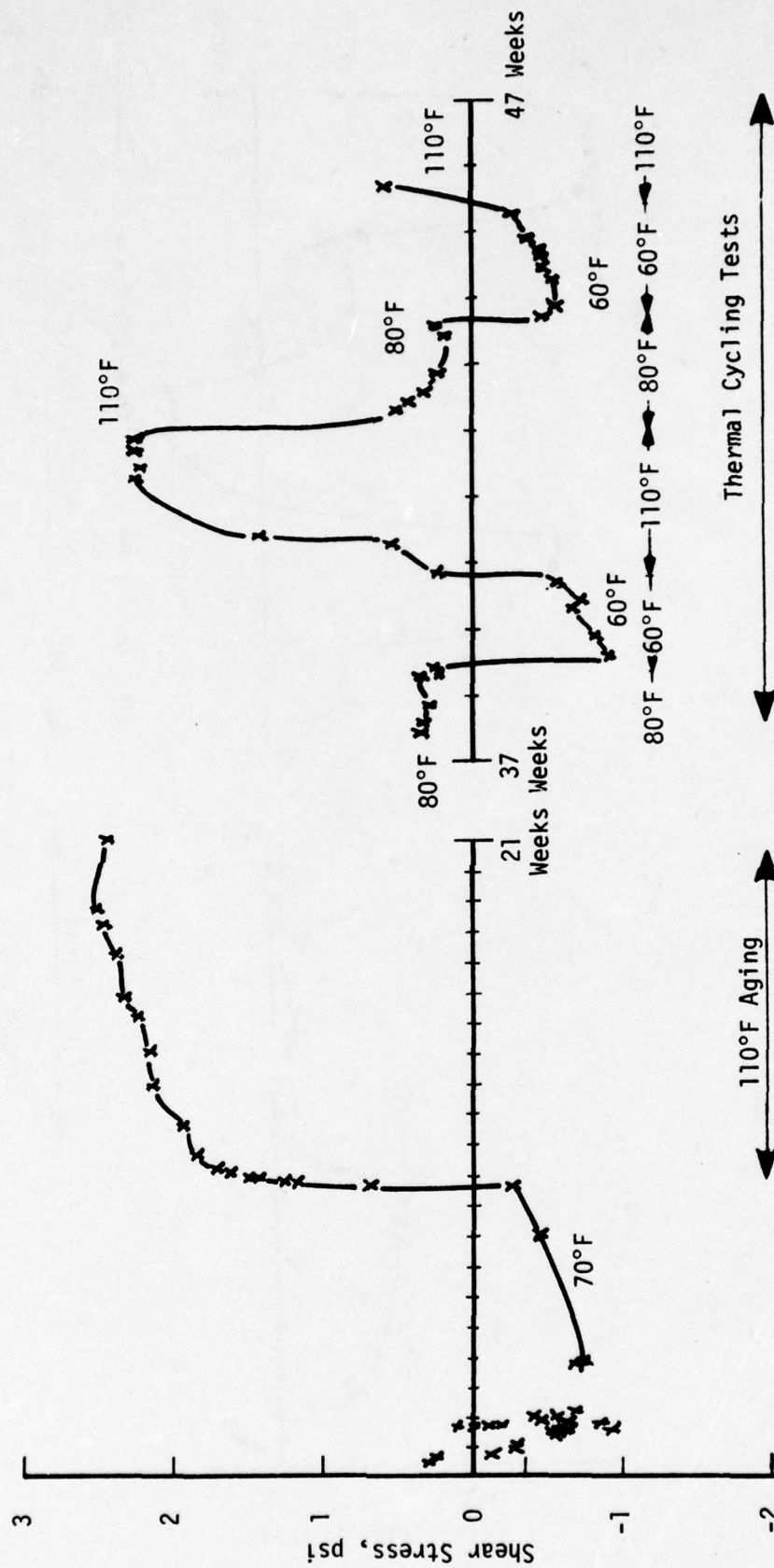


FIGURE H-46. FULL SCALE MOTOR NO. 1

GAGE SH-12

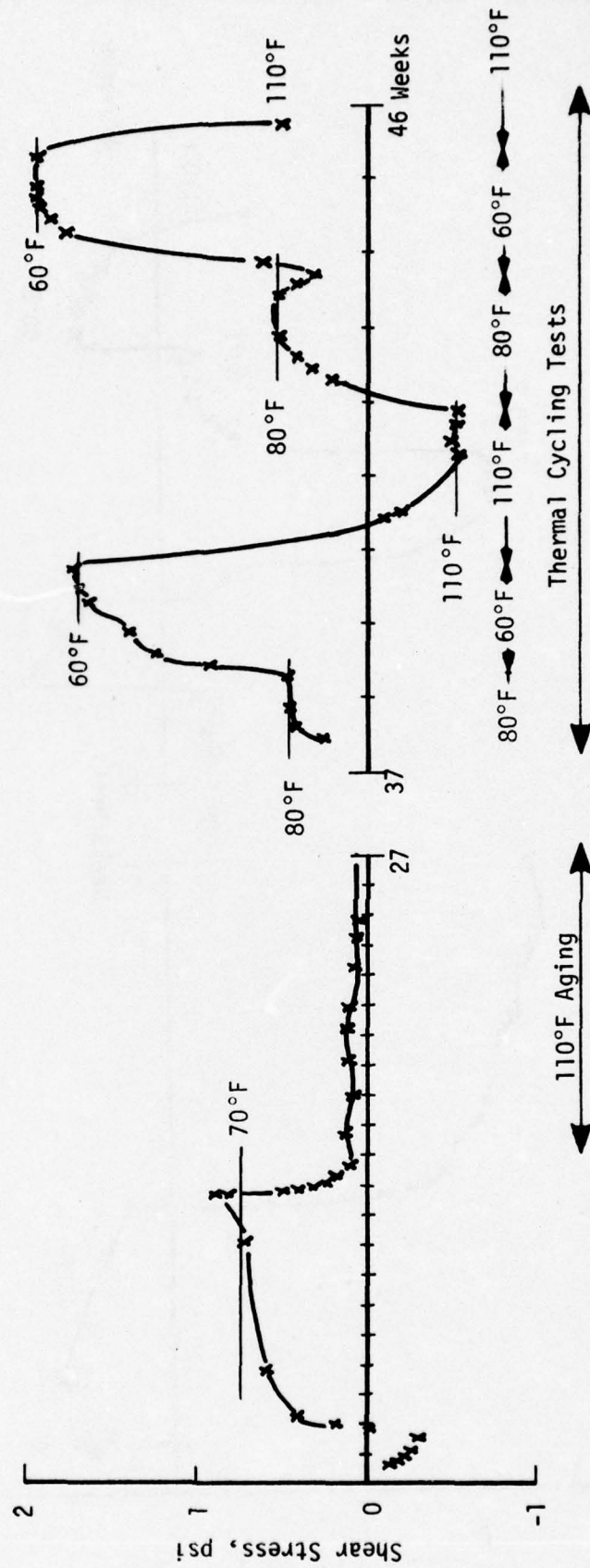


FIGURE H-47. FULL SCALE MOTOR NO. 1

GAGE SH14



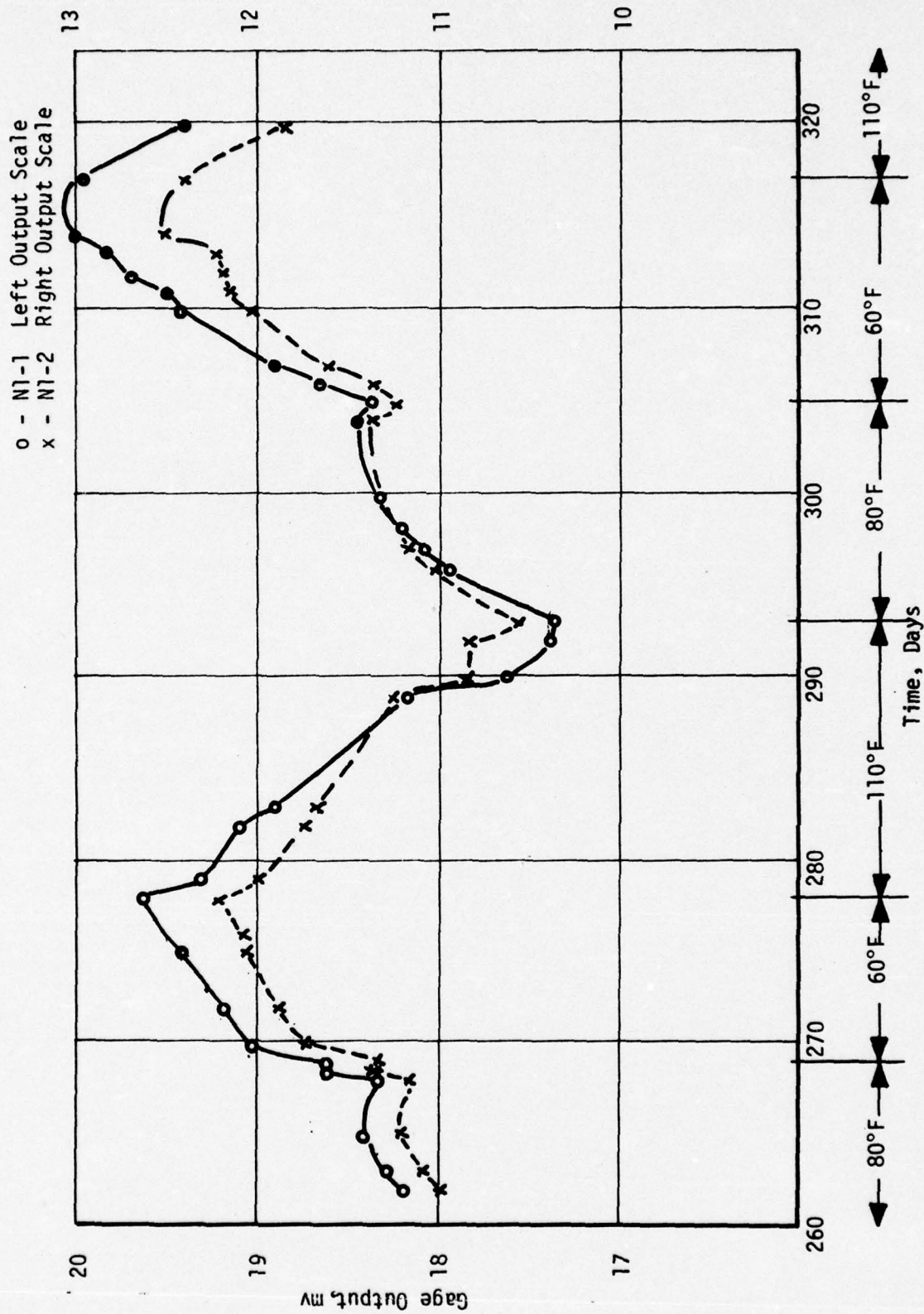


FIGURE H-48. FULL SCALE MOTOR NO. 1 - GAGES N1-1/2 OUTPUT (MV)

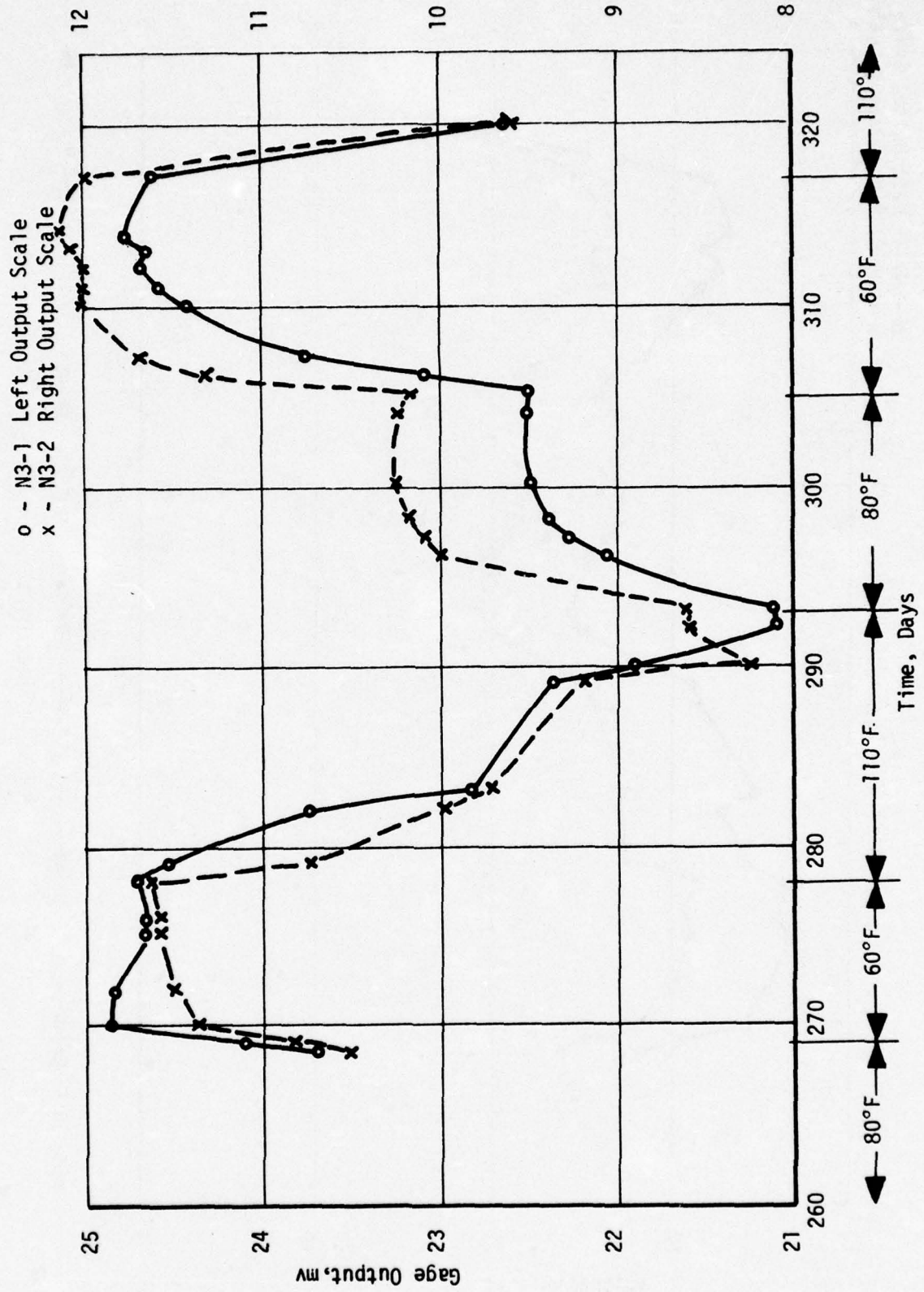


FIGURE H-50. FULL SCALE MOTOR NO. 1 - GAGES N3-1/2 OUTPUT (MV)

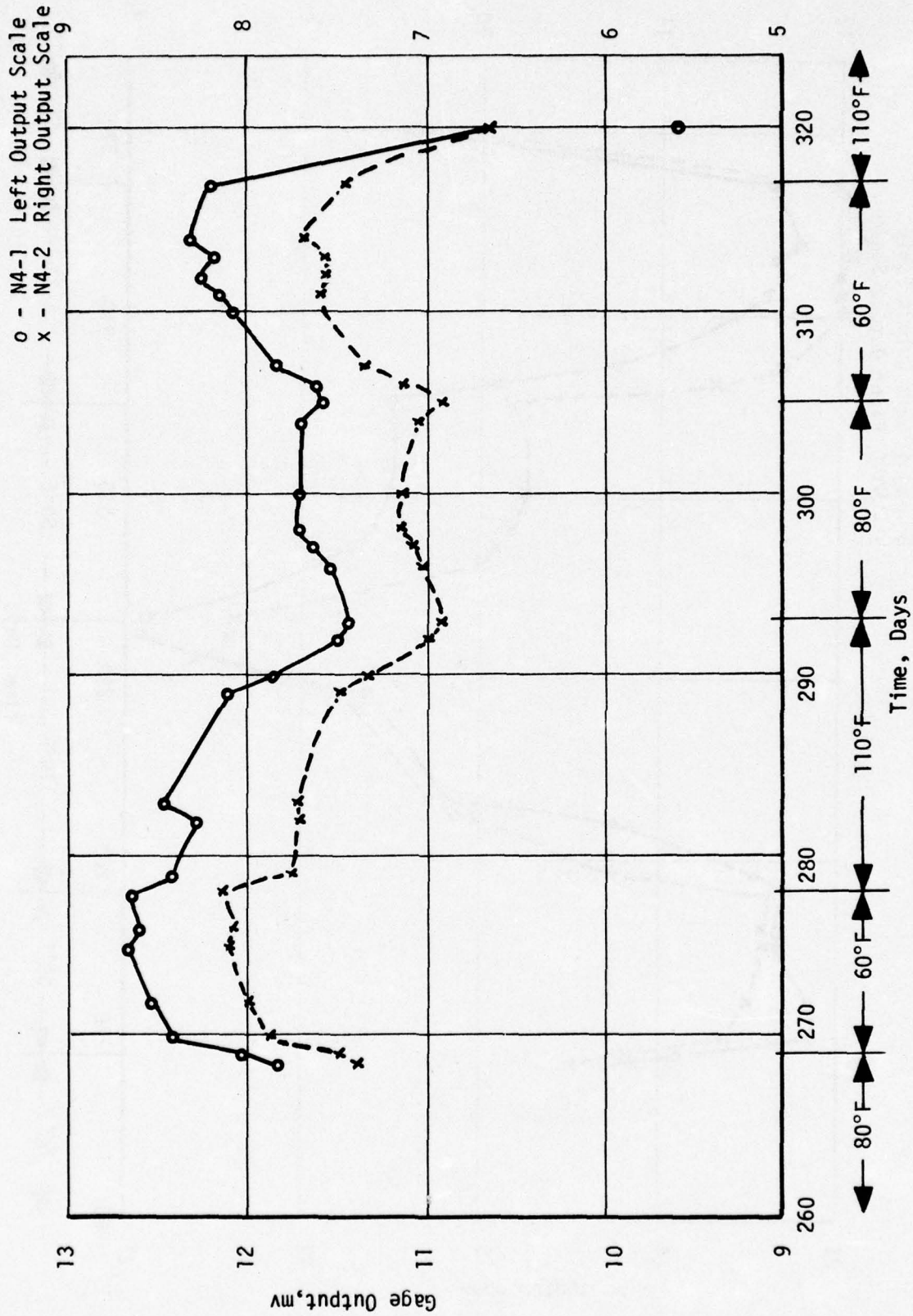


FIGURE H-51. FULL SCALE MOTOR NO. 1 - GAGES N4-1/2 OUTPUT



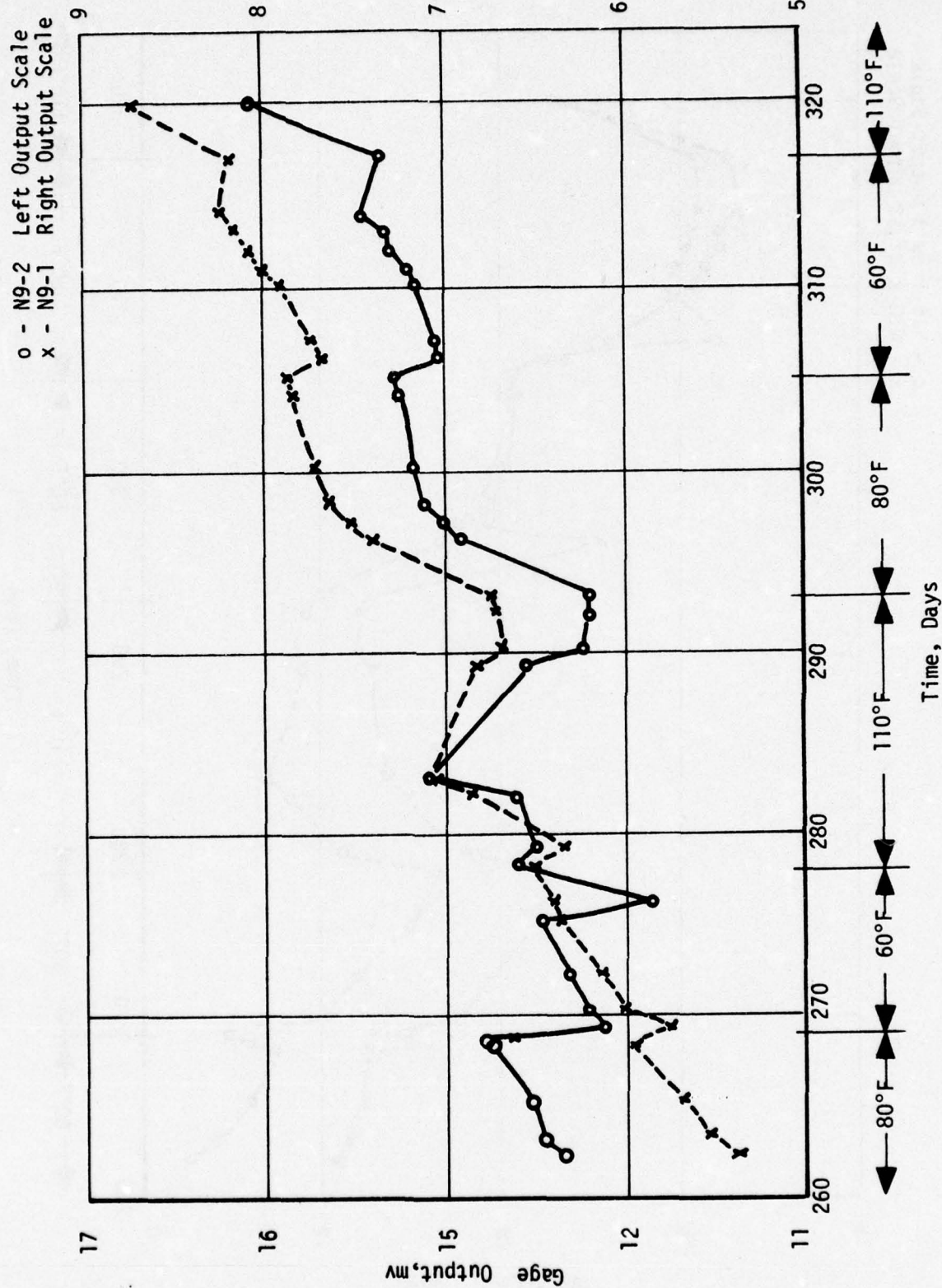


FIGURE H-52. FULL SCALE MOTOR NO. 1 - GAGES N9-1/2 OUTPUT (MV)

o - N10-1 Left Output Scale  
 x - N10-2 Right Output Scale

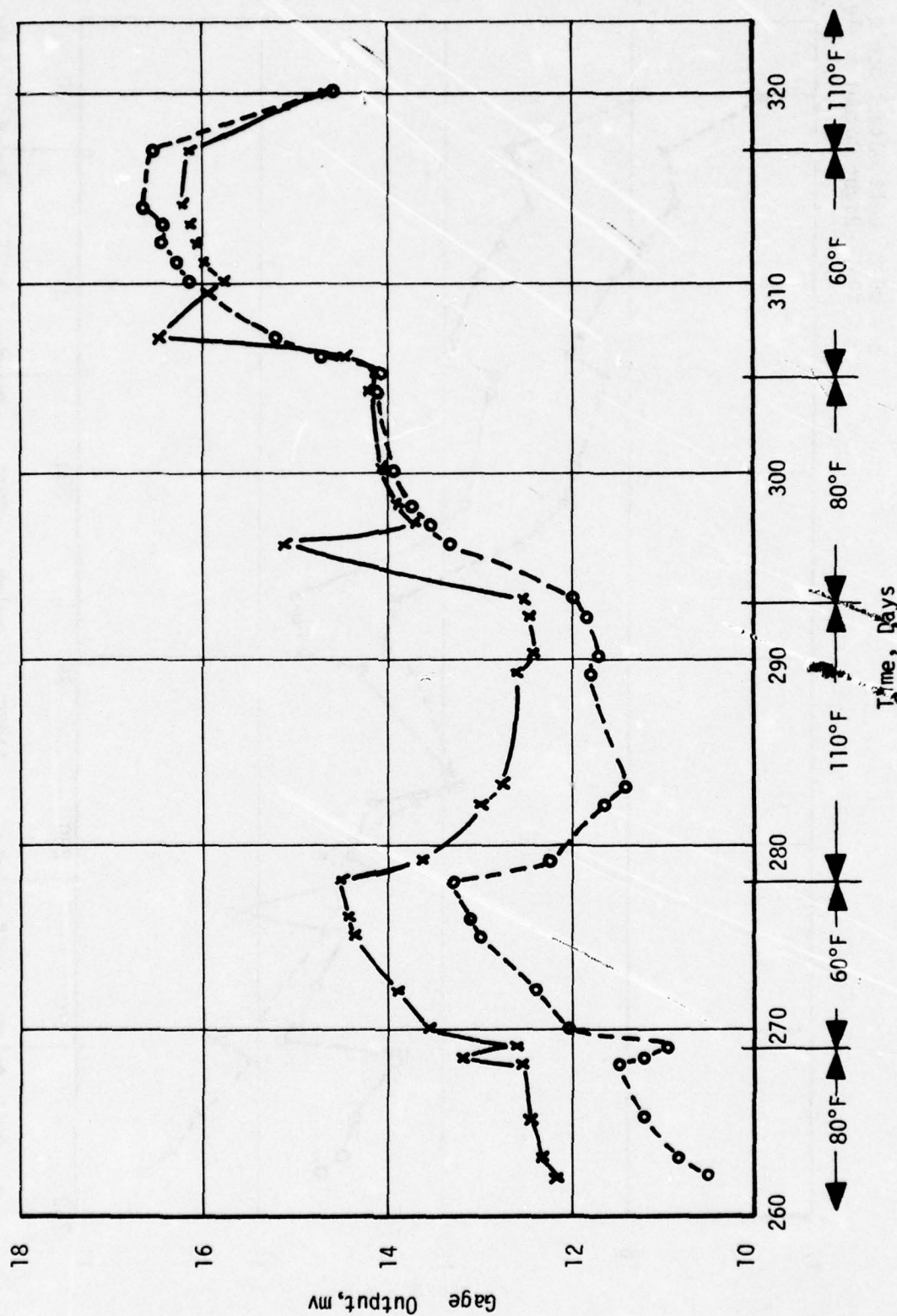


FIGURE H-53. FULL SCALE MOTOR NO. 1 - GAGES N10-1/2 OUTPUT (MV)





TABLE H-1. FULL SCALE MOTOR NO. 1 NORMAL GAGE OUTPUT LOG, MILLIVOLTS (CONT.)

[illegible]

TABLE H-1. FULL SCALE MOTOR NO. 1 NORMAL GAGE OUTPUT LOG, MILLIVOLTS (CONT.)

[illegible]

TABLE H-1. FULL SCALE MOTOR NO. 1 NORMAL GAGE OUTPUT LOG, MILLIVOLTS (CONT.)

[illegible]



TABLE H-1. FULL SCALE MOTOR NO. 1 NORMAL GAGE OUTPUT LOG, MILLIVOLTS (CONT.)

Test Propellant core cycle (cont'd)	Motor Conditions	Date	Time	Elapsed Time, Days	Temp., °F	Press., psi																				
							XI	XII	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15			
0800 5.5							-8.93	-4.36	.02	-93.02	-13.09	-7.54	-5.07	-4.48	4.05	-4.24										
							-8.99	-4.48	.02	-93.03	-13.10	-7.54	-5.06	-4.48	4.05	-4.24										
							-8.95	-4.37	.02	-93.02	-13.10	-7.54	-5.06	-4.47	4.02	-4.25										
							-8.95	-4.36	.02	-93.03	-13.09	-7.54	-5.06	-4.47	4.04	-4.25										
							-8.94	-4.35	.02	-93.03	-13.10	-7.54	-5.06	-4.47	4.05	-4.24										
							-8.95	-4.36	.02	-93.03	-13.09	-7.54	-5.06	-4.48	4.05	-4.24										
							-8.90	-4.43	.10	-92.87	-13.04	-7.43	-4.90	-3.31	4.38	-3.94										
							-8.82	-4.22	.11	-92.86	-13.03	-7.44	-4.90	-3.31	4.34	-3.96										
							-8.80	-4.20	.11	-92.86	-13.03	-7.44	-4.90	-3.31	4.38	-3.94										
							-8.80	-4.19	.12	-92.85	-13.02	-7.45	-4.92	-3.30	4.40	-3.92										
							-8.80	-4.19	.13	-92.85	-13.02	-7.45	-4.92	-2.94	4.40	-3.92										
							-8.79	-4.19	.13	-92.85	-13.02	-7.45	-4.91	-2.84	4.41	-3.91										
							-8.79	-4.18	.12	-92.86	-13.01	-7.45	-4.91	-2.84	4.41	-3.91										
							-8.79	-4.18	.12	-92.86	-13.01	-7.45	-4.91	-2.84	4.41	-3.91										
							-8.79	-4.18	.13	-92.85	-13.01	-7.45	-4.90	-2.84	4.42	-3.90										
11-8-72 0900 5.6							-8.73	-4.13	.14	-92.75	-13.01	-7.37	-4.83	-2.31	4.60	-3.73										
							-8.73	-4.13	.15	-92.75	-13.01	-7.37	-4.83	-2.31	4.60	-3.73										
							-8.73	-4.13	.13	-92.70	-13.00	-7.37	-4.83	-2.31	4.60	-3.73										
							-8.63	-4.00	.12	-92.41	-12.96	-7.34	-4.86	-1.38	4.90	-3.45										
							-8.65	-4.05	.14	-92.41	-12.96	-7.33	-4.78	-1.13	4.88	-3.46										
							-8.65	-4.11	.12	-92.43	-12.96	-7.30	-4.81	-1.18	4.88	-3.44										
							-8.64	-4.05	.07	-92.52	-12.96	-7.25	-4.80	-1.26	5.01	-3.34										
							-8.69	-4.16	.09	-92.51	-12.96	-7.25	-4.80	-1.26	5.01	-3.34										
							-8.63	-4.06	.09	-92.56	-12.95	-7.24	-4.76	-1.19	5.07	-3.28										
							-8.61	-4.04	.11	-92.56	-12.94	-7.25	-4.77	-1.19	5.08	-3.27										
							-8.61	-4.05	.11	-92.57	-12.93	-7.25	-4.78	-1.23	5.09	-3.26										
							-8.61	-4.06	.12	-92.57	-12.94	-7.26	-4.79	-1.20	5.10	-3.26										
							-8.61	-4.04	.12	-92.57	-12.93	-7.26	-4.79	-1.19	5.09	-3.25										
							-8.62	-4.06	.12	-92.57	-12.93	-7.26	-4.79	-1.19	5.08	-3.27										
							-8.61	-4.04	.13	-92.56	-12.93	-7.26	-4.78	-1.18	5.08	-3.27										
-8.60	-4.03	.13	-92.57	-12.93	-7.26	-4.79	-1.18	5.09	-3.25																	
2400 6.2							-8.49	-3.87	.18	-92.71	-12.92	-7.20	-4.66	-0.95	5.35	-3.02										
							-8.49	-3.87	.19	-92.69	-12.92	-7.20	-4.66	-0.93	5.36	-3.02										
							-8.49	-3.87	.20	-92.69	-12.91	-7.21	-4.66	-0.93	5.37	-3.01										
							-8.48	-3.86	.20	-92.68	-12.91	-7.21	-4.67	-0.93	5.37	-3.01										
							-8.48	-3.86	.21	-92.68	-12.90	-7.21	-4.67	-0.93	5.38	-3.00										
							-8.55	-4.04	.21	-92.69	-12.90	-7.21	-4.67	-0.92	5.38	-3.00										
							-8.55	-4.04	.21	-92.69	-12.90	-7.21	-4.67	-0.92	5.38	-3.00										
							-8.55	-4.04	.21	-92.69	-12.90	-7.21	-4.67	-0.92	5.38	-3.00										
							-8.55	-4.04	.21	-92.69	-12.90	-7.21	-4.67	-0.92	5.38	-3.00										
							-8.55	-4.04	.21	-92.69	-12.90	-7.21	-4.67	-0.92	5.38	-3.00										
							-8.55	-4.04	.21	-92.69	-12.90	-7.21	-4.67	-0.92	5.38	-3.00										
							-8.55	-4.04	.21	-92.69	-12.90	-7.21	-4.67	-0.92	5.38	-3.00										
							-8.55	-4.04	.21	-92.69	-12.90	-7.21	-4.67	-0.92	5.38	-3.00										
							-8.55	-4.04	.21	-92.69	-12.90	-7.21	-4.67	-0.92	5.38	-3.00										
							-8.55	-4.04	.21	-92.69	-12.90	-7.21	-4.67	-0.92	5.38	-3.00										
11-9-72 0400 6.3							-8.33	-3.68	.34	-93.11	-12.82	-7.14	-4.49	.21	5.61	-2.79										
							-8.33	-3.67	.35	-93.11	-12.82	-7.14	-4.50	.21	5.62	-2.78										
							-8.32	-3.67	.35	-93.12	-12.81	-7.14	-4.50	.22	5.62	-2.78										
							-8.32	-3.67	.35	-93.12	-12.81	-7.14	-4.50	.22	5.62	-2.78										
							-8.32	-3.66	.37	-93.11	-12.81	-7.15	-4.51	.22	5.64	-2.77										
							-8.38	-3.85	.37	-93.11	-12.81	-7.15	-4.51	.23	5.64	-2.77										
							-8.33	-3.68	.37	-93.11	-12.80	-7.16	-4.51	.23	5.61	-2.77										
							-8.32	-3.66	.37	-93.09	-12.80	-7.16	-4.51	.24	5.63	-2.77										
							-8.32	-3.66	.37	-93.10	-12.80	-7.15	-4.51	.23	5.63	-2.76										
							-8.32	-3.66	.37	-93.10	-12.80	-7.16	-4.52	.23	5.64	-2.76										
							-8.31	-3.66	.39	-93.09	-12.80	-7.16	-4.52	.22	5.64	-2.76										
							-8.31	-3.66	.38	-93.10	-12.80	-7.16	-4.50	.25	5.65	-2.76										
							-8.31	-3.65	.37	-93.11	-12.80	-7.16	-4.51	.24	5.65	-2.75										
							-8.31	-3.65	.37	-93.11	-12.80	-7.16	-4.51	.24	5.65	-2.75										
							-8.31	-3.66	.37	-93.11	-12.80	-7.16	-4.55	.17	5.65	-2.75										
-8.31	-3.66	.37	-93.11	-12.80	-7.16	-4.55	.17	5.65	-2.75																	
-8.31	-3.66	.37	-93.11	-12.80	-7.16	-4.55	.17	5.65	-2.75																	
-8.31	-3.66	.37	-93.11	-12.80	-7.16	-4.55	.17	5.65	-2.75																	
-8.31	-3.66	.37	-93.11	-12.80	-7.16	-4.55	.17	5.65	-2.75																	
-8.31	-3.67	.38	-93.11	-12.80	-7.16	-4.51	.25	5.62	-2.76																	
11-9-72 1100 6.6							-8.33	-3.61	.36	-93.08	-12.89	-7.06	-4.41	.27	5.89	-2.52										
							-8.33	-3.60	.37	-93.07	-12.90	-7.06	-4.42	.26	5.90	-2.51										
							-8.33	-3.60	.37	-93.07	-12.90	-7.06	-4.42	.26	5.90	-2.51										
							-8.33	-3.60	.37	-93.07	-12.90	-7.06	-4.42	.26	5.90	-2.51										
							-8.33	-3.60	.37	-93.07	-12.90	-7.06	-4.42	.26	5.90	-2.51										
							-8.33	-3.60	.37	-93.07	-12.90	-7.06	-4.42	.26	5.90	-2.51										
1610 6.9							-8.15	-3.50	.67	-93.96	-12.91	-7.06	-4.52	.11	5.94	-2.46										
							-8.15	-3.49	.67	-93.95	-12.91	-7.04	-4.52	.11	5.96	-2.45										
							-8.14	-3.49	.67	-93.95	-12.91	-7.04	-4.52	.11	5.96	-2.45										
							-8.14	-3.49	.67	-93.95	-12.91	-7.04	-4.52	.11	5.96	-2.45										
							-8.14	-3.49	.67	-93.95	-12.91	-7.04	-4.52	.11	5.96	-2.45										
							-8.14	-3.49	.67	-93.95	-12.91	-7.04	-4.52	.11	5.96	-2.45										

TABLE H-1. FULL SCALE MOTOR NO. 1 NORMAL GAGE OUTPUT LOG, MILLIVOLTS (CONT.)

Test Proellant Cure Cycle (cont'd)	Motor Conditions Alternate readings listed during this period.	Date	Time	Elapsed Time, Days	Temp., °F	Press., PSIG	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12
2000	7.0						-8.12	-3.49	70	-93.92	-12.90	-7.05	-4.54	.13	5.99	-2.42		
							-8.14	-3.50	68	-93.94	-12.91	-7.04	-4.53	.10	5.94	-2.45		
							-8.13	-3.48	68	-93.94	-12.89	-7.04	-4.53	.10	5.94	-2.45		
							-8.12	-3.47	68	-93.93	-12.89	-7.05	-4.54	.11	5.99	-2.42		
							-8.11	-3.46	69	-93.92	-12.89	-7.05	-4.54	.12	6.00	-2.41		
							-8.12	-3.48	70	-93.92	-12.89	-7.05	-4.53	.13	6.00	-2.41		
							-8.11	-3.46	69	-93.92	-12.89	-7.05	-4.54	.12	6.00	-2.41		
							-8.11	-3.47	69	-93.92	-12.88	-7.06	-4.54	.13	6.01	-2.41		
							-8.11	-3.47	68	-93.92	-12.88	-7.06	-4.54	.13	6.01	-2.41		
							-8.16	-3.40	75	-93.73	-12.86	-6.99	-4.48	.17	6.15	-2.28		
							-	-3.39	77	-93.72	-12.86	-7.00	-4.59	.03	6.15	-2.28		
							-8.11	-3.38	78	-93.70	-12.85	-7.00	-4.49	.17	6.16	-2.27		
2400	7.2						-8.11	-3.38	78	-93.69	-12.84	-7.01	-4.50	.19	6.13	-2.28		
							-8.04	-3.38	79	-93.68	-12.84	-7.01	-4.49	.20	6.16	-2.27		
							-8.03	-3.37	79	-93.68	-12.84	-7.01	-4.50	.20	6.17	-2.26		
							-8.03	-3.37	79	-93.68	-12.84	-7.01	-4.51	.18	6.17	-2.26		
							-8.03	-3.37	80	-93.67	-12.84	-7.01	-4.51	.18	6.17	-2.26		
							-8.03	-3.37	80	-93.67	-12.84	-7.01	-4.51	.19	6.17	-2.26		
							-8.02	-3.37	80	-93.67	-12.83	-7.01	-4.51	.19	6.17	-2.26		
							-7.95	-3.28	93	-93.24	-12.75	-6.94	-4.41	.30	6.36	-2.09		
							-7.95	-3.27	94	-93.24	-12.74	-6.95	-4.42	.29	6.37	-2.08		
							-7.94	-3.27	94	-93.24	-12.74	-6.95	-4.42	.30	6.37	-2.07		
							-7.93	-3.26	95	-93.23	-12.73	-	-4.43	.31	6.38	-2.07		
11-10-72	0840	7.5	114	15			-7.93	-3.26	95	-93.23	-12.73	-	-4.43	.32	6.38	-2.06		
							-7.94	-3.28	93	-93.21	-12.76	-	-4.41	.29	6.39	-2.05		
							-7.94	-3.28	94	-93.21	-12.75	-	-4.41	.30	6.39	-2.05		
							-7.93	-3.27	93	-93.24	-12.76	-6.96	-4.53	.11	6.42	-2.03		
							-7.92	-3.26	93	-93.24	-12.75	-6.96	-4.42	.29	6.43	-2.02		
							-7.92	-3.26	94	-93.23	-	-6.87	-4.42	.30	6.43	-2.02		
							-7.99	-3.45	94	-93.24	-12.74	-6.87	-4.42	.31	6.43	-2.02		
							-7.91	-3.27	96	-93.24	-12.74	-6.86	-4.40	.29	6.39	-2.01		
							-7.92	-3.26	95	-93.23	-12.74	-6.88	-4.45	.35	6.41	-2.05		
							-7.92	-3.26	95	-93.22	-12.74	-6.87	-4.41	.33	6.43	-2.01		
							-7.98	-3.51	95	-93.22	-12.73	-6.86	-4.43	.31	6.44	-2.00		
							-7.91	-3.25	95	-93.22	-12.73	-6.87	-4.42	.30	6.44	-2.00		
1300	7.7						-7.79	-3.13	108	-93.40	-12.68	-6.64	-4.31	.44	6.75	-1.72		
							-7.79	-3.13	108	-93.39	-12.67	-6.65	-4.34	.39	6.76	-1.71		
							-7.78	-3.12	110	-93.38	-12.66	-6.65	-4.32	.44	6.77	-1.70		
							-7.82	-3.13	110	-93.38	-12.66	-6.66	-4.32	.46	6.77	-1.71		
							-7.77	-3.11	110	-93.38	-12.66	-6.66	-4.32	.46	6.77	-1.70		
							-7.75	-3.08	112	-93.40	-12.63	-6.60	-4.26	.49	6.85	-1.62		
							-7.74	-3.07	111	-93.40	-12.62	-6.60	-4.26	.49	6.85	-1.60		
							-7.72	-3.05	116	-93.33	-12.61	-6.58	-4.25	.51	6.93	-1.56		
							-7.71	-3.04	117	-93.32	-12.60	-6.58	-4.26	.52	6.94	-1.55		
							-7.70	-3.03	118	-93.30	-12.60	-6.59	-4.28	.48	6.94	-1.54		
							-7.74	-3.12	118	-93.28	-12.59	-6.59	-4.26	.52	6.95	-1.54		
1500	7.8						-7.70	-3.01	119	-93.27	-12.58	-6.60	-4.26	.54	6.95	-1.54		
							-7.70	-3.02	119	-93.27	-12.58	-6.60	-4.26	.54	6.95	-1.53		
							-7.69	-3.02	119	-93.27	-12.58	-6.60	-4.26	.54	6.96	-1.50		
							-7.69	-3.02	119	-93.26	-12.58	-6.60	-4.27	.54	6.96	-1.52		
							-7.69	-3.02	120	-93.26	-12.58	-6.59	-4.27	.54	6.97	-1.52		
							-7.69	-3.02	119	-93.26	-12.58	-6.59	-4.26	.55	6.97	-1.52		
							-7.58	-2.90	130	-93.71	-12.47	-6.50	-4.13	.65	7.24	-1.27		
							-7.58	-2.90	129	-93.72	-12.46	-6.51	-4.15	.65	7.25	-1.26		
							-7.57	-2.89	130	-93.72	-12.46	-6.51	-4.15	.66	7.25	-1.26		
							-7.57	-2.89	130	-93.72	-12.46	-6.51	-4.15	.67	7.26	-1.25		
							-7.56	-2.88	130	-93.71	-12.45	-6.51	-4.16	.66	7.26	-1.24		
2000	8.0						-7.59	-	131	-93.70	-12.45	-6.52	-4.16	.67	7.28	-1.25		
							-7.57	-2.89	131	-93.70	-12.45	-6.52	-4.16	.68	7.24	-1.25		
							-7.57	-2.88	131	-93.71	-12.45	-6.52	-4.16	.68	7.26	-1.24		
							-7.56	-2.88	131	-93.71	-12.45	-6.52	-4.16	.68	7.26	-1.24		
							-7.56	-2.88	131	-93.71	-12.45	-6.52	-4.16	.68	7.26	-1.24		
							-7.56	-2.88	131	-93.71	-12.45	-6.52	-4.16	.68	7.26	-1.24		
							-7.56	-2.88	131	-93.71	-12.45	-6.52	-4.16	.68	7.26	-1.24		
							-7.56	-2.88	131	-93.71	-12.45	-6.52	-4.16	.68	7.26	-1.24		
							-7.56	-2.88	131	-93.71	-12.45	-6.52	-4.16	.68	7.26	-1.24		
							-7.56	-2.88	131	-93.71	-12.45	-6.52	-4.16	.68	7.26	-1.24		
							-7.56	-2.88	131	-93.71	-12.45	-6.52	-4.16	.68	7.26	-1.24		
2400	8.2						-7.29	-2.58	208	-93.24	-12.06	-6.41	-3.89	1.02	7.80	-	.75	
							-7.29	-2.57	208	-93.23	-12.05	-6.41	-3.89	1.04	7.81	-	.74	
							-7.26	-2.57	210	-93.22	-12.04	-6.42	-3.90	1.04	7.82	-	.73	
							-7.26	-2.56	210	-93.22	-12.04	-6.42	-3.90	1.04	7.82	-	.73	
							-7.26	-2.56	210	-93.22	-12.04	-6.42	-3.90	1.04	7.82	-	.73	
							-7.26	-2.56	210	-93.22	-12.04	-6.42	-3.90	1.04	7.82	-	.73	

$$\begin{array}{r} 110 \\ 111 \\ 112 \end{array} \quad \begin{array}{r} -1 \\ -2 \\ -3 \end{array}$$

Test	Motor Conditions	Date	Time, hrs.	Elapsed Time, Press.,	Propellant cure cycle (cont'd)
8.3	0400	11-11-72	8.3	-7.15 -2.40	2.41 -32.97 -11.72 -6.44 -3.76
				-7.14 -2.39	2.41 -32.97 -11.71 -6.45 -3.76
				-7.13 -2.38	2.42 -32.96 -11.70 -6.46 -3.76
				-7.12 -2.37	2.43 -32.96 -11.70 -6.47 -3.81
				-7.11 -2.36	2.42 -32.95 -11.70 -6.47 -3.79
				-7.10 -2.35	2.42 -32.95 -11.70 -6.47 -3.78
				-7.13 -2.38	2.43 -32.94 -11.69 -6.48 -3.78
				-7.12 -2.38	2.43 -32.94 -11.69 -6.48 -3.78
				-7.06 -2.31	2.52 -32.90 -11.59 -6.32 -3.71
				-7.05 -2.32	2.56 -32.78 -11.58 -6.33 -3.71
				-7.04 -2.30	2.67 -32.72 -11.56 -6.34 -3.73
				-7.04 -2.29	2.63 -32.65 -11.57 -6.34 -3.77
8.5	0800		8.5	-7.04 -2.30	2.63 -32.65 -11.57 -6.34 -3.77
				-7.03 -2.30	2.63 -32.65 -11.57 -6.34 -3.77
				-7.02 -2.29	2.63 -32.65 -11.57 -6.34 -3.77
				-7.01 -2.28	2.63 -32.65 -11.57 -6.34 -3.77
				-7.00 -2.27	2.63 -32.65 -11.57 -6.34 -3.77
				-6.99 -2.26	2.63 -32.65 -11.57 -6.34 -3.77
				-6.98 -2.25	2.63 -32.65 -11.57 -6.34 -3.77
				-6.97 -2.24	2.63 -32.65 -11.57 -6.34 -3.77
				-6.96 -2.23	2.63 -32.65 -11.57 -6.34 -3.77
				-6.95 -2.22	2.63 -32.65 -11.57 -6.34 -3.77
				-6.94 -2.21	2.63 -32.65 -11.57 -6.34 -3.77
				-6.93 -2.20	2.63 -32.65 -11.57 -6.34 -3.77
8.7	1200		8.7	-7.03 -2.46	1.92 -33.58 -11.74 -6.35 -3.81
				-7.02 -2.45	1.92 -33.58 -11.74 -6.35 -3.81
				-7.01 -2.44	1.94 -33.54 -11.73 -6.35 -3.82
				-7.00 -2.44	1.94 -33.54 -11.73 -6.35 -3.82
				-7.03 -2.44	1.94 -33.54 -11.73 -6.35 -3.82
				-7.02 -2.44	1.94 -33.54 -11.73 -6.35 -3.82
				-7.01 -2.43	1.94 -33.54 -11.73 -6.35 -3.82
				-7.00 -2.43	1.94 -33.54 -11.73 -6.35 -3.82
				-6.99 -2.43	1.94 -33.54 -11.73 -6.35 -3.82
				-6.98 -2.43	1.94 -33.54 -11.73 -6.35 -3.82
				-6.97 -2.43	1.92 -33.57 -11.72 -6.28 -3.84
				-6.96 -2.43	1.92 -33.58 -11.73 -6.28 -3.84
8.8	1600		8.8	-7.02 -2.40	1.93 -33.71 -11.84 -6.34 -3.79
				-7.01 -2.40	1.93 -33.71 -11.84 -6.34 -3.79
				-7.00 -2.40	1.93 -33.71 -11.84 -6.34 -3.79
				-6.99 -2.40	1.93 -33.71 -11.84 -6.34 -3.79
				-6.98 -2.40	1.93 -33.71 -11.84 -6.34 -3.79
				-6.97 -2.40	1.93 -33.71 -11.84 -6.34 -3.79
				-6.96 -2.40	1.93 -33.71 -11.84 -6.34 -3.79
				-6.95 -2.40	1.93 -33.71 -11.84 -6.34 -3.79
				-6.94 -2.40	1.93 -33.71 -11.84 -6.34 -3.79
				-6.93 -2.40	1.93 -33.71 -11.84 -6.34 -3.79
				-6.92 -2.39	1.98 -33.69 -11.83 -6.25 -3.82
				-6.91 -2.38	1.98 -33.69 -11.83 -6.25 -3.82
9.0	2000		9.0	-6.76 -2.03	2.61 -33.35 -11.66 -6.16 -3.49
				-6.75 -2.03	2.66 -33.32 -11.66 -6.17 -3.50
				-6.74 -2.03	2.64 -33.32 -11.65 -6.17 -3.49
				-6.73 -2.01	2.66 -33.30 -11.64 -6.17 -3.50
				-6.74 -2.01	2.65 -33.31 -11.64 -6.17 -3.50
				-6.73 -1.99	2.65 -33.32 -11.66 -6.18 -3.55
				-6.62 -1.88	2.92 -33.34 -11.53 -6.12 -3.38
				-6.56 -1.92	2.92 -33.34 -11.52 -6.11 -3.38
				-6.62 -1.97	2.90 -33.35 -11.52 -6.14 -3.38
				-6.58 -1.87	2.93 -33.33 -11.50 -6.13 -3.38
				-6.60 -1.86	2.94 -33.31 -11.50 -6.14 -3.39
				-6.60 -1.86	2.94 -33.33 -11.49 -6.14 -3.39
9.2	2400		9.2	-6.50 -1.86	2.95 -33.31 -11.49 -6.14 -3.39
				-6.50 -1.86	2.94 -33.32 -11.49 -6.14 -3.39
				-6.50 -1.76	3.22 -33.27 -11.34 -6.09 -3.28
				-6.51 -1.78	3.27 -33.25 -11.38 -6.09 -3.32
				-6.46 -1.76	3.21 -33.22 -11.35 -6.09 -3.27
				-6.48 -1.75	3.23 -33.26 -11.34 -6.10 -3.30
				-6.50 -1.77	3.27 -33.23 -11.35 -6.10 -3.27
				-6.45 -1.81	3.02 -33.19 -11.34 -6.08 -3.31
				-6.53 -1.77	3.06 -33.21 -11.37 -6.09 -3.32
				-6.53 -1.76	3.08 -33.18 -11.38 -6.08 -3.32
				-6.50 -1.78	3.02 -33.20 -11.36 -6.07 -3.34
				-6.51 -1.77	3.06 -33.22 -11.38 -6.12 -3.39
9.3	0400	11-12-72	9.3	-6.43 -1.76	3.02 -33.22 -11.35 -6.09 -3.32
				-6.43 -1.76	3.01 -33.22 -11.32 -6.09 -3.33
				-6.50 -1.76	3.22 -33.27 -11.34 -6.09 -3.28
				-6.51 -1.78	3.27 -33.25 -11.38 -6.09 -3.32
				-6.46 -1.76	3.21 -33.22 -11.35 -6.09 -3.27
				-6.48 -1.75	3.23 -33.26 -11.34 -6.10 -3.30
				-6.50 -1.77	3.27 -33.23 -11.35 -6.10 -3.27
				-6.45 -1.81	3.02 -33.19 -11.34 -6.08 -3.31
				-6.53 -1.77	3.06 -33.21 -11.37 -6.09 -3.32
				-6.53 -1.76	3.08 -33.18 -11.38 -6.08 -3.32
				-6.50 -1.78	3.02 -33.20 -11.36 -6.07 -3.34
				-6.51 -1.77	3.06 -33.22 -11.38 -6.12 -3.39
9.5	0800		9.5	-6.43 -1.76	3.02 -33.22 -11.35 -6.09 -3.32
				-6.43 -1.76	3.01 -33.22 -11.32 -6.09 -3.33
				-6.50 -1.76	3.22 -33.27 -11.34 -6.09 -3.28
				-6.51 -1.78	3.27 -33.25 -11.38 -6.09 -3.32
				-6.46 -1.76	3.21 -33.22 -11.35 -6.09 -3.27
				-6.48 -1.75	3.23 -33.26 -11.34 -6.10 -3.30
				-6.50 -1.77	3.27 -33.23 -11.35 -6.10 -3.27
				-6.45 -1.81	3.02 -33.19 -11.34 -6.08 -3.31
				-6.53 -1.77	3.06 -33.21 -11.37 -6.09 -3.32
				-6.53 -1.76	3.08 -33.18 -11.38 -6.08 -3.32
				-6.50 -1.78	3.02 -33.20 -11.36 -6.07 -3.34
				-6.51 -1.77	3.06 -33.22 -11.38 -6.12 -3.39
9.7	1200		9.7	-6.43 -1.76	3.02 -33.22 -11.35 -6.09 -3.32
				-6.43 -1.76	3.01 -33.22 -11.32 -6.09 -3.33
				-6.50 -1.76	3.22 -33.27 -11.34 -6.09 -3.28
				-6.51 -1.78	3.27 -33.25 -11.38 -6.09 -3.32
				-6.46 -1.76	3.21 -33.22 -11.35 -6.09 -3.27
				-6.48 -1.75	3.23 -33.26 -11.34 -6.10 -3.30
				-6.50 -1.77	3.27 -33.23 -11.35 -6.10 -3.27
				-6.45 -1.81	3.02 -33.19 -11.34 -6.08 -3.31
				-6.53 -1.77	3.06 -33.21 -11.37 -6.09 -3.32
				-6.53 -1.76	3.08 -33.18 -11.38 -6.08 -3.32
				-6.50 -1.78	3.02 -33.20 -11.36 -6.07 -3.34
				-6.51 -1.77	3.06 -33.22 -11.38 -6.12 -3.39
9.8	1600		9.8	-6.43 -1.76	3.02 -33.22 -11.35 -6.09 -3.32
				-6.43 -1.76	3.01 -33.22 -11.32 -6.09 -3.33
				-6.50 -1.76	3.22 -33.27 -11.34 -6.09 -3.28
				-6.51 -1.78	3.27 -33.25 -11.38 -6.09 -3.32
				-6.46 -1.76	3.21 -33.22 -11.35 -6.09 -3.27
				-6.48 -1.75	3.23 -33.26 -11.34 -6.10 -3.30
				-6.50 -1.77	3.27 -33.23 -11.35 -6.10 -3.27
				-6.45 -1.81	3.02 -33.19 -11.34 -6.08 -3.31
				-6.53 -1.77	3.06 -33.21 -11.37 -6.09 -3.32
				-6.53 -1.76	3.08 -33.18 -11.38 -6.08 -3.32
				-6.50 -1.78	3.02 -33.20 -11.36 -6.07 -3.34
				-6.51 -1.77	3.06 -33.22 -11.38 -6.12 -3.39



TABLE H-1. FULL SCALE MOTOR NO. 1 NORMAL GAGE OUTPUT LOG, MILLIVOLTS (CONT.)

Test	Motor Conditions	Date	Time	Elapsed Time, Days	Temp., Press., F. P.S.I.G.	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100	101	102	103	104	105	106	107	108	109	110	111	112	113	114	115	116	117	118	119	120	121	122	123	124	125	126	127	128	129	130	131	132	133	134	135	136	137	138	139	140	141	142	143	144	145	146	147	148	149	150	151	152	153	154	155	156	157	158	159	160	161	162	163	164	165	166	167	168	169	170	171	172	173	174	175	176	177	178	179	180	181	182	183	184	185	186	187	188	189	190	191	192	193	194	195	196	197	198	199	200	201	202	203	204	205	206	207	208	209	210	211	212	213	214	215	216	217	218	219	220	221	222	223	224	225	226	227	228	229	230	231	232	233	234	235	236	237	238	239	240	241	242	243	244	245	246	247	248	249	250	251	252	253	254	255	256	257	258	259	260	261	262	263	264	265	266	267	268	269	270	271	272	273	274	275	276	277	278	279	280	281	282	283	284	285	286	287	288	289	290	291	292	293	294	295	296	297	298	299	300	301	302	303	304	305	306	307	308	309	310	311	312	313	314	315	316	317	318	319	320	321	322	323	324	325	326	327	328	329	330	331	332	333	334	335	336	337	338	339	340	341	342	343	344	345	346	347	348	349	350	351	352	353	354	355	356	357	358	359	360	361	362	363	364	365	366	367	368	369	370	371	372	373	374	375	376	377	378	379	380	381	382	383	384	385	386	387	388	389	390	391	392	393	394	395	396	397	398	399	400	401	402	403	404	405	406	407	408	409	410	411	412	413	414	415	416	417	418	419	420	421	422	423	424	425	426	427	428	429	430	431	432	433	434	435	436	437	438	439	440	441	442	443	444	445	446	447	448	449	450	451	452	453	454	455	456	457	458	459	460	461	462	463	464	465	466	467	468	469	470	471	472	473	474	475	476	477	478	479	480	481	482	483	484	485	486	487	488	489	490	491	492	493	494	495	496	497	498	499	500	501	502	503	504	505	506	507	508	509	510	511	512	513	514	515	516	517	518	519	520	521	522	523	524	525	526	527	528	529	530	531	532	533	534	535	536	537	538	539	540	541	542	543	544	545	546	547	548	549	550	551	552	553	554	555	556	557	558	559	560	561	562	563	564	565	566	567	568	569	570	571	572	573	574	575	576	577	578	579	580	581	582	583	584	585	586	587	588	589	590	591	592	593	594	595	596	597	598	599	600	601	602	603	604	605	606	607	608	609	610	611	612	613	614	615	616	617	618	619	620	621	622	623	624	625	626	627	628	629	630	631	632	633	634	635	636	637	638	639	640	641	642	643	644	645	646	647	648	649	650	651	652	653	654	655	656	657	658	659	660	661	662	663	664	665	666	667	668	669	670	671	672	673	674	675	676	677	678	679	680	681	682	683	684	685	686	687	688	689	690	691	692	693	694	695	696	697	698	699	700	701	702	703	704	705	706	707	708	709	710	711	712	713	714	715	716	717	718	719	720	721	722	723	724	725	726	727	728	729	730	731	732	733	734	735	736	737	738	739	740	741	742	743	744	745	746	747	748	749	750	751	752	753	754	755	756	757	758	759	760	761	762	763	764	765	766	767	768	769	770	771	772	773	774	775	776	777	778	779	780	781	782	783	784	785	786	787	788	789	790	791	792	793	794	795	796	797	798	799	800	801	802	803	804	805	806	807	808	809	810	811	812	813	814	815	816	817	818	819	820	821	822	823	824	825	826	827	828	829	830	831	832	833	834	835	836	837	838	839	840	841	842	843	844	845	846	847	848	849	850	851	852	853	854	855	856	857	858	859	860	861	862	863	864	865	866	867	868	869	870	871	872	873	874	875	876	877	878	879	880	881	882	883	884	885	886	887	888	889	890	891	892	893	894	895	896	897	898	899	900	901	902	903	904	905	906	907	908	909	910	911	912	913	914	915	916	917	918	919	920	921	922	923	924	925	926	927	928	929	930	931	932	933	934	935	936	937	938	939	940	941	942	943	944	945	946	947	948	949	950	951	952	953	954	955	956	957	958	959	960	961	962	963	964	965	966	967	968	969	970	971	972	973	974	975	976	977	978	979	980	981	982	983	984	985	986	987	988	989	990	991	992	993	994	995	996	997	998	999	1000
Propellant cure cycle (Cont'd)																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																			

TABLE H-1. FULL SCALE MOTOR NO. 1 NORMAL GAGE OUTPUT LOG, MILLIVOLTS (CONT.)

Test	Date	Time	Elapsed Time, Days	Temp., °F	Press., PSI	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100	101	102	103	104	105	106	107	108	109	110	111	112	113	114	115	116	117	118	119	120	121	122	123	124	125	126	127	128	129	130	131	132	133	134	135	136	137	138	139	140	141	142	143	144	145	146	147	148	149	150	151	152	153	154	155	156	157	158	159	160	161	162	163	164	165	166	167	168	169	170	171	172	173	174	175	176	177	178	179	180	181	182	183	184	185	186	187	188	189	190	191	192	193	194	195	196	197	198	199	200	201	202	203	204	205	206	207	208	209	210	211	212	213	214	215	216	217	218	219	220	221	222	223	224	225	226	227	228	229	230	231	232	233	234	235	236	237	238	239	240	241	242	243	244	245	246	247	248	249	250	251	252	253	254	255	256	257	258	259	260	261	262	263	264	265	266	267	268	269	270	271	272	273	274	275	276	277	278	279	280	281	282	283	284	285	286	287	288	289	290	291	292	293	294	295	296	297	298	299	300	301	302	303	304	305	306	307	308	309	310	311	312	313	314	315	316	317	318	319	320	321	322	323	324	325	326	327	328	329	330	331	332	333	334	335	336	337	338	339	340	341	342	343	344	345	346	347	348	349	350	351	352	353	354	355	356	357	358	359	360	361	362	363	364	365	366	367	368	369	370	371	372	373	374	375	376	377	378	379	380	381	382	383	384	385	386	387	388	389	390	391	392	393	394	395	396	397	398	399	400	401	402	403	404	405	406	407	408	409	410	411	412	413	414	415	416	417	418	419	420	421	422	423	424	425	426	427	428	429	430	431	432	433	434	435	436	437	438	439	440	441	442	443	444	445	446	447	448	449	450	451	452	453	454	455	456	457	458	459	460	461	462	463	464	465	466	467	468	469	470	471	472	473	474	475	476	477	478	479	480	481	482	483	484	485	486	487	488	489	490	491	492	493	494	495	496	497	498	499	500	501	502	503	504	505	506	507	508	509	510	511	512	513	514	515	516	517	518	519	520	521	522	523	524	525	526	527	528	529	530	531	532	533	534	535	536	537	538	539	540	541	542	543	544	545	546	547	548	549	550	551	552	553	554	555	556	557	558	559	560	561	562	563	564	565	566	567	568	569	570	571	572	573	574	575	576	577	578	579	580	581	582	583	584	585	586	587	588	589	590	591	592	593	594	595	596	597	598	599	600	601	602	603	604	605	606	607	608	609	610	611	612	613	614	615	616	617	618	619	620	621	622	623	624	625	626	627	628	629	630	631	632	633	634	635	636	637	638	639	640	641	642	643	644	645	646	647	648	649	650	651	652	653	654	655	656	657	658	659	660	661	662	663	664	665	666	667	668	669	670	671	672	673	674	675	676	677	678	679	680	681	682	683	684	685	686	687	688	689	690	691	692	693	694	695	696	697	698	699	700	701	702	703	704	705	706	707	708	709	710	711	712	713	714	715	716	717	718	719	720	721	722	723	724	725	726	727	728	729	730	731	732	733	734	735	736	737	738	739	740	741	742	743	744	745	746	747	748	749	750	751	752	753	754	755	756	757	758	759	760	761	762	763	764	765	766	767	768	769	770	771	772	773	774	775	776	777	778	779	780	781	782	783	784	785	786	787	788	789	790	791	792	793	794	795	796	797	798	799	800	801	802	803	804	805	806	807	808	809	810	811	812	813	814	815	816	817	818	819	820	821	822	823	824	825	826	827	828	829	830	831	832	833	834	835	836	837	838	839	840	841	842	843	844	845	846	847	848	849	850	851	852	853	854	855	856	857	858	859	860	861	862	863	864	865	866	867	868	869	870	871	872	873	874	875	876	877	878	879	880	881	882	883	884	885	886	887	888	889	890	891	892	893	894	895	896	897	898	899	900	901	902	903	904	905	906	907	908	909	910	911	912	913	914	915	916	917	918	919	920	921	922	923	924	925	926	927	928	929	930	931	932	933	934	935	936	937	938	939	940	941	942	943	944	945	946	947	948	949	950	951	952	953	954	955	956	957	958	959	960	961	962	963	964	965	966	967	968	969	970	971	972	973	974	975	976	977	978	979	980	981	982	983	984	985	986	987	988	989	990	991	992	993	994	995	996	997	998	999	1000
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TABLE H-1. FULL SCALE MOTOR NO. 1 NORMAL GAGE OUTPUT LOG, MILLIVOLTS (CONT.)

[illegible]



TABLE H-1. FULL SCALE MOTOR NO. 1 NORMAL GAGE OUTPUT LOG, MILLIVOLTS (CONT.)

Test	Motor Conditions	Date	Time	Elapsed Time, Days	Temp., °F	Press., in. Hg	-11	-10	-9	-8	-7	-6	-5	-4	-3	-2	-1	0	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100	101	102	103	104	105	106	107	108	109	110	111	112	113	114	115	116	117	118	119	120	121	122	123	124	125	126	127	128	129	130	131	132	133	134	135	136	137	138	139	140	141	142	143	144	145	146	147	148	149	150	151	152	153	154	155	156	157	158	159	160	161	162	163	164	165	166	167	168	169	170	171	172	173	174	175	176	177	178	179	180	181	182	183	184	185	186	187	188	189	190	191	192	193	194	195	196	197	198	199	200	201	202	203	204	205	206	207	208	209	210	211	212	213	214	215	216	217	218	219	220	221	222	223	224	225	226	227	228	229	230	231	232	233	234	235	236	237	238	239	240	241	242	243	244	245	246	247	248	249	250	251	252	253	254	255	256	257	258	259	260	261	262	263	264	265	266	267	268	269	270	271	272	273	274	275	276	277	278	279	280	281	282	283	284	285	286	287	288	289	290	291	292	293	294	295	296	297	298	299	300	301	302	303	304	305	306	307	308	309	310	311	312	313	314	315	316	317	318	319	320	321	322	323	324	325	326	327	328	329	330	331	332	333	334	335	336	337	338	339	340	341	342	343	344	345	346	347	348	349	350	351	352	353	354	355	356	357	358	359	360	361	362	363	364	365	366	367	368	369	370	371	372	373	374	375	376	377	378	379	380	381	382	383	384	385	386	387	388	389	390	391	392	393	394	395	396	397	398	399	400	401	402	403	404	405	406	407	408	409	410	411	412	413	414	415	416	417	418	419	420	421	422	423	424	425	426	427	428	429	430	431	432	433	434	435	436	437	438	439	440	441	442	443	444	445	446	447	448	449	450	451	452	453	454	455	456	457	458	459	460	461	462	463	464	465	466	467	468	469	470	471	472	473	474	475	476	477	478	479	480	481	482	483	484	485	486	487	488	489	490	491	492	493	494	495	496	497	498	499	500	501	502	503	504	505	506	507	508	509	510	511	512	513	514	515	516	517	518	519	520	521	522	523	524	525	526	527	528	529	530	531	532	533	534	535	536	537	538	539	540	541	542	543	544	545	546	547	548	549	550	551	552	553	554	555	556	557	558	559	560	561	562	563	564	565	566	567	568	569	570	571	572	573	574	575	576	577	578	579	580	581	582	583	584	585	586	587	588	589	590	591	592	593	594	595	596	597	598	599	600	601	602	603	604	605	606	607	608	609	610	611	612	613	614	615	616	617	618	619	620	621	622	623	624	625	626	627	628	629	630	631	632	633	634	635	636	637	638	639	640	641	642	643	644	645	646	647	648	649	650	651	652	653	654	655	656	657	658	659	660	661	662	663	664	665	666	667	668	669	670	671	672	673	674	675	676	677	678	679	680	681	682	683	684	685	686	687	688	689	690	691	692	693	694	695	696	697	698	699	700	701	702	703	704	705	706	707	708	709	710	711	712	713	714	715	716	717	718	719	720	721	722	723	724	725	726	727	728	729	730	731	732	733	734	735	736	737	738	739	740	741	742	743	744	745	746	747	748	749	750	751	752	753	754	755	756	757	758	759	760	761	762	763	764	765	766	767	768	769	770	771	772	773	774	775	776	777	778	779	780	781	782	783	784	785	786	787	788	789	790	791	792	793	794	795	796	797	798	799	800	801	802	803	804	805	806	807	808	809	810	811	812	813	814	815	816	817	818	819	820	821	822	823	824	825	826	827	828	829	830	831	832	833	834	835	836	837	838	839	840	841	842	843	844	845	846	847	848	849	850	851	852	853	854	855	856	857	858	859	860	861	862	863	864	865	866	867	868	869	870	871	872	873	874	875	876	877	878	879	880	881	882	883	884	885	886	887	888	889	890	891	892	893	894	895	896	897	898	899	900	901	902	903	904	905	906	907	908	909	910	911	912	913	914	915	916	917	918	919	920	921	922	923	924	925	926	927	928	929	930	931	932	933	934	935	936	937	938	939	940	941	942	943	944	945	946	947	948	949	950	951	952	953	954	955	956	957	958	959	960	961	962	963	964	965	966	967	968	969	970	971	972	973	974	975	976	977	978	979	980	981	982	983	984	985	986	987	988	989	990	991	992	993	994	995	996	997	998	999	1000
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TABLE H-2  
POST CURE-PRESSURE CALIBRATION TEST FOR MOTOR #1

TEMP: 80 ± 5°F  
(Output in mv)

PSIG  
PRESSURE

	S-1	S-2	S-3	S-4	S-5	S-6	S-7	S-8	S-9	S-10
0	-2.248	+5.225	+814	+6.521	-14.316	+4.637	+7.353	+011	-9.413	-2.887
9.356	-.980	+4.887	-2.093	+6.597	-11.11	+3.437	+5.455	-.839	-5.033	-.768
19.84	-.057	+4.689	-4.492	+6.655	-7.410	+2.322	+3.732	-1.433	+827	+1.572
30.11	+.627	+4.595	-6.410	+6.667	-3.974	+1.571	+2.463	-1.893	+7.167	+3.865
41.03	+1.221	+4.529	-8.158	+6.712	-.581	+.96	+.723	-2.184	+14.27	+6.067
50.34	+1.693	+4.485	-9.787	+7.063	+2.138	+.380	-.799	-1.971	+20.82	+7.930

	S-11	S-12	S-13	S-14	N-1-1	N-1-2	N-2-1	N-2-2	N-3-1	N-3-1
0	+2.736	+2.933	+2.217	-1.839	-12.38	-5.2	-3.848	-99.79	-18.24	+8.514
9.356	+1.856	-.789	-6.839	-1.747	-9.718	-2.709	-1.800	-1.391	-1.611	+6.67
19.84	+.648	-4.42	-14.36	-1.666	-6.749	+.177	+.522	-3.647	-14.96	+4.909
30.11	-.591	-7.560	-21.16	-1.529	-3.861	+2.987	+2.658	-12.08	-12.66	+3.317
41.03	-2.006	-10.59	-28.67	-1.249	-.855	+5.905	+5.05	-7.86	-10.62	+1.633
50.34	-3.59	-13.12	-36.56	-.814	+1.749	+8.437	+6.962	-90.39	-8.24	-.160

TABLE H-2 (CONTINUED)  
POST CURE-PRESSURE CALIBRATION TEST FOR MOTOR #1  
(Output in mv)

PSIG PRESSURE	N-4-1	N-4-2	N-5-1	N-5-2	N-6-1	N-6-2
0	+0.000	-3.903	-9.84	-16.99	-7.53	-6.015
9.456	+6.928	-1.250	-2.545	-10.23	-.241	+1.041
19.84	+3.978	+1.674	+5.721	-2.582	+7.998	+9.263
30.11	+1.145	+4.509	+13.82	+4.922	+16.014	+17.322
41.03	-1.816	+7.444	+22.34	+12.81	+24.44	+25.759
50.34	-4.355	+9.982	+29.753	+19.721	+31.684	+32.895
	LVDT-1	LVDT-2	LVDT-3 (Output in Volts)*	LVDT-4	LVDT-5	LVDT-6
0	5.457	2.886	4.923	4.583	5.280	4.770
9.44	5.310	2.675	4.711	4.359	5.182	4.556
19.31	5.165	2.485	4.501	4.124	5.074	4.348
29.91	4.995	2.284	4.288	3.906	4.964	4.138
39.59	4.841	2.111	4.096	3.701	4.883	3.950
49.45	4.681	1.936	3.906	3.497	4.821	3.760

\* Scale factor 4.99 volts/inch (LVDT's 7 and 8 not connected because of aft plate interference).



TABLE H-3  
 FULL SCALE MOTOR #1 POST-CURE  
 PRESSURE TESTS OF NORMAL STRESS  
 TRANSDUCERS AT ABOUT 87°F

<u>Transducer</u>	<u>Apparent Gage Sensitivity, mv/psi</u>	
	<u>12-19-72</u>	<u>12-28-72</u>
N1-1	-	0.230
N1-2	-	0.233
N2-1	-	0.118
N2-2	-	-
N3-1	0.208	0.221
N3-2	0.187	0.192
N4-1	0.287	0.195
N4-2	0.259	0.223
N5-1	0.751	0.732
N5-2	0.716	0.670
N6-1	0.759	0.649
N6-2	0.759	0.743
N7-1	0.764	0.829
N7-2	0.765	0.816
N8-1	0.764	0.810
N8-2	0.751	0.789
N9-1	-	0.758
N9-2	-	0.728
N10-1	-	0.784
N10-2	-	0.740
N11-1	-	0.829
N11-2	-	0.764

TABLE H-4

## POST CURE PRESSURE CALIBRATION TEST

~~RESULTS FOR LVDT'S 1 TO 6~~

Pressure (psig)	Deflection (Inches) $T = 80 \pm 5^{\circ}\text{F}$					
	*LVDT No.					
	1	2	3	4	5	6
0	0	0	0	0	0	0
9.44	.030	.042	.042	.044	.020	.043
20.0	.063	.078	.085	.091	.043	.084
29.91	.093	.120	.127	.135	.063	.127
39.59	.124	.155	.165	.176	.080	.164
49.45	.156	.190	.203	.217	.092	.202

\* LVDT Nos. 7 and 8 were not installed because of pressure plate interferences.

TABLE H-5

FLEXIBLE CASE GRAIN INTERACTION MOTOR #1 - INITIAL 5 DAYS OF 110°F CONDITIONING  
(OUTPUT IN MV)

TEMP.	TIME/DATE	FUNCTION							
		N-1	N-1-1	N-2	N-2-1	N-3	N-3-1	N-4	N-4-1
64°F	1030 1-8-73	-11.90	-4.67	-4.19	-36.02	-19.15	-4.63	-8.68	-2.79
101°F	1530 1-8-73	-12.39	-5.69	-4.24	-36.37	-18.55	-2.91	-8.96	-3.61
109°F	0830 1-9-73	+12.69	-5.72	-3.43	-27.52	-17.57	-1.61	-9.08	-3.59
108°F	1545 1-9-73	+12.68	-5.69	-3.24	-24.99	-17.33	-1.35	-9.01	-3.51
108°F	0950 1-10-73	+12.59	-5.61	-2.68	-21.21	-16.49	-1.00	-8.84	-3.44
108°F	1530 1-10-73	+12.59	+5.59	-2.51	-18.19	-16.45	- .87	-8.82	-3.42
109°F	1530 1-11-73	+12.49	+5.49	-2.66	-20.94	-16.21	- .63	-8.65	-3.28
109°F	0920 1-12-73	+12.43	+5.42	-1.89	-22.07	-15.91	- .36	-8.52	-3.18
110°F	1430 1-12-73	+12.44	+5.42	-2.11	-22.31	-15.89	- .38	-8.52	-3.18
109°F	1-15-73 1505	+12.46	-5.49	-1.27	-22.19	-15.52	- .07	-8.53	-3.40



TABLE H-5 (CONTINUED)  
FLEXIBLE CASE GRAIN INTERACTION MOTOR #1 - INITIAL 5 DAYS OF 110°F CONDITIONING  
(Output in mv)

TEMP.	TIME/DATE	FUNCTION							
		N-5	N-5-1	N-6	N-6-1	N-7	N-7-1	N-8	N-8-1
64°F	1030 1-8-73	-10.77	-17.99	-7.52	-5.92	-2.52	+1.32	-1.74	+1.39
101°F	1530 1-8-73	-10.39	-17.62	-8.03	-6.77	+1.18	+2.35	+1.26	-3.19
109°F	0830 1-9-73	-9.93	-17.46	-7.86	-6.54	-	+3.64	+ .70	-7.52
108°F	1545 1-9-73	-9.79	-17.37	-7.75	-6.40	-3.97	+2.23	+ .54	-8.03
108°F	0950 1-10-73	-9.35	-17.01	-7.40	-6.11	-3.31	+2.93	+ .14	-9.05
108°F	1530 1-10-73	-9.30	-17.00	-7.36	-6.06	+3.22	+3.06	+ .08	-9.32
109°F	1530 1-11-73	-9.03	-16.85	-7.07	-5.72	+2.85	+3.52	+16.94	-10.02
109°F	0920 1-12-73	-8.88	-16.80	-6.87	-5.54	-2.64	+3.78	+ .35	-10.40
110°F	1430 1-12-73	-8.87	-16.82	-6.86	-5.53	-2.62	+3.79	+ .35	-10.46

TABLE H-5 (CONTINUED)  
FLEXIBLE CASE GRAIN INTERACTION MOTOR #1 - INITIAL 5 DAYS OF 110°F CONDITIONING  
(Output in mv)

TEMP.	TIME/DATE	FUNCTION									
		N-9	N-9-1	N-10	N-10-1	N-11	N-11-1	N-15	N-15-1		
64°F	1030 1-8-73	-8.15	-17.10	-10.55	-14.00	-2.44	-22.33	-11.80	-5.83		
101°F	1530 1-8-73	-8.54	-17.44	-10.35	-14.23	-1.73	-21.73	-12.92	-6.96		
109°F	0830 1-9-73	+8.62	-17.51	-9.56	-13.79	-000	-20.97	-12.24	-7.26		
108°F	1545 1-9-73	-8.50	-17.40	-9.33	-13.57	+ .27	-20.85	-14.19	-7.20		
108°F	0950 1-10-73	-8.12	-17.02	-8.91	-13.22	+ .81	-20.50	-13.35	-7.04		
108°F	1530 1-10-73	+8.09	+16.98	-8.83	-13.15	+ .82	-20.45	-13.85	-7.07		
109°F	1530 1-11-73	+7.85	+16.73	-8.50	-12.84	+1.09	-20.31	-13.49	-7.05		
109°F	0920 1-12-73	+7.68	+16.54	-8.30	-12.67	+1.25	-20.17	-14.23	-7.01		
110°F	1430 1-12-73	-7.66	-16.52	-8.29	-12.65	+1.26	-20.18	-15.57	-7.01		
109°F	1505 1-15-73	+7.62	-16.45	-8.32	-12.73	+1.36	-20.21	-15.75	-7.19		
107°F	1500 1-22-73	-7.23	-16.05	-8.15	-12.53	+1.96	-19.61	-19.11	-7.46		

TABLE H-5 (CONTINUED)  
FLEXIBLE CASE GRAIN INTERACTION MOTOR #1 - INITIAL 5 DAYS OF 110°F CONDITIONING  
(Output in mv)

TEMP.	TIME/DATE	FUNCTION							
		S-1	S-2	S-3	S-4	S-5	S-6	S-7	S-8
64°F	1030 1-8-73	+2.469	+5.903	+2.882	-.355	-11.99	+5.075	+8.170	-3.857
101°F	1530 1-8-73	- .65	+6.78	+ .833	+2.60	-11.15	+6.27	+8.88	+1.65
109°F	0830 1-9-73	-4.411	+9.09V	+ .694	+12.03	-14.40	+5.727	+9.496	+1.513
108°	1545 1-9-73	-4.632	↑	+ .574	+13.087	-14.91	+5.493	+9.638	+1.818
108°F	0950 1-10-73	-5.031	Data* Erratic	+ .604	+15.33	-15.70	+5.242	+9.91	+2.603
109°F	1530 1-11-73	-5.555	↓	+ .418	+16.93	-16.45	+5.008	+9.820	+3.357
108°F	1530 1-10-73	-5.174		+ .491	+15.59	-15.92	+5.135	+9.86	+2.732
109°F	0920 1-12-73	-5.708		+ .463	+17.61	-16.67	+4.904	+9.77	+3.70
110°F	1430 1-12-73	-5.745		+ .467	+17.74	-16.73	+4.862	+9.72	+3.76
109°F	1505 1-15-73	-6.027		+ .405	+19.06	-17.60	+4.738	+9.493	+4.585

\* Gage was disconnected for troubleshooting.  
Gage S-2 was intermittent.



TABLE H-5 (CONTINUED)  
FLEXIBLE CASE GRAIN INTERACTION MOTOR #1 - INITIAL 5 DAYS OF 110°F CONDITIONING  
(Output in mv)

TEMP.	TIME/DATE	FUNCTION					
		S-9	S-10	S-11	S-12	S-13	S-14
64°F	1030 1-8-73	-8.42	+8.99	+3.38	+1.129	-23.52	+1.110
101°F	1530 1-8-73	-8.69	+7.49	+1.92	+2.30	-17.95	- .579
109°F	0830 1-9-73	+9.09	+7.25	+1.488	+3.83	+ .021	-2.362
108°F	1545 1-9-73	+9.14	+7.36	+1.41	+4.05	+3.036	-2-617
108°F	0950 1-10-73	+8.98	+7.57	+1.65	+4.76	+8.81	-3.093
108°F	1530 1-10-73	+9.021	+7.59	+1.649	+4.899	+10.06	-3.238
109°F	1530 1-11-73	+8.951	+7.71	+1.649	+5.462	+13.78	-3.567
109°F	0920 1-12-73	+8.83	+7.73	+1.620	+5.672	+15.64	-3.734
110°F	1430 1-12-73	+8.82	+7.74	+1.61	+5.70	+1.60	-3.770
109°F	1-15-73 1505	+8.08	+7.68	+1.36	+6.16	+19.24	-4.132

TABLE H-6  
FLEXIBLE CASE GRAIN INTERACTION MOTOR #1  
INITIAL 6 WEEKS OF 110°F CONDITIONING  
(Output in mv)

DATE	S-1	S-2	S-3	S-4	S-5	S-6	S-7	S-8
1-22-73	-5.72	+ .56	+ .29	+17.97	-17.87	+4.92	+9.10	+3.82
1-31-73	-6.33	+ .63	+ .26	+17.57	-18.29	+5.15	+8.95	+3.84
2-8-73	-6.26	+ .68	+ .28	+16.56	-18.39	+5.10	+8.67	+3.26
2-16-73	-6.23	+ .65	+ .08	+15.89	-18.63	+5.22	+8.50	+2.80
2-21-73	-6.13	+ .65	+ .02	+15.54	-18.84	+5.40	+8.36	+2.58
3-2-73	-6.25	+1.2	+ .00	+15.17	-19.00	+5.30	+8.16	+2.54
	S-9	S-10	S-11	S-12	S-13	S-14		
1-22-73	-8.64	+7.44	+ .94	+6.43	+17.88	-4.03		
1-31-73	+8.74	+7.17	+ .55	+7.17	+18.52	-4.21		
2-8-73	+9.00	+7.19	+ .49	+7.27	+17.48	-4.11		
2-16-73	+9.31	+6.99	+ .28	+7.56	+16.44	-4.08		
2-21-73	-9.54	+6.87	+ .11	+7.88	+16.29	-4.13		
3-2-73	-9.65	+6.90	+ .12	+8.04	+16.99	-4.26		

TABLE H-6 (CONT.)

FLEXIBLE CASE GRAIN INTERACTION MOTOR #1  
INITIAL 6 WEEKS OF 110°F CONDITIONING  
(Output in mv)  
FUNCTION

DATE	N-1	N-1-1	N-2	N-2-1	N-3	N-3-1	N-4	N-4-1
1-22-73	-12.33	-5.46	-1.07	-20.14	-15.24	- .10	-8.32	-3.28
1-31-73	+12.27	+5.72	- .79	-19.56	-15.08	+ .03	-8.19	-3.08
2-8-73	+12.55	+6.43	- .64	-19.31	-15.07	+ .00	-8.33	-2.97
2-16-72	+12.88	+6.80	- .59	-19.32	-15.12	- .15	-8.56	-3.18
2-21-73	-12.82	-6.89	- .59	-19.47	-15.18	- .29	-8.60	-3.29
3-2-73	-13.20	-7.02	- .35	-119.30	-15.09	- .27	-8.76	-3.34
DATE	N-5	N-5-1	N-6	N-6-1	N-7	N-7-1	N-8	N-8-1
1-22-73	-8.85	-17.26	-6.31	-5.59	-2.19	+4.57	+ .63	-11.04
1-31-73	-8.54	-17.17	-5.81	-4.53	+1.70	+5.22	+1.08	-11.97
2-8-73	-8.43	-17.19	-5.50	-3.02	+1.44	+5.53	+ .86	-11.56
2-16-73	-8.62	-17.47	-5.32	-4.12	+1.30	+5.76	+ .54	-11.02
2-21-73	-8.75	-17.65	-5.29	-4.13	-1.29	+5.81	+ .77	-11.16
3-2-73	-8.47	-17.50	-4.85	-4.13	- .80	+6.31	+ .64	-11.14
DATE	N-9	N-9-1	N-10	N-10-1	N-11	N-11-1	N-15	N-15-1
1-22-73	-7.23	-16.05	-8.15	-12.53	+1.96	-19.61	-19.10	-7.45
1-31-73	-6.69	-15.49	-7.94	-12.28	+2.66	-18.90	-11.45	-7.74
2-8-73	+6.41	+15.21	+7.86	+12.15	+2.73	+18.81	-14.31	-7.88
2-16-73	+6.39	+15.17	-8.01	-12.24	+2.47	-19.07	-15.69	-8.27
2-21-73	-6.27	-15.04	-8.12	-12.31	+2.68	-18.89	-14.72	-8.29
3-2-73	-5.91	-14.68	-7.97	-12.15	+2.89	-18.67	-14.30	-8.24



TABLE H-7  
FLEXIBLE CASE GRAIN INTERACTION MOTOR #1  
MARCH CONDITIONING DATA AT 110°F  
(Output in mv)

DATE	S-1	S-2	S-3	S-4	S-5	S-6	S-7	S-8
3-9-73	-6.49	+1.19	-.003	+14.67	-18.89	+5.54	+8.28	+2.35
3-13-73	-5.92	+1.19	-.001	+14.79	-19.30	+5.47	+7.86	+2.29
3-29-73	-6.56	+ .68	-.06	+14.33	-18.16	+5.19	+7.38	+2.06
	S-9	S-10	S-11	S-12	S-13	S-14		
3-9-73	-9.77	+6.70	-.01	+8.38	+15.84	-4.32		
3-13-73	-9.97	+6.73	-.11	+8.52	+17.16	-4.40		
3-29-73	-10.32	+6.56	+ .33	+8.30	+15.52	-4.30		
DATE	N-1	N-1-1	N-2	N-2-2	N-3	N-3-1	N-4	N-4-1
3-9-73	-13.39	-7.17	-.24	-118.61	-15.23	-.43	-8.82	-3.35
3-13-73	-13.36	-7.14	-.18	-118.21	-15.16	-.59	-8.82	-3.39
3-29-73	-12.96	-6.94	-.20	-11.17	-15.26	-.69	-8.8	-3.36
	N-5	N-5-1	N-6	N-6-1	N-7	N-7-1	N-8	N-8-1
3-9-73	-8.71	-17.63	-4.77	-3.17	-.65	+6.56	+ .07	-11.41
3-13-73	-8.52	-17.68	-4.61	-4.06	-.54	+6.68	+ .87	-11.27
3-29-73	-8.25	-17.56	-4.18	-2.44	+ .11	+7.35	+1.03	-11.70
	N-9	N-9-1	N-10	N-10-1	N-11	N-11-1	N-15	N-15-1
3-9-73	-5.89	-14.65	-8.07	-12.21	+2.82	-18.71	-13.01	-8.50
3-13-73	-5.89	-14.65	-8.07	-12.21	+2.82	-18.71	-13.01	-8.50
3-29-73	-4.97	-13.20	-7.58	-11.34	+3.17	-17.12	-14.04	-8.23

TABLE H-8

NORMAL AND SHEAR GAGE OUTPUT DATA -  
MOTOR NO. 1 THERMAL CYCLING TEST  
(Output in mv)

Date	Temp.	Channel Function	N3-1 19 Days	N3-2 20 N-3-1	N4-1 21 N-4	N4-2 22 N-4-1	N5-1 23 N-5	N5-2 24 N-5-1	N6-1 25 N-6	N6-2 1A N-6-1	N7-1 2A N-7	N7-2 3A N-7-1	N8-1 4A N-8	N8-2 24A N-8-1	N9-1 6A N-9	N9-2 7A N-9-1	N10-1 8A N-10	N10-2 9A N-10-1	N11-1 10A N-11	N11-2 11A N-11-1
3-29-73	110°		-15.26	- .69	-8.8	-3.36	-8.25	-17.56	-4.18	-2.44	+ .11	+7.35	+1.03	-11.7	-4.97	-13.20	-7.58	-11.34	+3.17	+17.12
7-19-73			+0.00	+0.00	+0.00	+0.00	+0.56	+0.01	+0.00	+0.00	+0.01	+0.00	+0.02	--	+0.05	+0.11	+0.08	+0.14	+0.02	+0.16
7-20-73			+0.00	+0.01	+0.00	+0.00	+0.00	+0.00	+0.01	+0.00	+0.00	+0.01	+0.00	--	+0.04	+0.10	+0.08	+0.11	+0.02	+0.17
7-23-73			+0.00	+0.00	+0.00	+0.00	+0.00	+0.00	+0.00	+0.00	+0.00	+0.01	+0.01	--	+0.03	+0.07	+0.05	+0.06	+0.00	+0.12
7-24-73	80	262	+0.01	+0.00	+0.02	+0.00	+0.00	+0.02	+0.01	+0.00	+0.01	+0.01	+0.01	--	+4.46	+12.69	+10.61	+12.69	+1.15	+20.30
7-25-73	80	263	+0.01	+0.02	+0.01	+0.00	+0.00	+0.02	+0.01	+0.01	+0.01	+0.01	+0.01	--	+4.66	+12.81	+11.05	+12.90	+0.59	+20.79
7-27-73	80	265	+0.05	+0.05	+0.03	+0.03	+0.56	+0.04	+0.03	-0.00	+0.00	+0.00	+0.00	+0.00	+4.85	+12.90	+11.49	+13.04	+0.32	+21.14
7-30-73	80	268	+0.02	+0.02	+0.02	+0.02	+0.01	+0.02	+0.01	-0.00	+0.01	+0.00	+0.02	+0.01	+5.19	+13.17	+11.83	+13.18	-0.85	+21.01
7-30-73	80	268.5	-23.77	-10.14	-11.83	-7.39	+0.24	+0.22	+0.18	+0.23	+0.33	+0.21	-7.66	+7.00	+6.03	+13.22	+11.50	+14.04	+0.46	+21.03
7-31-73	80	269	-24.10	-10.52	-12.03	-7.49	+0.05	+0.05	+0.05	+0.06	+0.07	+0.05	-8.03	+8.73	+4.95	+12.40	+11.18	+13.24	+0.46	+20.95
8-1-73	60	270	-24.85	-11.21	-12.42	-7.88	+0.04	+0.04	+0.04	+0.05	+0.06	+0.04	-9.03	+10.27	+5.25	+12.52	+12.52	+14.45	+2.24	+22.35
8-3-73	60	272	-24.84	-11.37	-12.54	-8.00	+0.04	+0.04	+0.04	+0.05	+0.06	+0.04	-9.44	+10.75	+5.41	+12.64	+13.02	+14.86	+2.78	+22.79
8-6-73	60	275	-24.66	-11.47	-12.67	-8.11	+0.03	+0.04	+0.03	+0.05	+0.04	+0.04	-10.05	+11.56	+5.71	+12.84	+13.77	+15.46	+3.33	+23.26
8-7-73	60	276	-24.66	-11.46	-12.61	-8.07	-0.02	+0.03	-0.02	-0.02	-0.03	-0.02	-10.09	+11.58	+5.75	+12.06	+13.89	-15.52	+3.38	+23.31
8-9-73	60	278	-24.72	-11.53	-12.66	-8.14	+0.02	+0.03	+0.01	+0.01	+0.02	+0.01	-10.46	+12.01	+5.90	+12.99	+14.12	+15.64	+3.56	+23.45
8-10-73	110	279	-24.54	-10.41	-12.43	-7.75	+0.02	+0.01	+0.02	+0.02	+0.02	+0.02	-9.23	+7.35	+5.68	+12.87	+12.79	+14.51	+0.89	+21.40
8-13-73	110	282	-23.74	-9.47	-12.28	-7.72	+0.01	+0.03	+0.03	+0.02	+0.04	+0.03	-7.87	+2.05	+6.33	+13.00	+12.09	+13.72	+0.82	+20.62
8-14-73	110	283	-22.84	-9.14	-12.48	-7.73	+0.02	+0.06	+0.00	+0.03	+0.05	+0.03	+6.56	+3.15	+6.57	+13.60	+11.78	+13.42	-2.09	+20.07
8-20-73	110	289	-22.36	-8.48	-12.11	-7.49	+0.03	+0.03	+0.01	+0.02	+0.03	+0.03	+3.39	+7.74	+6.30	+12.94	+12.28	+13.23	-4.56	+18.28
8-21-73	110	290	-21.90	-7.34	-11.86	-7.34	+0.03	+0.03	+0.02	+0.03	+0.03	+0.03	+8.08	+7.87	+6.11	+12.53	+12.17	+13.02	+4.62	+17.64
8-23-73	110	292	-21.11	-7.75	-11.49	-6.99	-16.70	+28.27	-10.04	-11.75	+3.34	+3.04	+2.33	+8.95	+6.14	+12.50	+12.33	+13.09	-5.23	+17.12
8-24-73	110	293	-21.13	-7.77	-11.43	-6.92	-18.59	+28.21	-10.01	-11.70	+3.24	+3.11	+2.14	+4.23	+6.18	+12.49	+12.48	+13.15	-5.43	+16.91
8-27-73	80	296	-22.06	-9.50	-11.53	-7.02	-17.88	+29.09	-10.24	-11.89	+6.39	+0.39	+4.15	+0.83	+7.00	+13.37	+14.15	+16.39	+3.10	+18.01
8-28-73	80	297	-22.27	-9.62	-11.63	-7.09	-17.99	+29.16	-10.40	-12.16	+6.67	+0.15	+4.45	+0.52	+7.14	+13.49	+14.42	+14.63	+2.88	+18.16
8-29-73	80	298	-22.39	-9.72	-11.71	-7.14	-18.05	+29.20	+10.53	-12.28	+6.87	-0.03	-4.73	+0.13	+7.28	+13.63	+14.68	+14.86	+2.66	+18.30
8-31-73	80	300	-22.49	-9.80	-11.70	-7.14	-18.07	+29.22	+10.63	-12.37	+7.02	-0.17	-0.20	+0.34	+7.37	+13.69	+14.92	+15.05	+2.44	+18.47
9-4-73	80	304	-22.50	-9.81	-11.70	-7.05	-18.05	+29.18	+10.64	-12.40	+6.93	-0.05	-5.57	+0.67	+7.53	+13.80	+15.15	+15.21	+2.17	+18.77
9-5-73	80	305	-22.45	-9.69	-11.58	-6.91	-17.92	+29.07	-10.54	-12.30	+6.78	+0.25	-5.42	+0.35	+7.57	+13.84	+15.10	+15.16	-2.35	+18.66
9-6-73	60	306	-23.40	-11.13	-11.61	-7.12	-18.25	+29.63	-10.41	-12.09	+8.37	-1.38	-6.34	+5.70	+7.32	+13.51	+15.90	+15.59	+0.53	+19.60
9-7-73	60	307	-23.76	-11.59	-11.83	-7.33	-18.40	+29.79	-10.85	-12.47	+9.37	-2.27	-6.90	+6.55	+7.41	+13.55	+16.48	+18.11	+0.54	+20.42
9-10-73	60	310	-24.41	-12.01	-12.08	-7.56	-18.61	+30.06	-11.67	-13.22	+10.44	-3.41	-8.06	+8.09	+7.61	+12.67	+17.66	+17.21	+2.13	+21.79

TABLE H-8 (CONT.)

NORMAL AND SHEAR GAGE OUTPUT DATA -  
MOTOR NO. 1 THERMAL CYCLING TEST  
(Output in mv)

Date	Temp.	Channel Function	Days	N3-1	N3-2	N4-1	N4-2	N5-1	N5-2	N6-1	N6-2	N7-1	N7-2	N8-1	N8-2	N9-1	N9-2	N10-1	N10-2	N11-1	N11-2
				19	20	21	22	23	24	25	1A	2A	3A	4A	24A	6A	7A	8A	9A	10A	11A
		N-3	N-3-1	N-4	N-3-1	N-4	N-4-1	N-5	N-5-1	N-6	N-6-1	N-7	N-7-1	N-8	N-8-1	N-9	N-9-1	N-10	N-10-1	N-11	N-11-1
9-11-73	60	311		-24.57	-12.00	-12.15	-7.59	-18.62	+30.08	-11.75	-13.32	+10.83	-3.54	-8.36	+8.37	+7.74	+13.74	+17.86	+17.46	+2.31	+22.06
9-12-73	60	312		-24.67	-11.98	-12.25	-7.55	-18.82	+30.27	+12.02	-13.54	+10.92	-3.79	-8.50	+8.46	+7.84	+13.86	+18.04	+17.55	+2.56	+22.16
9-13-73	60	313		-24.64	-12.11	-12.18	-7.56	-18.83	+30.24	-11.90	+13.51	-24.06	-3.83	-8.76	+8.85	+7.92	+13.89	+18.05	+17.64	+2.74	+22.21
9-14-73	60	314		-24.77	-12.16	-12.32	-7.68	-18.99	+30.34	+12.07	-13.29	+10.97	-3.82	-9.05	+9.18	+8.02	-14.05	+18.32	+17.79	+2.93	+22.47
9-17-73	60	317		-24.61	-11.98	-12.20	-7.45	-18.64	+30.21	-11.86	-12.83	+10.73	-3.63	-9.17	+9.23	+7.96	+13.91	+18.12	+17.68	+2.94	+22.39
9-20-73	110	320		-22.69	-9.01	-9.58	-6.63	-17.67	+29.69	+10.89	-12.01	+5.79	+1.07	+6.61	-2.53	+8.64	+14.82	+15.77	+15.85	+1.22	+20.21

Approx. 70° Value  
for Vibration Test  
Checkout

+2

+15

+7

-5

-7

-11

-18

-7

-10

D.C. Levels at time of Vibration Testing

12-12-73 76 -24.9 -12.1 -11.6 -8.6 -18.5 -10.6 -7.9 -8.3 -13.4 -22.6 -6.29

12-12-73 76 N-15-1 N-15-2 3D-1 -2 -3 -4 -5 -6 -9.7 +45.6 +26.8 +15.3 +42.3 +57.0 +38.7



TABLE H-8 (CONT.)  
NORMAL AND SHEAR GAGE OUTPUT DATA - MOTOR NO. 1 THERMAL CYCLING TEST

Date	Temp.	Channel Function Days	12A N-15	13A N-15-1	14A 3D-1	15A 3D-2	16A 3D-3	17A 3D-4	18A 3D-5	19A 3D-6	LVDT- 1	LVDT- 2	LVDT- 3	LVDT- 4	LVDT- 5	SG-1	SG-2	SG-3	LVDT- 6
7-19-73			0.00	0.01	-	-	-	-	-	-									
7-20-73			+0.00	+0.01	-	-	-	-	-	-									
7-24-73			+0.00	+0.00	-	-	-	-	-	-									
7-25-73			+0.01	+0.00	-	-	-	-	-	-									
7-27-73			+0.01	+0.00	+0.00	+0.00	+0.01	+0.02	+0.02	+0.01						15790	15810	16162	
7-30-73			+0.01	+0.00	+0.01	+0.01	+0.01	+0.01	+0.01	+0.00						16410	16500	16740	
7-30-73			-14.28	-9.54	+62.69	+44.64	-41.95	-61.65	+82.98	+51.09									
7-31-73			-44.58	-11.11	+54.78	-36.03	-31.89	-52.33	+76.35	+43.56									
8-1-75			-52.40	-12.97	+50.26	-30.12	-23.21	-45.44	+71.59	+38.75									
8-3-73			-59.71	-13.36	+49.31	-28.79	-21.08	-46.82	+70.43	+37.6						16485	16590	16882	
8-6-73			-56.47	-13.91	+46.39	-25.07	-16.11	-39.83	+67.14	+34.42						16460	16550	16810	
8-7-73			-65.68	-13.99	-45.44	-23.98	-14.95	-38.75	+66.78	+33.70						16530	16600	16870	
8-9-73			-62.04	-14.11	+44.34	-22.61	-13.45	-37.42	+64.80	+32.09						16460	16600	16820	
8-10-73			-59.06	-14.13	+35.79	-13.65	-4.38	-28.58	+53.05	+22.51						16430	16545	16850	
8-13-73			-44.60	-13.29	+42.23	-22.90	-20.86	-41.00	+59.32	+29.24						16585	16700	17000	
8-14-73			-43.71	-13.44	+36.61	-18.40	-19.27	-37.58	+51.81	+23.14						16700	16740	16690	
8-20-73			-37.01	-12.36	+24.40	-35.21	-42.39	-56.67	+67.37	+38.30						17170	17340	17600	
8-21-73			-35.96	-11.98	+50.81	-35.48	-42.72	-56.55	+66.90	+38.43						17370	17430	17700	
8-23-73			-36.65	-10.98	+50.49	+35.48	-43.09	-56.56	+66.47	+38.18									
8-24-73			-36.19	-10.79	+50.56	-35.70	-43.50	-56.81	+66.53	+38.26						17130	17240	17520	
8-27-73			-23.68	-10.72	+55.83	-37.58	-37.91	-56.75	+74.02	+43.78						16860	16980	17240	
8-28-73			-44.63	-10.86	+53.61	-35.11	-34.25	-53.76	+71.59	+41.40						16780	16840	17100	
8-29-73			-46.93	-10.93	+52.23	-33.55	-31.83	-51.83	+70.05	+39.90						16660	16730	17070	
8-31-73			-47.29	-11.05	+50.50	-31.65	-28.91	-49.48	+68.14	+38.06						16860	16980	17260	
9-4-73			-48.07	-11.18	-	-31.35	-27.97	-48.94	+67.48	+37.79						16940	17080	17360	
9-5-73			-46.21	-10.94	+51.92	-33.06	-30.11	-50.84	+68.83	+39.29	5.147	2.592	4.541	5.106	3.929	16870	16950	17250	4.476
9-6-73			-45.32	-10.79	+58.97	-40.10	-35.44	-57.01	+79.35	+47.31	5.066	2.406	4.421	5.233	3.825	16630	16750	17000	4.353
9-7-73			-49.26	-11.28	+55.57	-35.95	-29.17	-51.94	+75.93	+43.69	4.837	2.193	4.161	5.169	3.624	16450	16490	16820	4.114
9-10-73			-52.74	-12.19	+51.53	-30.82	-21.08	-45.52	+71.36	+39.28	4.644	1.962	3.908	5.196	3.430	16810	16920	17220	3.860
9-11-73			-58.22	-	+50.86	-29.86	-19.89	-44.45	+70.54	+38.44	4.623	1.935	3.892	5.202	3.409	16740	16835	17100	3.828
9-12-73			-59.14	-12.51	+49.23	-28.15	-17.87	-42.63	+68.76	+36.93	4.587	1.903	3.836	5.183	3.378	16630	16720	17020	3.790
9-13-73			-62.79	-12.49	+48.51	-27.27	-16.68	-41.63	+67.81	+35.97	4.585	1.894	3.833	5.193	3.371	16650	16770	17020	3.778
9-14-73			-63.07	-12.58	+47.69	+26.38	+15.86	+40.86	+67.03	+35.12	4.575	1.880	3.818	5.196	3.360	16230	16280	16750	3.764
9-17-73			-65.53	-12.66	+44.22	-22.76	-12.03	-37.13	+63.30	+31.55						16900	16610	16520	
9-20-73			-63.89	-12.08	+26.99	+6.98	+1.22	-23.63	+42.48	+12.40									

TABLE H-8 (CONT.)

## NORMAL AND SHEAR GAGE OUTPUT DATA - MOTOR NO. 1 THERMAL CYCLING TEST

		(Output in mv)																	
Temp.	Channel Function Days	1	2	3	4	23A	6	7	8	9	10	11	12	13	14	15	16	17	18
		S-1	S-2	S-3	S-4	S-5	S-6	S-7	S-8	S-9	S-10	S-11	S-12	S-13	S-14	N-1-1	N-1-2	N-2-1	N-2-2
3-29-73	110																		
	7-18-73	+0.00	+0.00	+0.01	+0.00	+0.07	-0.02	-0.01	+0.02	+0.01	+0.01	-0.00	-0.01	-3.07	-0.01	-12.96	-6.94	- .20	-11.17
	7-20-73	+0.01	+0.02	+0.02	+0.01	+0.09	-0.03	-0.01	+0.02	+0.00	+0.00	+0.00	-0.02	-3.12	-0.02	-0.15	-0.07	-0.00	-0.11
	7-23-73	+0.01	+0.03	+0.01	-0.03	+0.05	-0.03	+0.02	+0.00	+0.01	+0.00	+0.00	+0.00	+1.78	+0.00	+0.08	+0.05	+0.00	-0.07
7-24-73	80	+0.01	+0.00	+0.01	+0.01	+10.28	-1.23	-1.95	+1.57	+0.01	+0.00	+0.07	+2.21	Over/ Scale	-2.59	-18.19	-10.98	-4.81	-0.38
7-25-73	80	+0.00	+0.02	+0.00	+0.02	+9.89	+1.73	+2.00	+2.16	+0.03	+0.01	+0.25	+2.11	"	-1.79	-18.29	-11.08	-4.88	-0.40
7-27-73	80	+0.02	+0.06	+0.00	+0.06	+9.62	-1.51	-1.98	+2.61	+0.04	+0.01	+0.25	+1.95	"	-1.69	-18.41	-11.21	-4.94	-0.27
7-30-73	80	+0.01	+0.03	+0.01	+0.03	+10.45	-1.65	-2.03	+1.64	-0.01	+0.00	-0.12	-2.21	"	-2.61	-18.34	-11.16	-4.95	-0.31
7-30-73	80	268.5	+ .12	+ .28	+0.37	+0.21	+7.55	+0.02	+2.20	+3.00	+0.06	+6.49	+1.81	357.7	-1.62	-18.61	-11.38	-4.88	-0.21
7-31-73	80	+ .04	+0.11	+0.09	+0.11	+6.87	-1.32	-3.09	+4.35	-0.01	+5.59	+0.99	-1.88	Over/ Scale	+0.52	-18.60	-11.33	-4.25	-0.42
8-1-73	60	+0.03	+0.05	+0.07	+0.06	+5.62	-3.07	-4.03	+4.31	+0.01	+2.76	+4.18	-0.83	"	+3.06	-19.03	-11.73	-4.30	-0.42
8-3-73	60	+0.03	+0.05	+0.06	+0.06	+5.37	-3.43	-4.19	+4.11	+0.01	+2.02	+4.97	-0.54	"	+3.75	-19.18	-11.87	-4.33	-0.03
8-6-73	60	+0.03	+0.03	+0.05	+0.04	+4.93	-3.89	-4.32	+3.60	+0.03	+0.95	+6.01	+0.01	"	+4.90	-19.41	-12.06	-4.15	-0.43
8-7-73	60	-0.02	-0.02	+0.03	-0.03	+4.89	-3.85	-4.16	+3.45	-0.12	-0.88	+5.96	-0.15	"	-5.07	-13.44	-12.06	-4.11	-0.95
8-9-73	60	+0.00	+0.01	+0.03	+0.03	+4.83	-3.73	-4.23	+6.40	+0.02	+ .90	+5.79	+0.47	"	+5.29	-19.62	-12.21	-4.10	-0.45
8-10-73	110	+0.00	+0.08	+0.03	+0.01	+6.46	-3.15	-2.50	+1.66	+0.09	+1.94	+2.74	+0.33	"	+1.61	-19.30	-11.98	-4.01	-0.46
8-13-73	110	+0.00	+0.01	+0.05	+0.01	+9.62	-1.87	-1.79	+0.95	+0.04	+4.93	+0.58	+1.40	"	-0.00	-19.09	-11.73	-4.44	-0.46
8-14-75	110	+0.02	+0.06	+0.05	+0.01	+12.39	-2.94	-2.76	+0.34	+0.10	+4.56	+0.02	+4.62	"	-5.10	-18.90	-11.67	-4.39	-0.48
8-20-73	110	+0.02	+0.00	+0.04	+0.02	+13.45	-3.37	-3.47	-0.49	+0.04	+6.14	-0.62	+7.60	"	-7.10	-18.16	-11.23	-3.37	-0.44
8-21-73	110	+0.01	+0.00	+0.04	+0.02	+13.12	-3.92	-3.38	-0.53	+0.03	+6.02	-0.54	+7.45	"	-0.95	-17.62	-10.84	-3.25	-0.43
8-23-73	110	-6.99	+0.51	Over/ Scale	+12.57	+13.20	+4.02	+3.45	-0.81	-12.50	+6.05	+0.38	+7.61	"	-7.04	-17.37	-10.83	-2.23	-0.09
8-24-73	110	-7.05	+0.55	"	+12.72	+13.22	-4.04	-3.47	-0.94	-12.51	+6.11	+0.34	+7.63	"	-7.06	-17.34	-10.55	-2.16	-0.08
8-27-73	80	-1.83	+1.32	"	+2.38	+8.10	-2.14	-1.87	+2.80	-11.00	+5.91	+0.17	+2.76	"	-2.79	-17.93	-11.00	-3.53	-0.03
8-28-73	80	-1.80	+1.38	"	+6.69	+7.96	-2.27	-1.82	+2.81	+10.77	+5.36	+0.72	+2.43	"	-2.25	-18.09	-11.16	-3.64	-0.06
8-29-73	80	-1.83	+1.50	"	+1.29	+7.88	+2.35	+1.81	+2.83	+10.63	+5.01	+1.10	-2.11	"	-1.83	-18.18	-11.21	-3.69	-0.04
8-31-73	80	-1.96	+1.65	"	+0.90	+7.87	+2.48	+1.81	+2.75	+10.51	+4.54	+1.41	-1.77	"	-1.31	-18.33	-11.30	-3.71	-0.05
9-4-73	80	-2.38	+1.63	"	+0.92	+8.20	-2.58	-1.97	+2.57	+10.59	+4.21	+1.40	+1.62	"	-1.28	-18.44	-11.35	-3.74	-0.03
9-5-73	80	-2.71	+1.39	"	+1.41	+8.47	-2.44	-1.72	+2.32	-10.81	+4.62	+0.95	-1.83	"	-1.84	-18.35	-11.22	-3.67	-0.27
9-6-73	60	+0.95	+3.29	"	+2.74	+5.98	+1.86	+2.28	+5.13	+9.31	+4.46	+2.23	+0.83	"	+1.31	-18.63	-11.33	-3.62	-0.40
9-7-73	60	+0.97	+4.49	"	-3.29	+5.26	-2.99	-3.01	+5.01	+8.07	+2.60	+4.19	+0.47	"	+0.04	-18.89	-11.61	-3.76	-0.84
9-10-73	60	+0.47	+5.50	"	-3.07	+4.41	-4.42	-3.94	+4.04	-6.68	+0.28	+7.03	+0.54	"	+5.52	-19.38	-12.01	-3.96	+0.32
9-11-73	60	+0.30	+5.70	"	+3.05	+4.39	+4.49	+6.02	+3.75	+6.44	+0.21	+7.28	+0.71	"	+5.93	-19.48	-12.14	-3.95	+0.46
9-12-73	60	+0.28	+5.72	"	+3.05	+4.34	+4.55	+4.08	+3.68	+6.50	+0.23	+7.33	+0.79	"	+6.23	-19.69	-12.15	-4.12	+0.35

TABLE H-8 (CONT.)  
NORMAL AND SHEAR GAGE OUTPUT DATA - MOTOR NO. 1 THERMAL CYCLING TEST

(Output in mv)

Temp.	Channel	1	2	3	4	23A	6	7	8	9	10	11	12	13	14	15	16	17	18
Function	S-1	S-2	S-3	S-4	S-5	S-6	S-7	S-8	S-9	S-10	S-11	S-12	S-13	S-14	S-15	N-1-1	N-1-2	N-2-1	N-2-2
Days																			
313	+0.22	+5.40	Over/ Scale	+2.98	+4.31	+4.52	+3.98	+3.44	-6.46	+0.23	+7.32	+0.86	Over/ Scale	+6.36	-19.79	-12.20	-4.12	-0.31	
314	+0.24	+5.86	+392.16	+3.02	+4.29	+4.46	-4.04	+3.40	+6.64	+0.24	+7.10	+1.09	"	+6.32	-19.98	-12.49	-4.16	-0.78	
317	-0.27	+5.64	Over/ Scale	-2.83	+4.40	-4.26	-3.72	+3.22	+6.71	+0.22	+6.79	+1.51	"	+6.35	-19.93	-12.38	-4.13	-0.94	
320	-6.60	+1.42	"	14.51	+11.30	+3.61	+2.62	+1.33	+10.61	+3.06	+1.93	+1.62	"	-2.60	-19.37	-11.82	-3.65	-0.84	
Approx. 70°F Value for Vib. Test Checkout	-2	+2	+7	+1	+8	-4	-4	+3	+11	+4	+1	-2	-30	-1	-11	-5			

D.C. Levels at time of vibration testing on 12 December 1973 T = 76°F

-6.0 +5.8 8.70 9.6 -1.89 +3.62 +5.18 +16.2 -8.5 +.91 +3.61 +1.56 -365 +5.38 -12.6 -4.8



APPENDIX I

SUMMARY OF VIBRATION TEST RESULTS

## SUMMARY OF VIBRATION TEST RESULTS

## A. OBJECTIVES

The overall objective of the thrust and transverse axis vibration survey tests of Minuteman III Stage III Motor No. 1 was to characterize the dynamic response functions resulting from the application of controlled constant acceleration and constant force sinusoidal input excitations.

Emphasis was directed in this phase of the test program towards the acquisition of high quality dynamic input and response data throughout the test frequency range of 10 to 300 cps. Special care was taken to ensure that the sinusoidal acceleration and force input excitation functions have a minimum wave form distortion over the specified test frequency range.

The specific objectives of the vibration survey tests of Motor No. 1 were:

1. To record on a magnetic tape recorder the dynamic responses of all designated dynamic transducers to constant acceleration and constant force sinusoidal excitation functions to be applied at a slow sweep rate of one octave/min. to the thrust (Y) and transverse (Z) axis of the motor over a test frequency range of 10 to 300 cps.
2. To identify all significant resonant and anti-resonant frequencies of the thrust (Y) axis and transverse (Z) axis motor test configurations occurring in the 10-300 cps test frequency range.
3. To determine the mode shapes and corresponding dynamic normal and shear stress measurements at each significant resonant and anti-resonant frequency detected during the Y and Z axis surveys.

The detailed test procedures for conducting the low input level, constant acceleration and constant force survey tests in the Y (thrust) and Z (transverse) axis of the full scale third stage Minuteman motor are specified in test plan 1826-26-TP.

## B. DESCRIPTION OF MOTOR TEST CONFIGURATIONS

## 1. General

During the thrust (Y) axis and the transverse (Z) axis survey tests the motor (including a fired nozzle attached for c.g. considerations) was suspended in a horizontal attitude from an overhead cable support system. These cables were attached to motor handling rings that are bolted to the forward and aft motor skirts. The angular orientation of the motor during both the Y and Z axis survey tests was such that the  $0^\circ$  -  $180^\circ$  (X) axis of the motor is vertical (perpendicular to the floor) with the  $0^\circ$  location of the motor in the "down" position.

## 2. Thrust (Y) Axis Test Configuration

The vibration survey tests of the Y-axis test configuration of the motor were conducted with the use of one Ling Model A-249 electrodynamic exciter. The Y axis vibratory excitation was applied to the motor forward skirt through a rigid conical drive fixture that was bolted to the motor forward skirt. An impedance head was positioned between the moving element of the electrodynamic exciter and the small-diameter end of the conical drive fixture. A schematic drawing of the Y-axis test setup is shown in Figure I-1.

## 3. Transverse Z (90° - 270°) Axis Test Configuration

The vibration survey tests of the Z-axis motor test configuration was conducted with the use of two Ling Model A-249 electrodynamic exciters. One exciter was attached to the forward skirt motor handling ring and the other was attached to the aft skirt motor handling ring. The Z-axis vibratory excitation was applied transversely to the motor through the two motor skirt handling rings along an axis that is parallel to the 90° - 270° axis of the motor. An impedance head (or dynamic force gage) was installed between each exciter moving element and motor skirt vibration input point. A schematic drawing of the test setup used in conducting the transverse Z-axis vibration survey tests of motor No. 1 is shown in Figure I-2.

Both in-phase and out-of-phase Z-axis vibration survey tests were conducted by operating the two exciters in-phase and 180° out-of-phase with respect to each other.

## C. INSTRUMENTATION AND DATA ACQUISITION REQUIREMENTS

### 1. Dynamic Instrumentation Requirements

#### a. Thrust (Y) Axis Vibration Survey Tests

Forty-six channels of dynamic response data were continuously recorded on a magnetic tape recorder and oscillographs during the Y-axis vibration test. The channels of Y-axis dynamic response data and dynamic transducers were:

• Impedance Head	1 Channel
• Accelerometers	20 Channels
• Normal Stress Gages	11 Channels
• Shear Stress Gages	14 Channels



Detailed descriptions of the locations of the dynamic transducer during the Y-axis vibration survey tests are shown in Figure I-1.

b. Transducer Z-Axis Vibration Survey Tests

Forty-six channels of dynamic response data were continuously recorded on a magnetic tape recorder and oscillographs during the Z-axis vibration survey test. The channels of Z-axis dynamic response data were the same as for the Y-axis tests. The locations of the dynamic transducers during the Z-axis vibration survey tests are shown in Figure I-2.

D. TESTS

1. Thrust Axis

The first series of tests to be conducted was in the thrust axis. The types of vibration tests conducted in the thrust axis of the motor were listed as follows:

Run 001	10-100 HZ Sweep, 5K Constant Force
001A	10-100 HZ Sweep, 1K Constant Force
002	10-300 HZ Sweep, .5 g Constant Acceleration
003	30-56 HZ Dwell, .5 g Constant Acceleration
004	10-300 HZ Sweep, 1 g Constant Acceleration
005	44 HZ Dwell, 20 sec, 1 g Constant Acceleration
Run 006	36.7 HZ Dwell, 20 sec, 1 g Constant Acceleration
007	10-27 HZ at 1.5 g (Wave form was bad)
008	10-300 HZ Sweep, 1.5 g Constant Acceleration
009	44.4 HZ Dwell, 1.5 g Constant Acceleration
010	38.4 HZ Dwell, 1.5 g Constant Acceleration
011	10-59 HZ, (Bad cycle), 2 g Constant Acceleration
012	30-37 HZ, Rerun, 2 g Constant Acceleration
013	30-38 HZ, Rerun, 2 g Constant Acceleration
014	46-306 HZ Sweep, 2 g Constant Acceleration
015	45.1 HZ Dwell, 2 g Constant Acceleration

Results of the series of survey tests conducted in the transverse axis indicated that the propellant axial shear mode is between 36 to 44 HZ.

The sine sweep survey test consisted of one sine sweep from 10 to 300 cps using a logarithmic sweep rate of one octave/min. The resonant frequency dwell test consisted of 20-second discrete frequency dwell test conducted at a certain acceleration input and at the resonant frequency determined during the sine sweep survey test. The input control accelerometers for all constant acceleration tests were accelerometers G-30-Z and G-31-Z.

## 2. Transverse Axis

The series of survey tests conducted in the transverse axis were listed in the following test sequence (a total of 21 runs were completed).

<u>Run Nos.</u>	<u>Z (Transverse Axis)</u>
016	1 g sine sweep survey (10-300 Hz)
017	(No oscillator data - just repeat for accel. plots)
017	1 g sine sweep survey (103-00 Hz) - the repeat 1 g sweep was required in order to plot accelerometer data from tape - D.C. to frequency conditioning amplifier overloaded on run 016 resulting in compressed information.
018	1,000# force sine sweep (10-300 Hz)
019	32.7 Hz at 1,000# force (anti-resonance)
020	45.7 Hz at 1 g (propellant ray)
021	77.5 Hz at 1,000#/per shaker force (principal resonance)
022	77.5 Hz at 1.0 g
023	123.7 Hz at 800#/per shaker force (propellant due to nozzle)
024	123.7 at 1 g
025	1,000# force sine sweep (180° out of phase - 10-300)
026	1 g sine sweep survey (180° out of phase - 10-300)
027	40.9 Hz at 1 g (180° out of phase)
028	82.3 Hz at 1 g (180° out of phase)

<u>Run Nos.</u>	<u>Z (Transverse Axis)</u>
029	Nozzle - 115.4 Hz at 1 g (180° out of phase)
030	Nozzle - 153.2 Hz at 1 g (180° out of phase)
031	2 g sine sweep survey (in phase) 10-300 Hz
032	44.6 Hz at 2 g (in phase)
033	110.4 Hz at 2 g (in phase)
034	73.8 Hz at 2 g (in phase)
035	124.8 Hz at 2 g (in phase)
036	124.8 Hz - G30Z at 1 g, G31Z at 2 g (in phase)

The large number of discrete frequency runs made during the transverse axis testing were required to provide additional data to define the dynamic characteristics of the motor and its components. As previously described, 15 runs were conducted in the Y (thrust) axis.

The vibration testing of full scale Motor No. 1 was essentially conducted in accordance with test plan 1826-26-TP, dated September 1972. Deviations in procedure are described in the following paragraph.

The force transducers available for the test (Endevco Model 2110 impedance heads) have a 10,000 pound maximum dynamic range. During the 1/2 g pre-run tests, it was determined that the force necessary to produce 1/2 g acceleration was approximately 7,000 lbs, therefore the force to produce 1 g acceleration would be almost double and exceed the capacity of the 10,000 impedance head. To prevent damage, therefore, the transducers were removed after completion of the 1/2 g constant acceleration and the constant force surveys in the Y axis and after the 1 g survey in the Z axis.

#### E. VIBRATION TEST DATA

Typical response data obtained from the normal stress and shear gages during the vibration runs are presented in Figures I-3 through I-5. Similar data for the accelerometers are given in Figures I-6 through I-11. The complete data package from the vibration testing is located in the documentary file.

Sine sweep survey tests were conducted over a 10 to 300 Hz frequency range at each of the following acceleration levels: 0.5 g, 1.0 g, 1.5 g and 2.0 g. Input excitation was controlled using accelerometer G-8Y, mounted on the forward skirt of the motor adjacent to the drive ring. The data plots indicate primary propellant responses occurred in the 45 Hz fre-



quency range in the thrust (Y) axis. Figure I-6 shows the relative responses on the chamber near the forward "Y" joint (G-10Y), at Station 26 (G-11Y), and near the aft "Y" joint (G-12Y). Figures I-7 through I-9 show the propellant response (solid line) versus the chamber response (dotted line) at Stations 38, 26 and 9 for a 1 g input at G-8Y. The "Q" of the propellant response at Station 38, 26 and 9 are approximately 5.6, 3.6, and 3.2 respectively. Typical responses at the igniter boss (G-13Y) and on the aft motor dome (G-14Y) are illustrated in Figures I-10 and I-11. The igniter boss shows a Q of 14, and the aft dome shows a Q of 2.8.

The shear stresses measured at 2 g longitudinal acceleration during the resonance frequency dwell at 45 Hz (Table I-1) show a gradual increase in magnitude from the aft end where the grain is attached to the case to the forward end where the end is booted and free to move. It is interesting to point out that during the 1.5 g survey tests, two frequencies were of interest, 38 Hz and 45 Hz. It was estimated that the 38 Hz is an anti-resonance frequency and the 45 Hz is the principal resonance frequency of the grain.

The normal and shear gage outputs for the vertical transverse vibration mode for runs No. 020 and No. 032, which were both at the resonant frequency of approximately 45 Hz in this mode of vibration are shown in Figure 47 in the text. The two runs were at  $\pm 1$  g and  $\pm 2$  g acceleration levels and the increase in signal with g level is apparent. However, most of the gages do not give a linear response as a function of acceleration. This may be because of the slight shift in frequency from 45.7 Hz to 44.6 Hz from the 1 g to the 2 g run.

The data from the 3D gage situated near the head end flap termination are also given in Figure 47 of Volume I. These data seem to correspond fairly well with the other gage data in similar locations. The large output from shear gage SH-13 and from the 3D gage in the  $\tau_{rz}$  plane suggests that there is a good deal of motion of the star points in this mode of vibration. There also seems to be some coupled fore and aft motion in the motor.

Figure 48 in the text shows similar data from the axial vibration tests. These data were measured during runs 05 and 09 at 1.0 and 1.5 g level respectively. The resonant frequency in this direction was 44.4 Hz.

These data seem very good in the sense that the stresses from the gages show a maximum response at the aft end adjacent to the knuckle area. The stress becomes smaller along the mid section of the grain and are smallest just short of the end of the forward flap. The axial shear stresses are generally the largest stresses along the center of the motor but around the aft end dome, the shear stresses gradually change into fore and aft tension/compression stresses. Thus at the rear end of the motor, the normal stresses from gages N2 and N3 situated at the knuckle are the most significant. Gages N1 and N4 further round the aft dome exhibit smaller tensile/compressive stresses and after this point the major stresses are the shear stresses. As might be anticipated the greatest shear stress occurs at the flap termination and is noted by gage SH-13. This stress is approximately 1.5 x the shear stress at the middle of the grain (gages SH-9 and SH-11).

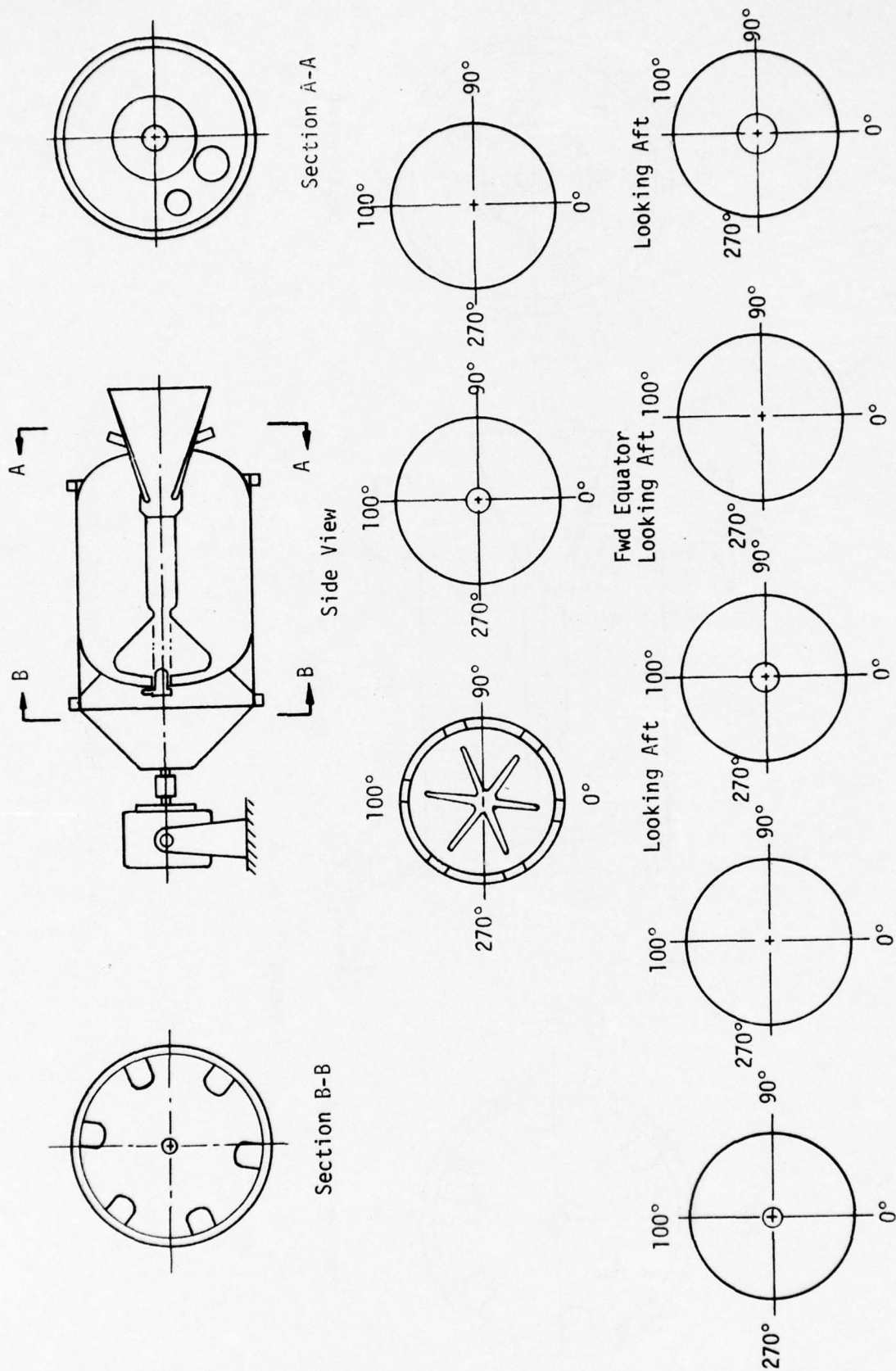


FIGURE I-1. THRUST (Y) AXIS TEST SET-UP

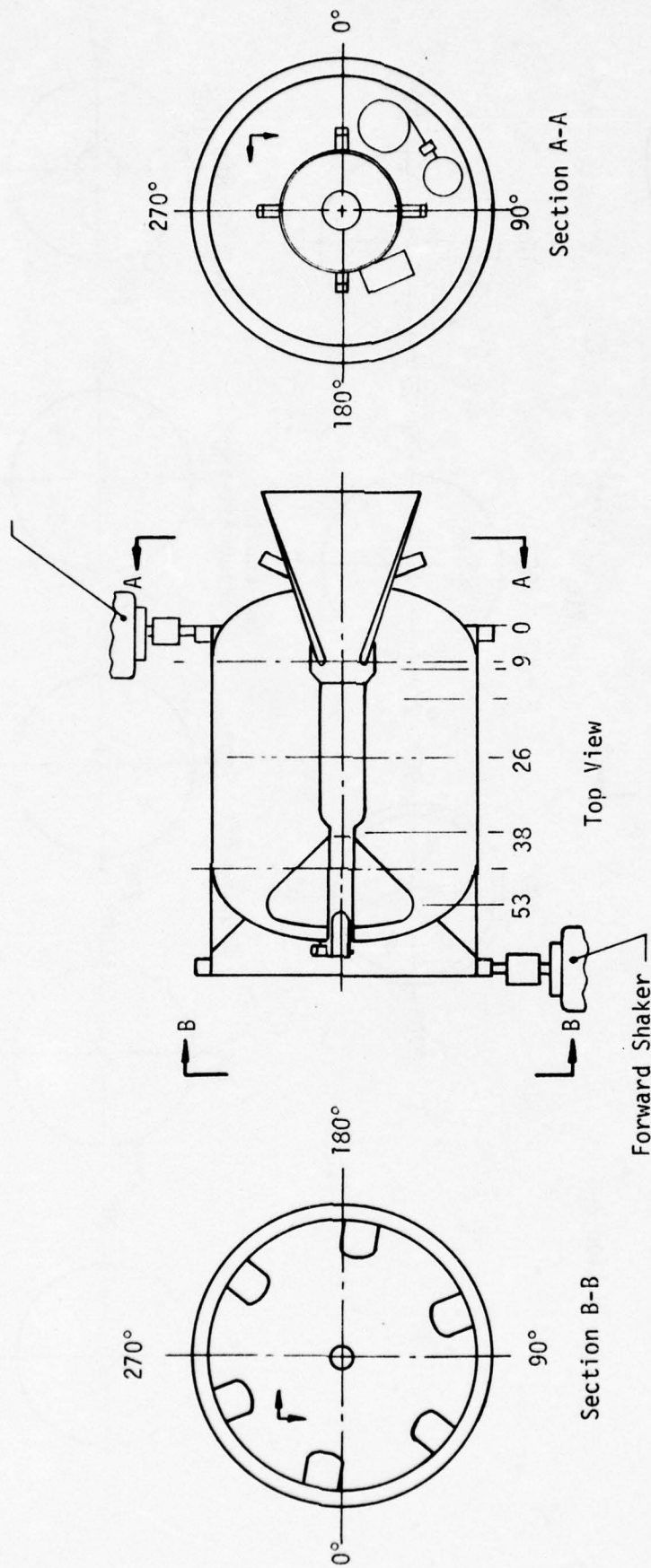


FIGURE I-2. TRANSVERSE (Z) AXIS TEST SET-UP



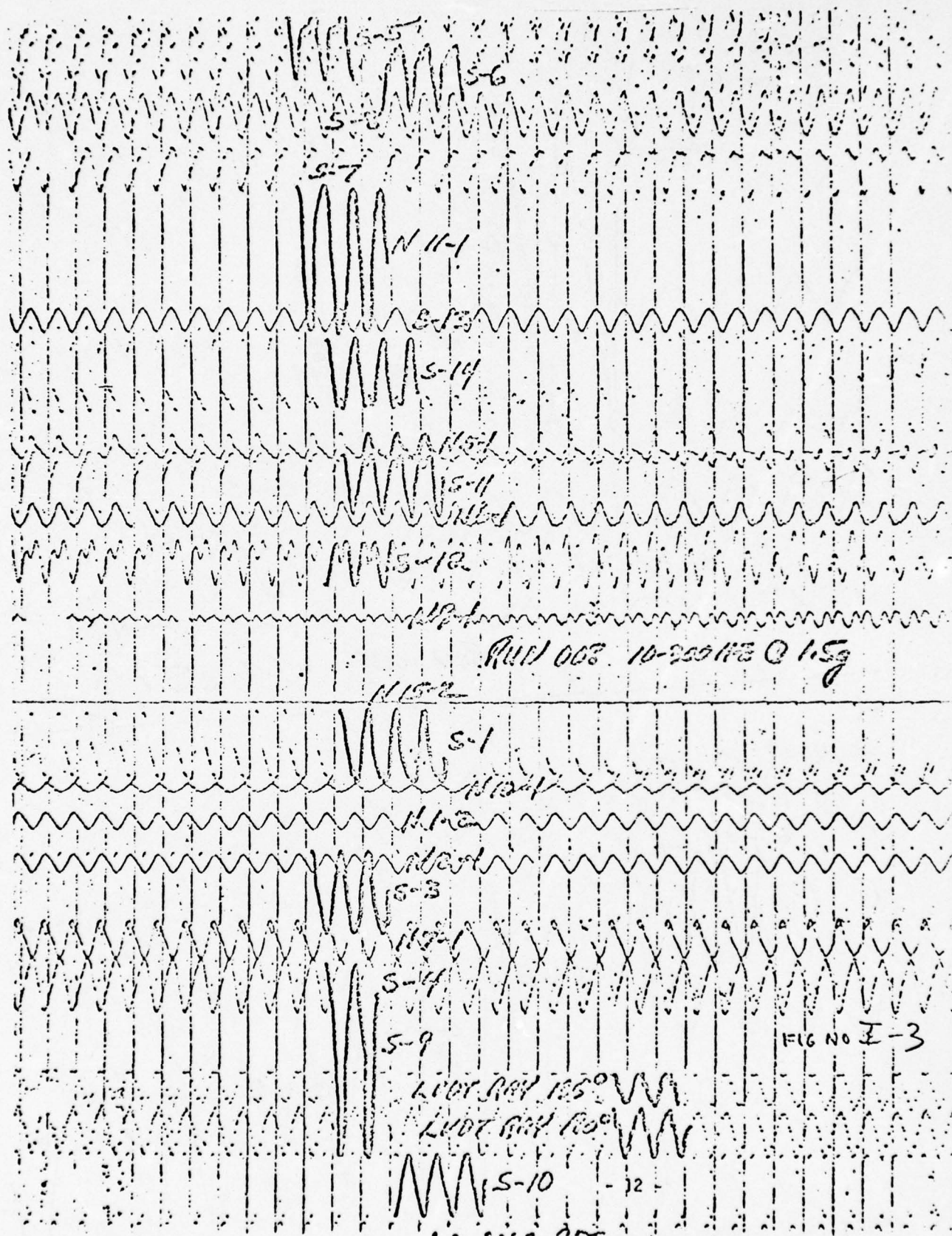


FIGURE I-3. STRESS GAGE RESPONSE DATA FOR 10 TO 300 HZ SWEEP AT 1.5 G ACCELERATION

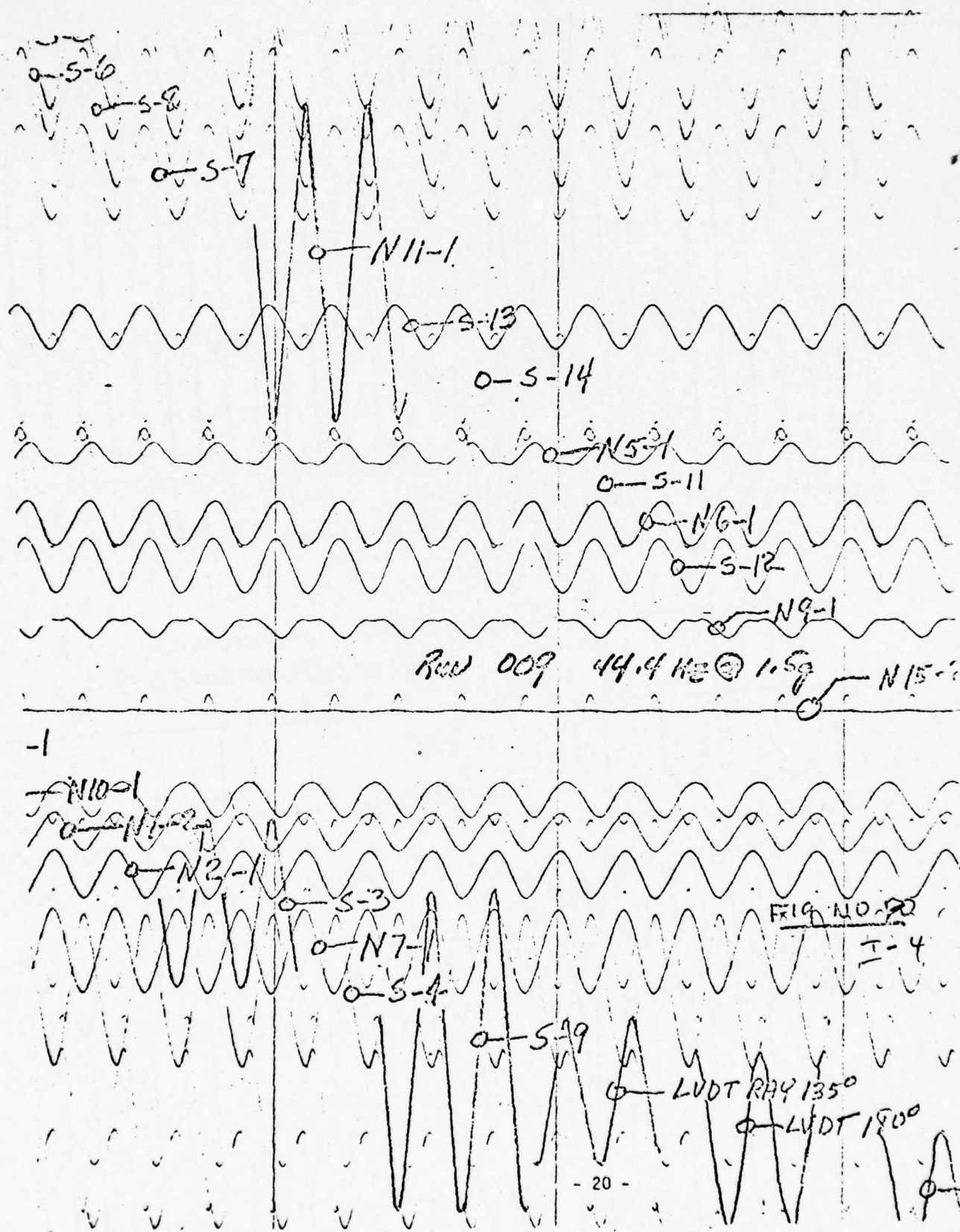


FIGURE I-4. STRESS GAGE RESPONSE DATA FOR 44.4 HZ DWELL TEST AT 1.5 G CONSTANT ACCELERATION

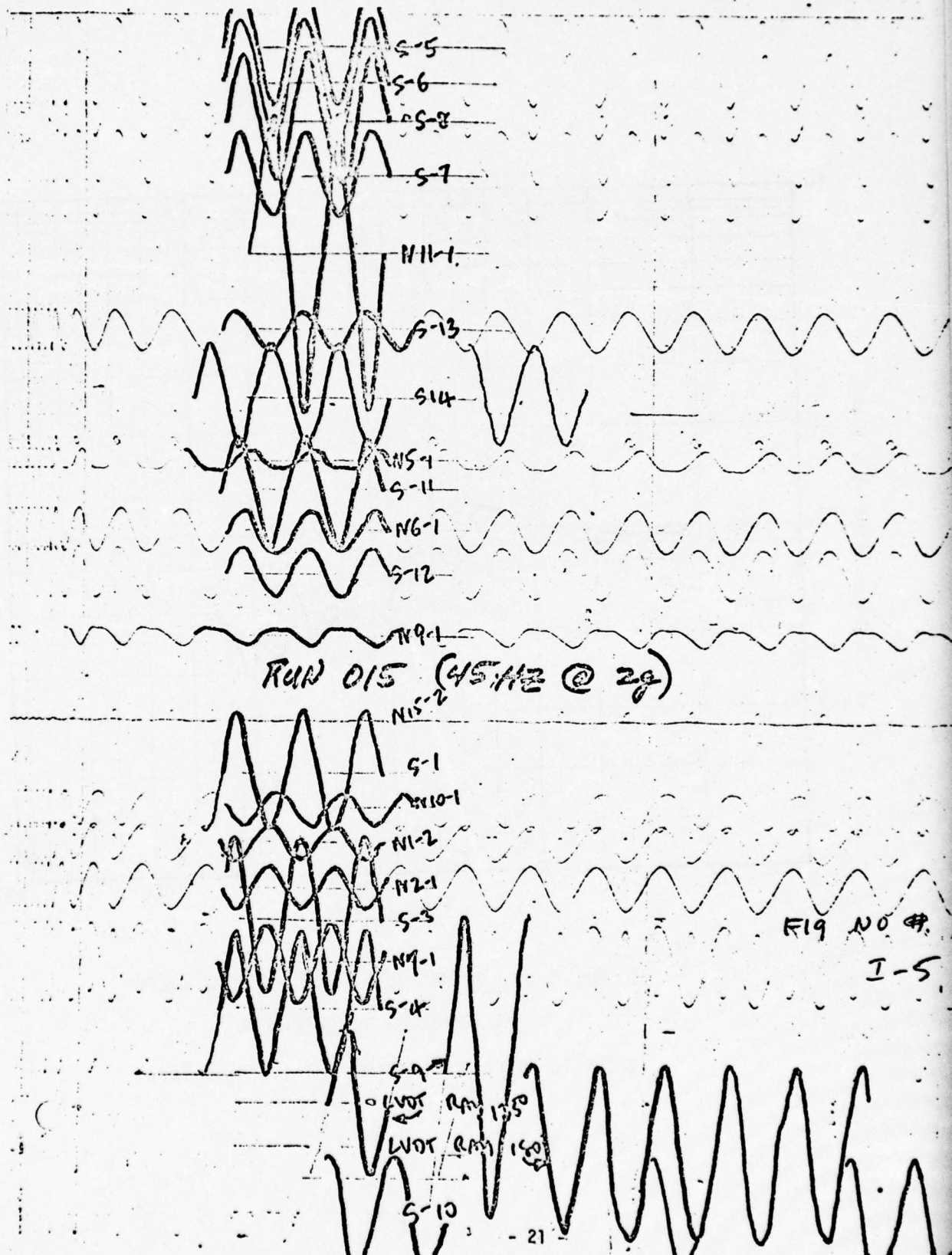


FIGURE I-5. STRESS GAGE RESPONSE DATA FOR 45.1 HZ DWELL TEST  
AT 2 G CONSTANT ACCELERATION



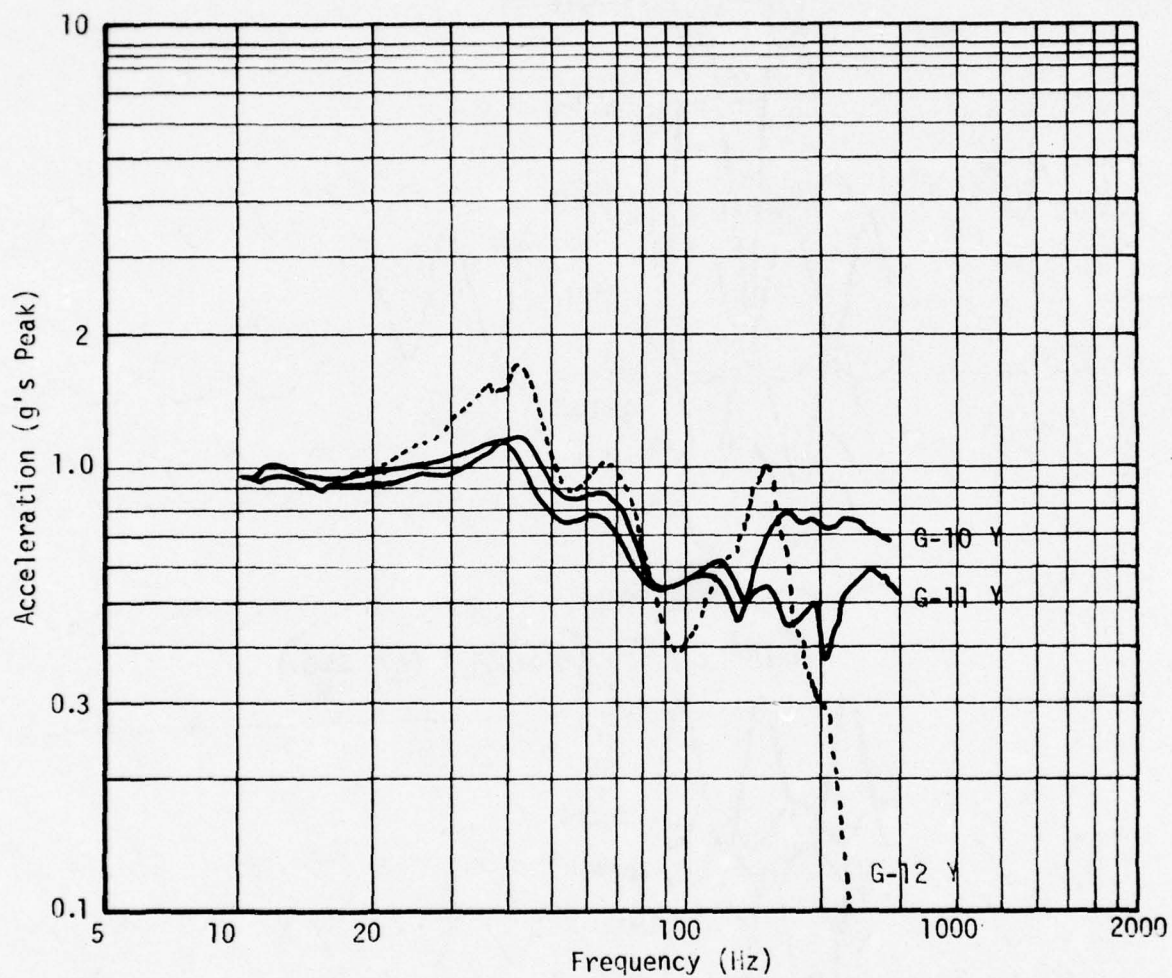


FIGURE I-6. RUN NO. 004 - 10 TO 300 HZ SWEEP  
 AT 1G CONSTANT ACCELERATION  
 (ACCELEROMETER NOS. G-10Y, G-11Y, AND G-12Y)

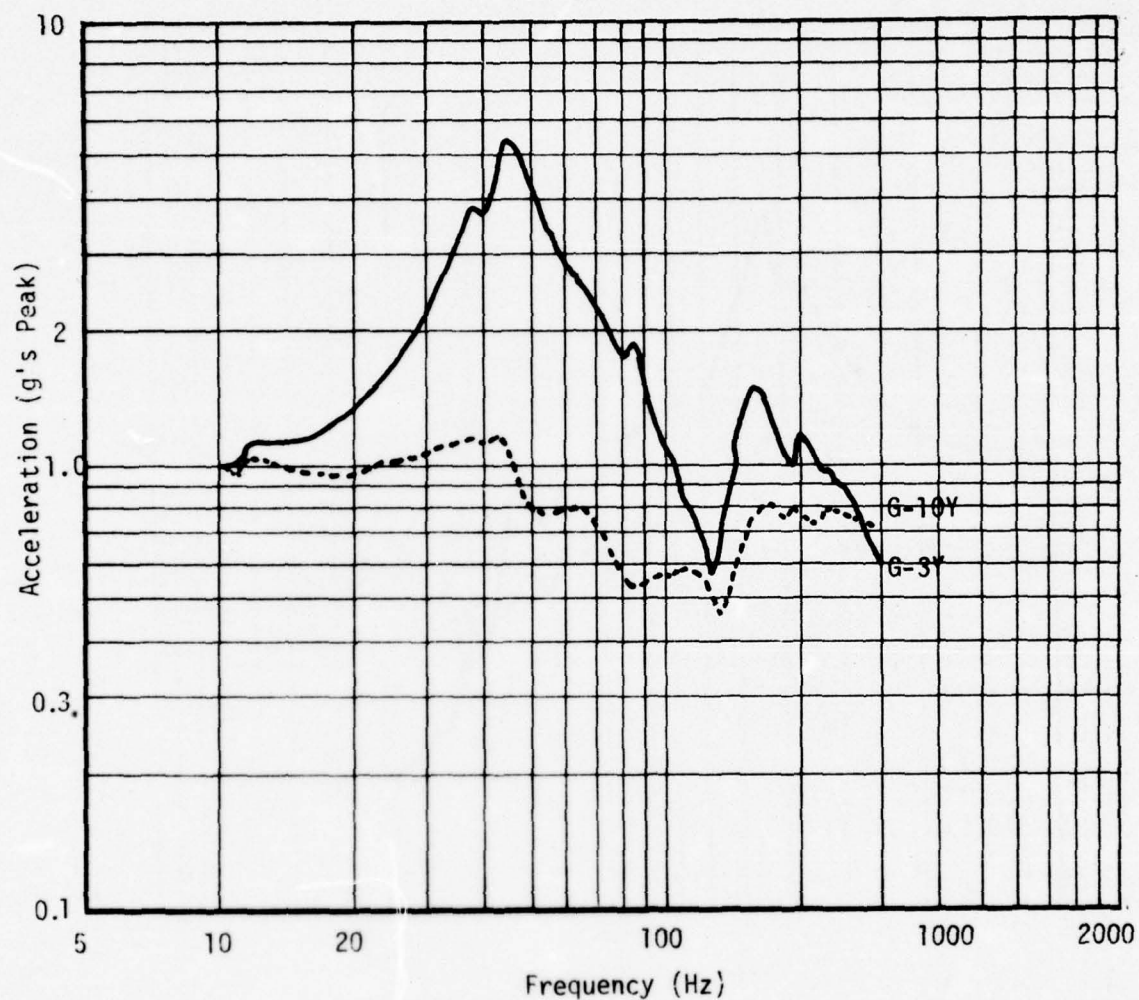


FIGURE I-7. RUN NO. 004 - 10 TO 300 HZ SWEEP  
AT 1G CONSTANT ACCELERATION  
(ACCELEROMETER NOS. G-3Y AND G-10Y)

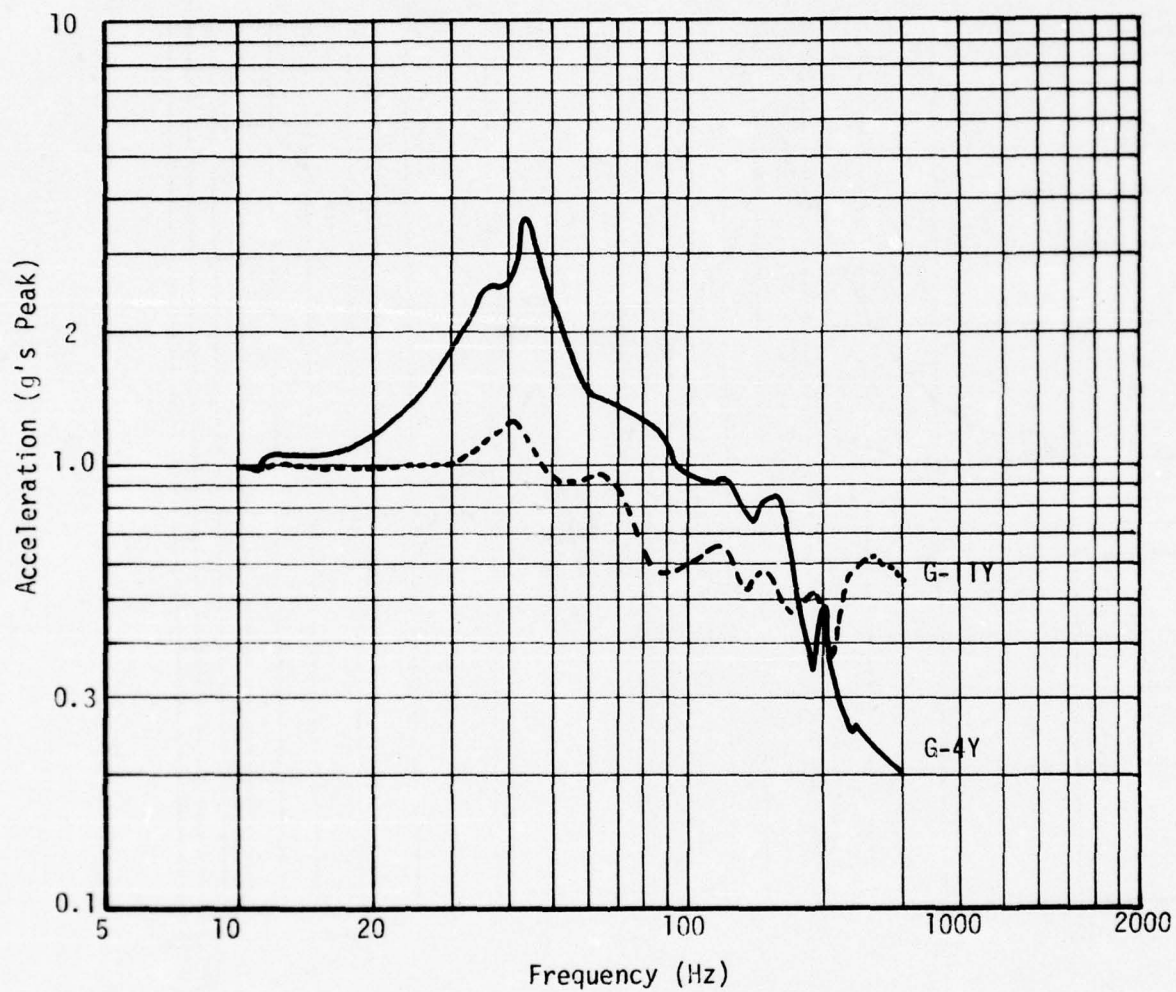


FIGURE I-8. RUN NO. 004 - 10 TO 300 HZ SWEEP  
AT 1G CONSTANT ACCELERATION  
(ACCELEROMETER NOS. G-4Y AND G-11Y)



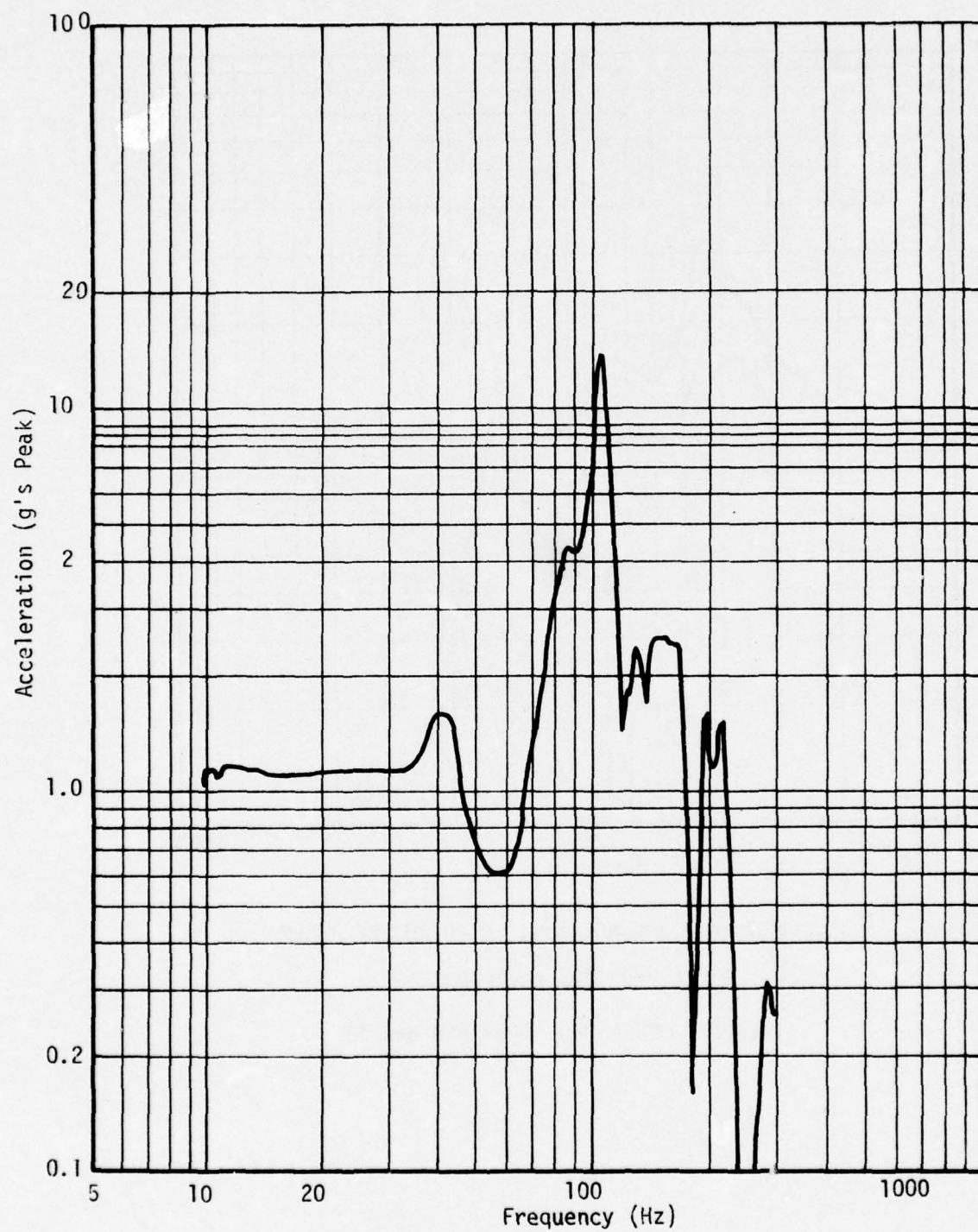


FIGURE I-9. RUN NO. 004, 10 TO 300 HZ SWEEP  
AT 1G CONSTANT ACCELERATION  
(ACCELEROMETER NO. G-13Y)

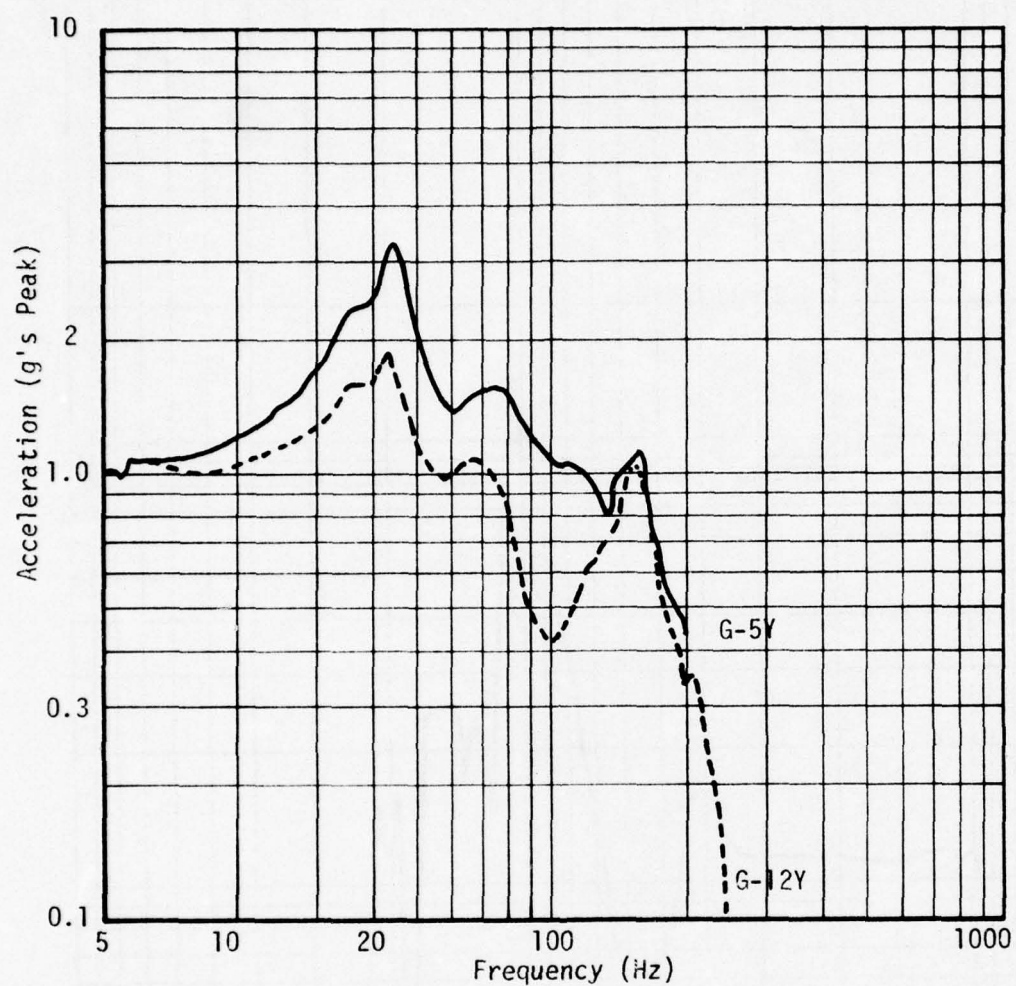


FIGURE I-10. RUN NO. 004, 10 TO 300 HZ SWEEP  
AT 1G CONSTANT ACCELERATION  
(ACCELEROMETER NOS. G-5Y AND G-12Y)

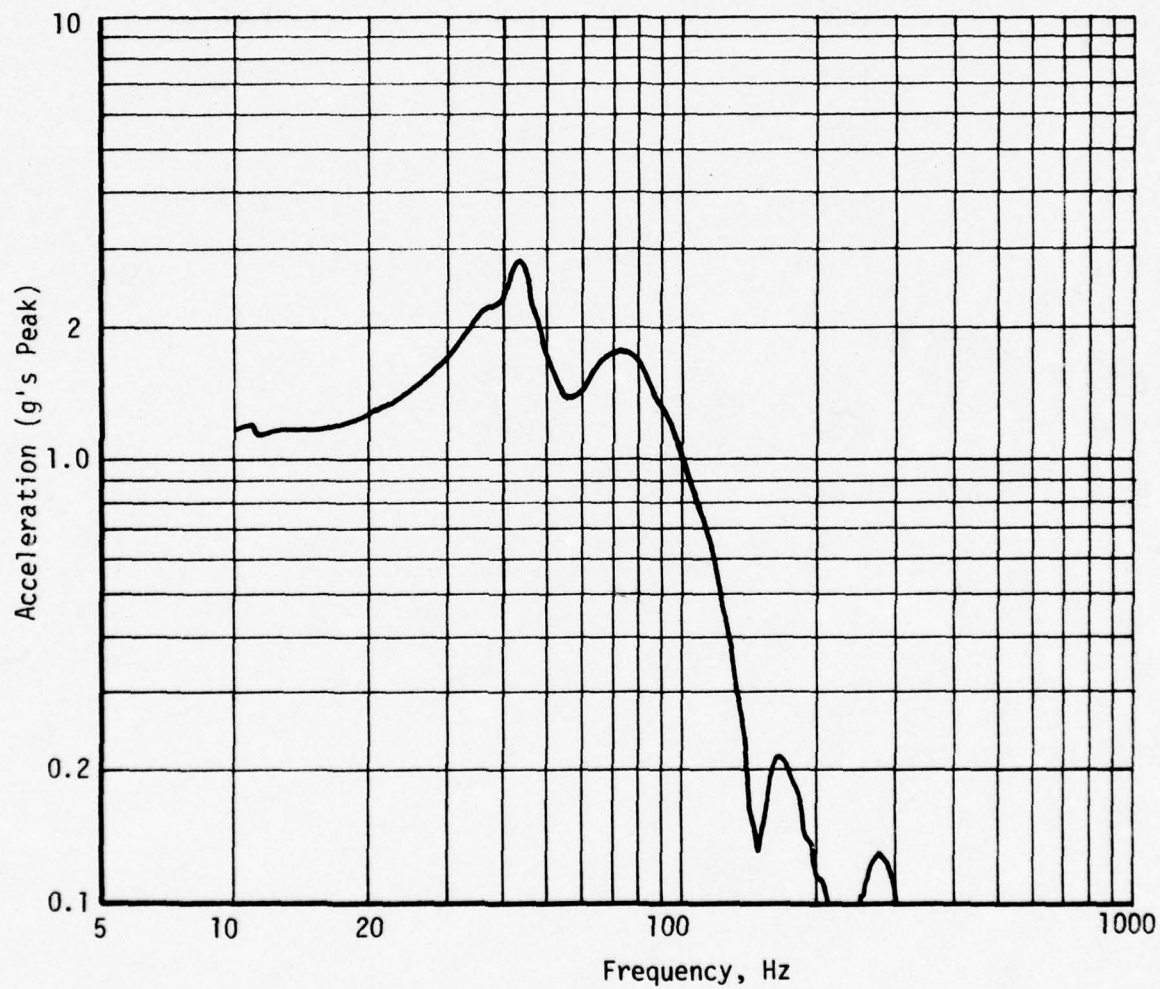


FIGURE I-11. RUN NO. 004 - 10 TO 300 HZ SWEEP AT 1 G  
CONSTANT ACCELERATION (ACCEL. NO. G-14 G)



TABLE I-1  
 VIBRATION STRESSES MOTOR NO. 1 AT RESONANCE 45.1 Hz:  
 2 g ACCELERATION LEVEL  
 LONGITUDINAL VIBRATION TEST

<u>Gage No.</u>	<u>Stress PK-PK</u>	<u>Gage No.</u>	<u>Stress PK-PK</u>
S-1	1.92	N1-2	5.7
S-3	1.76	N2-1	7.58
S-4	1.65	N5-1	1.36
S-5	1.07	N6-1	2.75
S-6	1.82	N7-1	4.63
S-7	1.40	N9-1	1.28
S-8	.95	N10-1	1.93
S-9	3.33	N11-1	15.92
S-10	4.09	N15-2	--
S-13	6.21		

APPENDIX J

SUMMARY OF THE HANDLING AND  
TRANSPORTATION TESTS

## HANDLING AND TRANSPORTATION TESTS

## A. HANDLING TESTS

The handling tests were conducted to subject the motor to all of the required handling conditions: lifting, tip over and rolling. During each of the tests performed in the Manufacturing Area, the propellant stress/strain gages were monitored continuously on an oscillograph.

The primary objective of this test was to simulate the conditions to which a typical motor was subjected during processing operations and to verify that the stresses produced during handling are of a low magnitude and, therefore, relatively unimportant to overall motor design.

## 1. Description and Test Procedures For Handling Tests

## a. Lifting Operations

(1) With Motor No. 1 in a horizontal position on a trailer, transfer the motor to an area accessible to an overhead hoist (Figure J-1).

(2) Zero and balance all instrument channels and record zero reading.

(3) Turn on recorders prior to the lifting operation; monitor and record all channels continuously until end of the operation.

## b. Rolling Operation

(1) Transfer Motor No. 1 to the roller fixture permanently attached to the building.

(2) Zero and balance all channels and record zero readings.

(3) Turn on recorders prior to the start of the rolling operation; monitor and record all channels continuously until five complete revolutions have been made and the test is stopped.

## c. Tipping Operations

(1) Transfer the motor from the roller fixture to the T-416438 transport trailer shown in Figure J-2.

(2) Move the motor to area accessible to an overhead hoist and attach hoist to yoke on the aft-end trailer.



(3) Zero and balance all instrument channels and record zero readings.

(4) Turn on recorder prior to tipping operation. Monitor and record all channels continuously until end of tipping operation with the longitudinal axis of the motor elevated to an angle of 90°.

## 2. Summary of Handling Tests

The rolling test was conducted in the Manufacturing area using a Polaris rolling fixture. The test involved three complete rolling cycles of the motor. Twenty-four channels of gages were recorded dynamically onto two oscillographs. The horizontal lifting test was conducted and monitored continuously with the 24 channels of dynamic recording. The test was repeated in slow motion to enable monitoring the internal LVDT's and D.C. pots which are recorded on a static recording system. The tipping and vertical lifting were conducted and monitored continuously with the dynamic recording system. That operation was repeated in slow motion in order to record the channels on the static recording system. This was done by stopping at every 10 degree increment to 90 degrees or to the vertical position. At the vertical position, the motor was lifted vertically and all channels were recorded.

The data obtained from the roll tests performed as a part of the handling tests on the motor are summarized in Figures 45 and 46 of Volume I. The stress changes appear to be reasonable in view of the calculated lateral stresses given in Appendix P. The greatest shear stress was obtained from gage SH-5 (Figure 45 of Volume I). This was unusual since gage SH-5 was positioned in a fore and aft direction and should not respond significantly to the roll test. Both shear gages SH-12 and SH-14 which were rotated through 90° so as to measure these types of stress were inoperative during the roll test. Gage SH-12 failed during the test and gage SH-14 was erratic.

With the exception of SH-5, a typical shear stress variation during a roll test was about 0.3 psi from peak to peak. The normal stress gages show larger outputs as would be expected. From these gages it appeared that roll stresses about 2.5 to 3.0 psi peak-to-peak may be anticipated. The stresses due to handling must be added to the existing thermal and gravitational stresses.

The results of the vertical tilt test from a horizontal position to a vertical position are given in Table J-1. All of the stress differences were small except for gage N11-1 which is at the aft equator. Without a stress analysis it is hard to make a good assessment. But, these stresses are less than was anticipated.

## B. TRANSPORTATION TESTS

The primary objective of this test was to simulate the conditions to which the stage II Minuteman motor was subjected to during transportation and to monitor the stresses produced during transportation.

The first test involved the transfer from the storage vehicle and consisted of lifting and placing the motor in a horizontal position from the cart (T-103457) shown in Figure J-1 to the transporter shown in Figure J-2. This test is very similar to the lifting series of the handling test. The second half of the transportation test consisted of the transfer of the motor from the transporter (TPC Harness) to a utility van. The motor was lifted horizontally with forward end facing the van and placed on transfer rails on the loading platform. It was then towed along the tracks of the platform to the tracks leading into the van with a motor driven winch. The recorders were turned on prior to lifting in both cases and all channels which were selected to be monitored, were recorded continuously until the end of operation.

### 1. Description of Test Vehicles

#### a. Transfer From Storage to Transporter

A Boeing Company TBC operational harness and an ASPC semi-trailer were used for this test of Motor No. 1.

#### b. Transfer Motor From Transporter to Transporter

A semi-trailer and a utility van were used in this test where Motor No. 1 was transferred from its original position in the semi-trailer to the utility van.

### 2. Test Procedures

#### a. Transfer From Storage to Transporter

During this operation the stresses of the propellant grain were monitored. The test procedures were as follows:

- (1) Hookup and connect instrumentation.
- (2) Zero and balance all channels. Record all zero readings.
- (3) Monitor and record all channels continuously until end of operation.

b. Transfer of Motor From Transporter to Transporter

During this operation the stresses of the propellant grain were monitored. The test procedures were as follows:

(1) Safety: The motor must be grounded to van at all times after installation.

(2) Safety: Do not detach winch cable or brake cable until motor is tied down.

(3) Ensure van brakes are set. Install two jacks beneath the rear end of the van, one jack under each van rail.

(4) Clean interior of van thoroughly.

(5) Remove the aft spreader track from the aft end of van.

(6) Attach the transfer rails and "A" frame assembly to trailer rails, then adjust jacks under van and adjust "A" frame as required to level. Level in both the longitudinal and transverse directions.

(7) Instrumentation, monitoring schedule, and data acquisition system were identical to the storage-to-transporter test.

(8) Position the tow cable through the pulley in van and secure to the carriage forward end. Secure the other cable to the aft end and take up slack in both cables.

(9) Zero and balance all instrumentation channels. Record zero readings.

(10) Using the winch, tow the motor into the van and position it over forward tie down area.

(11) Monitor and record all channels continuously until end of operation.

(12) Stop recorder, rezero instrumentation.

(13) Position motor installed on its trailer adjacent to transfer rails and accessible to overhead hoist; then using overhead hoist and T-1021847 spanner beam with four 1/2" dia x 40-1/2" long cables and 3/4" shackles connected to carriage hoist points, lift motor clear of its trailer and position on the transfer rails with forward end of motor facing toward van. Turn recorders on prior to lifting and monitor all channels continuously until end of operation.

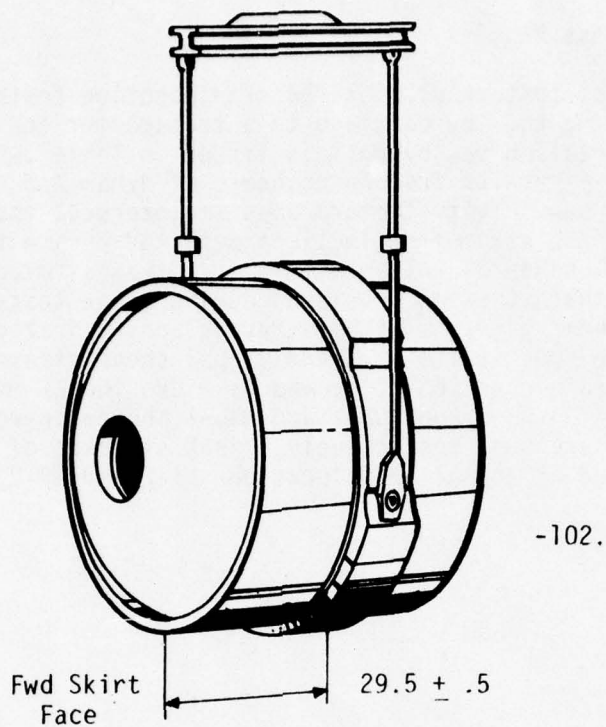
NOTE: ENSURE BOTH CABLES ARE TAUT DURING MOTOR TRANSFER



### 3. Test Results

The test results of the transportation tests which consist of transferring the TBC carriage to a transporter and subsequently to the MM transportation van by rail is listed in Table J-2. Table J-2 summarizes the test results from 24 channels of dynamic data and it lists the voltage sensitivity factors used to interpret the data on the oscillograms, the maximum deflections detected during these tests, and the resulting change of voltages (mv) and strains (micro inch). Results indicate that stresses developed during these tests were negligible. For the shear gages, maximum stresses occurred at gage locations S-9 and S-10 where approximately .2 and .1 psi shear stresses occurred, respectively. Strain gage SCAT-2 showed 41 micro inches/inch of strain in the forward head. Normal gages N8-1 and N10-1 showed approximately 1.5 and 1.0 psi peak stresses, respectively. Peak stresses of approximately 5 to 7 psi occurred at normal gage locations N1-2 and N4-2.

T-1017442  
or  
T-1020611



T-1020611 02  
T-1013442  
Pneumatic Ring

Fwd Skirt  
Face

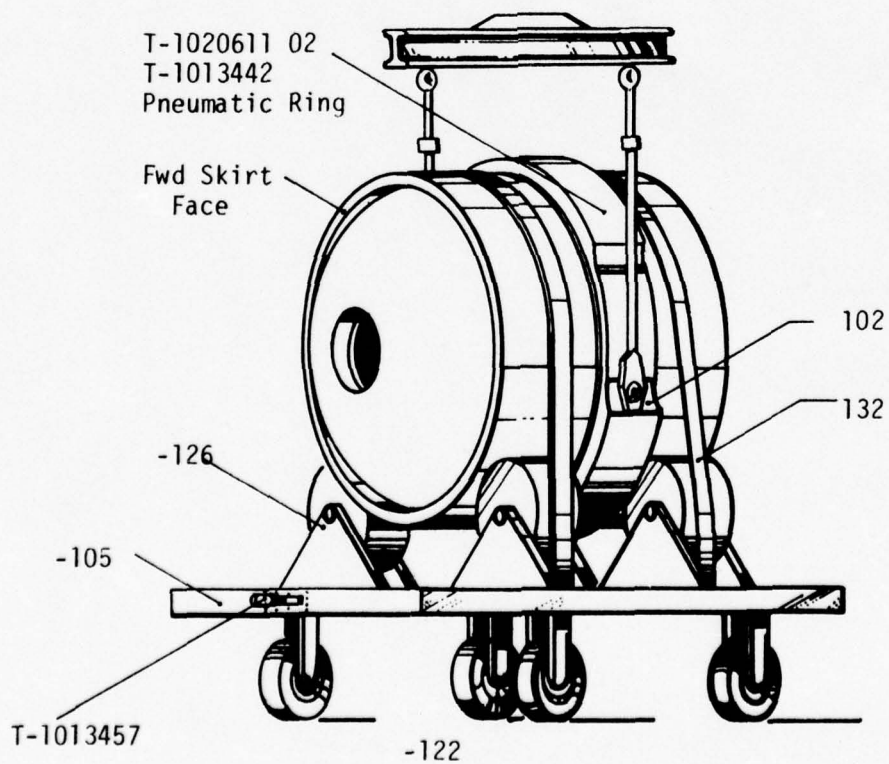


FIGURE J-1. LIFTING OPERATIONS

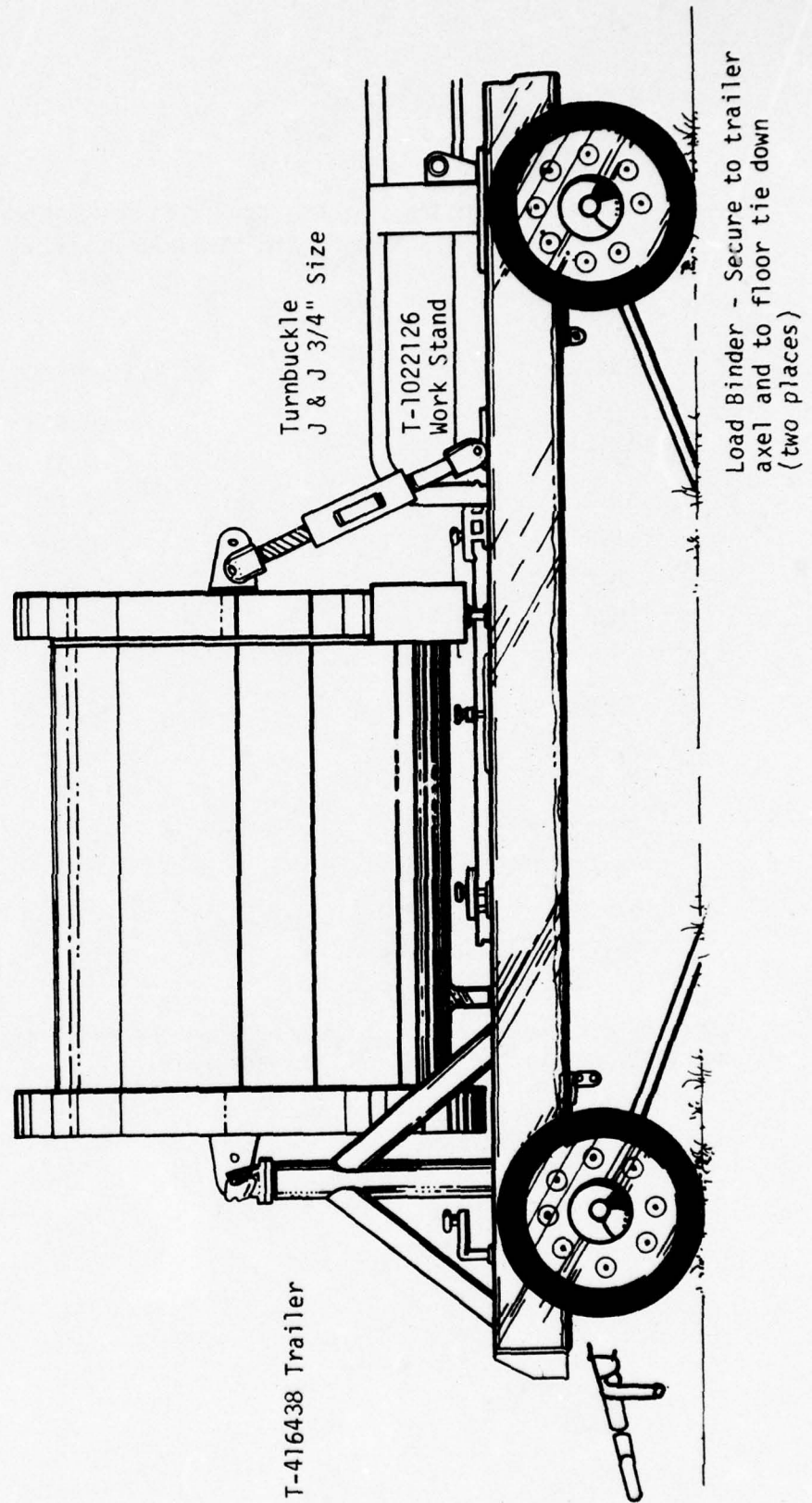


FIGURE J-2. T-416438 TRANSPORT TRAILER FOR HANDLING AND TRANSPORTATION TESTS



TABLE J-1

CHANGE IN GAGE OUTPUTS IN PSI FOR TILTING MOTOR AXIS  
FROM THE HORIZONTAL TO THE VERTICAL POSITION

<u>Gage No.</u>	<u>Stress Change, psi</u>
SH-4 }	+0.67
SH-8 }	+0.31
SH-3 }	+0.70
SH-7 }	+0.38
SH-2 }	Failed
SH-6 }	+0.18
SH-1 }	-0.35
SH-5 }	+0.56
SH-9 }	-0.69
SH-11 }	+0.88
SH-12 (90°)	+0.32
SH-14 (90°)	-0.60
SH-10	+0.75
N11-1	+1.67

Note the different stress changes between the pairs of shear gages nominally at similar locations in the grain.

TABLE J-2  
SUMMARY OF RESULTS FROM TRANSPORTATION TESTS

<u>Normal Gages</u>	<u>Sensitivity Factors Used on Oscillographs (Inches/mv except where noted)</u>	<u>Max. Deflections From Start of Test (Inches)</u>	<u>Voltage Changes (Strain Changes Noted) (mv)</u>
N1-2	.73	1.5	2.05
N4-2	.65	2.4	3.7
N5-2	.70	.2	.28
N6-1	.60	.1	.17
N7-1	.34	.1	.30
N8-1	.60	.8	1.33
N9-1	.57	.28	.50
N10-1	.34	.30	.86
N11-1	.45	.20	.44
N15-2	-	-	-
<u>Shear Gages</u>			
S-1	.25	.025	.10
S-3	.45	.075	.12
S-4	.25	.025	.10
S-5	.75	.050	.065
S-6	.55	.050	.09
S-7	.55	.050	.09
S-8	.67	.050	.08
S-9	.55	.300	.54
S-10	.35	.100	.29
S-11	.45	-	-
S-12	.76	-	.13
S-14	.51	0	0
<u>Strain Gages</u>			
SCAT-1	.48 in/375 $\mu$ "/"	.025	15 micro inch/inch
SCAT-2	.45 in/375 $\mu$ "/"	.050	41 micro inch/inch

APPENDIX K

SUMMARY OF TEST MEASUREMENTS  
FOR MOTOR NO. 2



SUMMARY LOG OF TEST MEASUREMENTS  
FOR MOTOR NO. 2

## A. INSTRUMENTATION

The instrumentation of Motor No. 2 was similar to that of Motor No. 1 except that additional stress gages were installed in the aft dome to measure stresses at locations considered to be critical in motor firings.

## B. TEST SCHEDULE

The calibration and test schedule for Motor No. 2 is shown in Figure K-1.

## C. TESTS AND CALIBRATIONS

The stress gages installed in the motor were calibrated before and after the propellant grain was cast. These calibrations consisted of pressurizing the chamber (precast) or motor (cast) with gaseous nitrogen to 50 psig in incremental steps of 10 psig while recording the stress gage outputs. The precast pressure calibration curves for the normal gages at ambient temperature are shown in Figure K-2. The post-cast pressure calibration curves for the 450 psig and 150 psig normal stress gages and shear gages are shown in Figures K-3, K-4 and K-5, respectively. The measured pressure (stresses) as a function of applied pressure for some of the normal gages are shown in Figures K-6 and K-7. Additionally, a pressure and thermal calibration for the installed gages were performed by pressurizing the chamber (precast) to 15 psig in steps of 5 psig at 30°F, 77°F and 110°F. This pressure exercise was conducted when the gages were initially installed to verify proper polarity, proper hookup and to provide a baseline calibration for future data analyses. The results of all the above tests were indicated in Table K-1, which shows the test data presented in terms of zero stress conditions and gage sensitivity (mv/psi) at temperatures of 30°F, 77°F and 110°F. The checkout data (zero stress-precast) for Motor No. 2 at 77°F is shown in Table K-2.

After gage installation, checkout and calibration, the chamber was processed through the Manufacturing Operations. Test measurements were made during the cast, cure and cooldown operations. The cure and cooldown stresses are shown in Figures K-8 through K-11. Stresses produced during core removal were monitored. Approximately 1500 lbs of load was required to remove the core. Upon completion of the casting and trimming operations, and after achieving temperature equilibrium, the gages were then pressure calibrated at temperatures of 33, 77, and 130°F. These calibration results were also included in Table K-1.

With the emphasis placed on Motor No. 1, only a few data were reduced from the measurements made on Motor No. 2. The data in Table K-3 give a comparison of the thermal stresses at the end of cure and after cooldown between the two motors. Some differences are noted; particularly in the shear stress measurements.

The motor was subsequently X-rayed on 17 July 1973 in conformance with AGC-32188. The results of the radiographic inspection report were similar to most third stage Minuteman motors cast and are shown in Table K-4. Motor No. 2 is presently being stored and aged in Building No. 131 in the Manufacturing Area at  $80 \pm 20^{\circ}\text{F}$ . The purpose of this storage was to simulate field environments to which the third stage Minuteman motors are subjected. Final disposition of Motor No. 2 will be determined by AFRPL and ASPC.

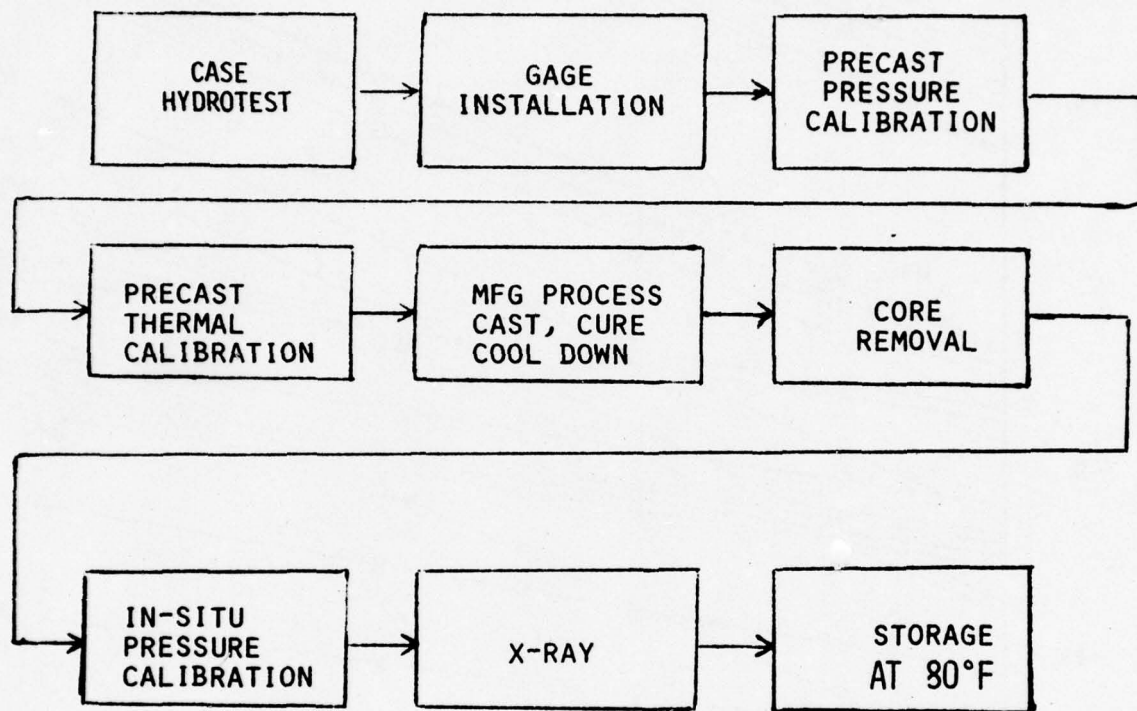


FIGURE K-1. CALIBRATE & TEST FLOW DIAGRAM FOR MOTOR NO. 2



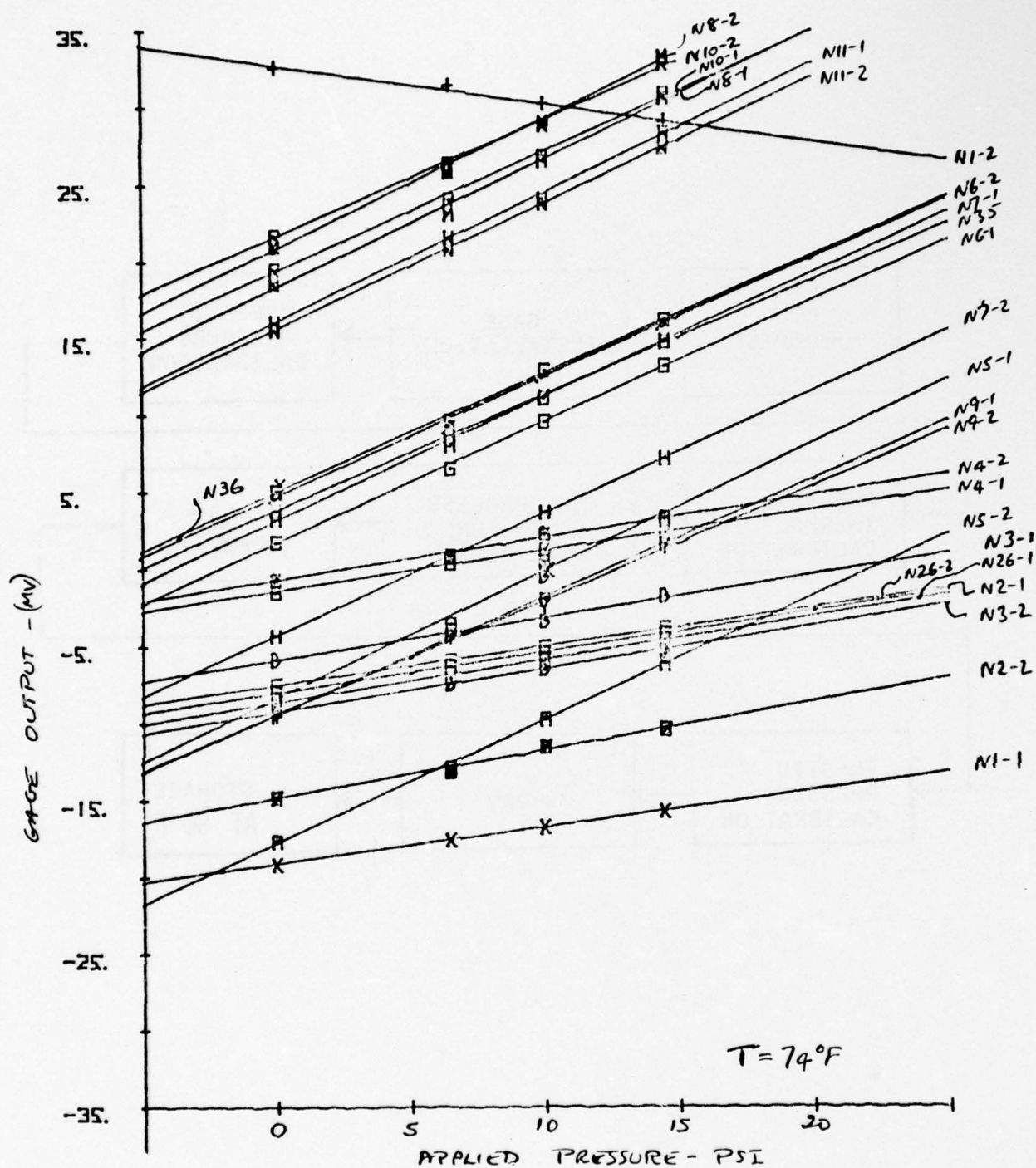


Figure K-2. Pre-Casting Pressure Calibrations ASPC Motor# 2

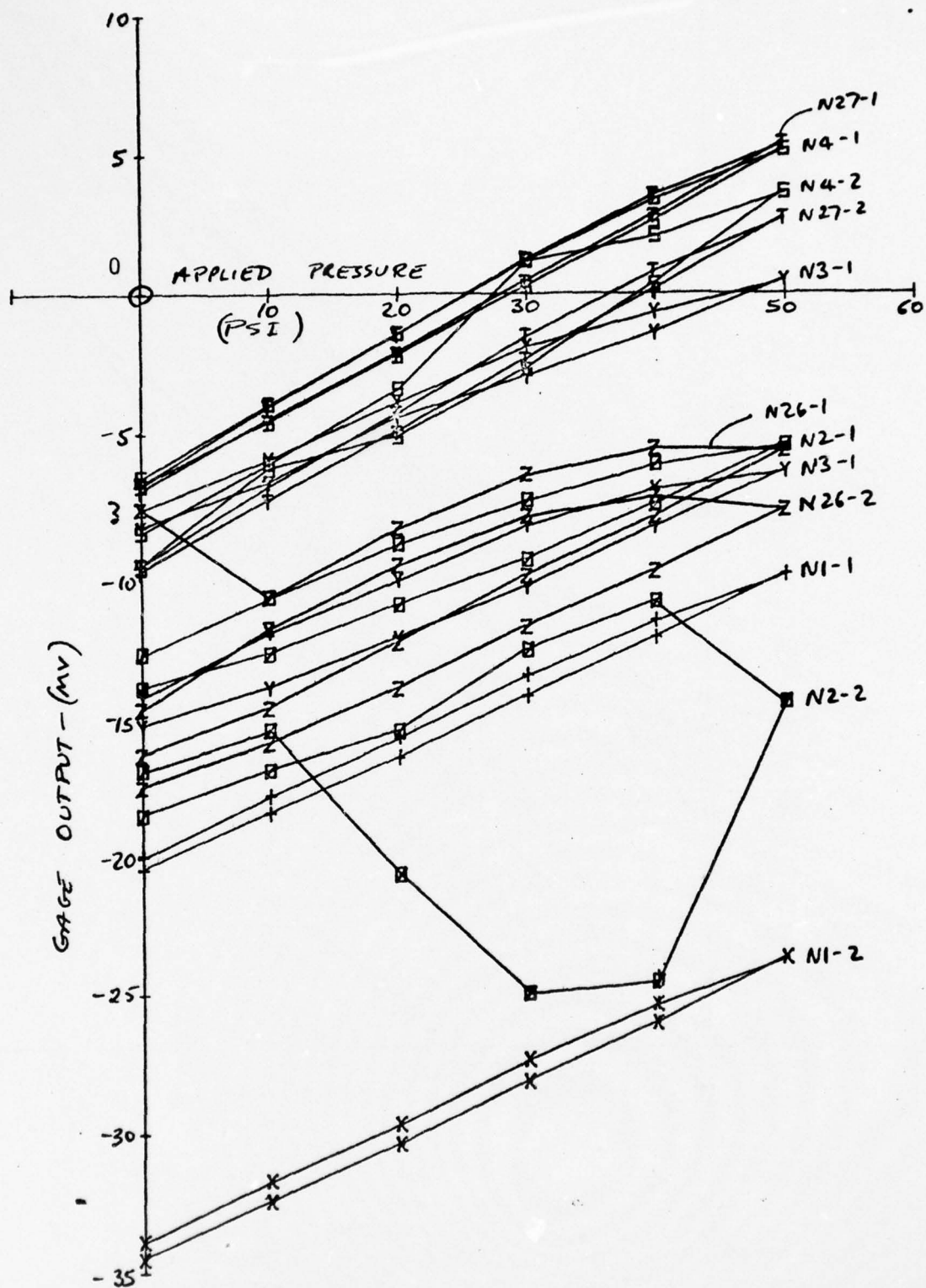


Figure K-3. Post Casting Pressure Test Data ASPC Motor # 2  
450 psi Normal Stress Gages

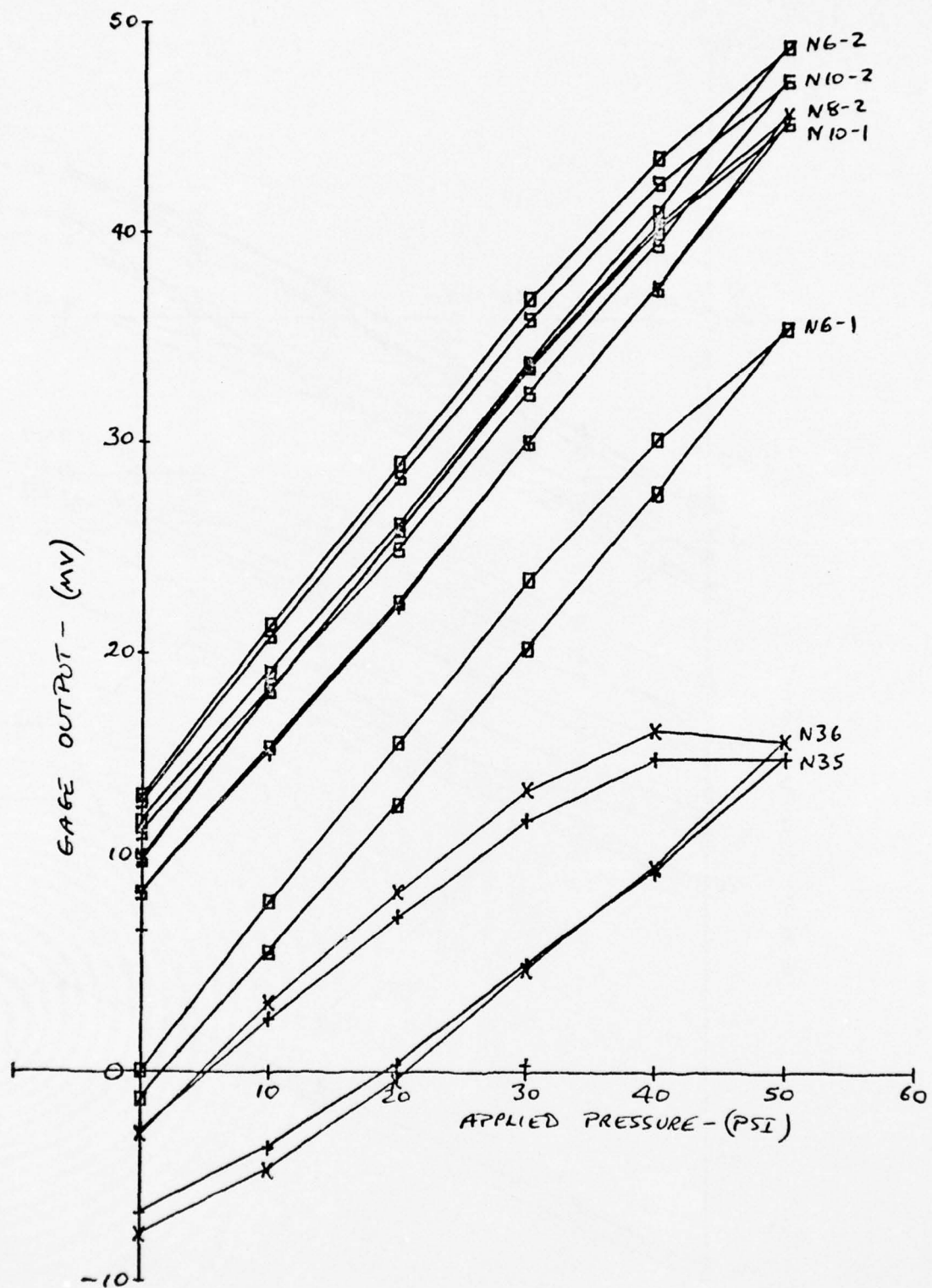


Figure K-4. Post Casting Pressure Test Data 150 psi Gages in ASPC Motor #2



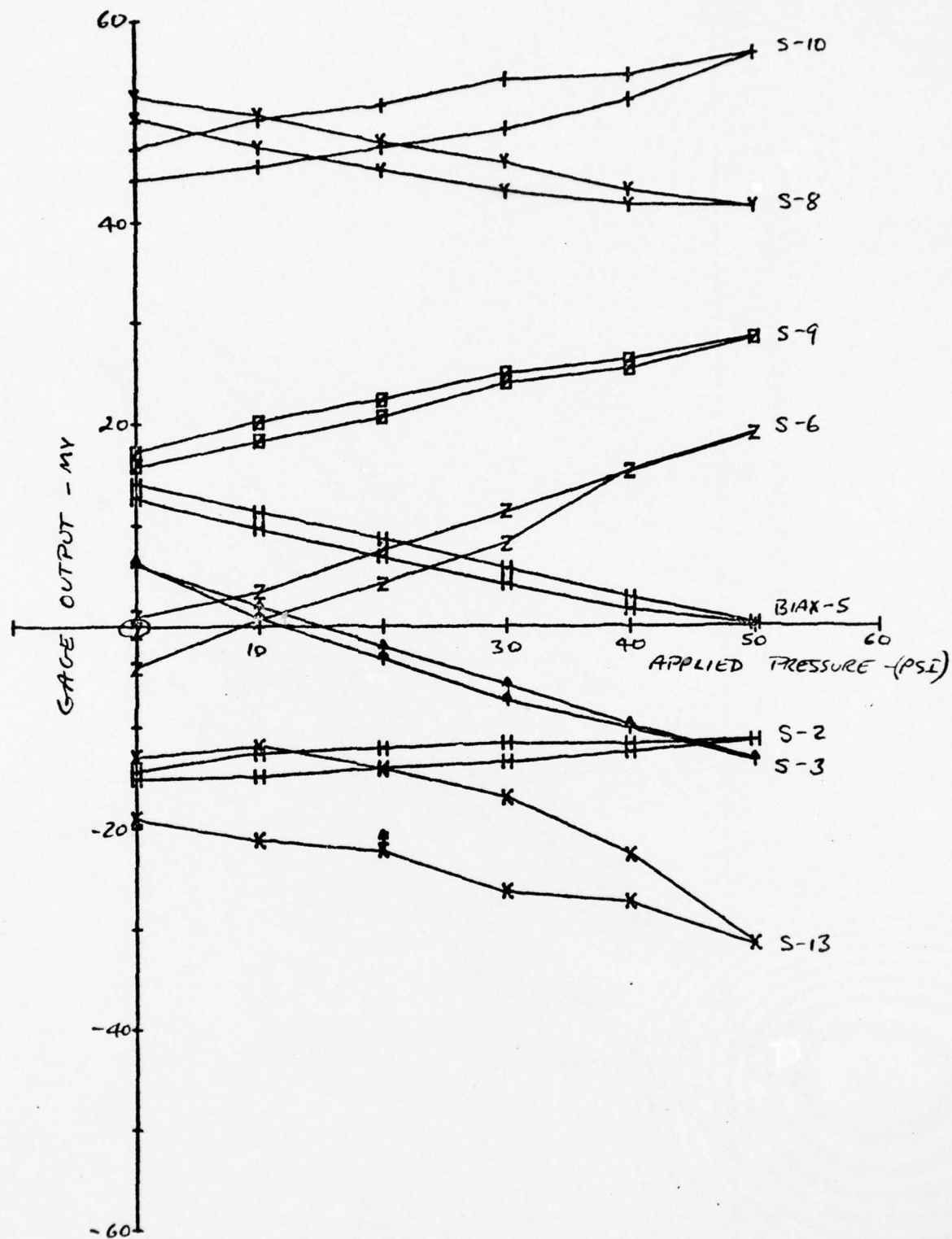


Figure K-5. Post Casting Pressure Test Data from Shear Gages in ASPC Motor # 2

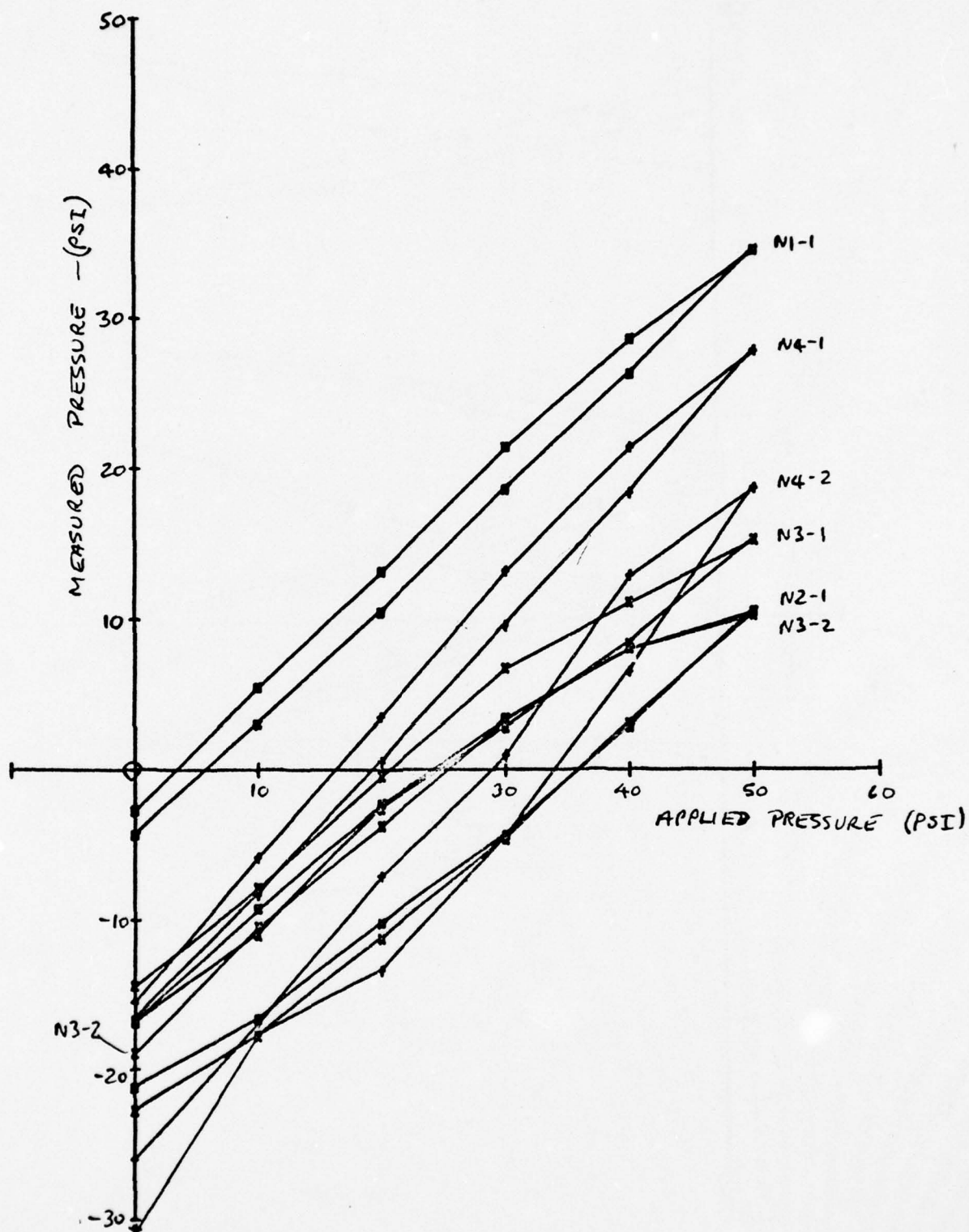


Figure K-6. Post Casting Pressure Stresses from 450 psi Gages in ASPC Motor # 2

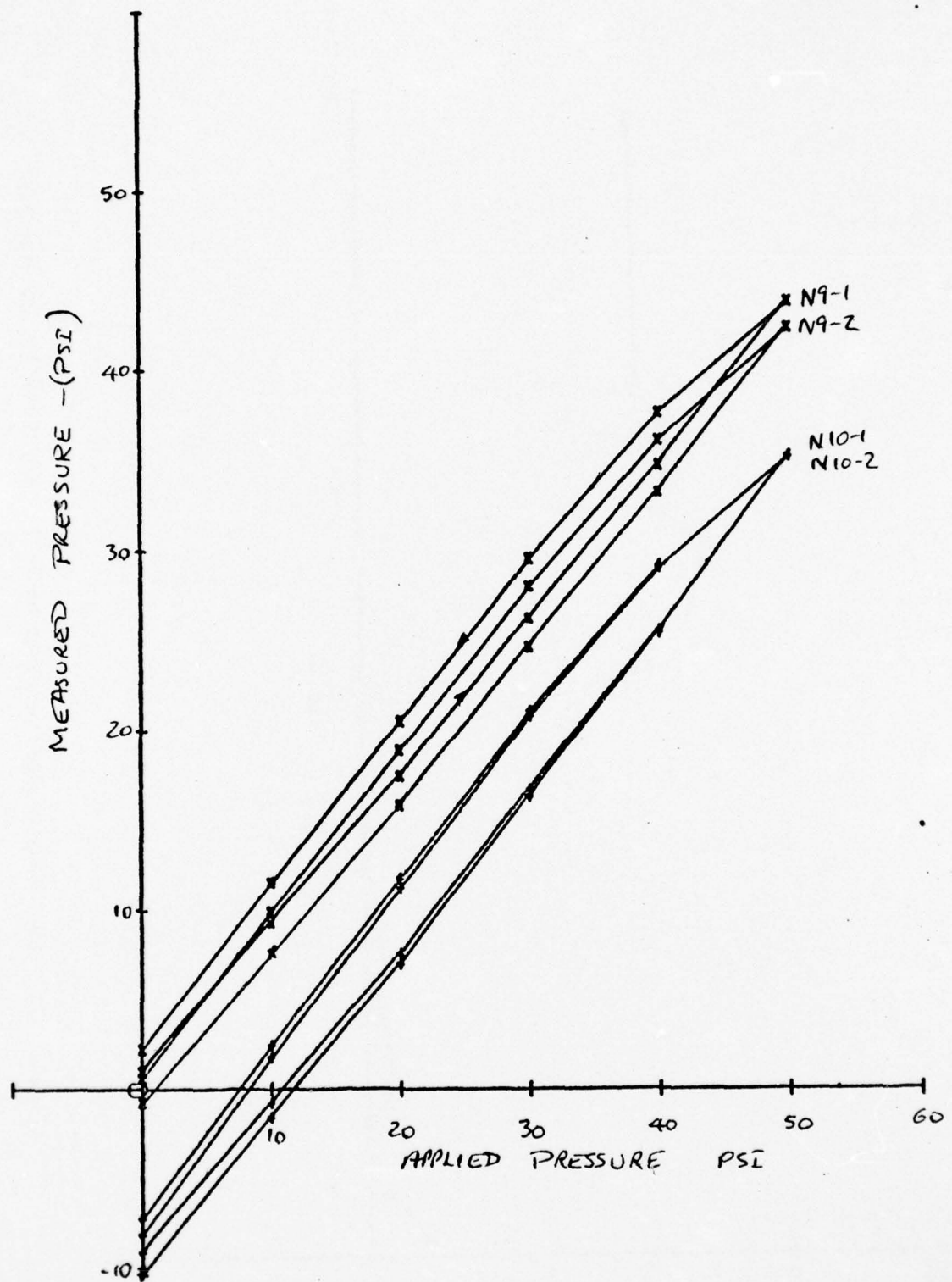


Figure K-7. Post Casting Pressure Stresses from 150 psi Gages in ASPC Motor #2



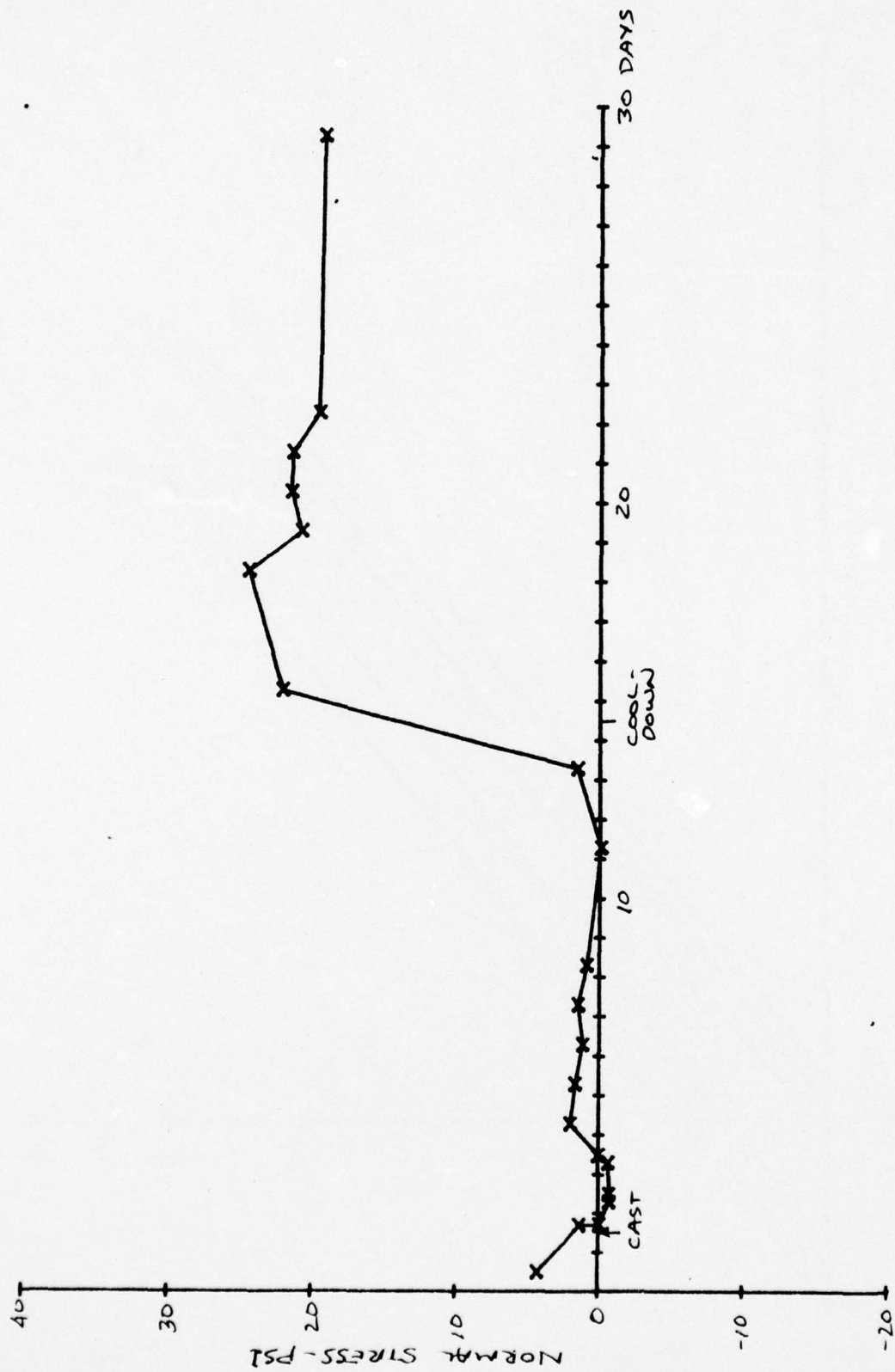


Figure K-8. Cure and Cooldown Stresses on Full Scale Motor No. 2; Gage N2-1

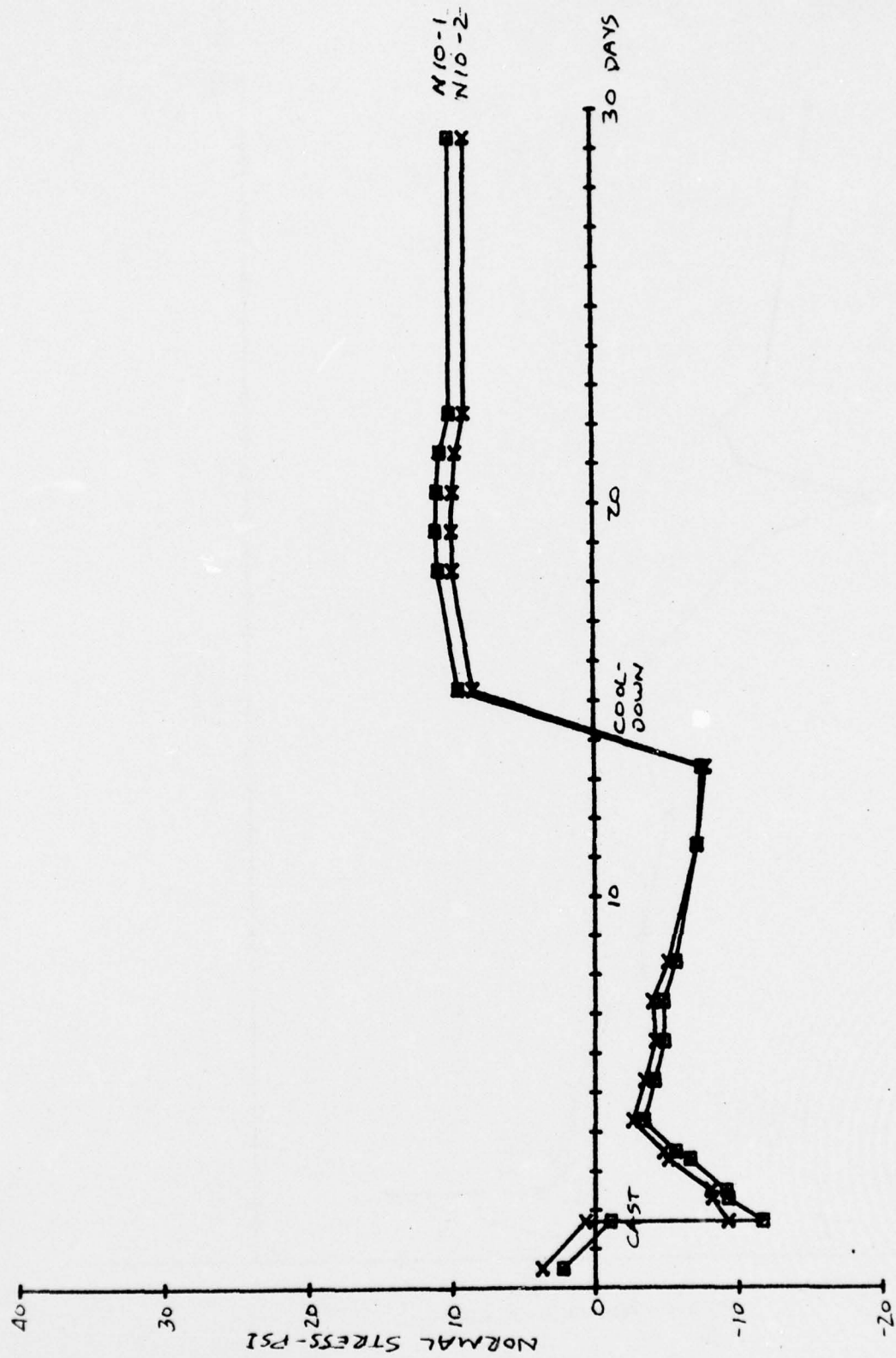
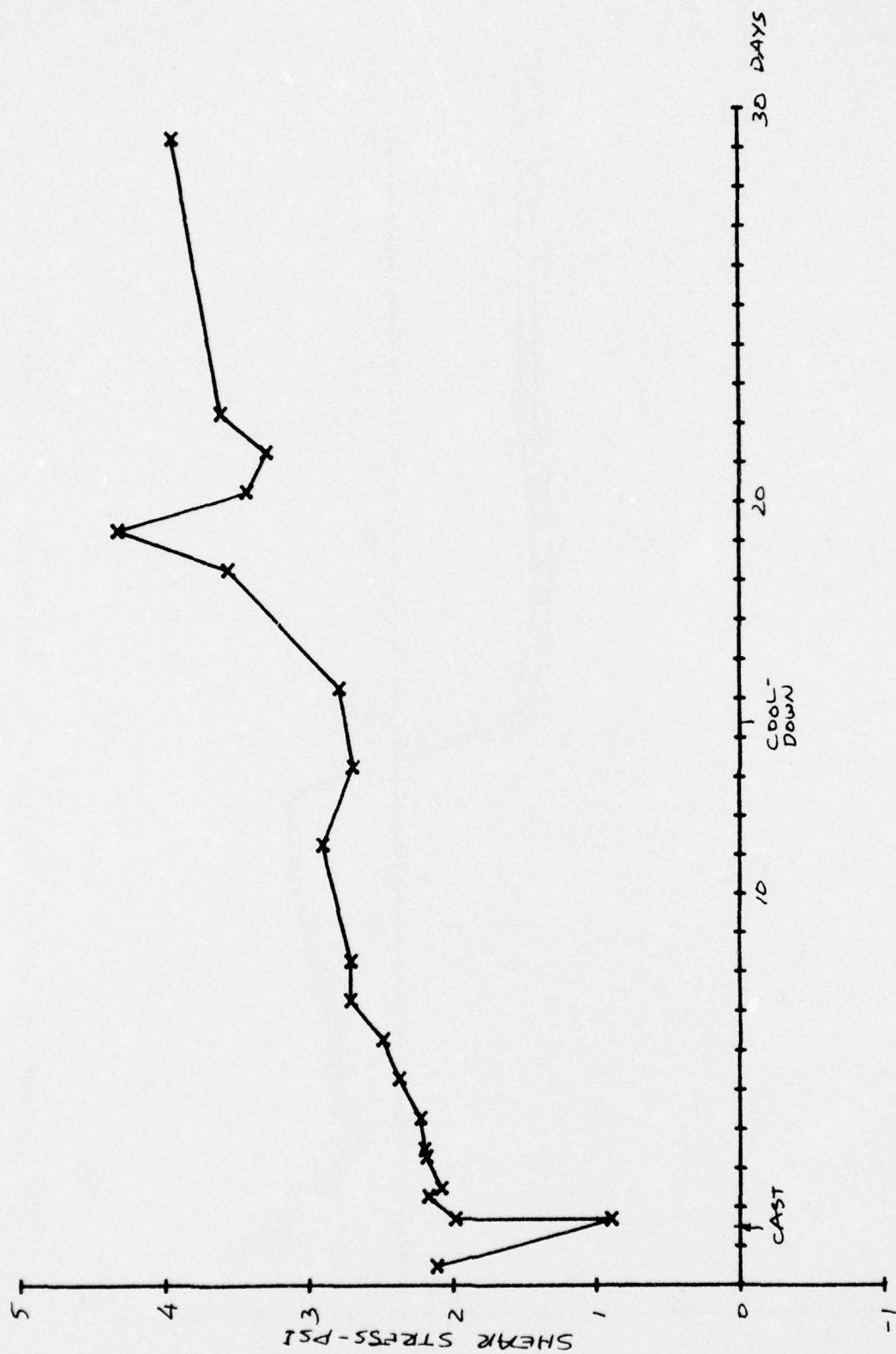


Figure K-9. Cure and Cooldown Stresses on Full Scale Motor No. 2; Gages N10-1 and N10-2



K-14

Figure K-10. Cure and Cooldown Stresses on Full Scale Motor No. 2; Gage S-10



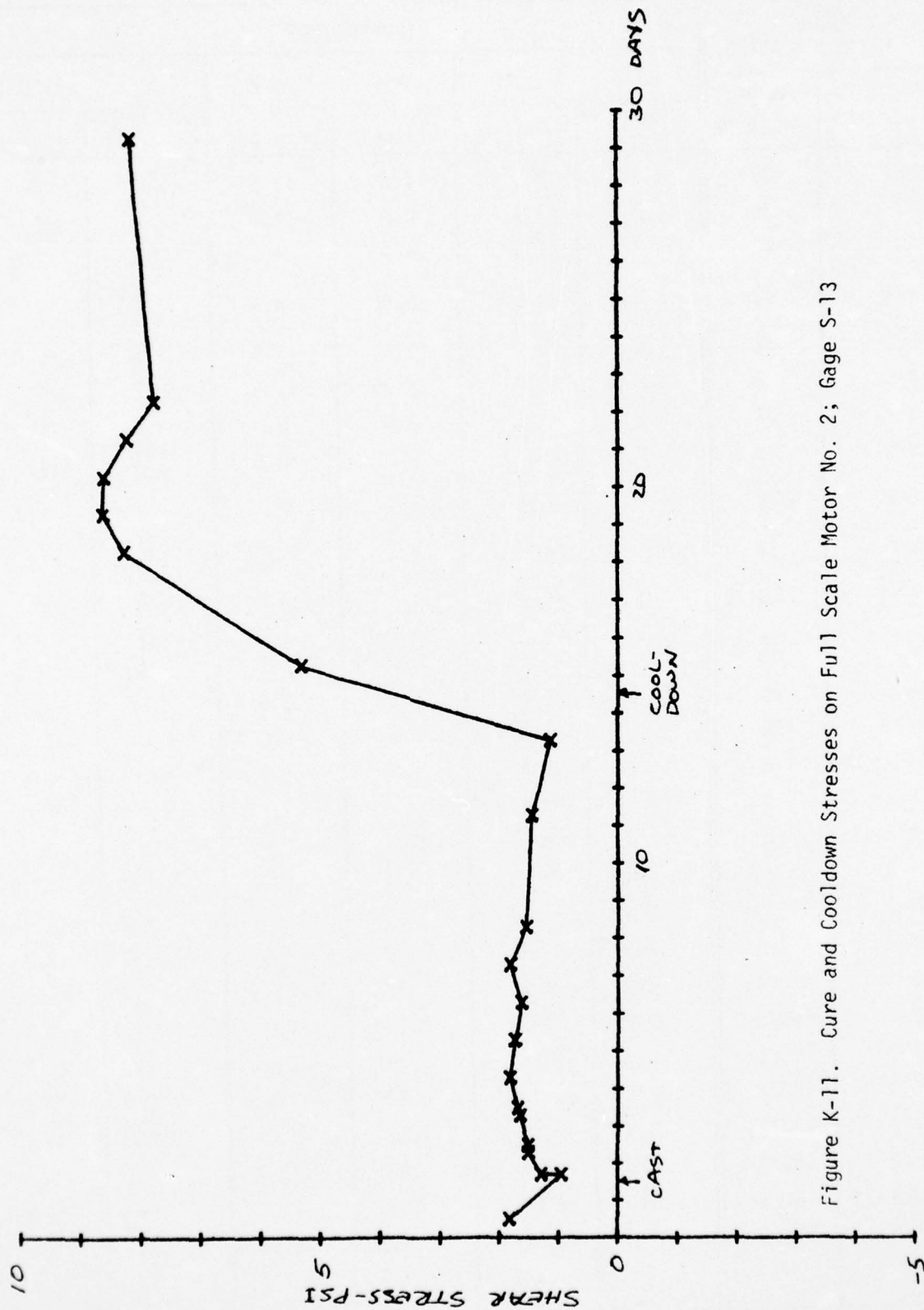


Figure K-11. Cure and Cooldown Stresses on Full Scale Motor No. 2; Gage S-13

TABLE K-1  
FULL SCALE MOTOR NO. 2 GAGE DATA

GAGE IDENTIFICATION		TEMPERATURE					
Motor No.	KI or HL&A No	33°F	30°F	77°F	74°F	130°F	110°F
		SENSITIVITY MV/PSI	ZERO RDG.	SENSITIVITY MV/PSI	ZERO RDG.	SENSITIVITY MV/PSI	ZERO RDG.
S-1	30	4.35	+2.1 mv	4.6	+1.8	4.8	+2.7
S2	26	4.0	-7.3	4.8	+1.8	4.6	+6.7
S3	27	3.8	-1.4	4.1	-1.8	3.8	-2.0
S-4	42	3.05	-3.2	3.65	+4.5	(+144) 3.7	+11.1
S-5	32	3.6	+0.2	3.6	+3.5	3.5	+5.2
S-6	29	3.6	+8.3	3.85	+1.3	3.75	-4.3
S-7	31	4.0	-6.5	3.9	0.0	3.76	-1.2
S-8	33	2.0	+10.0	2.0	+6.5	(144°F) 2.0	+4.5
S-9	S-B	*	-0.5	*	0.0	*	-1.8
S-10	34	2.6	+42.0	2.4	+37.5	(144°F) 2.2	+35.0
S-11	13	3.2	+5.8	3.0	-0.5	3.5	-3.5
S-12	35	2.6	+4.0	2.3	+1.2	(144°F) 2.45	+1.4
S-13	25	2.8	-2.4	3.1	-2.6	3.2	-1.7
S-14	36	2.1	-0.7	1.9	-1.0	1.6	-1.8
Shear A	S-A	*	-3.7	*	-1.5	*	-0.4
Shear C	S-C	*	+0.2	*	0.0	*	-1.0
Shear D	S-D	*	+0.1	*	-1.0	*	-0.3
N1-1	450/3-1	.268	-19.2	.269	-19.2	.271	-20.2
N1-2	450/3-2	.271		.272		.272	
N2-1	450/11-1	.269	-12.0	.271	-8.6	.269	-7.2
N2-2	450/11-2	.272	-14.9	.274	-14.8	.272	-15.3
N3-1	450/4-1	.275	-10.2	.274	-4.0	.272	-4.3
N4-1	450/22-1	.270	-3.0	.270	-2.5	.270	-1.3
N4-2	450/22-2	.272	-2.1	.272	-1.6	.274	-0.3
N5-1	150/21-1	.823	-7.5	.821	-8.4	.818	-8.6
N5-2	150/21-2	.820	-18.4	.821	-17.7	.826	-17.8
N6-1	150/6-1	.811	+2.0	.807	-1.7	.807	-0.9
N6-2	150/6-2	.812	+5.9	.811	+5.1	.811	+3.4

\*Viscoelastic Calibration. See Curves

TABLE K-1  
FULL SCALE MOTOR NO. 2 GAGE DATA (Cont)

GAGE IDENTIFICATION		TEMPERATURE					
Motor No.	KI or HL&A No.	33°F	30°F	77°F	74°F	130°F	110°F
		SENSITIVITY MV/PSI	ZERO RDG.	SENSITIVITY MV/PSI	ZERO RDG.	SENSITIVITY MV/PSI	ZERO RDG.
N7-1	150/29-1	.816	+3.7	.814	+3.2	.816	+4.0
N7-2	150/29-2	.820	-5.1	.816	-4.3	.818	-3.8
N8-1	150/2-1	.825	+18.7	.819	+18.5	.813	+17.3
N8-2	150/3-2	.818	+20.3	.818	+21.0	.812	+2.07
N9-1	150/28-1	.811	-8.0	.813	-10.5	.811	-11.5
N9-2	150/23-2	.801	-9.5	.806	-10.7	.807	-11.2
N10-1	450/7-1	.270	+18.0	.271	+16.7	.268	+18.0
N10-2	450/7-2	.272	+20.0	.271	+18.7	.274	+20.5
N11-1	150/4-1	.841	+16.8	.839	+15.2	.841	+14.8
N11-2	150/4-2	.806	+15.3	.806	+15.5	.805	+15.2
N26-1	450/26-1	(-75) .268	-10.9	.269	-8.0	(+180) .270	-6.3
		(-75) .270				(+180) .270	
N26-2	450/26-2	(-75) .270	-10.7	.270	-7.5	(+180) .270	-6.3
N27-1	450/27-1	(-75) .270	-0.7	.266	+2.8	(+180) .267	+4.3
		(-75) .268				(+180) .270	
N27-2	450/27-2	.268	+2.6	.266	-1.7	.270	-1.8
BI-7SH	3D5	*	+13.0		+10.1		+7.6
BI-5SH	3D5	*	+7.0		+20.7		+25.7
6A-D	3D5	*	-12.2		+11.0		+23.0
6B-D	3D5	*	+5.1		+18/7		+25.0
5A-D	3D5	*	+23.8		+28.7		+30.6
7A-D	3D5	*	-3.8		+0.4		+1.6
2+	3D6-4	*	+7.0		-1.8		-11.8
2-	3D6-3	*	-10.3		-20.5		-13.0
3+	3D6-6	*	-9.0		-19.0		-29.0
3-	3D6-5	*	-13.8		-27.6		-37.5
1+	3D6-2	*	-13.5		-18.0		-24.5
1-	3D6-1	*	+16.3		+11.3		+3.0
N35	150/35-1	(0°F) .761		.759		(150°F) .762	
		(0°F) .792				(150°F) .812	
N36	150/36-1	.792		.804		.812	

\*Viscoelastic Calibration. See Curves



TABLE K-2  
MOTOR NO. 2 CHECKOUT DATA (ZERO STRESS - PRECAST)  
(NORMAL STRESS GAGES)

<u>Normal Gages</u>	<u>S/N</u>	<u>Output (MV) Reading at 77°F</u>
N1-1	3	-18.7
N1-2	3	34
N2-1	11	- 9.6
N2-2	11	-14.6
N3-1	4	- 4.7
N3-2	4	- 9.4
N4-1	22	- 2.4
N4-2	22	- 1.4
N5-1	21	- 8.2
N5-2	21	-17.7
N6-1	6	1.8
N6-2	6	5.1
N7-1	29	3.2
N7-2	29	- 4.8
N8-1	3	18.3
N8-2	3	20.7
N9-1	28	-10
N9-2	28	-10.7
N10-1	7	+17
N10-2	7	+19.4
N11-1	4	+16.0
N-11-2	4	+16.0

TABLE K-3

THERMAL STRESS COMPARISON AT END OF CURE  
AND AFTER COOLDOWN: FULL SCALE THIRD STAGE  
MINUTEMAN MOTORS NO. 1 AND 2

GAGE #	STRESSES IN FS # 1			STRESSES IN FS # 2		
	END OF CURE	AT 70°F	STRESS CHANGE	END OF CURE	AT 70°F	STRESS CHANGE
N-10	-8 psi	9 psi	17 psi	-8 psi	10 psi	18 psi
N-2/1	0.0	19	19	1	21	20
SH-10	0.3	-3.3	-3.6	2.7	3.4	0.7
SH-13	0.5	-9.0	-9.5	1.5	8.5	7.0

TABLE K-4

## RADIOGRAPHIC INSPECTION MINUTEMAN MOTOR, STAGE III

Motor No. 002 , TC-30113

Date: July 17, 1973

<u>CASE/INSULATION SEPARATION</u>			
Forward Zone	In <sup>2</sup>	<u>18</u>	
Cylindrical Zone	In <sup>2</sup>	<u>None</u>	
Aft Zone	In <sup>2</sup>	<u>None</u>	
<u>PROPELLANT/INSULATION SEPARATION</u>			
Forward Zone	In <sup>2</sup>	<u>None</u>	
Cylindrical Zone	In <sup>2</sup>	<u>None</u>	
Aft Zone	In <sup>2</sup>	<u>None</u>	
<u>BOOT/INSULATION JOINT</u>			
Forward:	Minimum Bond	Satisfactory	
Aft :	Minimum Bond	N/A	
<u>PROPELLANT DEFECTS</u>			
Cracks	-	None	
Voids	-	None	
Other	-	None	
REMARKS:			
<ol style="list-style-type: none"> <li>1. Motor contains a moderate amount of liner ridging noted in the aft knuckle area.</li> <li>2. Motor contains case/insul. separation adjacent to the igniter boss. This separation ranges from 0.25" to 1.9" long and is noted at six (6) different tangent points. Total of 18 In.<sup>2</sup>.</li> <li>3. Motor contains numerous sensing devices and wires attached to the motor at different places.</li> </ol>			
Motor conforms to AGC 32188F		C.F. Broman Radiographer Department 5254	



AD-A032 637

AEROJET SOLID PROPULSION CO SACRAMENTO CALIF  
FLEXIBLE CASE-GRAIN INTERACTION IN BALLISTIC WEAPON SYSTEMS. VO--ETC(U)  
OCT 76 K W BILLS, S W JANG, H LEEMING  
F04611-72-C-0055

F/6 21/9.2

UNCLASSIFIED

ASPC-1953-81-F-VOL-3

AFRPL-TR-76-57-VOL-3

NL

4 OF 4  
AD  
A032637



APPENDIX L

SOFTWARE DOCUMENTATION

AND USER'S MANUAL

MULTIPLEXER DRIVER PROGRAM

FOR THE VARIAN 620i COMPUTER

SOFTWARE DOCUMENTATION AND USER'S MANUAL  
MULTIPLEXER DRIVER PROGRAM  
FOR THE VARIAN 620i COMPUTER

A. THE OVERALL DATA ACQUISITION SYSTEM

A Varian 620i computer with 4096 X 16 bits of memory and a teletype was interfaced to an A/D converter, then to a differential multiplexer capable of multiplexing 64 input channels. A block diagram of this data acquisition system is given in Figure L-1.

B. ELECTRA PHYSICS MULTIPLEXER AND A/D CONVERTER

The specifications for this equipment are briefly stated below.

1. Interface with Varian 620i computer, 3 commands.

- (1) Transmit MPX address and gain.
- (2) Sense if A/D conversion is complete.
- (3) Read A/D converter.

2. Multiplexer

- (1) 64 differential input channels with  $\pm 15$  V maximum allowed voltage on any input lead.
- (2) Turn on time for the multiplexer switches is 1.5 microseconds.

3. Amplifier

- (1) Thermal drift 0.6 microvolts/degree C.
- (2) Slew rate and time constant adjustable by changes of capacitance, nominal .01 seconds.
- (3) Long term and short term amplifier drift of less than 10 microvolts over a one-year period.

(4) Gain values selected for the  $\mu$ A725 amplifier.

		<u>Max. Signal In</u>	<u>Value of LSB</u>
Highest Gain	24.10	$\pm 409.6$ mv	.05 mv
Intermediate	12.05	$\pm 819.2$ mv	-1 mv
Lowest	1.205	$\pm 8.192$ Volts	1.0 mv
		$\pm 8192$ mv	



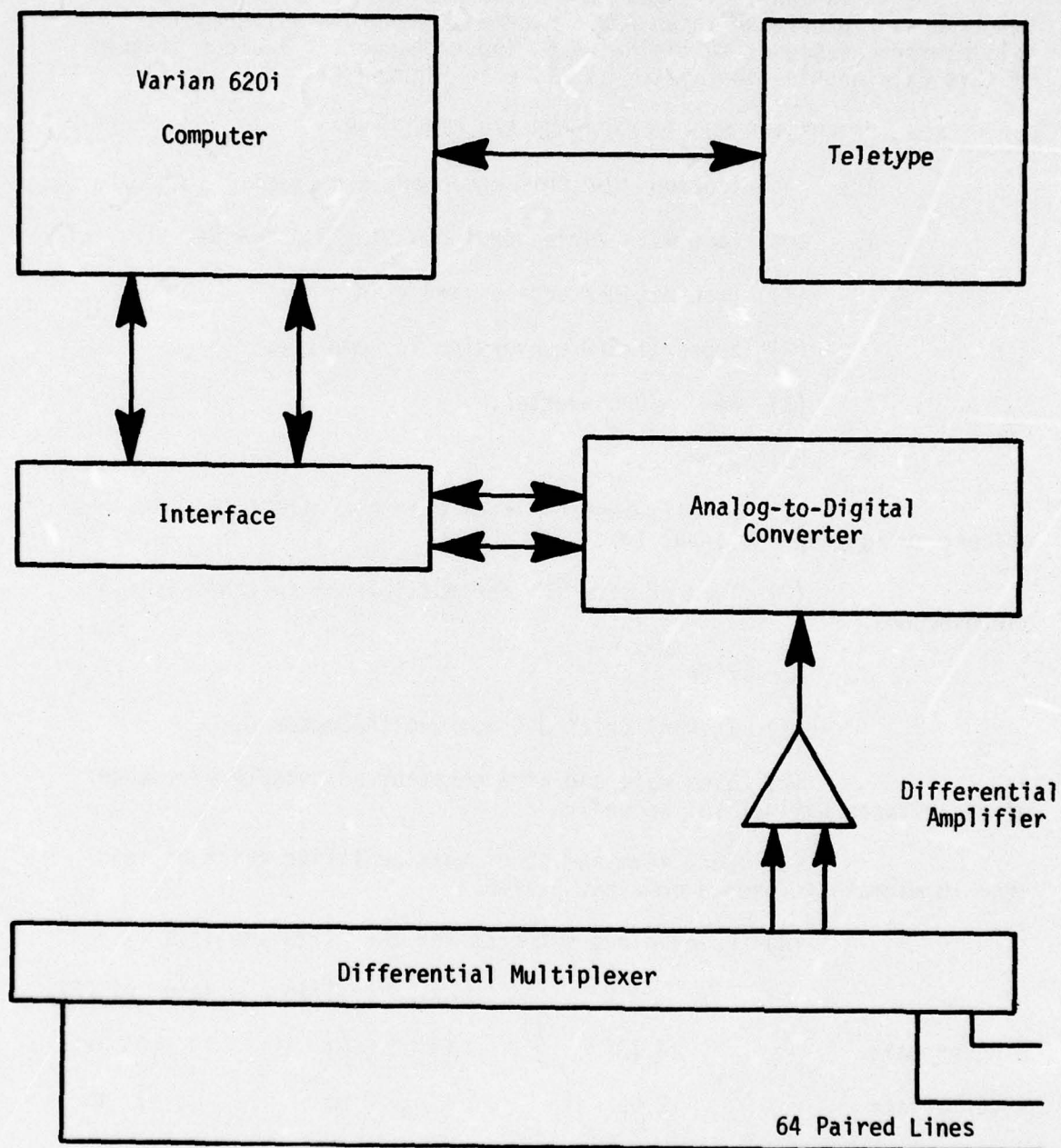


FIGURE L-1. BLOCK DIAGRAM OF DATA ACQUISITION SYSTEM

### C. OPERATING PROCEDURES

The program was designed to scan, in whatever order desired, any selected sequence of the multiplexer channels. It was also designed to read each channel N times and print the average result. N is selectable by the user.

Starting at location 400 (octal)\* a list of 64 multiplexer addresses and gains are placed as data.

- Rules:
- (1) Always start with 177.
  - (2) The next octal digit sets gain  
7 = 24.1, 6 = 12.05 and 5 = 1.205.
  - (3) The last 2 digits designate the channel - 1 in octal.

#### Examples

(Octal)		Decimal	Decimal
177700	=	channel 1 gain	24.10
177600	=	channel 1 gain	12.05
177500	=	channel 1 gain	1.205
177706	=	channel 7 gain	24.10
177722	=	channel 19 gain	24.10
177740	=	channel 32 gain	24.1
177540	=	channel 32 gain	1.205

Space is made available for more than 64 designated channels in case that the operator wishes to repeat selected channels. Any order or arrangement of addresses and gain is feasible in location 400<sub>8</sub> through 747<sub>8</sub>. LIMC, the word in location 773<sub>8</sub>, designates the number of locations to scan. The program will start at 400 and go to 400+ the contents of LIMC.

---

\* Note all numbers in the program and this appendix are in the octal or base 8 designation, unless otherwise designated.

The program starts at location 1000 with a timer which waits for approximately the number of seconds set in T1 (location 761) between each scan of the address list. The program then moves to location 1025 and begins scanning.

Each channel is read for the number of times set in LIM (location 1022) and the resultant number is divided by the contents of LIM and printed.

The output format is in decimal and is 2 spaces, 2 integers which are the channel #, 2 spaces, 2 more integers which are the gain exponent, 2 spaces, + or - sign 3 digits, decimal point and the 2 final digits of the reading.

Example:

05 00 + 323.25

03 01 - 224.55

04 02 - 32.44

Intepretation:

Channel 5 decimal OK +323.25 millivolts

Channel 3 move decimal  
1 right -2245.5 millivolts

Channel 4 move decimal  
point 2  
right -3244 millivolts

For loading program and/or making changes to the parameters refer to the computer instruction manuals by Varian. The program listing is given on pages L-7 through L-23.



## SYMBOLS

0	001763	R	SUBB
0	001762	R	SUBA
0	001752	R	SUBT
0	001740	R	CHEX
0	001737	R	BB
0	001733	R	B
0	001712	R	BIB
0	001677	R	XBTD
1	001672	R	XDSU
0	001641	R	XDAD
0	001616	R	QSIN
0	001615	R	DSIN
0	001614	R	DVDN
0	001613	R	DVSR
0	001612	R	XR
0	001611	R	K14
1	001606	R	XDIV
0	001603	R	XERR
0	001561	R	ADJ
0	001550	R	TEST
0	001546	R	NEGU
0	001521	R	P0SU
0	001502	R	T0P
0	001465	R	XDC0
0	001452	R	XB88
0	001405	R	XB8D
0	001376	R	PL
0	001375	R	MIN
0	001374	R	PD
0	001373	R	SVX
0	001372	R	SP
0	001371	R	TEMC
0	001370	R	TEMB
0	001367	R	TEMA
0	001361	R	SCL5
0	001350	R	SCXT
0	001325	R	SCAL
0	001322	R	REC0
0	001306	R	D0
0	001303	R	D0N
0	001302	R	S2
0	001301	R	S1
1	001270	R	ARIT
0	001263	R	PLUS
0	001245	R	SIGN
0	001177	R	0PTB
0	001157	R	0PTA
0	001156	R	LF
0	001155	R	CR
0	001144	R	CRLF

1	001137	R	ØTP1
0	001135	R	ØUT
0	001101	R	PRIN
0	001040	R	GØRU
0	001026	R	FØRE
0	001025	R	READ
0	001024	R	SUM2
1	001023	R	SUM
1	001022	R	LIM
0	001021	R	BLK
0	001011	R	GØ
0	001003	R	FSTE
0	001002	R	RUNE
0	001001	R	INNE
0	001000	R	STAR
0	000774	R	KØN1
0	000773	R	LIMC
0	000772	R	TEMP
0	000771	R	CØN3
0	000770	R	ØNE
0	000767	R	IMS1
0	000766	R	IMS2
0	000765	R	CØN2
0	000764	R	CØN1
0	000763	R	T3
0	000762	R	T2
0	000761	R	T1
0	000760	R	NCTR
0	000757	R	IMSK
1	000756	R	TENK
0	000755	R	SVEB
0	000754	R	LIT5
0	000753	R	LIT4
0	000752	R	LIT3
0	000751	R	LIT2
0	000750	R	LIT1
0	000400	R	CHGN
0	000001		SPU
0	000001		SLIS
0	000000		SMDV
0	000020		NBIT

000020	NB11	,SET	,16
000000	\$MDV	,SET	,0
000001	\$LIS	,SET	,1
000001	\$PU	,SET	,1

\*

\*

MPX SCAN AND A TO D CVI

\*

000400		,ORG	,0400
000400	177700	CHGN	,DATA ,0177700,0177701,0177702,0177703,017770
000401	177701		
000402	177702		
000403	177703		
000404	177704		
000405	177705		
000406	177706	,DATA	,0177706,0177707,0177710,0177711,017771
000407	177707		
000410	177710		
000411	177711		
000412	177712		
000413	177713		
000414	177714	,DATA	,0177714,0177715,0177716,0177717,017772
000415	177715		
000416	177716		
000417	177717		
000420	177720		
000421	177721		
000422	177722	,DATA	,0177722,0177723,0177724,0177725,017772
000423	177723		
000424	177724		
000425	177725		
000426	177726		
000427	177727		
000430	177730	,DATA	,0177730,0177731,0177732,0177733,017773
000431	177731		
000432	177732		
000433	177733		
000434	177734		
000435	177735		
000436	177736	,DATA	,0177736,0177737,0177740,0177741,017774
000437	177737		
000440	177740		
000441	177741		
000442	177742		
000443	177743		
000444	177744	,DATA	,0177744,0177745,0177746,0177747,017775
000445	177745		
000446	177746		
000447	177747		
000450	177750		
000451	177751		
000452	177752	,DATA	,0177752,0177753,0177754,0177755,017775
000453	177753		
000454	177754		
000455	177755		



```

000456 177756
000457 177757
000460 177760      ,DATA ,0177760,0177761,0177762,0177763,017776
000461 177761
000462 177762
000463 177763
000464 177764
000465 177765
000466 177766      ,DATA ,0177766,0177767,0177770,0177771,017777
000467 177767
000470 177770
000471 177771
000472 177772
000473 177773
000474 177774      ,DATA ,0177774,0177775,0177776,0177777
000475 177775
000476 177776
000477 177777

```

\*

\* TIMER AND START

\*

```

000750      ,ORG ,0750
000750 000200 LIT1 ,DATA ,128
000751 000002 LIT2 ,DATA ,2
000752 000004 LIT3 ,DATA ,4
000753 000060 LIT4 ,DATA ,48
000754 177777 LIT5 ,DATA ,0177777
000755 000000 SVEB ,DATA ,0
000756 023420 TENK ,DATA ,10000
000757 077777 IMSK ,DATA ,077777
000760 000000 NCTR ,DATA ,0
000761 001760 T1 ,DATA ,1000      SECONDS WAIT BETWEEN READS
000762 001744 T2 ,DATA ,996
000763 000135 T3 ,DATA ,93
000764 140000 CON1 ,DATA ,0140000
000765 020000 CON2 ,DATA ,0020000
000766 000300 IMS2 ,DATA ,0300
000767 000077 IMS1 ,DATA ,077
000770 000001 ONE ,DATA ,01
000771 037777 CON3 ,DATA ,037777
000772 000000 TEMP ,DATA ,0
000773 000100 LIMC ,DATA ,0100
000774 000077 KON1 ,DATA ,077
001000      ,ORG ,01000
001000 020761 START ,LDB ,T1      BEGIN TIMER
001001 010762 INNER ,LDA ,T2
001002 030763 RUNER ,LDX ,T3
001003 005000 FSTER ,NOP ,
001004 005344      ,DXR ,
001005 001040      ,JXZ ,G0
001006 001011 R
001007 001000      ,JMP ,FSTER
001010 001003 R

```

\* 10.8 MICROSEC INNER LOOP

```

001011 005311 GØ ,DAR ,
001012 001002 ,JAP ,RUNER
001013 001002 R
001014 005322 ,DBR ,
001015 001020 ,JBZ ,READY
001016 001025 R
001017 001000 ,JMP ,INNER
001020 001001 R
*
* SET FIRST CHANNEL
* THEN READ ALL CHANNELS
*
001021 000000 BLK ,DATA ,0
001022 000100 LIM ,DATA ,64
001023 000000 SUM ,DATA ,0
001024 000000 SUM2 ,DATA ,0
001025 031021 READY ,LDX ,BLK
001026 015400 FØRE ,LDA ,CHGN,1
001027 005211 ,CPA ,
001030 103162 ,ØAR ,062
001031 101062 ,SEN ,062,*
001032 001031 R
001033 005001 ,TZA ,
001034 051023 ,STA ,SUM
001035 051024 ,STA ,SUM2
001036 141022 ,SUB ,LIM
001037 050760 ,STA ,NCTR
*
* MAIN READ LOOP READS LIM TIMES
*
001040 015400 GØRUN ,LDA ,CHGN,1
001041 005211 ,CPA ,
001042 103162 ,ØAR ,062
001043 101062 ,SEN ,062,*
001044 001043 R
001045 005000 ,NOP ,
001046 005000 ,NOP ,
001047 005000 ,NOP ,
001050 005000 ,NOP ,
001051 005000 ,NOP ,
001052 005000 ,NOP ,
001053 005000 ,NOP ,
001054 005000 ,NOP ,
001055 005000 ,NOP ,
001056 005000 ,NOP ,
001057 005000 ,NOP ,
001060 102562 ,CIA ,062
001061 001010 ,JAZ ,GØRUN
001062 001040 R
001063 150771 ,ANA ,CØN3
001064 005012 ,TAB ,
001065 005001 ,TZA ,
001066 002000 ,CALL ,XDAD,SUM
001067 001641 R

```

```

001070 001023 R
001071 051023      ,STA  ,SUM
001072 061024      ,STB  ,SUM2
001073 040760      ,INR  ,NCTR
001074 010760      ,LDA  ,NCTR
001075 001010      ,JAZ  ,PRINT
001076 001101 R
001077 001000      ,JMP  ,G0RUN
001100 001040 R
*      PRINT THE LINE OF DATA
*
001101 002000      PRINT ,CALL ,CRLF
001102 001144 R
001103 015400      ,LDA  ,CHGN,1
001104 150767      ,ANA  ,IMSI
001105 120770      ,ADD  ,ONE
001106 002000      ,CALL ,XBTD
001107 001677 R
001110 002000      ,CALL ,OPTA
001111 001157 R
*      THE CHANNEL NO IS PRINTED
001112 015400      ,LDA  ,CHGN,1
001113 005211      ,CPA  ,
001114 150766      ,ANA  ,IMS2
001115 004306      ,ASRA ,6
001116 002000      ,CALL ,XBTD
001117 001677 R
001120 002000      ,CALL ,OPTA
001121 001157 R
*      THE EXPONENT IS PRINTED
001122 002000      ,CALL ,SCALE
001123 001325 R
001124 002000      ,CALL ,OPTB
001125 001177 R
001126 005144      ,IXR  ,
001127 005041      ,TXA  ,
001130 140773      ,SUB  ,LIMC
001131 001010      ,JAZ  ,START
001132 001000 R
001133 001000      ,JMP  ,FORE
001134 001026 R
*
*      OUTPUT ONE CHAR FROM A TO TTY
*
001135 103101      OUT  ,OAR  ,01
001136 001000      ,JMP  ,*
001137 001136 R
001137      OTPT  ,BES  ,0
001140 101101      ,SEN  ,0101,OUT
001141 001135 R
001142 001000      ,JMP  ,*-2
001143 001140 R
*
*      CARRIAGE RETURN AND LINE FEED

```



```

*
001144 000000      CRLF ,ENTRY,
001145 011155          ,LDA  ,CR
001146 002000          ,CALL ,0TPT
001147 001137 R
001150 011156          ,LDA  ,LF
001151 002000          ,CALL ,0TPT
001152 001137 R
001153 001000          ,JMP* ,CRLF
001154 101144 R
001155 000015      CR   ,DATA ,015
001156 000012      LF   ,DATA ,012
*
*      OUTPUT2SPACES AND NCHARS FROM A N IN B
*
001157 000000      0PTA ,ENTRY,
001160 061367          ,STB  ,TEMA
001161 011372          ,LDA  ,SP
001162 002000          ,CALL ,0TPT
001163 001137 R
001164 011372          ,LDA  ,SP
001165 002000          ,CALL ,0TPT
001166 001137 R
001167 021367          ,LDB  ,TEMA
001170 004450          ,LLRL ,8
001171 061371          ,STB  ,TEMC
001172 020751          ,LDB  ,LIT2
001173 002000          ,CALL ,D0N
001174 001303 R
001175 001000          ,JMP* ,0PTA
001176 101157 R
*
*      OUTPUT THE DATA WORD
*
001177 000000      0PTB ,ENTRY,
001200 051367          ,STA  ,TEMA
001201 061370          ,STB  ,TEMB
001202 002000          ,CALL ,CHEX
001203 001740 R
001204 011372          ,LDA  ,SP
001205 002000          ,CALL ,0TPT
001206 001137 R
001207 011372          ,LDA  ,SP
001210 002000          ,CALL ,0TPT
001211 001137 R
001212 002000          ,CALL ,SIGN
001213 001245 R
001214 021370          ,LDB  ,TEMB
001215 011367          ,LDA  ,TEMA
001216 002000          ,CALL ,XDIV,TENK
001217 001606 R
001220 000756 R
001221 051370          ,STA  ,TEMB
001222 005021          ,TBA  ,

```

```

001223 120753      ,ADD  ,LIT4
001224 002000      ,CALL ,0TPT
001225 001137 R
001226 011370      ,LDA  ,TEMB
001227 002000      ,CALL ,XBID
001230 001677 R
001231 061371      ,STB  ,TEMC
001232 020751      ,LDB  ,LIT2
001233 002000      ,CALL ,D0N
001234 001303 R
001235 011374      ,LDA  ,PD
001236 002000      ,CALL ,0TPT
001237 001137 R
001240 020751      ,LDB  ,LIT2
001241 002000      ,CALL ,D0N
001242 001303 R
001243 001000      ,JMP* ,0PTB
001244 101177 R

```

\*

\* FORM THE SIGN + OR -

\*

```

001245 000000      SIGN ,ENTRY,
001246 011367      ,LDA  ,TEMA
001247 001002      ,JAP  ,PLUS
001250 001263 R
001251 021370      ,LDB  ,TEMB
001252 002000      ,CALL ,XDC0
001253 001465 R
001254 061370      ,STB  ,TEMB
001255 051367      ,STA  ,TEMA
001256 011375      ,LDA  ,MIN
001257 002000      ,CALL ,0TPT
001260 001137 R
001261 001000      ,JMP* ,SIGN
001262 101245 R
001263 011376      PLUS ,LDA  ,PL
001264 002000      ,CALL ,0TPT
001265 001137 R
001266 001000      ,JMP* ,SIGN
001267 101245 R

```

\*

\* MULTIPLY AB BY 5

\*

```

001270 000000      ARITH ,ENTRY,
001271 051301      ,STA  ,S1
001272 061302      ,STB  ,S2
001273 004402      ,LASL ,2
001274 002000      ,CALL ,XDAD,S1
001275 001641 R
001276 001301 R
001277 001000      ,JMP* ,ARITH
001300 101270 R
001301 000000      S1 ,DATA ,0
001302 000000      S2 ,DATA ,0

```

\*  
\* PRINT N CHARS FROM B  
\*

001303	000000	D0N	,ENTRY,	
001304	071373		,STX	,SVX
001305	005024		,TBX	,
001306	005001	D0	,TZA	,
001307	021371		,LDB	,TEMC
001310	004444		,LLRL	,4
001311	061371		,STB	,TEMC
001312	120753		,ADD	,LIT4
001313	002000		,CALL	,0TPT
001314	001137	R		
001315	005344		,DXR	,
001316	001040		,JXZ	,REC0
001317	001322	R		
001320	001000		,JMP	,D0
001321	001306	R		
001322	031373	REC0	,LDX	,SVX
001323	001000		,JMP*	,D0N
001324	101303	R		

\*  
\* GET PROPER SCALE  
\*

001325	000000	SCALE	,ENTRY,	
001326	011023		,LDA	,SUM
001327	021024		,LDB	,SUM2
001330	002000		,CALL	,XDIV,LIM
001331	001606	R		
001332	001022	R		
001333	060772		,STB	,TEMP
001334	005001		,TZA	,
001335	051023		,STA	,SUM
001336	010765		,LDA	,C0N2
001337	140772		,SUB	,TEMP
001340	051024		,STA	,SUM2
001341	001002		,JAP	,SCXT
001342	001350	R		
001343	010754		,LDA	,LIT5
001344	051023		,STA	,SUM
001345	011024		,LDA	,SUM2
001346	150757		,ANA	,IMSK
001347	051024		,STA	,SUM2
001350	015400	SCXT	,LDA	,CHGN,1
001351	005211		,CPA	,
001352	150766		,ANA	,IMS2
001353	001010		,JAZ	,SCL5
001354	001361	R		
001355	021024		,LDB	,SUM2
001356	011023		,LDA	,SUM
001357	001000		,JMP*	,SCALE
001360	101325	R		
001361	011023	SCL5	,LDA	,SUM
001362	021024		,LDB	,SUM2



```

001363 002000          ,CALL ,ARITH
001364 001270 R
001365 001000          ,JMP* ,SCALE
001366 101325 R
001367 000000 TEMA ,DATA ,0
001370 000000 TEMB ,DATA ,0
001371 000000 TEMC ,DATA ,0
001372 000040 SP ,DATA ,32
001373 000000 SVX ,DATA ,0
001374 000256 PD ,DATA ,'. '
001375 000255 MIN ,DATA ,'- '
001376 000253 PL ,DATA ,'+ '
*
*   FIXED POINT DP INTEGER CONVERSION
*
001377 011453          ,LDA ,XB88+1 BIN 10 DEC CONVERSION
001400 002000          ,CALL ,XBTD
001401 001677 R
001402 005041          ,TXA ,
001403 031454          ,LDX ,XB88+2
001404 001000          ,PZE ,001000
001405 000000 XB8D ,ENIR ,
001406 002004          ,JANM ,XDC0
001407 001465 R
001410 051452          ,STA ,XB88
001411 061453          ,STB ,XB88+1
001412 071454          ,STX ,XB88+2
001413 006030          ,LDXI ,077774
001414 077774
001415 071445          ,STX ,XB88-5
001416 006030          ,LDXI ,XB88+1
001417 001453 R
001420 071434          ,STX ,**12
001421 007400          ,R0F ,
001422 005004          ,TZX ,
001423 005041          ,TXA ,
001424 004244          ,LRLA ,4
001425 005014          ,TAX ,
001426 011452          ,LDA ,XB88
001427 021453          ,LDB ,XB88+1
001430 041434          ,INR ,**4
001431 041434          ,INR ,**3
001432 002000          ,CALL ,XDSU,*
001433 001672 R
001434 001434 R
001435 001004          ,JAN ,**7
001436 001444 R
001437 005144          ,IXR ,
001440 051452          ,STA ,XB88
001441 061453          ,STB ,XB88+1
001442 001000          ,JMP ,*-8
001443 001432 R
001444 006040          ,INRi ,
001445 000000

```

```

001446 001001      ,J0F      ,XB8D-6
001447 001377 R
001450 001000      ,JMP      ,*-21
001451 001423 R
001452 000000      XB88 ,DATA ,0,0,0,0461,013200,036,041100
001453 000000
001454 000000
001455 000461
001456 013200
001457 000036
001460 041100
001461 000003      ,DATA ,3,03240,0,023420
001462 003240
001463 000000
001464 023420

```

\* XDC0 FIXED POINT DOUBLE PRECISION COMPLEMENT

```

*
001465 000000      XDC0 ,ENTR ,
001466 005211      ,CPA ,
001467 001020      ,JBZ ,**8
001470 001477 R
001471 005222      ,CPB ,
001472 005122      ,IBR ,
001473 004041      ,LRLB ,1
001474 004141      ,LSRB ,1
001475 001000      ,JMP* ,XDC0
001476 101465 R
001477 005111      ,IAR ,
001500 001000      ,JMP* ,XDC0
001501 101465 R

```

\* XDIV SOFTWARE DIVIDE

```

*
* A,B/MB [QUOTIENT] < A [REMINDER]
* A REG MEMORY X IS UNCHANGED
* QUOTIENT IS ALWAYS TRUE
* REMAINDER IS SIGN OF DIVIDEND [UNLESS R>0]
*

```

```

001502 071612      TOP ,STX ,XR      SAVE XR
001503 005304      ,DECR ,4      SET SIGN INDICATOR
001504 001002      ,JAP ,POSU      SET DIVIDEND POS
001505 001521 R
001506 005244      ,CPX , SET DSIN>NEG
001507 005222      ,CPB , LO ORDER TWO,S COMPL
001510 005122      ,IBR ,
001511 004041      ,LRLB ,1      SIGN>0
001512 004141      ,LSRB ,1
001513 005211      ,CPA , HI ORDER TWO,S COMPL
001514 001020      ,JBZ ,**4
001515 001520 R
001516 001000      ,JMP ,**3
001517 001521 R
001520 005111      ,IAR ,

```

001521	071615	P0SU	,STX	,DSIN	SAVE DIVDN SIGN
001522	051614		,STA	,DVDN	SAVE DIVDN
001523	031606		,LDX	,XDIV	GET ADDR OF CALL SEQ
001524	035000		,LDX	,0,1	GET ADDR OF PARAM
001525	015000		,LDA	,0,1	GET DIVISOR
001526	031615		,LDX	,DSIN	GET DIVDN SIGN
001527	001002		,JAP	,**5	SET DIVISOR P0S
001530	001534	R			
001531	005244		,CPX	, SET QUOTIENT SIGN	
001532	005211		,CPA	, TWO,S COMPL	
001533	005111		,IAR	,	
001534	051613		,STA	,DVSR	SAVE DIVISOR
001535	011614		,LDA	,DVDN	GET DIVDN
001536	071616		,STX	,QSIN	SAVE QUOT SIGN
001537	031611		,LDX	,K14	SET CYCLE COUNT
001540	004041		,LRLB	,1	ADJUST L0 ORDER (DELETE
001541	141613		,SUB	,DVSR	SUB DIVSOR
001542	007401		,SOF	,	
001543	001002		,JAP	,XERR	JMP IF OVERFLOW ERROR
001544	001603	R			
001545	007400		,R0F	,	
001546	004441	NEGU	,LLRL	,1	DEVELOP 14 QUOTIENT BITS
001547	121613		,ADD	,DVSR	(NON RESTORING ALGORITHM
001550	001040	TEST	,JXZ	,ADJ	JMP IF COMPLETE
001551	001561	R			
001552	005344		,DXR	, COUNT BITS	
001553	001004		,JAN	,NEGU	JUMP IF NEG REMAINDER
001554	001546	R			
001555	004441		,LLRL	,1	SHIFT QUOTOREM
001556	141613		,SUB	,DVSR	SUBTRACT DIVSR
001557	001000		,JMP	,TEST	G0 TEST
001560	001550	R			
001561	004041	ADJ	,LRLB	,1	GET LAST QUOTIENT BIT
001562	001002		,JAP	,**4	JMP IF 0R
001563	001566	R			
001564	005122		,IBR	, ELSE SET L0B	
001565	121613		,ADD	,DVSR	RESTORE REMAINDER
001566	031616		,LDX	,QSIN	GET TRUE QUOTIENT
001567	001040		,JXZ	,**4	JMP IF NEGATIVE QUOT
001570	001573	R			
001571	005222		,CPB	, ELSE SET POSITIVE	
001572	005322		,DBR	,	
001573	005122		,IBR	,	
001574	031615		,LDX	,DSIN	GET TRUE REMAINDER
001575	001040		,JXZ	,**4	JMP IF REMAINDER NEG
001576	001601	R			
001577	001000		,JMP	,**4	ELSE LEAVE P0S
001600	001603	R			
001601	005211		,CPA	,	
001602	005111		,IAR	,	
001603	041606	XERR	,INR	,XDIV	SET RETURN
001604	031612		,LDX	,XK	
001605	001000		,JMP*	,XDIV	A,BOM* B>QUOT A>REM
001606	101606	R			



001606		XDIV	,BES	,0	ENTRY
001607	001000		,JMP	,10P	
001610	001502 R				
001611	000016	K14	,DATA	,14	
001612		XR	,BSS	,1	TEMP STORAGE
001613		DVSR	,BSS	,1	
001614		DVDN	,BSS	,1	
001615		DSIN	,BSS	,1	
001616		QSIN	,BSS	,1	
*					
* XDAD		FIXED POINT DOUBLE PRECISION ADD/SUBTRACT			
*					
001617	071644	,STX	,XDAD+3		SAVE XR
001620	007400	,R0F	, RESET 0F		
001621	031641	,LDX	,XDAD		
001622	035000	,LDX	,0,1		XR-ADDR 0F HI B
001623	051645	,STA	,XDAD+4		SAVE HI A
001624	005021	,TBA	, GET L0 A		
001625	125001	,ADD	,1,1		ADD L0 B
001626	006150	,ANAI	,077777		MASK SIGN
001627	077777				
001630	005012	,TAB	, SAVE RESULT		
001631	005001	,TZA	,		
001632	005511	,A0FA	, GET CARRY		
001633	007400	,R0F	, RESET 0F		
001634	121645	,ADD	,XDAD+4		ADD HI A
001635	125000	,ADD	,0,1		ADD HI B
001636	041641	,INR	,XDAD		SET RETURN
001637	031644	,LDX	,XDAD+3		RESTORE XR
001640	001000	,JMP	,0		RETURN
001641	000000				
	001641 R	XDAD	,EQU	,*-1	ENTRY
001642	001000		,JMP	,*-19	
001643	001617 R				
001644	000000	,DATA	,0,0		TEMP STORAGE
001645	000000				
*					
* XDSU		FIXED POINT DOUBLE PRECISION SUBTRACT			
*					
001646	071675	,STX	,XDSU+3		SAVE XR
001647	007400	,R0F	, RESET 0F		
001650	031672	,LDX	,XDSU		
001651	035000	,LDX	,0,1		XR-ADDR 0F HI B
001652	051676	,STA	,XDSU+4		SAVE HI A
001653	005021	,TBA	,		
001654	006110	,0RAI	,0100000		SET SIGN FOR CARRY
001655	100000				
001656	145001	,SUB	,1,1		SUB L0 B
001657	006150	,ANAI	,077777		MASK SIGN
001660	077777				
001661	005012	,TAB	, SAVE RESULT		
001662	005001	,TZA	,		
001663	005711	,S0FA	, GET CARRY		
001664	007400	,R0F	, RESET 0F		

001665	121676		,ADD	,XDSU+4	ADD HI A
001666	145000		,SUB	,0,1	SUB HI B
001667	041672		,INR	,XDSU	SET RETURN
001670	031675		,LDX	,XDSU+3	RESTORE XR
001671	001000		,JMP	,0	RETURN
001672	000000				
001672			,ORG	,*-1	
001672	000000	XDSU	,ENTR	, ENIRY	
001673	001000		,JMP	,*-21	
001674	001646	R			
001675	000000		,DATA	,0,0	TEMP STORAGE
001676	000000				
*					
* XBTD					FIXED POINT INTEGER BIN TO DEC CONVERSION
*					
001677	000000	XBTD	,ENIR	,	
001700	051733		,STA	,B	
001701	071734		,STX	,B+1	
001702	001002		,JAP	,**4	JUMP IF POSITIVE
001703	001706	R			
001704	005211		,CPA	, ELSE COMPLEMENT	
001705	005111		,IAR	, AND ADD ONE	
001706	005012		,TAB	,	
001707	006030		,LDXI	,3	INITIALIZE COUNT
001710	000003				
001711	005001		,TZA	,	
001712	002000	BIB	,CALL	,XDIV,BB	
001713	001606	R			
001714	001737	K			
001715	061735		,STB	,**16	SAVE BIN VALUE
001716	021736		,LDB	,**16	GET PREVIOUS DIGITS
001717	004544		,LLSR	,4	ATTACH DIGIT TO RESULT
001720	001040		,JXZ	,**7	JUMP IF COMPLETE
001721	001727	R			
001722	005344		,DXR	, ELSE COUNT DIGITS	
001723	061736		,STB	,**11	SAVE DIGITS ASSEMBLED
001724	021735		,LDB	,**9	GET BIN VALUE
001725	001000		,JMP	,BIB	
001726	001712	R			
001727	011733		,LDA	,**4	RESTORE AR
001730	031734		,LDX	,**4	RESTORE XR
001731	001000		,JMP*	,XBTD	RETURN
001732	101677	R			
001733	000000	B	,DATA	,0,0,0,0	TEMP STORAGE
001734	000000				
001735	000000				
001736	000000				
001737	000012	BB	,DATA	,10	CONSTANT
001740	000000	CHEX	,ENTRY,		
001741	015400		,LDA	,CHGN,1	IF CHI SAVE
001742	150774		,ANA	,K0N1	
001743	001002		,JAP	,SUBT	
001744	001752	R			
001745	011367		,LDA	,TEMA	

001746	051762		,STA	,SUBA	
001747	061763		,STB	,SUBB	
001750	001000		,JMP*	,CHEX	
001751	101740	R			
001752	011367		SUBT	,LDA	,TEMA NOT CH1 SUBTRACT
001753	002000			,CALL	,XDSU,SUBA
001754	001672	R			
001755	001762	R			
001756	051367		,STA	,TEMA	
001757	061370		,STB	,TEMB	
001760	001000		,JMP*	,CHEX	
001761	101740	R			
001762	000000		SUBA	,DATA	,0
001763	000000		SUBB	,DATA	,0
	000000			,END	,

LITERALS

POINTERS

SYMBOLS

1	001763	R	SUBB
1	001762	R	SUBA
1	001752	R	SUBT
1	001740	R	CHEX
1	001737	R	BB
1	001733	R	B
1	001712	R	B1B
1	001677	R	XBTD
1	001672	R	XDSU
1	001641	R	XDAD
1	001616	R	QSIN
1	001615	R	DSIN
1	001614	R	DVDN
1	001613	R	DVSR
1	001612	R	XR
1	001611	R	K14
1	001606	R	XDIV
1	001603	R	XERR
1	001561	R	ADJ
1	001550	R	TEST
1	001546	R	NEGU
1	001521	R	P0SU
1	001502	R	T0P
1	001465	R	XDC0
1	001452	R	X988
1	001405	R	XB8D
1	001376	R	PL
1	001375	R	MIN
1	001374	R	PD
1	001373	R	SVX
1	001372	R	SP



1	001371	R	TEMC
1	001370	R	TEMB
1	001367	R	TEMA
1	001361	R	SCL5
1	001350	R	SCXT
1	001325	R	SCAL
1	001322	R	KEC0
1	001306	R	D0
1	001303	R	D0N
1	001302	R	S2
1	001301	R	S1
1	001270	R	ARIT
1	001263	R	PLUS
1	001245	R	SIGN
1	001177	R	0PTB
1	001157	R	0PTA
1	001156	R	LF
1	001155	R	CR
1	001144	R	CRLF
1	001137	R	0IPT
1	001135	R	0UT
1	001101	R	PRIN
1	001040	R	G0RU
1	001026	R	F0RE
1	001025	R	READ
1	001024	R	SUM2
1	001023	R	SUM
1	001022	R	LIM
1	001021	R	BLK
1	001011	R	G0
1	001003	R	FSTE
1	001002	R	RUNE
1	001001	R	INNE
1	001000	R	STAR
1	000774	R	K0N1
1	000773	R	LIMC
1	000772	R	TEMP
1	000771	R	C0N3
1	000770	R	0NE
1	000767	R	IMS1
1	000766	R	IMS2
1	000765	R	C0N2
1	000764	R	C0N1
1	000763	R	T3
1	000762	R	T2
1	000761	R	T1
1	000760	R	NCTR
1	000757	R	IMSK
1	000756	R	TENK
1	000755	R	SVEB
1	000754	R	LIT5
1	000753	R	LIT4
1	000752	R	LIT3
1	000751	R	LIT2

PAGE 000017

1	000750	R	LIT1
1	000400	R	CHGN
0	000001		\$PU
0	000001		\$LIS
0	000000		\$MDV
0	000020		NBIT

APPENDIX M

CALIBRATION PROCEDURE FOR DATA  
ACQUISITION SYSTEMS



## CALIBRATION PROCEDURE FOR DATA ACQUISITION SYSTEMS

## A. PURPOSE

The purpose of this calibration procedure is to certify that the data acquisition systems used to record data for the Flexible Case/Grain Interaction Test Program meet the accuracy requirements specified in Paragraph 3.8.1.2 Of the program work statement.\*

## B. DESCRIPTION OF THE CALIBRATION PROCEDURE

This procedure shall be divided into three parts. First, the recording system shall be subjected to known voltages and its measured value compared against a laboratory standard voltmeter, whose calibration is traceable to the National Bureau of Standards. Secondly, a transducer simulator shall be calibrated to verify its stability and balance. Thirdly, as a data acquisition system check, the transducer simulator shall be installed in place of the transducer. A two step calibration step shall be applied from the simulator. The two values recorded shall be repeated a prescribed number of times to verify repeatability.

Calibration of the recording system alone and transducer simulator shall be conducted every three months. This interval may be extended to a maximum of 4-1/2 months. Calibration of the data acquisition system with simulator shall be conducted every week. If, after four calibrations, the data are acceptable, the calibration interval may be increased to one month.

## C. APPLICABLE DOCUMENTS

Government Documents

Mil Q-9858A	Program Quality Assurance Requirements
Mil C-45662A	Calibration Requirements

ASPC

QCI-0203(d)	System Calibrations
QCI-0204(a)	Laboratory Calibrations
QCI-0205(c)	Laboratory Operations
QCI-0206(a)	Calibration Intervals

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\* This procedure was instituted at the time that a more stringent DAS accuracy requirement was levied and the new DAS (described in Appendix L) was assembled. The procedure was not used for the DAS described in Appendix G or in connection with any of the data discussed in this volume of the report

ASPC (cont)

QCI-0207	Calibration Procedures
QCI-2001(e)	Qualification Tags and Stickers
ACP 2259B	System Calibration
ACP 2260	Control Room Operations

D. APPLICABLE EQUIPMENT

<u>Manufacture</u>	<u>Model</u>	<u>Description</u>
Hewlett-Packard	3450B	Multifunction Meter
John P. Fluke	332A	D.C. Voltage Standard
John P. Fluke	335A	D.C. Voltage Standard
Leeds & Northrup	4232B	Wheatstone Bridge
Leeds & Northrup	8686	Potentiometer
Brooklyn, Inc.	-	Thermometer, Gages (0 to 120°F)

E. CALIBRATION OF THE RECORDING SYSTEMS (WITHOUT BRIDGE COMPLETION)

1. ASPC Test Operations Data Acquisition System

This system shall be calibrated in accordance with Aerojet Calibration Procedure 2259B and 2260.

2. ASPC Propellant Development Lab DAS Procedure

a. Using the shop standard precision voltage source apply voltages of 0.0,  $\pm 1.00$ ,  $\pm 2.00$ ,  $\pm 3.00$ ,  $\pm 4.00$ ,  $\pm 5.00$ ,  $\pm 6.00$ ,  $\pm 7.00$ ,  $\pm 8.00$ ,  $\pm 9.00$ ,  $\pm 10.00$ ,  $\pm 20.00$ ,  $\pm 30.00$ ,  $\pm 40.00$ ,  $\pm 50.00$ ,  $\pm 60.00$ ,  $\pm 70.00$ ,  $\pm 80.00$ ,  $\pm 90.00$ ,  $\pm 100.00$  millivolts  $\pm 0.02$  millivolts, to a randomly selected channel. All other channels shall be subjected to 0.00,  $\pm 5.00$ ,  $\pm 10.00$ ,  $\pm 50.00$ , and  $\pm 100.00$  millivolts,  $\pm 0.02$  millivolts.

b. Repeat Paragraph 1 above after a 24 hour period twice, without adjustments.

c. Cross talk shall be checked on a randomly selected channel by subjecting adjacent channels to plus and minus overscale input voltages and recording any changes to the reading of the selected channel.

d. Common Mode Rejection

(1) D. C.

A common mode voltage of 10 volts D.C. shall be applied to an unbalanced input of 1,000 ohms of a randomly selected channel.

(2) A.C.

A common mode voltage of 5 volts RMS, 60 hz shall be applied to an unbalanced input of 1,000  $\Omega$  of a randomly selected channel.

e. Acceptance Criteria

(1) Zero effects must be less than 2.00 millivolts. The DAS must be capable of reproducing the input voltage to within  $\pm 0.1$  millivolt or  $\pm 1.5\%$  whichever is greater.

(2) Cross talk shall not exceed  $+ .1$  millivolt of the value read on the selected channel just prior to the application of the overscale voltages.

(3) Common mode rejection shall be -80 db or better for both AC and DC.

F. CALIBRATION OF THE SIMULATOR

1. Purpose

The transducer simulator shall consist of two balanced 500 ohm resistors and a 550 ohm bridge unbalance resistor. The purpose of the simulator is to serve as a stable source which can be installed in place of the transducer so that an expected output can be applied.

2. Calibration

The purpose of the calibration is to verify that the simulator is stable over time. Therefore, an excitation voltage of 10.0000,  $\pm 0.0002$  volts shall be applied and the voltage at the midpoint of the simulator shall be measured with and without the shunt resistor. Ten readings shall be taken for each 24 hours for three consecutive days. These readings shall be taken at  $+30$ ,  $+60$ , and  $+90^\circ\text{F}$ .

3. Specifications

The voltage measurements taken at each temperature shall not vary more than  $\pm 0.2$  mv for any corresponding readings taken. The same shall be true for the measurements taken with the shunt step applied.

G. SYSTEM SIMULATION

1. Purpose

The purpose of the system simulation is to provide a cursory check of the system excluding the transducer. It will serve to check system components and verify proper operation of the bridge completion network. It also will provide a means of monitoring and checking system drift.



## 2. Calibration

### a. First System Simulation

To assure that the common mode voltage does not exceed 5 volts RMS AC, the AC voltage on all input lines will be measured with respect to ground when the DAS is first attached to the motor instrumentation. As a further check dummy bridges will be limit checked to  $\pm 0.1$  mv. The DAS will be programmed to check these bridges and flag the system if any of the outputs exceed the limits.

- (1) Install the simulator in place of the transducer.
- (2) Read the voltage from each output leg to the negative input leg.
- (3) Take ten readings of the zero and shunt step. (A reading is defined as the average of 100 samples or greater).
- (4) Record all data.
- (5) Repeat for each channel.
- (6) Wait 24 hours.
- (7) Repeat paragraphs (1) through (6) three times.

### b. Subsequent System Simulations

Perform paragraphs (1) through (5) above one time.

## 3. Specifications

### a. First System Simulation

- (1) Calculate the mean and standard deviation of the sampled readings at the zero and the shunt step values.

$$\bar{X} = \frac{1}{n} \sum_{i=1}^n X_i, \quad S = \sqrt{\frac{\sum_{i=1}^n (X_i - \bar{X})^2}{n}}$$

- (2) "S" must be less than  $\pm 0.2$  millivolt or 4% whichever is greater.

- (3) The value of  $\bar{X}_0$  and  $\bar{X}_{\text{shunt}}$  and S then is to be recorded and compared with each subsequent system simulation.

b. Subsequent Simulations

(1) Repeat paragraph a(1) above.

(2) The recorded values of  $\bar{X}_0$  and  $\bar{X}_{shunt}$  shall be within  $\pm 0.2$  millivolts or 3.0% whichever is greater of the original values of  $\bar{X}_0$  and  $\bar{X}_{shunt}$ . The value of  $S_0$  and  $S_{shunt}$  shall be checked to determine the standard deviation and shall not exceed  $\pm 0.2$  mv or 3.0% whichever is greater.

H. FAILURE TO MEET SPECIFICATIONS

1. Recording System

If the recording system fails to meet specifications, it shall be adjusted and/or repaired until it does meet specifications. Or, if it is determined that, once the system is in operation, it is unfeasible to move the equipment to the laboratory, the system and/or data shall be adjusted to account for obvious shifts in calibration.

2. Transducer Simulation

If the transducer simulator fails to meet specifications, it (they) shall be reworked or replaced.

3. System Simulation

Should the system simulation fail to meet specifications, the recording system shall be checked to verify its proper operation. If the recording system is within specifications, an attempt will be made to repair the bridge completion network and/or its associated wiring. If the channel cannot be brought into specification, it shall be suspended from service.

APPENDIX N

PRESSURE CALIBRATION AND STABILITY

TESTING OF GAGES



## PRESSURE CALIBRATION AND STABILITY TESTING OF GAGES

This appendix describes the steps followed while conducting the pressure calibrations and stability tests of the gages in Motor No. 1.

## A. MOTOR PREPARATION AND TEST SET-UP

1. Connect gage and test circuits to the DAS according to the channel assignments given in Table N-1.
2. Connect thermocouples to a separate strip-chart recorder according to the channel assignments given in Table N-2.
3. Calibrate and connect 100 psi pressure transducer onto pressure line and to DAS channel 60. Provide shunt step for certain % of F.S.
4. Connect Heise gage near pressure controls.
5. Test pressure system to ensure that it will hold and maintain 0 to 50 psi nitrogen gas pressure for times up to 50 minutes. Conduct leak check at 15 psi gas pressure.

## B. DAS STABILITY TESTS

1. Take readings according to the Integrated Test Instructions (I.T.I.) given below, (Table N-3). This I.T.I. will be initiated each week during the stability tests.
2. Conduct tests six times over a period of 15 weeks.
3. Reduce test data in terms of means ( $\bar{X}$ ) and standard deviation (S).
4. Analyze data for drift prior to continuing.

## C. PRESSURE CALIBRATION OF GAGES

1. Take readings according to the Integrated Test Instructions given below.
2. Record all zero data five times with the motor vented to the atmosphere. (Data sampling time: approximately 1 hour).
3. Record all zero data two times in closed pressure system. (Data sampling time: 22 minutes).
4. Conduct the first set of pressure calibration tests according to the following sequence of test pressures: 0 psi, 5 psi, 10 psi, 20 psi, 30 psi, 40 psi, 50 psi, 20 psi, 10 psi, 0 psi. (Each pressure increment took 25 to 40 minutes to stabilize).

5. After system is stabilized, hold for 10 minutes prior to first data recordings.

6. Take two sampling runs (22 minutes) at given pressure.

7. Cognizant engineer will evaluate data to determine need for possible additional data sampling.

8. Continue to next pressure level and repeat steps e through g.

9. After completing the test series (see d above) allow one day for grain recovery

10. Conduct second set of pressure calibration tests.

11. Reduce the test data as discussed in Section 11 of the report.

#### D. INTEGRATED TEST INSTRUCTIONS

A copy of these detailed test instructions are given in Table N-3.

TABLE N-1  
DAS CHANNEL ASSIGNMENTS FOR GAGES AND TEST CIRCUITS

<u>Channel No.</u>	<u>Gage on Circuit</u>	<u>Channel No.</u>	<u>Gage on Circuit</u>
1	Zero Reference Channel	36	N7-1
2	Empty	37	S-10
3	N2-2	38	S-1
4	N10-2	39	N3-2
5	N5-2	40	Power supply voltage divider
6	N3-1	41	Power supply voltage divider
7	Empty	42(+11.40V)	Dummy Resistor
8	Empty	43(-11.00V)	Full Bridges
9	N9-2	44(-10.80V)	"
10	N4-1	45(+11.05V)	"
11	N1-1	46 392.75	Volt Div.
12	Empty	47 400 mv	Std. Voltage
13	N11-2	48	Empty
14	S-8	49	N7-2
15	N2-1	50	M6-2
16	S-11	51	LVDT-1
17	S-6	52	LVDT-2
18	N8-2	53	LVDT-3
19	S-14	54	LVDT-4
20	N9-1	55	LVDT-5
21	N11-1	56	LVDT-6
22	S-9	57	Potentiometer-7 (0°)
23	N15-1	58	Potentiometer-8 (135°)
24	N10-1	59	Potentiometer-9 (180°)
25	N1-2	60	Pressure transducer
26	S-12		
27	S-5		
28	N4-2		
29	N5-1		
30	N15-2		
31	S-7		
32	N8-1		
33	N6-1		
34	S-3		
35	S-4		

Gages S-2 and S-13 were bad gages and were not connected.



TABLE N-2

## RECORDER CHANNEL ASSIGNMENTS FOR THERMOCOUPLES

<u>Channel Number</u>	<u>Thermocouple Number</u>
1	1
2	2
3	3
4	4
5	5
6	6
7	7
8	8
9	9
10	10
11	11
12	Blank *
13	20
14	21
15	22
16	16
17	Blank **
18	18
19	19
20	12
21	14
22	15

---

\* Thermocouple 13 not installed

\*\* Thermocouple 17 bad

AEROJET SOLID PROPULSION COMPANY  
INTEGRATED TEST INSTRUCTIONS

PAGE	1	OF	4
PREPARED BY	A. I. Imagine		
DATE	7/7/75		
APPROVED			
APPROVED			
REFERENCE	DAS System Certification		

ITI NO.	00845
FACILITY	
TEST NO	
UNIT S. N	

TABLE N-3  
FLEXIBLE CASE GRAIN  
INTERACTION  
GAGE STABILITY TESTS

INITIAL		OPERATION
T.D.	INSP.	
<p><u>GENERAL INSTRUCTIONS</u></p> <p>As each step is completed, the individual performing the test or examination shall initial the space provided as verification of completion. All entries shall be made in a legible manner, in ink, preferably black.</p> <p>When a characteristic of this checklist is not complied with, record its number in the operations column and notify the cognizant engineer and Inspection, who will initiate an I. R. Record on the I.R. information that is necessary to give a complete description of the discrepancy.</p> <p>During actual processing, revisions to the checklist or requirements contained therein will be made in the operations column. Changes will be initiated by the cognizant test engineer and entered in ink. Quality Engineering approval of the change must be obtained prior to use of the data for acceptance purposes.</p> <p>Upon completion of this checklist, it shall be delivered to the Project Engineer.</p> <p><u>SPECIAL INSTRUCTIONS</u></p> <p>Notify AFPRO/QE J. Simperman, 355-3885, at beginning of test.</p> <p>1.0 PROCEDURE</p> <p>1.1 <u>System Simulated</u></p> <p>1.1.1 The DAS shall be checked with the simulated transducer networks each week until four consecutive checks indicate that the data remains unchanged, at which time the interval shall be changed to one month. The mean value shall not deviate from the baseline data more than <math>\pm 0.10</math> mv or <math>\pm 1.5\%</math> of reading, whichever is larger. Ten readings shall be taken at each simulation.</p> <p>1.2 <u>Transducer Output Measurement</u></p> <p>1.2.1 The transducer excitation voltage must be applied for a minimum of 16 hours but not more than 24 hours prior to taking measurements. The excitation to the transducers must be off for a minimum of 24 hours prior to application of voltage.</p>		

INITIAL		OPERATION
T.O.	INSP.	
		<p>1.3 <u>Sequence of Operation</u></p> <p>1.3.1 The weekly sequence of operation shall be as follows:</p> <p>1.3.1.1 On Monday (or Tuesday) of each week system simulation shall be performed. There is no warm-up on-off period restriction. The data shall be compared to data established during system certification, if the data falls within specification. Measure and record common mode voltage.</p> <p>1.3.1.2 Re-hook up the transducers and at 1600 hours turn on excitation voltage. Insure that it is 28.00 VDC.</p> <p>1.3.1.3 On the day following simulation and at least 16 hours but not more than 24 hours after application of excitation voltage take five readings of each channel. Record time for each reading. Record barometric pressure to the nearest hundredth of an inch (Mather AFB 364-4377). Attach temperature recorder chart to data. Turn power off.</p> <p>1.3.1.4 Wait 24 hours and repeat steps 2 and 3.</p> <p>1.4 This I.T.I. shall be initiated each week.</p> <p>2.0 SYSTEM SIMULATION</p> <p>2.0.1 Date and time started _____ Date Time</p> <p>2.0.2 Verify DAS is in calibration.</p> <p>2.0.3 Hook DAS inputs to transducer simulators.</p> <p>2.0.4 Turn on bridge excitation. Set to 28.00 VDC.</p> <p>2.0.5 Measure AC voltage on each input line to ground. Verify that the common mode voltage is less than 5V. RMS.</p> <p>2.0.6 Take ten (10) readings on simulated channels.</p> <p>2.0.7 Secure system. Turn off power.</p> <p>2.0.8 Reterminate DAS inputs to transducers.</p> <p>Date and time completed _____ Date Time</p>



INITIAL		OPERATION
T.O.	INSP.	
		<p>2.0.9 Verify that system simulation <math>\bar{x}</math> and sigma values meet specification requirements of certification data + _____ mv.</p> <p>2.1 <u>Transducer Output Measurement</u></p> <p>NOTE: This part of the procedure is not to be started unless the transducer excitation has been turned off for at least 24 hours. When excitation is turned on readings may not be taken until the excitation has been on for a minimum of sixteen (16) hours but not more than twenty-four (24) hours.</p> <p>2.1.1 First readings -</p> <p>2.1.1.1 At 1600 hours on the day prior to taking readings turn on transducer excitation set at +28.00 VDC.</p> <p>Date and time turned on _____ Date Time</p> <p>Date and time system will be ready for readings _____ Date Time</p> <p>2.1.1.2 At time and date system is ready to record (Para. 2.1.1.1 above) take five readings on DAS.</p> <p>Record date and time on printout.</p> <p>2.1.1.3 Call Mather AFB 364-4377 for barometric pressure and record _____ psia.</p> <p>2.1.1.4 Remove temperature record noting date and time removed and attach to printout.</p> <p>2.1.1.5 Shut-off excitation. Time and date _____ Date Time</p> <p>2.1.1.6 Deliver data to instrument engineer in charge.</p>

INITIAL		OPERATION
T. D.	INSP.	
		<p>2.1.2 Second readings -</p> <p>2.1.2.1 Verify power has been off for a minimum of 24 hours (Para. 2.1.1.4). Record date and time _____.</p> <p style="text-align: center;">Date                      Time</p> <p>2.1.2.2 Turn on transducer excitation. Set for 28.00 VDC. Record date and time power turned on _____.</p> <p style="text-align: center;">Date                      Time</p> <p>2.1.2.3 System ready for readings at _____ Date _____.</p> <p style="text-align: center;">Time</p> <p>_____ (16 hours from date and time in 2.1.2.2).</p> <p>2.1.2.4 Take readings at time specified in Para. 2.1.2.3 above. Record all channels five times. Note date and time on printout.</p> <p>Date and time data recorded _____.</p> <p style="text-align: center;">Date                      Time</p> <p>2.1.2.5 Record barometric pressure (call Mather AFB 364-4377) on printout.</p> <p>2.1.2.6 Strip temperature record and affix to printout.</p> <p>2.1.2.7 Shut off excitation. Record date and time _____.</p> <p style="text-align: center;">Date                      Time</p> <p>2.1.2.8 Deliver data to instrumentation engineer.</p> <p>3.0 Verify ITI complete and signed off and N/A initialed.</p> <p style="text-align: right;">Test Engineer _____ Date _____</p>

APPENDIX O  
TRANSDUCER STABILITY

LETTER FROM  
E. KONIGSBERG

JUNE 24, 1976



# KONIGSBERG INSTRUMENTS, INC.

2000 East Foothill Boulevard Pasadena, California 91107

Telephone: 213 449-0016

June 24, 1976

Mr. Ken Bills  
Aerojet Solid Propulsion Co.  
Post Office Box 13400  
Sacramento, California 95813

Dear Mr. Bills:

Thank you for giving me an opportunity to comment on transducer stability: how to approach getting it, our experience with it, and to bring you up to date on some new developments in our laboratory. My comments will be brief; I shall be happy to amplify any portion of them at an appropriate forum.

To achieve stability (to whatever degree) the following should be done: (the order of presentation does not indicate importance; we discuss herein bonded strain gage devices although comparable criteria can be set forth for other devices we make, such as capacitance transducers)

1. Specify the degree of stability desired at the beginning of the program. Designing for stability, whether in instrument design or test program protocol must start with a hard number, whether as a design goal or a minimum requirement.

This figure must specify load conditions, environmental conditions, duration of desired stability; differentiation between allowable short term and long term drift rates; differentiation between conformance to a known, unidirectional drift rate and random excursions from an unpredictable baseline.

Such detail is not necessary if one can accept a large transducer (say a thin film type), with only moderate overload conditions, which is produced in large commercial quantities, and which can be replaced by the manufacturer if it falls out of specification. But if one wants to resolve 1 psi in a 450 psi transducer the size of a shirt button, with a redundant backup sensor, in an inaccessible location, and such devices must be manufactured to order in limited quantities, very great care must be taken to define what is desired.

2. Regardless of how transducer stability is defined, the tests to be performed which define acceptable instruments should be set forth. I differentiate between "desired stability" and "tests performed" so that the nature of the



Mr. Ken Bills

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compromises which must be made between economically feasible testing and sufficiently predictive testing are set forth, and the implicit limitations be realized. Is 0.1% F.S. stability per month under no load, room temperature conditions, unpotted, in the second month after manufacture predictive of 1% F.S. per month stability after one year in the field, in situ, under load conditions, after prior proof pressure testing? How does one know, at reasonable cost, the relationship between test and field conditions, for a new design?

3. Regardless of how stability or test conditions are defined, we believe that for miniaturized, hand assembled, limited quantity production devices, test conditions should include

- a. Separate test beams, instrumented by the same techniques, to separate the variables of transducer design from proper sensor installation techniques.
- b. 100% stability testing, all transducers, all lots. We test for at least one month -- ! -- on our commercial implantable products, (which are installed in "inaccessible" locations), we have found it pays, and we will be expanding the scope of our testing program. Of necessity, this involves additional delay in delivery of from 2 - 3 months.

4. Field measurements of transducer characteristics (not output, but bridge impedance, comp resistor values, megohm leakage to ground, forward and reverse resistance) and of signal conditioner characteristics (using very high stability dummy bridges) must be routinely incorporated in test procedures, and transducer anomalies reported regularly to the manufacturer. Recovered failed units should have thoroughgoing failure analysis by both user and manufacturer.

5. A long term transducer development and supply program should be instituted. Long term does not mean sequential short term, rapid procurement contracts. I suggest 2 to 5 years with one or two sources is realistic. The small size and comparatively low cost (as compared to test program costs) of individual transducers does not mean that the instruments are not complex. Further, if experience in design and manufacture of instruments is diffuse or intermittent, programs may, perforce, always have to start anew.

6. All desired parameters of transducers besides stability should be specified in mechanical or electrical terms. It is not enough to state that a transducer is not interactive with some propellant, nor that it be intrinsically safe. The stiffness of a diaphragm, whether in  $\mu\text{in}/\text{psi}$  deflection at the center, or volumetric compliance ( $\text{in}^3/\text{psi}$ ), or ratio of deflection to radius, or whatever, the relationship between each gage strain and diaphragm distortion, etc., should be stated unambiguously. Ditto voltages, current, heat rise, etc.



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For whatever it is worth, I have yet to see a transducer specification which has adequately addressed itself to all the parameters significant to propellant stress testing. One of the nice cop-outs is to specify an "error-band," so one can be a purist and say that was (was not) considered in the specification. I do not mean to be contentious, but avoidance of detail in specification does avoid tediousness in reading or writing of the specification, does specify a simplistic and (perhaps) attractive error band, but promulgates downstream exceptions to the specification or arguments as to just what the error band meant.

Now a word about our more recent experience:

- a. As noted above, we 100% stability test all our pressure transducers, whether or not customer required, as an internal control device.
- b. We now EB weld all transducers required to be stable. EB welding does not improve stability; it does seem to assure that initially observed stability is continued.
- c. Our standard (epoxied gages) implantable line (Titanium) seems to slip between 1 - 2  $\mu\epsilon$ /month (top 25%), 2 - 4  $\mu\epsilon$ /month (top 50%). I would expect better results from epoxy gaged steel or Kovar, but we have no current data.
- d. We believe anodic bonding (strain gages to glass to Kovar) offers great promise for stability: approximately 0.5  $\mu\epsilon$  per year seems a realistic goal. We are now test bonding gages in our clean room lab.
- e. We have developed miniature Titanium diaphragm/Kovar beam structures with great sensitivity. In a 2 mm. x 8 mm. dia. housing (.08 x .31 inch dia.) we have achieved outputs of 50 mV/psi with good overrange protection. Our 6 psi design is better than 0.25% accurate, our 1 psi design is 0.5% accurate, using epoxied strain gages. Stability tests on the beamed designs have not been run.
- f. We are building miniature (.10 x .28 inch dia.) capacitance transducers, some with internal hybrid electronic circuitry. Results will be duly reported.

I should note that our anodic bonding development is less concerned with the basic process itself than with its adaptation to very small devices, which has required development of smaller strain gages, miniature beams, special assembly fixtures and the like.

As you know, our recent R & D work has been more intensive in multi axis micro-miniature accelerometers, to sense both linear and angular motion. This has



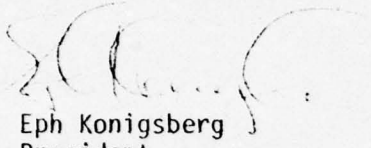


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required us to move to computer design approaches, the results of which should shortly be applicable to our pressure transducer developments. Please let me know if our recent experience can be helpful to you in your own development work.

Yours very truly,



Eph Konigsberg  
President

EK/mar



APPENDIX P

CALCULATED INTERFACE STRESSES  
FOR THE FLEXIBLE CASE MOTOR  
UNDER 50 PSIG INTERNAL  
PRESSURIZATION AND ONE-G  
LATERAL ACCELERATION

CALCULATED INTERFACE STRESSES FOR THE FLEXIBLE  
CASE MOTOR UNDER 50 PSIG INTERNAL PRESSURIZATION AND ONE-G  
LATERAL ACCELERATION

Stress analyses were made of the Minuteman third stage motor when subjected to an internal pressure of 50 psi for up to 10 minutes at temperatures of 30°F, 77°F and 110°F. In addition to stress analysis of this motor subjected to 1 g lateral acceleration (resting in a horizontal position) at 110°F was made.

The pressurization analyses were performed using the TEXGAP finite element elastic computer program for the grid shown in Figure P-1 and the SA011 1-dimensional viscoelastic computer program on a cross section of the motor at mid length. The 1 g lateral acceleration analysis was performed using the TEXGAP finite element elastic computer program using the same grid as for the pressurization.

The results are to be compared with the gage readings and consequently only the results at the gages are given. Figure P-2 shows the correlation between the gage location code numbers and the gage numbers. Figure P-3 shows the interface radial stress at gage location code No. 4 versus time for internal pressure of 50 psi held for 10 minutes using average aged propellant properties ( $T = 30^{\circ}\text{F}$ ).

Table P-1 shows the interface normal and shear stresses at all the gage location code numbers for 50 psi internal pressure held for 10 minutes at 30°F, 77°F and 110°F. Table P-2 shows the interface normal and shear stresses at all the gage location code numbers for 1 g lateral acceleration at 110°F (0° azimuth is at the bottom).



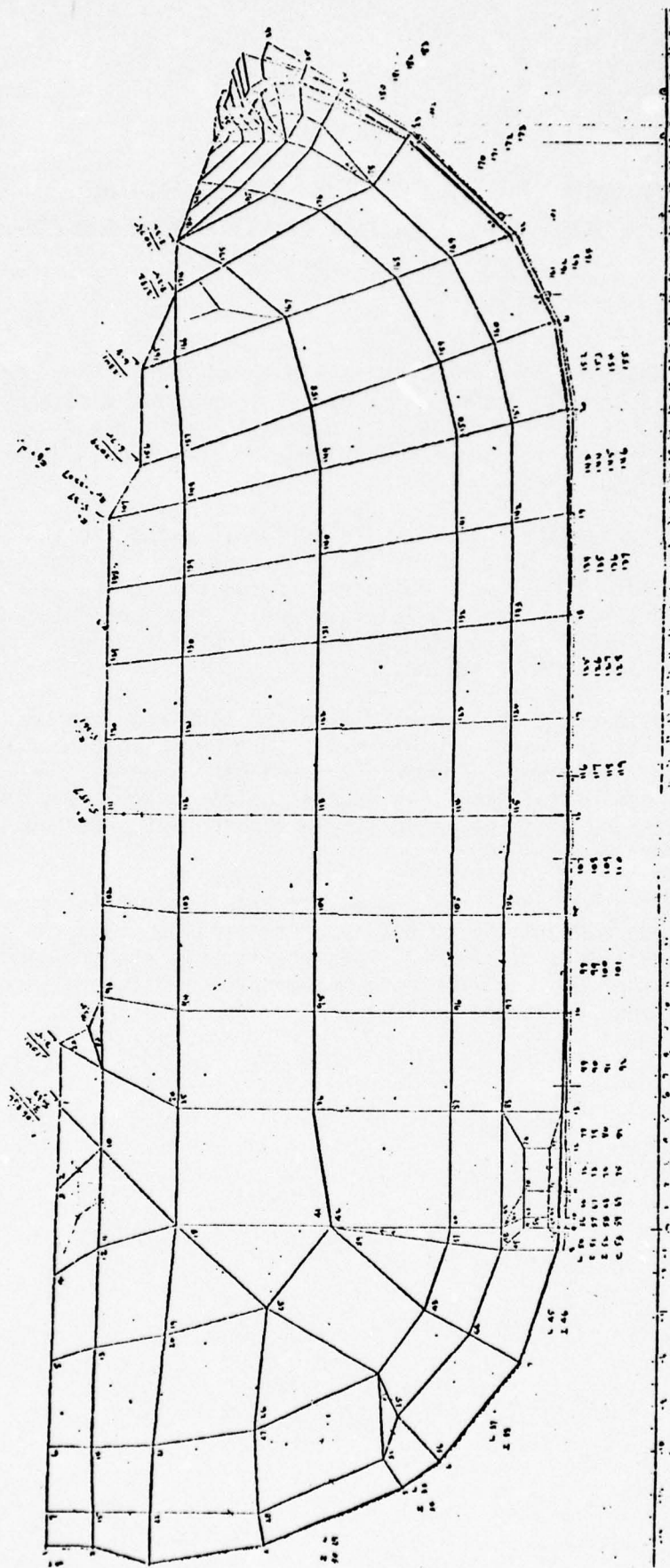


FIGURE P-1. FINITE ELEMENT GRIDWORK USED IN THE ANALYSIS OF THE MINUTEMAN III  
THIRD STAGE MOTOR

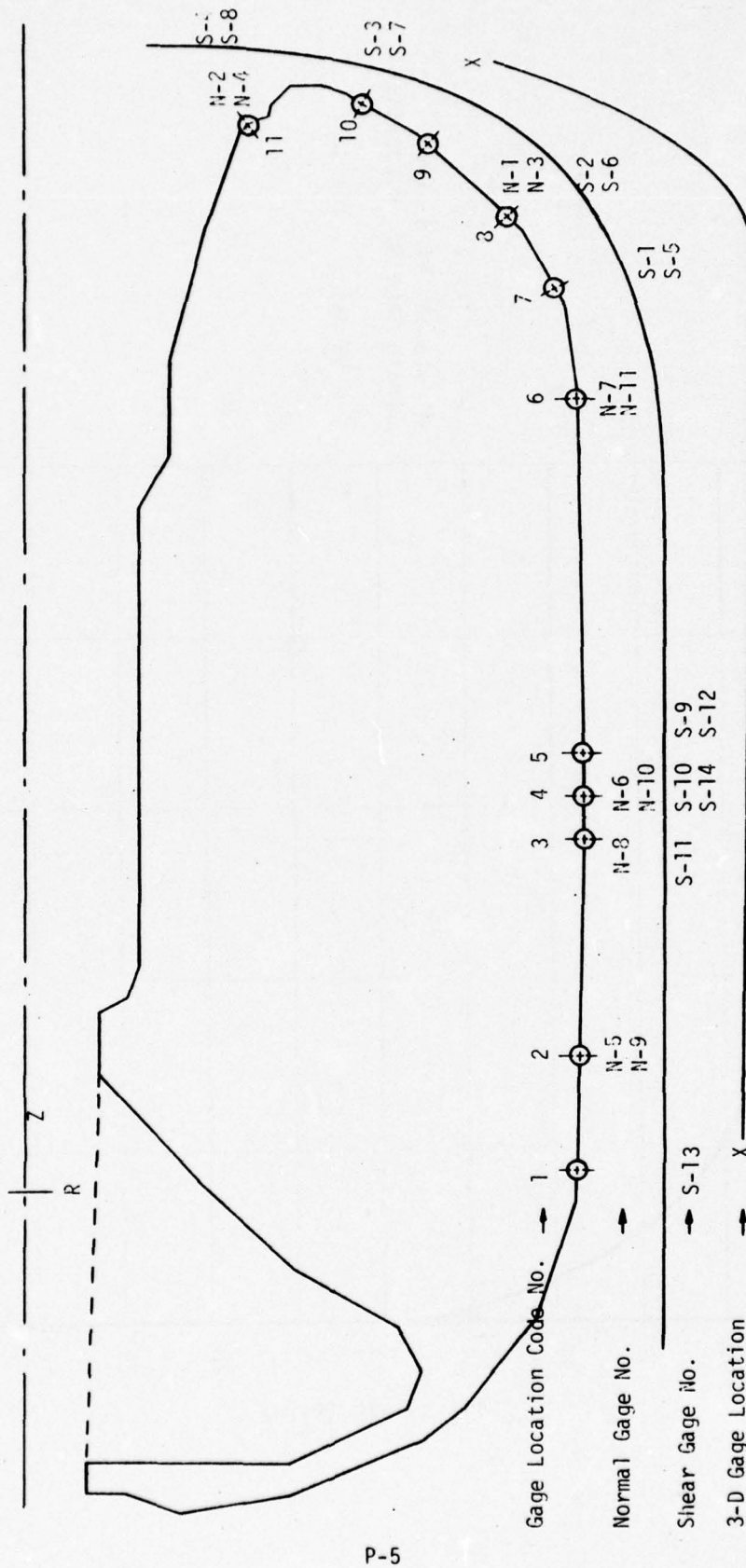


FIGURE P-2. CORRELATION BETWEEN GAGE LOCATION CODE NUMBERS AND GAGE NUMBERS

Minuteman Third Stage  
Strain Gage No. 4 Location  
P = 50 psi      T = 30°F

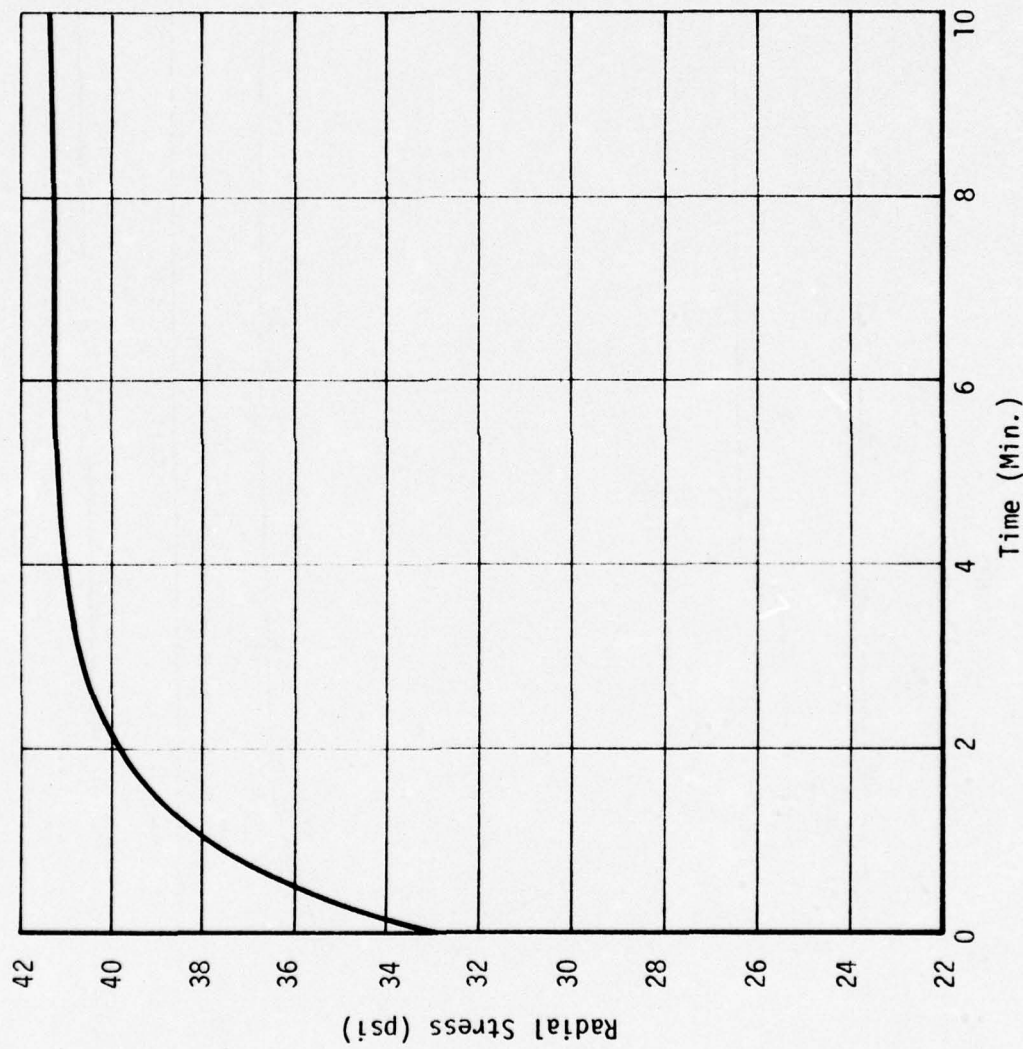


FIGURE P-3. INTERFACE RADIAL STRESS VS TIME FOR AVERAGE  
AGED PROPELLANT PROPERTIES



TABLE P-1

INTERFACE NORMAL AND SHEAR STRESSES AT ALL THE  
GAGE LOCATION CODE NUMBERS FOR 50 P.S.I.  
INTERNAL PRESSURE HELD FOR 10 MINUTES  
AT 30°F, 77°F AND 110°F

Gage Location Code No.	$\sigma_N(\text{psi})$	$\tau_{rz}(\text{psi})$	$\sigma_N(\text{psi})$	$\tau_{rz}(\text{psi})$	$\sigma_N(\text{psi})$	$\tau_{rz}(\text{psi})$	
	+1 $\sigma$ Modulus		Mean Modulus		-1 $\sigma$ Modulus		
	(Ep = 619 psi)		(Ep = 530 psi)		(Ep = 441 psi)		
1	-47.90	5.012	-48.3	4.32	-48.53	3.60	Temperature = 30°F
2	-46.31	2.66	-46.8	2.30	-47.3	1.91	
3	-45.84	1.64	-46.4	1.72	-47.0	1.19	
4	-45.66	1.47	-46.2	1.28	-46.9	1.07	
5	-45.47	1.31	-46.1	1.14	-46.7	.95	
6	-43.95	.71	-44.2	.51	-45.2	.43	
7	-44.55	.30	-47.0	.27	-47.5	.23	
8	-45.16	-.12	-45.6	-.08	-46.3	-.06	
9	-44.9	-.12	-44.9	-.08	-45.7	-.06	
10	-45.2	+.3	-45.2	+.27	-46.0	+.23	
11	-34.5	3.7	-36.0	3.11	-38.4	2.59	
<hr/>							
	(Ep = 369 psi)		(Ep = 316 psi)		(Ep = 263 psi)		Temperature = 77°F
1	-48.8	3.01	-49.0	2.57	-49.2	2.14	
2	-47.8	1.60	-48.1	1.37	-48.4	1.14	
3	-47.5	1.20	-47.9	1.03	-48.2	0.85	
4	-47.4	0.89	-47.7	0.76	-48.1	0.64	
5	-47.3	0.79	-47.7	0.68	-48.1	0.56	
6	-46.0	0.36	-46.5	0.30	-47.1	0.25	
7	-47.9	0.19	-48.2	0.16	-48.5	0.13	
8	-46.9	-0.06	-47.4	-0.05	-47.8	-0.04	
9	-46.4	-0.06	-46.9	-0.05	-47.5	-0.04	
10	-46.7	0.19	-47.1	0.16	-47.6	0.13	
11	-40.3	2.17	-41.6	1.85	-43.1	1.54	
<hr/>							
	(Ep = 306 psi)		(Ep = 262 psi)		(Ep = 218 psi)		Temperature = 110°F
1	-49.0	2.49	-49.2	2.14	-49.3	1.78	
2	-48.2	1.33	-48.4	1.14	-49.5	0.95	
3	-47.9	0.99	-48.2	0.85	-48.5	0.71	
4	-47.8	0.74	-48.1	0.63	-48.4	0.53	
5	-47.7	0.66	-48.1	0.56	-48.4	0.47	
6	-46.6	0.29	-47.1	0.25	-47.6	0.21	
7	-49.3	0.16	-48.5	0.13	-48.8	0.11	
8	-47.5	-0.05	-47.8	-0.04	-48.2	-0.03	
9	-47.1	-0.05	-47.5	-0.04	-47.9	-0.03	
10	-47.2	0.16	-47.6	0.13	-48.0	0.11	
11	-41.9	1.80	-43.1	1.54	-44.2	1.28	

TABLE P-2  
INTERFACE NORMAL AND SHEAR STRESSES FOR 1g  
LATERAL ACCELERATION LOADING CONDITION T = 110°F

Gage Location Code No.	0° Azimuth			180° Azimuth			90° and 270° Azimuths			
	$\sigma_N$ (psi)	$\tau_{rz}$ (psi)	$\tau_{r\theta}$ (psi)	$\sigma_N$ (psi)	$\tau_{rz}$ (psi)	$\tau_{r\theta}$ (psi)	$\sigma_N$ (psi)	$\tau_{rz}$ (psi)	$\tau_{r\theta}$ at 90° (psi)	$\tau_r$ at 270° (psi)
1	-3.686	-0.72	0	3.686	0.72	0	0	0	0.657	-0.657
2	-1.98	-0.111	0	1.98	0.111	0	0	0	0.196	-0.196
3	-1.663	+0.033	0	1.663	-0.033	0	0	0	0.069	-0.069
4	-1.628	0.038	0	1.628	-0.038	0	0	0	0.063	-0.063
5	-1.594	0.044	0	1.594	-0.044	0	0	0	0.057	-0.057
6	-1.50	0.035	0	1.50	-0.035	0	0	0	0.061	-0.061
7	-1.3	0.020			-0.020					
8	-1.08	0.030			-0.030					
9	-1.0	0.044			-0.044					
10	-0.80	+0.050			-0.050					
11	-0.562	-0.255			+0.255					

APPENDIX Q

NONLINEAR GAP PROGRAM

(TEXGAP-2D NONLINEAR)

MATRIX DERIVATIONS

AND

INPUT INSTRUCTIONS



NONLINEAR GAP PROGRAM  
(TEXGAP-2D NONLINEAR) MATRIX DERIVATIONS AND  
INPUT INSTRUCTIONS

A. INTRODUCTION

This appendix presents supporting information for Volume I, Section 14B, Geometrically Nonlinear Analysis of Axisymmetric Shells. Included are the derivations of the pertinent matrices for the geometric nonlinear analysis modifications and the input instructions which describe the nonlinear option cards required to execute the new program.

For the convenience of the reader the input instructions are given first, then the more detailed matrix derivations are presented.

B. INPUT INSTRUCTIONS

Reference Q-1 describes the input required for the basic TEXGAP program; only the command card format with the following changes are required to execute the nonlinear analysis.

COMMAND MODE CARD (Page 11 of Reference Q-1)

AXISYM, nldinc, [jprint], [ $a_r$ ,  $a_\theta$ ,  $a_z$ ,  $\omega$ ]

nldinc <0> = Number of load increments (20 max)

nldinc <0 = Only the linear analysis is performed.

if nldinc >0 = Two additional cards are required.

- 1) Control Parameters card
- 2) Load Scale Factors card

[jprint <4>] is the level of printed output required

jprint >2 stresses are output

jprint >3 strains are output

jprint >4 displacements are output

[ $a_r$ ,  $a_\theta$ ,  $a_z$ ,  $\omega$ , <0>] are the body force accelerations associated with radial, hoop, axial, and spin effects.

CONTROL PARAMETERS CARD (Skip if nldinc  $\leq 0$ )

nsinc  $\langle 1 \rangle$ , nnewt  $\langle 0 \rangle$ , nmnewt  $\langle 0 \rangle$ , itmax  $\langle 20 \rangle$ , tol  $\langle 0.001 \rangle$ , iout  $\langle 0 \rangle$

nsinc = Number of successive load increments during the loading path, at the end of which iterations are required to satisfy equilibrium.

nnewt = Number of successive Newton iterations following an incremental loading or a Modified Newton iteration.

nmnewt = Number of successive Modified Newton iterations following a Newton iteration.

itmax = Maximum number of iterations allowed at each specified load level.

Note:  $itmax \geq nnewt + nmnewt + nnewt + nmnewt + \dots$

tol = Error tolerance. Iterations stop when either the number of iterations exceed itmax or when:

$$\max \left| \frac{u_i^n - u_i^{n-1}}{u_i^n} \right| \leq tol$$

where:  $u_i^n$  is the displacement of  $i^{th}$  node at  $n^{th}$  iteration.

iout = Output option

- if iout =
- 0: No intermediate prints are required.
  - 1: Print displacements after each incremental loading and after each convergence.
  - 2: Print displacements after each incremental loading and after each iteration.

LOAD SCALE FACTORS CARD (Skip if nldinc  $\leq 0$ )

$f_1, f_2, f_3, \dots, f_{nldinc}$

Note: The load on the structure at  $i^{th}$  load increment will be  $f_i \times$  applied load.

B. MATRIX DERIVATIONS

Axisymmetric Problems

Geometrically Nonlinear Analysis

Strain Displacement Relations

$$\begin{aligned} e_{ij} &= \frac{1}{2} (u_{i|j} + u_{j|i}) + \frac{1}{2} u^k_{|i} u_{k|j} \\ &= e_{ij}^L + e_{ij}^{NL} \end{aligned} \quad (\text{Tensor Components})$$

where

$$e_{ij}^L = \frac{1}{2} (u_{i|j} + u_{j|i})$$

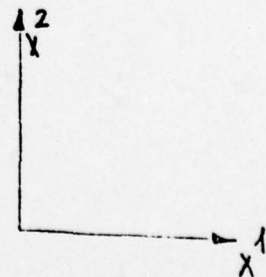
$$e_{ij}^{NL} = \frac{1}{2} u^k_{|i} u_{k|j}$$

For Axisymmetric Solids

$$e_{\alpha\beta} = \frac{1}{2} (u_{\alpha,\beta} + u_{\beta,\alpha} + u^{\mu}_{|\alpha} u_{\mu,\beta}) \quad \alpha, \beta, \mu = 1, 2$$

$$e_{\alpha 3} = 0$$

$$\begin{aligned} e_{33} &= \frac{r^2}{2} \left( 2 \frac{u_1}{r} + \frac{u_1^2}{r^2} \right) \\ &= r^2 \left( \frac{u_1}{r} + \frac{u_1^2}{2r^2} \right) \end{aligned}$$





Physical Components of Strain

$$\epsilon_{ij} = \sqrt{g^{ii}} \sqrt{g^{jj}} e_{ij}$$

$$u_i = u_i \sqrt{g^{ii}}$$

$$x^1 = r \quad x^2 = z \quad x^3 = \theta$$

$$g^{11} = 1 \quad g^{22} = \frac{1}{r^2} \quad g^{33} = 1$$

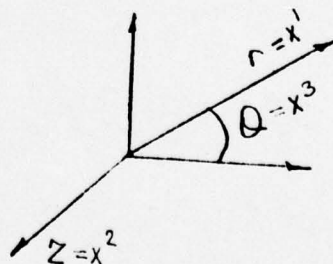
$$u_r = u_1 \quad u_z = u_2 \quad u_\theta = u_3 / r$$

$$\epsilon_{rr} = \epsilon_{rr}^L + \epsilon_{rr}^{NL} = \frac{\partial u_r}{\partial r} + \frac{1}{2} \left[ \left( \frac{\partial u_r}{\partial r} \right)^2 + \left( \frac{\partial u_z}{\partial r} \right)^2 \right]$$

$$\epsilon_{zz} = \epsilon_{zz}^L + \epsilon_{zz}^{NL} = \frac{\partial u_z}{\partial z} + \frac{1}{2} \left[ \left( \frac{\partial u_r}{\partial z} \right)^2 + \left( \frac{\partial u_z}{\partial z} \right)^2 \right]$$

$$\epsilon_{rz} = \frac{1}{2} \left( \frac{\partial u_r}{\partial z} + \frac{\partial u_z}{\partial r} \right) + \frac{1}{2} \left( \frac{\partial u_r}{\partial r} \frac{\partial u_r}{\partial z} + \frac{\partial u_z}{\partial r} \frac{\partial u_z}{\partial z} \right)$$

$$\epsilon_{\theta\theta} = \frac{u_r}{r} + \frac{u_r^2}{2r^2}$$



$$\underline{\underline{\epsilon}}_L^T = [ \epsilon_{rr}^L, \epsilon_{zz}^L, \epsilon_{\theta\theta}^L, 2\epsilon_{rz}^L ]$$

$$\underline{\underline{\epsilon}}_{NL}^T = [ \epsilon_{rr}^{NL}, \epsilon_{zz}^{NL}, \epsilon_{\theta\theta}^{NL}, 2\epsilon_{rz}^{NL} ]$$

Strain Energy

$$U = \frac{1}{2} \int_V \sigma_{ij} \epsilon_{ij} dV = \frac{1}{2} \int_V E_{ijkl} \epsilon_{kl} \epsilon_{ij} dV$$

$$= \frac{1}{2} \int_V \underline{\underline{\sigma}}^T \underline{\underline{\epsilon}} dV$$

$$\underline{\underline{\sigma}}^T = [ \sigma_{rr}^L, \sigma_{zz}^L, \sigma_{\theta\theta}^L, \sigma_{rz}^L ] + [ \sigma_{rr}^{NL}, \sigma_{zz}^{NL}, \sigma_{\theta\theta}^{NL}, \sigma_{rz}^{NL} ]$$

$$= \underline{\underline{D}} \underline{\underline{\epsilon}}$$

Therefore,

$$U = \frac{1}{2} \int_V \left( \underline{\underline{\epsilon}}_L^T \underline{\underline{D}} \underline{\underline{\epsilon}}_L + 2 \underline{\underline{\epsilon}}_L^T \underline{\underline{D}} \underline{\underline{\epsilon}}_{NL} + \underline{\underline{\epsilon}}_{NL}^T \underline{\underline{D}} \underline{\underline{\epsilon}}_{NL} \right) dV$$

Nodal point displacement vector

$$\underline{\underline{S}} = \begin{bmatrix} \underline{\underline{u}} \\ \underline{\underline{v}} \end{bmatrix}_{2n \times 1}$$

Where

$$\underline{\underline{u}}^T = [u_r^1, u_r^2, u_r^3, \dots, u_r^n]$$

$$\underline{\underline{v}}^T = [u_z^1, u_z^2, \dots, u_z^n]$$

n = total # of nodes in an element

$$U = \underline{\underline{S}}^T \left[ \frac{1}{2} \underline{\underline{K}} + \frac{1}{6} \underline{\underline{K}}_1 + \frac{1}{12} \underline{\underline{K}}_2 \right] \underline{\underline{S}}$$

Introducing the shape functions  $N^\alpha$  such that

$$u_i = N^\alpha u_i^\alpha$$

$$\alpha = 1, 2, \dots, n$$

$$i = 1, 2$$

or  $u_r = \underline{\underline{N}}^T \underline{\underline{u}}$

$$u_z = \underline{\underline{N}}^T \underline{\underline{v}}$$



$$\underline{\underline{C}}_L = \begin{bmatrix} C_{rr}^L \\ C_{zz}^L \\ C_{\theta\theta}^L \\ 2C_{rz}^L \end{bmatrix} = \begin{bmatrix} \underline{\underline{N}}_{,r}^T & 0 \\ 0 & \underline{\underline{N}}_{,z}^T \\ \frac{1}{r} \underline{\underline{N}}^T & 0 \\ \underline{\underline{N}}_{,z}^T & \underline{\underline{N}}_{,r}^T \end{bmatrix} \begin{bmatrix} \underline{\underline{u}} \\ \underline{\underline{v}} \end{bmatrix} = \underset{4 \times 2n}{\underline{\underline{B}}} \underset{2n \times 1}{\underline{\underline{S}}}$$

or

$$\underline{\underline{B}} = \begin{bmatrix} \underline{\underline{N}}_{,r}^T & 0 \\ 0 & \underline{\underline{N}}_{,z}^T \\ \frac{1}{r} \underline{\underline{N}}^T & 0 \\ \underline{\underline{N}}_{,z}^T & \underline{\underline{N}}_{,r}^T \end{bmatrix}_{4 \times 2n}$$

$$\underline{\underline{C}}_{NL} = \begin{bmatrix} C_{rr}^{NL} \\ C_{zz}^{NL} \\ C_{\theta\theta}^{NL} \\ 2C_{rz}^{NL} \end{bmatrix} = \frac{1}{2} \begin{bmatrix} \underline{\underline{u}}^T \underline{\underline{N}}_{,r} \underline{\underline{N}}_{,r}^T \underline{\underline{u}} + \underline{\underline{v}}^T \underline{\underline{N}}_{,r} \underline{\underline{N}}_{,r}^T \underline{\underline{v}} \\ \underline{\underline{u}}^T \underline{\underline{N}}_{,z} \underline{\underline{N}}_{,z}^T \underline{\underline{u}} + \underline{\underline{v}}^T \underline{\underline{N}}_{,z} \underline{\underline{N}}_{,z}^T \underline{\underline{v}} \\ \frac{1}{r^2} \underline{\underline{u}}^T \underline{\underline{N}} \underline{\underline{N}}^T \underline{\underline{u}} \\ \underline{\underline{u}}^T (\underline{\underline{N}}_{,r} \underline{\underline{N}}_{,z}^T + \underline{\underline{N}}_{,z} \underline{\underline{N}}_{,r}^T) \underline{\underline{u}} + \underline{\underline{v}}^T (\underline{\underline{N}}_{,r} \underline{\underline{N}}_{,z}^T + \underline{\underline{N}}_{,z} \underline{\underline{N}}_{,r}^T) \underline{\underline{v}} \end{bmatrix}_{4 \times 1}$$

$$= \frac{1}{2} \underset{4 \times 2n}{\underline{\underline{H}}}(\underline{\underline{S}}) \underline{\underline{S}} \quad \text{or } \frac{1}{2} \underline{\underline{S}}^T \underline{\underline{H}}^T$$

Note:  $\frac{\partial \underline{\underline{C}}_{NL}}{\partial \underline{\underline{S}}} = \underline{\underline{H}}(\underline{\underline{S}}) \quad \frac{\partial \underline{\underline{C}}_{NL}}{\partial \underline{\underline{S}}^T} = \underline{\underline{H}}^T(\underline{\underline{S}})$

Where

$$\underline{H}(\underline{s}) = \begin{bmatrix} \underline{s}^T \underline{H}_{rr} \\ \underline{s}^T \underline{H}_{rz} \\ \underline{s}^T \underline{H}_{\theta\theta} \\ \underline{s}^T \underline{H}_{rz} \end{bmatrix} = \begin{bmatrix} \underline{\bar{H}}_{rr}^T \\ \underline{\bar{H}}_{rz}^T \\ \underline{\bar{H}}_{\theta\theta}^T \\ \underline{\bar{H}}_{rz}^T \end{bmatrix}_{4 \times 2n}$$

$$\underline{H}_{rr} = \begin{bmatrix} \underline{N}_{r,r} \underline{N}_{r,r}^T & \underline{0}_{n \times n} \\ \underline{0} & \underline{N}_{r,r} \underline{N}_{r,r}^T \end{bmatrix}_{2n \times 2n}; \quad \underline{\bar{H}}_{rr}^T = \begin{bmatrix} \underline{u}^T \underline{N}_{r,r} \underline{N}_{r,r}^T & \underline{v}^T \underline{N}_{r,r} \underline{N}_{r,r}^T \end{bmatrix}_{1 \times 2n}$$

$$\underline{H}_{zz} = \begin{bmatrix} \underline{N}_{z,z} \underline{N}_{z,z}^T & \underline{0}_{n \times n} \\ \underline{0} & \underline{N}_{z,z} \underline{N}_{z,z}^T \end{bmatrix}_{2n \times 2n}; \quad \underline{\bar{H}}_{zz}^T = \begin{bmatrix} \underline{u}^T \underline{N}_{z,z} \underline{N}_{z,z}^T & \underline{v}^T \underline{N}_{z,z} \underline{N}_{z,z}^T \end{bmatrix}_{1 \times 2n}$$

$$\underline{H}_{rz} = \begin{bmatrix} \underline{N}_{r,z} \underline{N}_{z,z}^T + \underline{N}_{z,z} \underline{N}_{r,r}^T & \underline{0}_{n \times n} \\ \underline{0}_{n \times n} & \underline{N}_{r,z} \underline{N}_{z,z}^T + \underline{N}_{z,z} \underline{N}_{r,r}^T \end{bmatrix}_{2n \times 2n}; \quad \underline{\bar{H}}_{rz}^T = \begin{bmatrix} 2\underline{u}^T \underline{N}_{r,z} \underline{N}_{z,z}^T & 2\underline{v}^T \underline{N}_{r,z} \underline{N}_{z,z}^T \end{bmatrix}_{1 \times 2n}$$

$\underline{u}^T (\underline{N}_{r,r} \underline{N}_{z,z}^T + \underline{N}_{z,z} \underline{N}_{r,r}^T)$

$$\underline{H}_{\theta\theta} = \frac{1}{r^2} \begin{bmatrix} \underline{N} \underline{N}^T & \underline{0} \\ \underline{0} & \underline{0} \end{bmatrix}_{2n \times 2n}; \quad \underline{\bar{H}}_{\theta\theta}^T = \frac{1}{r^2} \begin{bmatrix} \underline{u}^T \underline{N} \underline{N}^T & \underline{0} \end{bmatrix}_{1 \times 2n}$$

$$\bar{K} = \bar{K}_\sigma + \bar{K}_r$$

Where  $\bar{K}_\sigma$  is the first order initial stress matrix and  $\bar{K}_r$  is the first order initial displacement (or rotation) matrix.

$$\bar{K}_\sigma = \begin{bmatrix} \bar{K}_\sigma & 0_{n \times n} \\ 0_{n \times n} & \bar{K}_\sigma \end{bmatrix}_{2n \times 2n}$$

where

$$\bar{K}_\sigma = \int_V \left[ \sigma_{rr}^L \tilde{N}_{,r} \tilde{N}_{,r}^T + \sigma_{zz}^L \tilde{N}_{,z} \tilde{N}_{,z}^T + \sigma_{rz}^L (\tilde{N}_{,r} \tilde{N}_{,z}^T + \tilde{N}_{,z} \tilde{N}_{,r}^T) + \frac{1}{r^2} \sigma_{\theta\theta}^L \tilde{N} \tilde{N}^T \right] dV$$

$$\bar{K}_\sigma = \int_V \left[ \sigma_{rr}^L \tilde{N}_{,r} \tilde{N}_{,r}^T + \sigma_{zz}^L \tilde{N}_{,z} \tilde{N}_{,z}^T + \sigma_{rz}^L (\tilde{N}_{,r} \tilde{N}_{,z}^T + \tilde{N}_{,z} \tilde{N}_{,r}^T) \right] dV$$

or

$$\bar{K}_\sigma = \bar{K}_\sigma + \int_V \frac{1}{r^2} \sigma_{\theta\theta}^L \tilde{N} \tilde{N}^T dV$$



$$\underline{\underline{K}}_1 = \int_V [\underline{\underline{H}}^T \underline{\underline{D}} \underline{\underline{B}} + \underline{\underline{B}}^T \underline{\underline{D}} \underline{\underline{H}}] dV$$

Similarly

$$\underline{\underline{K}}_2 = \underline{\underline{K}}_{2G} + \underline{\underline{K}}_{2r}$$

where  $\underline{\underline{K}}_{2r}$  is the second order initial stress matrix and  $\underline{\underline{K}}_{2G}$  is the second order initial displacement (or rotation) matrix

and

$$\underline{\underline{K}}_{2G} = \begin{vmatrix} \underline{\underline{K}}_{2G} \\ \underline{\underline{K}}_{2G} \end{vmatrix}$$

$$\underline{\underline{K}}_{2G} = \left( \frac{1}{2} \right) \int_V \left[ \sigma_{rr}^{NL} \underline{\underline{N}}_r \underline{\underline{N}}_r^T + \sigma_{zz}^{NL} \underline{\underline{N}}_z \underline{\underline{N}}_z^T + \sigma_{rz}^{NL} (\underline{\underline{N}}_r \underline{\underline{N}}_z^T + \underline{\underline{N}}_z \underline{\underline{N}}_r^T) \right] dV$$

$$\underline{\underline{K}}_{2G} = \underline{\underline{K}}_{2G} + \left( \frac{1}{r^2} \right) \int_V \sigma_{\theta\theta}^{NL} \underline{\underline{N}} \underline{\underline{N}}^T dV$$

$$\underline{\underline{K}}_{2r} = \int_V \underline{\underline{H}}^T(\underline{\underline{s}}) \underline{\underline{D}} \underline{\underline{H}}(\underline{\underline{s}}) dV$$

For Orthotropic Materials

$$\underline{D} = \begin{bmatrix} D_{11} & D_{12} & D_{13} & D_{14} \\ D_{21} & D_{22} & D_{23} & D_{24} \\ D_{31} & D_{32} & D_{33} & D_{34} \\ D_{41} & D_{42} & D_{43} & D_{44} \end{bmatrix}$$

$$\epsilon_{rr}^L = D_{11} \epsilon_{rr}^L + D_{12} \epsilon_{zz}^L + 2D_{14} \epsilon_{rz}^L + D_{13} \epsilon_{\theta\theta}^L$$

$$\epsilon_{zz}^L = D_{21} \epsilon_{rr}^L + D_{22} \epsilon_{zz}^L + 2D_{24} \epsilon_{rz}^L + D_{23} \epsilon_{\theta\theta}^L$$

$$\epsilon_{\theta\theta}^L = D_{31} \epsilon_{rr}^L + D_{32} \epsilon_{zz}^L + 2D_{34} \epsilon_{rz}^L + D_{33} \epsilon_{\theta\theta}^L$$

$$\epsilon_{rz}^L = D_{41} \epsilon_{rr}^L + D_{42} \epsilon_{zz}^L + 2D_{44} \epsilon_{rz}^L + D_{43} \epsilon_{\theta\theta}^L$$

where

$$\epsilon_{rr}^L = \underline{N}_{,r}^T \underline{u}$$

$$\epsilon_{zz}^L = \underline{N}_{,z}^T \underline{v}$$

$$2\epsilon_{rz}^L = (\underline{N}_{,z}^T \underline{u} + \underline{N}_{,r}^T \underline{v})$$

$$\epsilon_{\theta\theta}^L = \frac{1}{r} \underline{N} \underline{u}$$

Similarly

$$\underline{\underline{G}}_{NL} = \underline{\underline{D}} \underline{\underline{G}}_{NL}$$

where

$$G_{rr}^{NL} = \frac{1}{2} (\underline{u}^T \underline{N}_{,r} \underline{N}_{,r}^T \underline{u} + \underline{v}^T \underline{N}_{,r} \underline{N}_{,r}^T \underline{v})$$

$$G_{zz}^{NL} = \frac{1}{2} (\underline{u}^T \underline{N}_{,z} \underline{N}_{,z}^T \underline{u} + \underline{v}^T \underline{N}_{,z} \underline{N}_{,z}^T \underline{v})$$

$$G_{rz}^{NL} = \frac{1}{2} (\underline{u}^T \underline{N}_{,r} \underline{N}_{,z}^T \underline{u} + \underline{v}^T \underline{N}_{,r} \underline{N}_{,z}^T \underline{v})$$

$$G_{\theta\theta}^{NL} = \frac{1}{2r^2} \underline{u}^T \underline{N} \underline{N}^T \underline{u}$$

$$G_{rr}^{NL} = D_{11} G_{rr}^{NL} + D_{12} G_{zz}^{NL} + 2 D_{14} G_{rz}^{NL} + D_{13} G_{\theta\theta}^{NL}$$

$$G_{zz}^{NL} = D_{21} G_{rr}^{NL} + D_{22} G_{zz}^{NL} + 2 D_{24} G_{rz}^{NL} + D_{23} G_{\theta\theta}^{NL}$$

$$G_{\theta\theta}^{NL} = D_{31} G_{rr}^{NL} + D_{32} G_{zz}^{NL} + 2 D_{34} G_{rz}^{NL} + D_{33} G_{\theta\theta}^{NL}$$

$$G_{rz}^{NL} = D_{41} G_{rr}^{NL} + D_{42} G_{zz}^{NL} + 2 D_{44} G_{rz}^{NL} + D_{43} G_{\theta\theta}^{NL}$$



Strain Energy:

$$U = \underline{\underline{S}}^T \left[ \frac{1}{2} \underline{\underline{K}} + \frac{1}{6} \underline{\underline{K}}_1(\underline{\underline{S}}) + \frac{1}{12} \underline{\underline{K}}_2(\underline{\underline{S}}^2) \right] \underline{\underline{S}}$$

Equilibrium Equation:

$$\left[ \underline{\underline{K}} + \frac{1}{2} \underline{\underline{K}}_1(\underline{\underline{S}}) + \frac{1}{3} \underline{\underline{K}}_2(\underline{\underline{S}}^2) \right] \underline{\underline{S}} = \underline{\underline{f}}$$

Incremental Loading Method:

$$\left[ \underline{\underline{K}} + \underline{\underline{K}}_1(\underline{\underline{S}}_n) + \underline{\underline{K}}_2(\underline{\underline{S}}_n^2) \right] \Delta \underline{\underline{S}}_{n+1} = \Delta \underline{\underline{f}}_{n+1}$$

where for  $(n + 1)$  st loading

$$\Delta \underline{\underline{S}}_{n+1} = \underline{\underline{S}}_{n+1} - \underline{\underline{S}}_n$$

$$\Delta \underline{\underline{f}}_{n+1} = \underline{\underline{f}}_{n+1} - \underline{\underline{f}}_n$$

$$n = 0, 1, \dots, N_{\max}$$

Newton's Method (Iterative):

$$\left[ \underline{\underline{K}} + \underline{\underline{K}}_1(\underline{\underline{S}}_i) + \underline{\underline{K}}_2(\underline{\underline{S}}_i^2) \right] \Delta \underline{\underline{S}}_{i+1} = \underline{\underline{f}} - \left[ \underline{\underline{K}} + \frac{1}{2} \underline{\underline{K}}_1(\underline{\underline{S}}_i) + \frac{1}{3} \underline{\underline{K}}_2(\underline{\underline{S}}_i^2) \right] \underline{\underline{S}}_i$$

Where  $i$  = iteration number and iterations said to close when  $\Delta \underline{\underline{S}}_{i+1} \approx 0$

To calculate

$$\underline{\underline{B}}^T \underline{\underline{D}} \underline{\underline{H}}$$

$$\underline{\underline{B}}^T_{2n \times 4}$$

Multiply

$$(\underline{\underline{B}}^T \underline{\underline{D}})_{2n \times 4}$$

$$\underline{\underline{B}} = \underline{\underline{B}}^T \underline{\underline{D}} = \begin{bmatrix} \underline{\underline{N}}_{1,r} & 0 & \frac{1}{r} \underline{\underline{N}} & \underline{\underline{N}}_{1,z} \\ 0 & \underline{\underline{N}}_{1,z} & 0 & \underline{\underline{N}}_{1,r} \end{bmatrix}_{2n \times 4}$$

$D_{11}$	$D_{12}$	$D_{13}$	$D_{14}$
$D_{21}$	$D_{22}$	$D_{23}$	$D_{24}$
$D_{31}$	$D_{32}$	$D_{33}$	$D_{34}$
$D_{41}$	$D_{42}$	$D_{43}$	$D_{44}$

$4 \times 4$

$$\underline{\underline{B}} = \begin{bmatrix} \underline{\underline{B}}_{11} & \underline{\underline{B}}_{12} & \underline{\underline{B}}_{13} & \underline{\underline{B}}_{14} \\ \underline{\underline{B}}_{21} & \underline{\underline{B}}_{22} & \underline{\underline{B}}_{23} & \underline{\underline{B}}_{24} \end{bmatrix}_{2n \times 4}$$

$$\bar{\bar{B}}_{11} = D_{11} \tilde{N}_{,r} + D_{41} \tilde{N}_{,2} + \frac{D_{31}}{r} \tilde{N}$$

$$\bar{\bar{B}}_{12} = D_{12} \tilde{N}_{,r} + D_{42} \tilde{N}_{,2} + \frac{D_{32}}{r} \tilde{N}$$

$$\bar{\bar{B}}_{13} = D_{13} \tilde{N}_{,r} + D_{43} \tilde{N}_{,2} + \frac{D_{33}}{r} \tilde{N}$$

$$\bar{\bar{B}}_{14} = D_{14} \tilde{N}_{,r} + D_{44} \tilde{N}_{,2} + \frac{D_{34}}{r} \tilde{N}$$

$$\bar{\bar{B}}_{21} = D_{21} \tilde{N}_{,2} + D_{41} \tilde{N}_{,r}$$

$$\bar{\bar{B}}_{22} = D_{22} \tilde{N}_{,2} + D_{42} \tilde{N}_{,r}$$

$$\bar{\bar{B}}_{23} = D_{23} \tilde{N}_{,2} + D_{43} \tilde{N}_{,r}$$

$$\bar{\bar{B}}_{24} = D_{24} \tilde{N}_{,2} + D_{44} \tilde{N}_{,r}$$

Then multiply

$$\underline{\underline{\bar{B} \tilde{H}}}$$



REFERENCES FOR APPENDIX Q

- Q-1. Dunham, R. S. and Becker, E. B., "TEXGAP - The Texas Grain Analysis Program", The University of Texas, TICOM Report No. 73-1 (August 1973).